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PREDICTION OF UNSTEADY AERODYNAMIC LOADINGS OF NON-PLANAR WINGS AND WING-TAIL CONFIGURATIONS IN SUPERSONIC FLOW

Part II Computer Program Description

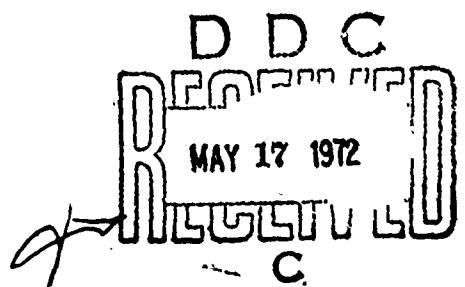
GORDON D. KRAMER

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THE BOEING COMPANY
COMMERCIAL AIRPLANE GROUP

TECHNICAL REPORT AFFDL-TR-71-108, PART II

MARCH 1972



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Flight Dynamics

AFFDL-TR-71-108, PART II

PREDICTION OF UNSTEADY AERODYNAMIC LOADINGS OF NON-PLANAR WINGS AND WING-TAIL CONFIGURATIONS IN SUPERSONIC FLOW

Part II Computer Program Description

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FOREWORD

Part II of this report was prepared by Boeing Computer Services, Inc., in conjunction with The Boeing Company, Commercial Airplane Group, Renton, Washington for the Aerospace Dynamics Branch, Vehicle Dynamics Division, Air Force Flight Dynamics Laboratory, Wright-Patterson Air Force Base, Ohio, under Air Force Contract AF 33615-70-C-1126, "Unsteady Aerodynamics and Flutter of Wing Horizontal Tail Configurations in Supersonic Flow". This work was conducted under Project No. 1370, "Dynamic Problems in Military Flight Vehicles" and Task No. 137003, "Prevention of Dynamic Aeroelastic Instabilities in Advanced Military Aircraft". Part II of this report is intended to describe in detail the computer program developed under the above contract, and is subsidiary to Part I, Theoretical Development, Program Usage, and Application.

Mr. William S. Rowe of The Boeing Company served as supervisor of the work covered under this contract. The co-authors of this part of the report worked from engineering analyses prepared and documented in Part I by Dr. J. M. Iii, C. J. Borland and J. R. Hogley. Mr. Lawrence J. Huttell of the Aerospace Dynamics Branch (FYS) was Project Engineer. Appreciation is expressed to Mr. H. Huffman and Mr. R. Hirst for graphical work done in this report.

Manuscript was released by the authors in August, 1971 for publication as an AFFDL report.

This report has been reviewed and approved.

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ABSTRACT

The Mach box technique has been extended to include wing and tail with dihedral angles and vertical separation. A digital computer program, written in FORTRAN, is presented. The program provides for up to nine sweep angles of the leading and trailing edges of each surface. First order piston theory thickness correction is available as an option, and two refinement procedures are provided, subdivision with averaging and velocity potential smoothing. For a maximum of twenty oscillatory mode shapes the program calculates normal washes, velocity potentials, lifts, pressures and generalized forces matrices. If only one surface is being analyzed, sampling of wake up-wash, side-wash and longitudinal wash is available.

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NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
a		length/time	Speed of Sound = U/M
A(n,m)		area	Integration area of box n,m
b ₁	B1	length	Chordwise dimension of Mach box.
b _{1S}	B1S	length	b ₁ /N _S = chordwise dimension of a subdivided box.
C _{D̄μλ}	C	non-dimensional	Velocity potential spatial aerodynamic influence coefficient (AIC).
C _{D̄μλ} ^(xy)	C	non-dimensional	Spatial AIC giving velocity potential at a point on surface "xy" due to constant outward normal wash over a box on surface "ab"; possible values for superscripts are x or a { R-right L-left y or b { W-wing T-tail
C _{D̄μo}	PKERNL	non-dimensional	Velocity potential planar AIC
C _{D̄μo} ^s	SKERNL	non-dimensional	Planar AIC defined for subdivided sending boxes.
ΔC _{P,j} ^{n,m}	DELCP	1/length	Pressure coefficient difference at box n,m for the j th mode (program output)
l _c , c	--	length	Local reference chord

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
$C_l^m_j$	--	1/length	Local lift coefficient per unit span for the j^{th} mode
$C_m^m_j$	SECMOM	$1/(\text{length})^2$	Local moment coefficient per unit span for the j^{th} mode
c_{rW}	---	length	Wing root section chord length.
c_{rT}			Tail
$\frac{D}{Dt}$	--	1/time	Substantial derivative; $\frac{D}{Dt} = \frac{\partial}{\partial t} + U \frac{\partial}{\partial x}$
$f_j(x,y)$	DEFSL(1,L)	non-dimensional	j^{th} mode shape deflection at location (X,Y).
f_{ij}	---	non-dimensional	Deflection of i^{th} lumped mass in mode j
$\frac{\partial f_j(x,y)}{\partial x}$	DEFSL(2,L)	1/length	Slope of j^{th} mode shape function.
$\bar{f}_j^{n,m}$	---	time	Scaled modal displacement at box n,m $\bar{f}_j^{n,m} = \frac{b_1}{U} f_j^{n,m}$
j		1/length	j^{th} mode shape deflection / s
i	---	mass x length ²	Moment of inertia about the elastic axis of the i^{th} lumped mass
i,j	---	force/length	Generalized stiffness
s	XKS	Non-dimensional	Reduced frequency based on leading planform semi-span, $k_s = \frac{\omega s}{U}$

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
k_1	XKI,K1	non-dimensional	Reduced frequency based on the chordwise dimension of the Mach box $k_1 = \frac{\omega b_1}{U}$
\bar{k}_1	K1BAR	non-dimensional	$k_1 M^2 / \beta^2$
L	CAPL	non-dimensional	Vertical separation of the center lines of the 2 surfaces, positive upward measuring from the wing to the tail.
$L_j^{n,m}$	--	force/length	lift on box n,m for the j^{th} mode
\bar{L}_j^m	--	force/length	lift on the m^{th} chordwise strip of boxes for the j^{th} mode.
\bar{L}_j	--	force/length	lift on a complete half-surface or half-airplane for the j^{th} mode.
$\bar{L}_j^{n,m}$	--	force/length	amplitude of box lift $L_j^{n,m}$
\bar{L}_j^m	--	force/length	amplitude of section lift L_j^m
\bar{L}_j	--	force/length	amplitude of total lift L_j
$\bar{L}_j^{n,m}$	BXLIFT	non-dimensional	Non-dimensional amplitude of box lift (program output) $\bar{L}_j^{n,m} = L_j^{n,m} e^{-i\omega t - i\phi(\frac{b_1}{\rho})} L_j^{n,m} e^{i\omega t}$

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
\underline{L}_j^m	SLIFT	non-dimensional	nondimensional amplitude of section lift (program output) $\underline{L}_j^n = \bar{\underline{L}}_j^n e^{i\omega t} = q \left(\frac{b_1}{\beta} \right) \underline{L}_j^n e^{i\omega t}$
\underline{L}_j^t	TLIFT	non-dimensional	Nondimensional amplitude of total lift (program output) $L_j = \bar{L}_j e^{i\omega t} = q \left(\frac{b_1}{\beta} \right) L_j e^{i\omega t}$
ℓ	EL	non-dimensional	The ℓ_c coordinate location of a pulse receiving point, i.e., the perpendicular distance from the sending plane to receiving point.
M	XMACH	- non-dimensional	-Mach number
M_{jj}	--	mass	Generalized mass for the j^{th} mode
m	--	non-dimensional	The m_c coordinate location of a pulse receiving point.
m_i	--	mass	i^{th} lumped mass
$N_{xyz}^{n,m}$	ENRUS ENRLS, EN, ENSUBD	non-dimensional	Normal wash at box n,m on surface "xyz" due to local source strength, where possible subscript values are: $x = \begin{cases} R-\text{right} \\ L-\text{left} \end{cases}$ $y = \begin{cases} U-\text{upper} \\ L-\text{lower} \end{cases}$ $z = \begin{cases} W-\text{wing} \\ T-\text{tail} \end{cases}$ e.g. $N_{RUW}^{n,m}$ means normal wash on the right upper wing at box n,m

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
$\hat{N}^{n,m}$ xyz abc	ENRULU, ENRLLL, ENRURW, ENRULW	non-dimensional	Normal wash at box n,m on surface "xyz" due to remote source strengths on surface "abc", where possible subscript values for a,b, and c are the same as for x,y, and z, respectively, defined above; e.g. $\hat{N}^{n,m}$ means normal RUT LLW wash at box n,m on the right upper tail due to source strengths on the left lower wing.
N_s	NSUBDV	non-dimensional, odd integer	No. of "sub-boxes" (chordwise and spanwise) to be used in the subdivision improvement technique.
n	--	non-dimensional	the n_c coordinate location of a pulse receiving point.
n_c, m_c, l_c	---	---	Sending Surface Coordinate System
$\bar{n}_c, \bar{m}_c, \bar{l}_c$	---	---	Receiving Point Coordinate System
$p, p(x,y,t)$	---	force/area	local static pressure
p_∞	---	force/area	Free stream static pressure
$\Delta p(x,y,t)$	---	force/area	pressure difference between upper and lower surfaces at point (x,y) at time t $\Delta p(x,y,t) = p(x,y,t)_{\text{upper}} - p(x,y,t)_{\text{lower}}$

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
$\Delta \bar{P}(x,y)$	--	force/area	Amplitude of pressure difference: $\Delta p(x,y,t) = \Delta \bar{P}(x,y)e^{i\omega t}$
Q_{ij}		force/length	Generalized force due to the deformation in the i^{th} elastic mode and loading for the j^{th} modal deflections
\bar{Q}_{ij}	--	force/length	Amplitude of generalized force
\hat{Q}_{ij}	GENAF	non-dimensional	Non-dimensional generalized force (program output); $Q_{ij} = \bar{Q}_{ij} e^{i\omega t} = q \frac{b_i}{\beta} \hat{Q}_{ij} e^{i\omega t}$
\hat{Q}_{ij}	QAGARD	$1/(\text{length})^2$	Generalized force in the AGARD notation
Q'_{ij}, Q''_{ij}	GENAFC	$1/(\text{length})^2$	Real and imaginary parts of \hat{Q}_{ij} in the AGARD definition (program output)
q	--	force/area	dynamic pressure
$q_j(t)$	--	length	Generalized coordinate relating physical deflection to j^{th} modal deflections: $z(x,y,t) = \sum_j f_j(x,y) q_j(t)$
\bar{q}_j	--	length	Amplitude of j^{th} generalized coordinate
s	S	length	Wing semi-span.

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
S	--	length ²	Area of integration. Bounded by edge of planform plus diaphragm and lying inside the forward Mach cone of the receiving point.
t	--	time	Time
U	--	length / time	Free stream velocity.
$U_{\bar{v} \bar{x}}$	---	non-dimensional	Velocity spatial AIC for perturbation velocity parallel to the free stream.
U	---	length / time	Perturbation velocity in the streamwise direction, positive downstream.
$V_{\bar{v} \bar{x}}$	V	non-dimensional	Velocity spatial aerodynamic influence coefficient (AIC) for velocity normal to the free stream and parallel to the sending surface.
$V_{\bar{v} \bar{a} \bar{b}}$	V	non-dimensional	Spatial AIC giving velocity normal to the free stream and parallel to surface "ab" at a point on surface "xy" or in the flowfield (FF), due to constant normal wash over a box on surface "ab"
v		length / time	Perturbation velocity in the spanwise direction, positive right (looking upstream).
$W_{\bar{v} \bar{x}}$	W	non-dimensional	Velocity spatial aerodynamic influence coefficient (AIC) for velocity normal to the sending plane.

NOMENCLATURE

Mathe- matical symbol	FORTRAN Symbol	Dimension	Definition
$\frac{(xy)}{w_{ab}}$	w	non- dimensional	Spatial AIC giving velocity normal to surface "ab" at a point on surface "xy" or in the flow field (FF) due to a constant normal wash over a box on surface "ab".
w		length/ time	Perturbation velocity in the vertical direction, positive upward.
x, Y, Z		length	Reference (global) coordinate system, X positive aft, Y positive right, Z positive upward.

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
x_w, y_w, z_w			Wing local coordinate system.
x_T, y_T, z_T			Tail local coordinate system
x_{WLE}	XWLE	length	The location of a leading edge definition point of the wing planform, measured along the x_w co-ordinate.
x_{TLE}	XTLE	length	Same as above for the tail planform, measured along the x_T co-ordinate.
x_{WTE}	XWTE	length	The location of a trailing edge definition point of the wing planform.
x_{TTE}	XTTE	length	Same as above for the tail, measured along the x_T co-ordinate.
$\left\{ \begin{array}{l} x_F \\ x_c \end{array} \right\}$	$\left\{ \begin{array}{l} XEDGE \\ XCENTR \end{array} \right\}$	length	Location of the $\left\{ \begin{array}{l} \text{edge} \\ \text{center} \end{array} \right\}$ of a Mach box used for the placement of the box pattern, measured along the x_w co-ordinate.
$\left\{ \begin{array}{l} x_{CW} \\ x_{CT} \end{array} \right\}$			The location of the most upstream row of boxes on the $\left\{ \begin{array}{l} \text{wing} \\ \text{tail} \end{array} \right\}$ measured along the $\left\{ \begin{array}{l} x_w \\ x_T \end{array} \right\}$ co-ordinate.

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
$\begin{Bmatrix} Y_{WLE} \\ Y_{TLE} \end{Bmatrix}$	$\begin{Bmatrix} YWLE \\ YTLE \end{Bmatrix}$	length	The location of a leading edge definition point of the wing/tail planform measured along coordinate $\begin{Bmatrix} Y_W \\ Y_T \end{Bmatrix}$.
\bar{y}	YBAR	non-dimensional	Offset of receiving chord from the nearest sending chord.
$\begin{Bmatrix} z_u \\ z_L \end{Bmatrix}(x,y,t)$ --		length	$\begin{Bmatrix} \text{Upper} \\ \text{Lower} \end{Bmatrix}$ surface deflection at (x,y) as a function of time
$z_m(x,y,t)$ --		length	mean surface deflection: $z_M(x,y,t) = f_j(x,y)e^{i\omega t}$
$z_\tau(x,y)$ --		length	Local thickness at (x,y)
$\frac{\partial z_\tau}{\partial x}(x,y)$ TSLFN		non-dimensional	Local thickness slope at (x,y)
\bar{z}_τ --		non-dimensional	Thickness slope piston theory correction; $\bar{z}_\tau = 1 + \frac{Y+1}{2} M \frac{\partial z_\tau}{\partial x}$
$\alpha_{n,m}$	ALPHA	non-dimensional	Edge box area ratio for box(n,m).
β	BETA	non-dimensional	$\sqrt{M^2 - 1}$

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
b_1/β	BIBETA	length	Spanwise dimension of the Mach box.
b_{1s}/β	BIBTAS	length	Spanwise dimension of a subdivided Mach box.
γ	GAMMA	non-dimensional	Ratio of specific heats, = 1.4
ζ	ZETA	non-dimensional	Dummy variable in the z_w or z_T coordinate.
$\bar{\eta}$	ETABAR	non-dimensional	Dummy variable of integration in the \bar{m}_c coordinate.
θ	THETA	radians	$\sin^{-1} \frac{\bar{\eta}}{\zeta} = \sin^{-1} \frac{\bar{\eta}}{\sqrt{\bar{\zeta}^2 - \bar{\lambda}^2}}$
θ_{ij}	--	radians/length	Torsion of i^{th} lumped mass in mode j .
$\bar{\lambda}$	--	non-dimensional	\bar{l}_c coordinate location of a pulse sending box.
μ	MU, YMU	non-dimensional	m_c coordinate location of a pulse sending box.
$\bar{\mu}$	YMUBAR	non-dimensional	\bar{m}_c coordinate location of a pulse sending box.

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
ν	NU, XNU	non-dimensional	n_c coordinate location of a pulse sending box.
$\bar{\nu}$	XNUBAR	non-dimensional	\bar{n}_c coordinate location of a pulse sending box.
ξ	XI	non-dimensional	Dummy variable of integration in the x_w or x_T coordinate.
$\bar{\xi}$	XIBAR	non-dim dimensional	Dummy variable of integration in the \bar{n}_c coordinate.
ρ, ρ_∞	--	Mass/volume	Free stream density.
τ	TAU	non-dimensional	$\sqrt{\xi^2 - \lambda^2}$
$\emptyset(x, y, t)$	--	length ² /time	Disturbance velocity potential at point (x, y) and time t , defined so that $\frac{\partial \emptyset}{\partial x_1}$ is velocity, positive in positive x_1 direction, where $x_1 = X, Y, \text{ or } Z$
$\Delta \emptyset(x, y, t)$	--	length ² /time	Disturbance velocity potential difference between the top and bottom side of the surface at point (x, y) and time t :
			$\Delta \emptyset(x, y, t) = \emptyset_{\text{upper}} - \emptyset_{\text{lower}}$

NOMENCLATURE

Mathematical Symbol	FORTRAN Symbol	Dimension	Definition
$\Delta \bar{\theta}_j(x, y)$	--	length / time	Amplitude of velocity potential difference at point (x, y) or for box n, m for the j^{th} mode
$\Delta \bar{\theta}_j(x, y)$	DELPHI	non-dimensional	Non-dimensional velocity potential difference due to the unit j^{th} generalized coordinate (program output);
$\Delta \bar{\theta}_j^{n, m}$			$\Delta \theta_j^{n, m} = \bar{\theta}_j^{n, m} e^{i\omega t} = \frac{U}{b_1} \left(\frac{b_1}{\beta} \right) \bar{\theta}_j^{n, m} e^{i\omega t}$
$\Delta \bar{\theta}_j(x_{TE}, y)$	TVP	non-dimensional	Trailing edge velocity potential difference.
$\Delta \bar{\theta}_j^m$	TE		
$\left\{ \begin{matrix} \psi_w \\ \psi_T \end{matrix} \right\}$	$\left\{ \begin{matrix} \text{PSIW} \\ \text{PSIT} \end{matrix} \right\}$	degrees	Dihedral angle of $\left\{ \begin{matrix} \text{wing} \\ \text{tail} \end{matrix} \right\}$, radians, positive upwards from horizontal.
ω	---	radians/time	Circular frequency
ω_j	---	radians/time	Circular frequency of mode j

Superscripts

(n, m) } Box location
 (v, μ) }

Subscripts

L	Lower limit of Integration; Left-hand surface; Lower surface
R	Right-hand Surface
S	Subdivided
T	Tail
U	Upper limit of Integration; Upper Surface
W	Wing
FF	Flowfield

GLOSSARY OF TERMINOLOGY

Aftmost Box - Each chord on each planform and diaphragm combination has one such box. It is the aftmost box on that chord for which AIC arrays must be calculated and may be on the planform or diaphragm.

AIC - Aerodynamic Influence Coefficient

Area Ratio - On-planform fraction of a box which is cut by the planform boundary.

Apex Box - The box on the sending surface which encloses the apex of the Mach hyperbola associated with the receiving box.

Box Grid - Non-dimensionalized geometric array of boxes whose extent is determined by the geometric properties of the planforms. The term "grid" embraces the arrays on both surfaces.

Control Point - The location at which a receiving box is deemed to be influenced by other boxes. In general, the center of the receiving box.

Effective Area - A concept which relates entirely to the sub-division technique. It is composed of those boxes sufficiently close to the receiving box that their influence on it is large enough for the subdivision refinement to affect results significantly. The size has been arbitrarily set to include the N_{BOX}/N_S rows immediately ahead of the receiving box.

Global Co-ordinate System - An overall reference system of co-ordinates. For example, the airplane co-ordinate system X ~ aft, Y~right, Z~up. Y = 0 at centerline of airplane.

Leading Edge Diaphragm - All diaphragms on which $\Delta\theta = 0$.

Local Co-ordinate System - A co-ordinate system lying in the plane of the surface. x~aft, y~root to right tip. y = 0 at center line of airplane.

Longitudinal Separation - Streamwise distance between the trailing edge of the wing and the leading edge of the tail, measured along the centerline.

GLOSSARY OF TERMINOLOGY

Mach Asymptote - The asymptote of the Mach hyperbola.

Mach Hyperbola - The intersection of the sending plane and the forward Mach cone of the receiving point. Since this is always non-dimensionalized, it is a rectangular hyperbola.

Map - A condensed description of a large amount of data which can be used to locate any desired data element. A map of a banded sparse matrix might consist of two numbers per row, the first being the first non-zero column of that row and the second being the band width for that row. The matrix itself could then be stored as band elements only.

Normal Offset - The l_c distance between the sending box and the receiving point.

Parallel Offset - The m_c distance between the sending box center and the receiving point.

Partial Box - A sending box which is cut by the Mach hyperbola but which is neither an apex box nor an edge box.

Planar A.I.C. - An A.I.C defined by the geometric relation between a sending box and receiving box which lies in the same plane. $C_{\bar{\nu}, \bar{\mu}, 0}$ only.

Receiving Box - In defining the relationship between two boxes the receiving box is the box which can be influenced by the other box.

Receiving Chord - Those receiving boxes which lie on the same chord. The receiving chord is significant in that all the boxes lying on it use AIC arrays which are a subset of those for the aftmost box lying on that chord.

Sending Box - In defining the relationship between two boxes, the sending box is the box which influences the other box (c.f. Receiving Box).

Spatial A.I.C. - An AIC defined by the geometrical relationship between two boxes which do not lie in the same plane. $C_{\bar{\nu}, \bar{\mu}, \bar{\lambda}}, V_{\bar{\nu}, \bar{\mu}, \bar{\lambda}}, W_{\bar{\nu}, \bar{\mu}, \bar{\lambda}}$.

GLOSSARY OF TERMINOLOGY

Sub-box - A member of the array of boxes formed when the grid of sending boxes is subdivided. Note it refers to the small box which is a fraction of the large box, and not to a large box which has been subdivided.

Tail - The downstream surface.

Vertical Separation - The vertical distance between the center lines of the two surfaces. Positive if the second surface is above the first.

Wake Diaphragm - That part of the diaphragm where $\Delta P = 0$ due to the influence of a surface.

Wing - Upstream Surface - (E.g. a Canard could be referred to as a wing);

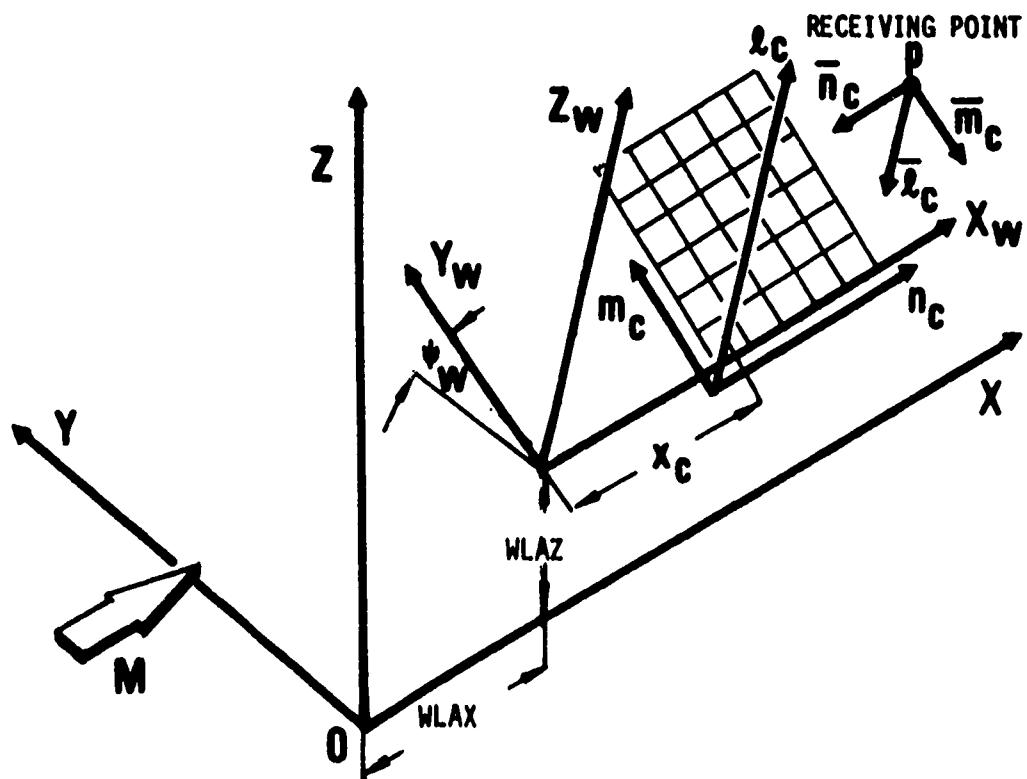


Figure 1 Coordinate Systems For A Right Wing

Symbol	Transformation	Definition	Dimension
X Y Z	Global or Reference Coordinate System. X positive Aft, Y positive Right, Z positive Upward		Length
x_w y_w z_w	$x_w = WLAX$ $y_w = Y \cos \phi_w + (Z - WLAZ) \sin \phi_w$ $z_w = (Z - WLAZ) \cos \phi_w - Y \sin \phi_w$	Wing Local Coordinate System, used to define wing leading and trailing edges. x_T, y_T, z_T are similarly defined for the tail local axes	Length
n_c m_c l_c	$(x_w - x_c)/b_1 + 1$ $y_w/(b_1/s) + 1/2$ $z_w/(b_1/s)$	Sending Surface Coordinate System used to define box grid. The (n_c, m_c) plane lies within the plane of the sending surface, in this case the right wing.	Non-dimensional

Figure 1 (Cont'd)

Symbol	Transformation	Definition	Dimension
\bar{n}_c \bar{m}_c \bar{l}_c	$-(n_c - n)$ $-(m_c - m)$ $-(l_c - l)$	Receiving Point Coordinate System parallel to the n_c, m_c, l_c coordinates but opposite in sign and having their origin at the pulse receiving point (n, m, l in the n_c, m_c, l_c coordinates)	non-dimensional

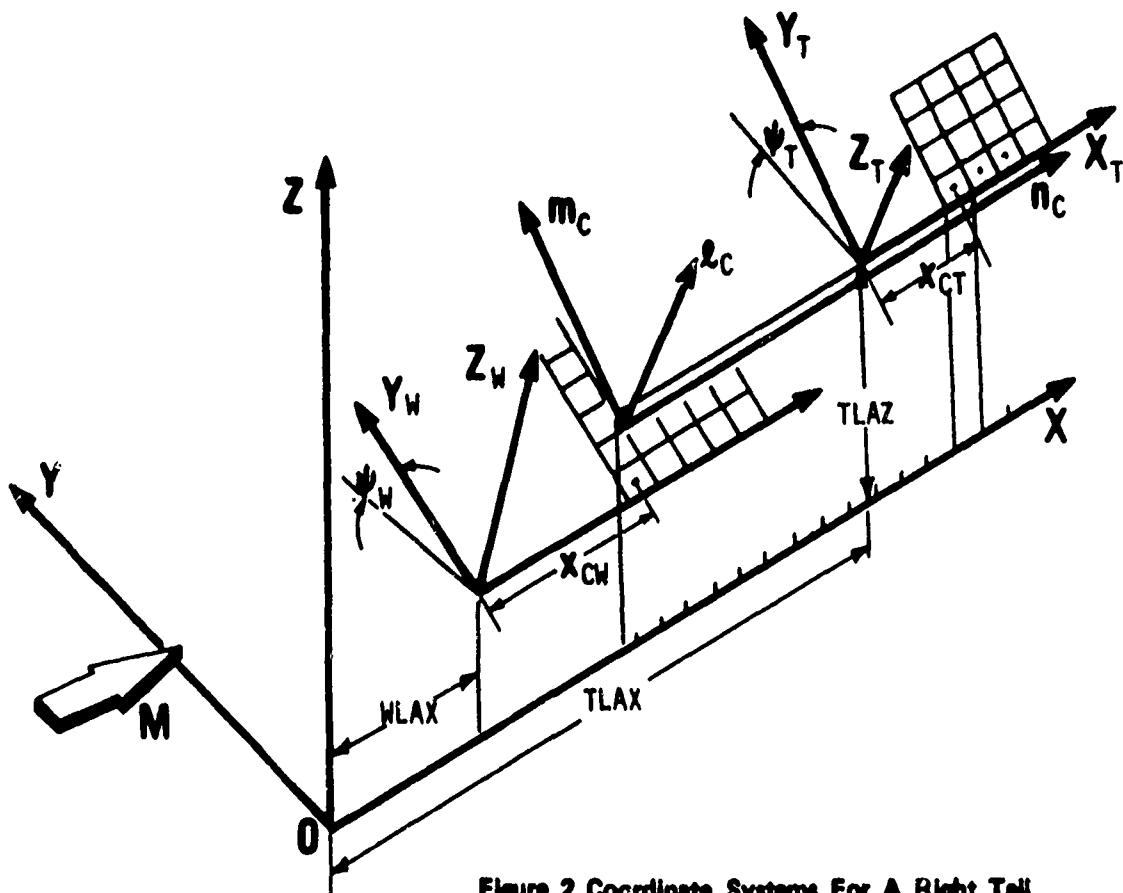


Figure 2 Coordinate Systems For A Right Tail

SYMBOL	TRANSFORMATION	DEFINITION	DIMENSION
x_T y_T z_T	$x = TLAX$ $y = \cos \psi_T + (z - TLAZ) \sin \psi_T$ $z = (z - TLAZ) \cos \psi_T - y \sin \psi_T$	Tail Local Coordinate System used to define tail leading and trailing edges.	length
n_C m_C β_C	$n_C = \{x_T + TLAX - (WLAX + X_C)\} / b_1 + 1$ $m_C = y_T / (b_1 / \beta) + 1/2$ $\beta_C = z_T / (b_1 / \beta)$	Sending Surface Coordinate System. In this case the right tail is shown as the sending surface.	non-dimensional

SECTION I

INTRODUCTION

Part II of this report describes the computer program written according to the analysis of Part I. Part II refers implicitly to Part I, Section III, Computer Program Usage, and material covered there is not repeated here. The program computes generalized unsteady air forces on a wing or wing and tail in supersonic flow, given geometric details of the surfaces and the oscillatory mode shapes of the surfaces. The surfaces may be coplanar, may have dihedral angles, and may be separated vertically. The Mach box technique may be used "straight", or three refinements may be applied: 1) Sub-division of the Mach boxes to improve velocity potentials, 2) Least-squares smoothing of calculated velocity potentials to eliminate roughness due to box representation of surface edges, 3) Piston theory correction for airfoil thickness. The refinements may be applied in any combination. As intermediate results, normal-washes, velocity potentials and wake sampling of upwash, sidewash and longitudinal washes may be printed, all at box center locations. The box lifts, pressure distribution, section lifts and total lift are also available for each mode.

SECTION II

COMPUTER PROGRAM DESCRIPTIONS

1. GENERAL DESCRIPTION

The computer program consists of a main (0,0) overlay, one primary level overlay, and eight secondary level overlays (see fig. (3)). The main (0,0) overlay is a general purpose driver, and easily can be used to incorporate other compatible programs with this one as a complete flutter system. Its sole function is to set up buffers and any other system oriented parameters, then call the primary level overlay, CONTROL.

Overlay CONTROL contains the basic logic of the program. It first calls secondary overlay DATAPP, which reads and processes the basic card data necessary for execution of the program. The resulting parameters are stored in labelled common blocks, accessible to all other overlays.

If PRVGEOM = .FALSE., overlay GEOMBX is next called. The planform geometry is read and processed to yield a disk file IGEOOSC containing all internally necessary geometric parameters.

If PRVMODE = .FALSE., overlay CONTROL next calls overlay MODES. This area processes the three forms of modal data and places the results, evaluated at box centers, on scratch file MODESC.

Overlay CONTROL next enters a loop on reduced frequency. Each pass through the loop first executes overlay VICMAIN, which computes (or reads from previously saved tapes) all AIC arrays needed at the current reduced frequency. Next overlay NWVPMBX is called, to compute normal-washes, velocity potentials, and optional sample washes. If SMOOTH = .TRUE. overlay SMTH is called to do a least-squares surface fit of the resulting $\Delta\phi$ arrays. If CRDFIT = .TRUE. overlay CHORDF is called to smooth the $\Delta\phi$ values a chord at a time. The final overlay, FORCES, then computes box lifts, section lifts and generalized forces for any smoothed $\Delta\phi$ values first, then for the unsmoothed values. The desired results are printed as they are computed. The loop on reduced frequencies terminates at this point.

Overlay CONTROL reads the termination card which causes a transfer back to the execution of DATAPP (Recycle), the call of another overlay (if available), RETURN to the main (0,0) overlay, or EXIT to control cards.

The following sections give a more detailed description of all of the overlay main programs, and the major subroutines called by each.

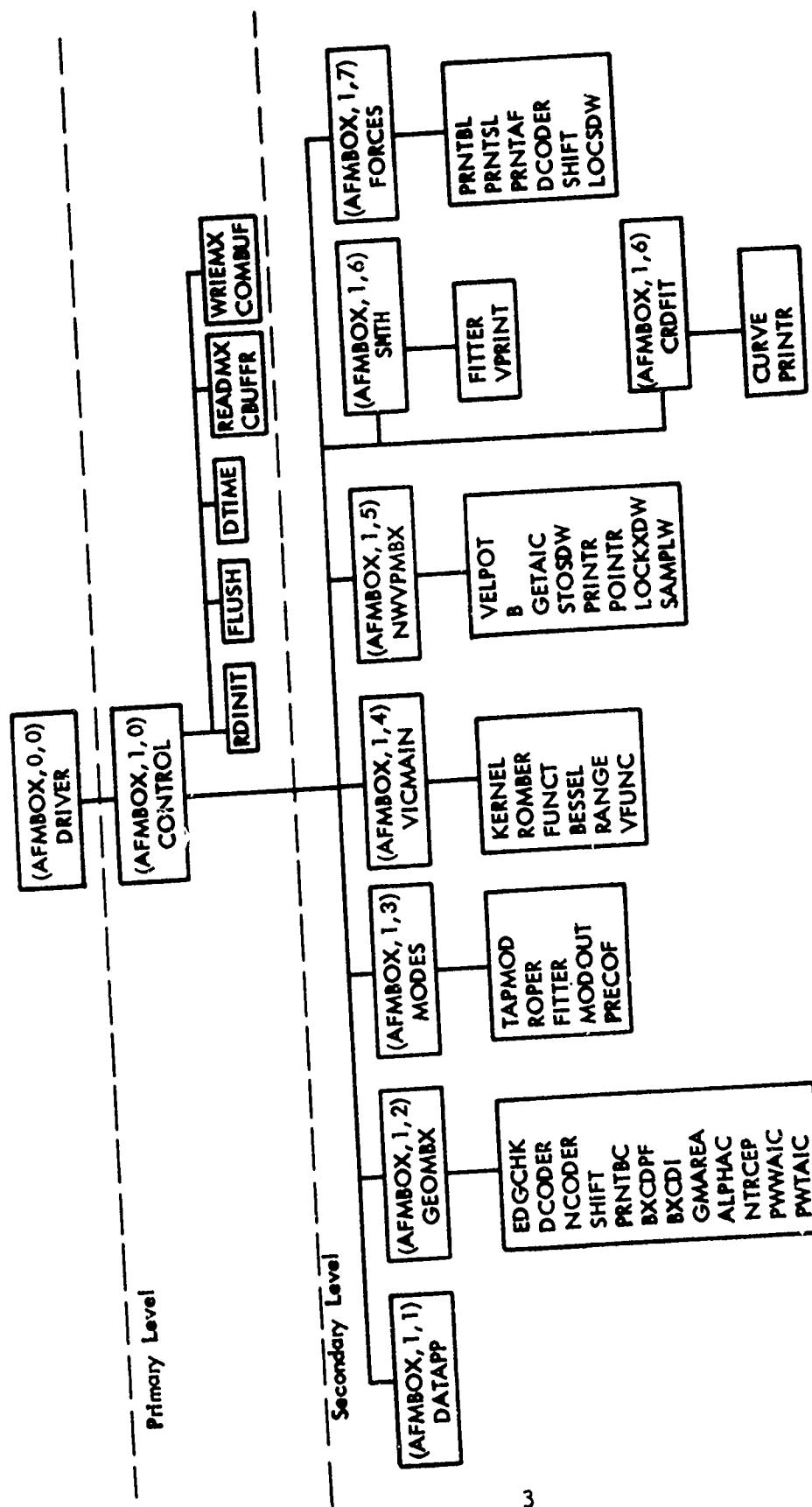


Figure 3 Program Overlay Structure

1. GENERAL PURPOSE SUBROUTINES

Fortran Callable COMPASS Function SHIFT

Author: G. E. Keylon

Purpose: To shift the contents of a word left or right a specified number of bits, identical to the Fortran Extended capability.

Method: The word and the number of bits to be shifted are stored in machine registers. The word is then left circular shifted the number specified. This causes the word to be shifted left circular if the number is positive and right with sign extension if the number is negative. The result is left in register X6 so that this routine must be used as a function subroutine.

Usage: INTEGER SHIFT

.

.

.

IWORD = SHIFT (NWORD,N)

Input

NWORD - The word to be shifted

N - The number to shift the word

If N is positive shift left circular.

If N is negative shift right with sign extension.

Output

IWORD has the results of the shift on NWORD.

Fortran Subprogram WRTEMX

Author: G. E. Keylon

Purpose: To write a matrix on a tape or disk file.

Method: The matrix is placed row-wise into a buffer in labelled common RWBUFF with all of the unused areas of its array omitted. The buffer is then written onto the specified tape or disk file with the Fortran BUFFER OUT statement. A 16 word header record is written in the same manner before each matrix. The header record contains matrix size, name and optional parameters.

Usage: CALL WRTEMX (IOUTFL, MXWRIT, RANDOU, NFS, NMS, LS, NMR, LWS, K, ID, A, ITYPE, M, N, PARM, IRR)

Input

IOUTFL - Tape number or left-justified file name.
MXWRIT - Logical variable, not used.
RANDOU - .T. Random File (not used)
 .F. Sequential File
NFS - Number of files to space before writing
NMS - Number of matrices to space before writing
LS - Level number to space (not used)
NMR - Name or number in random index (not used)
LWS - Level number of this matrix (not used)
K - Row dimension of array A.
 if K≤ matrix is already in /RWBUFF/
ID - Array containing matrix nome.
A - Array containing matrix
ITYPE - TYPE of matrix (i.e., real, complex,
 integer, null, mixed)
M - Row dimension of matrix
N - Column dimension of matrix
PARM - 10-word parameter array

Output

IRR - Error return
 0, no error
 1, matrix spacing is negative
 2, File spacing is negative
 4, M*N dimensions greater than buffer size
 1500+I, encountered EOF after matrix I
 while skipping matrices.

Fortran Subprogram COMBUF

Author: G. E. Keylon

Purpose: To put a complex matrix into a buffer prior to writing on file.

Method: The matrix is placed row-wise into a buffer with all of the unused areas of the matrix omitted. All of the real parts are stored in the first part of the buffer then all of the imaginary parts are placed immediately following the real.

Usage: CALL COMBUF (A, K2, M, N, BUFF)

Input:

A - Array that contains matrix, typed complex
K2 - 2x (row dimension of A)
M - Number of rows in matrix (not array size)
N - Number of columns in matrix (not array size)

Output:

BUFF - Buffer that will contain matrix

Fortran Subprogram READMX

Author: G. E. Keylon

Purpose: To read a matrix from tape or disk file.

Method: A 16-word header record and a matrix record are read from the specified file with BUFFERIN statements. The 16-word header record contains matrix size, name and optional parameters. The matrix is then placed in a given array in correct Fortran storage.

Usage: CALL READMX(INFILE, MXREAD, RANDIN, NFS, NMS, LS, NMR, K, NID, ID, ITYPE, LRS, A, M, N, PARM, IRR)

Input:

INFILE - Tape number or left justified alphanumeric file name
MXREAD - Logical variable (not used)
RANDIN - .T. Random File (not used)
 .F. Sequential File
NFS - Number of files to space before reading
NMS - Number of matrices to space before reading
LS - Level number to space (not used)
NMR - Random name or number (not used)
K - Row dimension of array A
 If $K \leq 0$ matrix will be left in /RWBUFF/
NID - Number of words available in ID array

In/Out:

ID - Identification array
ITYPE - Real, diagonal, null, mixed, complex

Output:

LRS - Level number of matrix read (not used)
A - Array containing matrix
M - Row dimension of matrix
N - Column dimension of matrix
PARM - Array of numerical parameters stored with the matrix in the 16 word header record
IRR - Error return
 0, no error
 1, matrix spacing is negative
 2, file spacing is negative
 4, matrix dimensions illegal
 5, $M > K$
 1500 + I, encountered EOF after matrix I while skipping matrices.

Fortran Subprogram CBUFFR

Author: G. E. Keylon

Purpose: To move a complex matrix from a buffer to a Fortran array.

Method: The matrix assumed stored row-wise in the buffer with all of the real parts followed by all of the imaginary parts. The conversion leaves the matrix in the array in typical Fortran storage.

Usage: CALL CBUFFER(A,K2,M,N,BUFF)

Input:

K2 - 2x (row dimension of array A)
M - Number of rows in matrix (not array size)
N - Number of columns in matrix (not array size)
BUFF - Buffer that contains matrix

Output:

A - Array that will contain matrix in complex storage

3. MAIN CONTROL PROGRAM

Fortran Program CONTROL

Author: G. E. Keylon, G. D. Kramer

Purpose: To control the flow of the program to the various lower level overlay section.

Method: The program has all of the labeled common blocks so that information can be passed from lower levels to this program which will determine the program flow.

Usage: The CONTROL Program is a main routine. It is the only primary overlay section in the program. It calls all of the lower level or secondary overlay sections. It is called from the initial or main overlay section as follows:

```
CALL OVERLAY(6HAFMBOX, 1, 0, 0)
```

Common Input and Output:

This program does not input or create common values. It is the means by which common values are passed between the secondary overlays of the program.

4. DATA INPUT PROCESSOR

Fortran Program DATAPP

Author: G. E. Keylon

Purpose: To read most of the input data and set flags and options for use throughout the program. It prints the title and options for each run.

Method: The title and all the input options are read in. The heading is printed. The options are read under a NAMELIST format and flags set to default options unless read in.

Usage: The DATAPP program is the main program of a secondary overlay of the Mach Box program. It is called as an overlay section as follows:

```
CALL OVERLAY(6HAFMBOX, 1, 1, 0)
```

All input and output is through labeled common blocks.

Common Input:

```
PREVEX  
OMACH  
DEFAULT
```

Common Output:

TITLE	ERR	SYM
PRVGEOM	XKUAL	MTYPEW
PRVMODE	OPLAIC	MTYPET
DIHW	OSPAIC	COPLAN
DIHT	WTGEOM	NSUBDV
XMACH	WTGNAF	NSURF
NKVALS		
XKI		
XKS		
NT5	WTBL	SMOOTH
NT6	PRBOX	NDEG
INTAPE	PRPAIC	DPPCPR
INFSP	PRSAIC	
ISMPLW		GEOCPR
NPLAIC	PRMODS	MODCPR
NSPAIC	PRCOEF	
	PRNW	AICCP
NOUTP	PRUW	
IOUFSP	PRSW	NWSCPR
	PRVP	
OSAIC	PRBL	GAFCPR
	PRSL	
	PRGNAF	
	PRDCP	
	PRGNAC	

5. GEOMETRY PROCESSOR

Fortran Program GEOMBX

Author: G. D. Kramer

Purpose: To read geometric data from cards and compute all necessary geometric parameters.

Method: Cards F through L are read in this section. As they are read they are printed, then checked for inconsistent or missing data, with suitable diagnostics. The leading and trailing edge data is checked in EDGCHK, then transformed to non-dimensional coordinates. Planform and diaphragm box code patterns are determined in BXCDPF and BXCDI, and optionally printed by PRNTBC. The fractional on-planform portion of all boxes cut by a planform edge is determined by GMAREA, which in turn calls ALPHAC and NTRCEP. If spatial AIC's are necessitated by non-zero dihedral angles or vertical separation of wing and tail, integer arrays MUAIC are determined for each AIC set (C,W,V). These serve as a map, so that only those AIC values needed will be calculated. The MUAIC arrays are computed in PWWAIC and PWTAIC. All resulting arrays are written on scratch file IGEOSC.

Usage: The GEOMBX program is the main program of a secondary overlay. It is called by:

```
CALL OVERLAY (6HAFMB0X, 1, 2, 0)
```

Common Input:

OMACH	NSUBDV
TITLE	NSURF
PRVGEOM	MYBW
DIHW	PRBOX
DIHT	GEOCPY
XMACH	

Common Output:

COPLAN	MXBW	MXBT	FSMPLW
XSUBDV	MXBBW	MYBT	ICHORD
NSUBDV	MYBBW	MYBBT	IBOXF
NSUBD2	MXBSW	MXBST	IBOXL
NSUBCN	MYBSW	MYBST	ZLOC
B1	MYBBSW	MYBBST	
B1BETA	IXBW	IXBT	
B1S	XCENTR	IXBST	
B1ETAS	TLAX	CAPL	

WLAX	TLAZ	NSPATK
WLAZ	PSIT	
PSIW		

Arrays output on scratch file IGEOOSC:

- IBOXW - Wing box codes (Wing and tail if COPLAN = .TRUE.)
- IBOXT - Tail Box codes
- FEXLOC - Leading edge locations at chord centers
- TEXLOC - Trailing edge locations at chord centers
- ALPHA - Fractional areas of boxes cut by a planform edge
- IJALPH - Locations of cut boxes, of the form $(1000 \cdot J + I)_8$
- KPT - Table of contents for the MUAIC arrays (and AIC's)
- MUAIC - Pointer array indicating where contributing boxes will be found for one spatial AIC set.

Fortran Subroutine EDGCHK

Author: G. D. Kramer

Purpose: Given the leading or trailing edge values, to check for illegal combinations.

Method: Either a leading or trailing edge is checked for monotonic increasing y-values, starting at zero. The last trailing edge value is compared with the previous last value. A leading edge is checked for monotonically increasing x-values.

Usage: DIMENSION XEDGE(10), YEDGE(10)

CALL EDGCHK (XEDGE, YEDGE, NEDGE, IEDGE, IRR)

Input Parameters:

XEDGE - Array of X-values for edge location points
YEDGE - Array of Y-values for edge location points
NEDGE - Number of points to check
IEDGE - =1, leading edge
 =2, trailing edge

Output Parameter:

IRR =0, Successful
 = 1, Non-monotonic y-values
 = 2, Non-monotonic x-values, leading edge only
 = 4, Y-values not starting at zero
 = 8, Tip y-values not agreeing
 Other, additive combination of above conditions

Fortran Subroutine BXCDPF

Author: G. D. Kramer

Purpose: To generate on-planform box codes for one surface, and store them in a compressed format.

Method: For each (subdivided) chord, the location of the leading edge and trailing edge (FEXLOC, TEXLOC) is determined. Codes for all boxes between those values are then set to 1 in subroutine NCODER. The remainder of the box code array is not changed.

Usage: The subroutine is called by:
CALL BXCDPF(XLE, YLE, NLE, XTE,YTE,NTE, LSROWS, IBOX)

Input Parameters:

XLE } = { x } locations of leading edge, measured in
YLE } { y } n_c, m_c, l_c system

XTE } = { x } locations of trailing edge, measured
YTE } { y } in n_c, m_c, l_c system

NLE } = Number of { leading } edge definition points.
NTE } { trailing }

LSROWS = Maximum number of subdivided rows allowed.

In/Out Parameters:

IXBW } = Input: 0 indicates the { wing } is to be done.
IXBT } Output: Subdivided row of first unsub-
divided box center on the surface.

Output Parameters:

IBOX Compressed box codes, 1 for on-planform
boxes found, unchanged elsewhere. See
Figure 4.

Output Common Parameters:

MXBS { W } Maximum X (aft) extension of the subdivided
{ wing } pattern
{ tail }

MYBS { W } Maximum Y (outward) extension of the sub-
divided on-planform { wing } pattern
{ tail }

MXB { W } Maximum X extension of the unsubdivided
{ wing } pattern
{ tail }

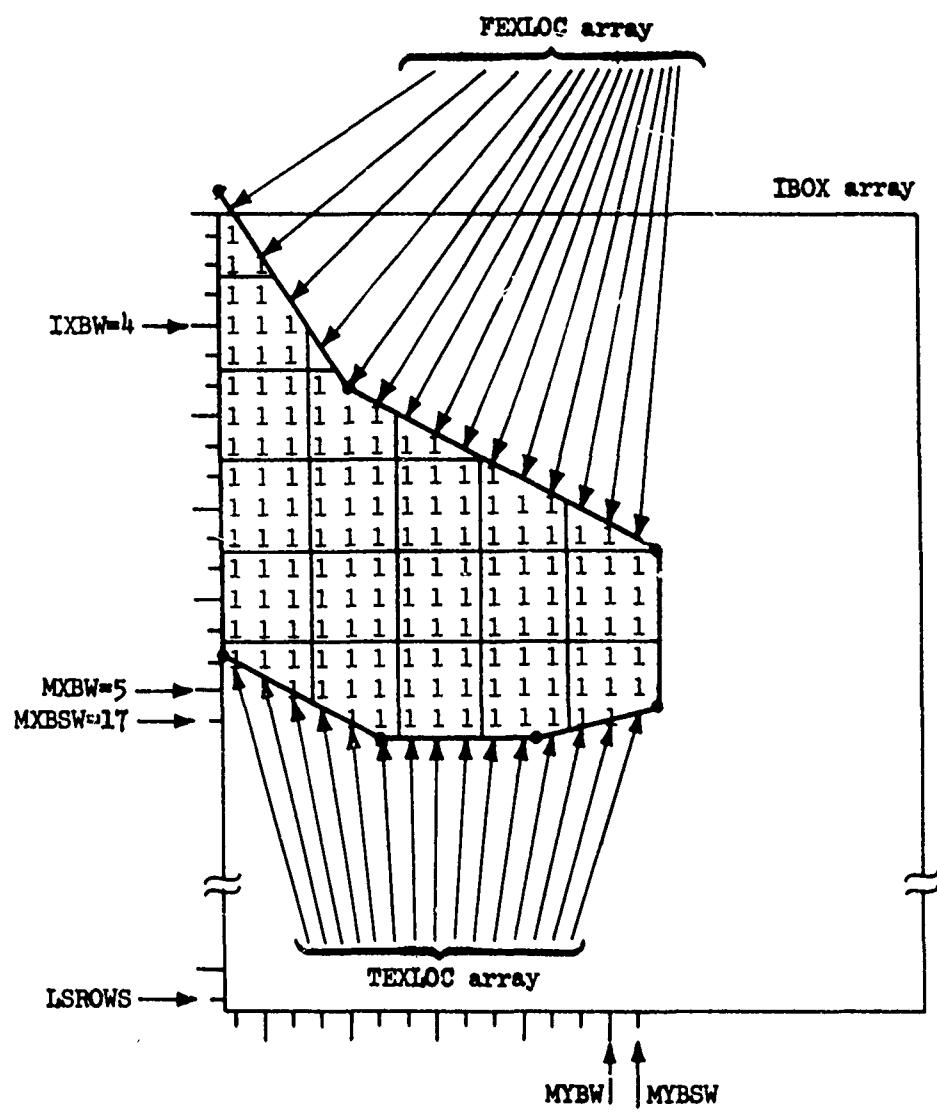


FIGURE 4 Output from BXCDPF, Called for a Wing (NSUBDV=3)

MYB $\left\{ \begin{matrix} W \\ T \end{matrix} \right\}$ Maximum Y extension of the unsubdivided
 wing tail pattern

FEXLOC } = $\left\{ \begin{matrix} \text{Leading} \\ \text{Trailing} \end{matrix} \right\}$ edge X location at (subdivided)
 TEXLOC chord centers.

Fortran Subroutine BXCDI

Author: G. D. Kramer

Purpose: Given an array indicating a pattern of on-planform Mach boxes, to determine the associated off-planform diaphragm boxes

Method: Leading edge diaphragm boxes are first determined, followed by wake diaphragm boxes. The tip diaphragm is then determined as a function of the tip chord. For the wing, an integer array is interrogated to determine whether additional wake areas (and tip diaphragm) are needed for wing-tail interference.

Usage: The subroutine is called by:
CALL BXCDI (IWAKE, LSROWS, LSCHDS, IBOX)

Input Parameters:

IWAKE: Array of locations on the wing for aft-most unsubdivided box in each chord affecting a tail surface. Not used for the tail surface, first element = 0

LSROWS: Maximum number of subdivided rows allowed

LSCHDS: Maximum number of subdivided chords allowed

Input/Output

IBOX: Array of subdivided box codes, previously set 1 at planform locations by subroutine BXCDPF. See figure 5.

Common Input

MXBBSW } Maximum X extension of the subdivided {wing}
MXBBST } box pattern, including diaphragm
IXBST X-location of the first subdivided tail row
MYBSW } Maximum Y extension of the subdivided plan-
MYBST } form {wing} pattern {tail}
MYBBSW } Maximum Y extension of the subdivided {wing}
MYBBST } pattern, including diaphragm
NSUBDV Number of subdivisions

Common Output

MYBBSW } Modified, if necessary
MYBBST }
MYBBW } Maximum Y extension of the unsubdivided
MYBBT } {wing} pattern, including diaphragm {tail}

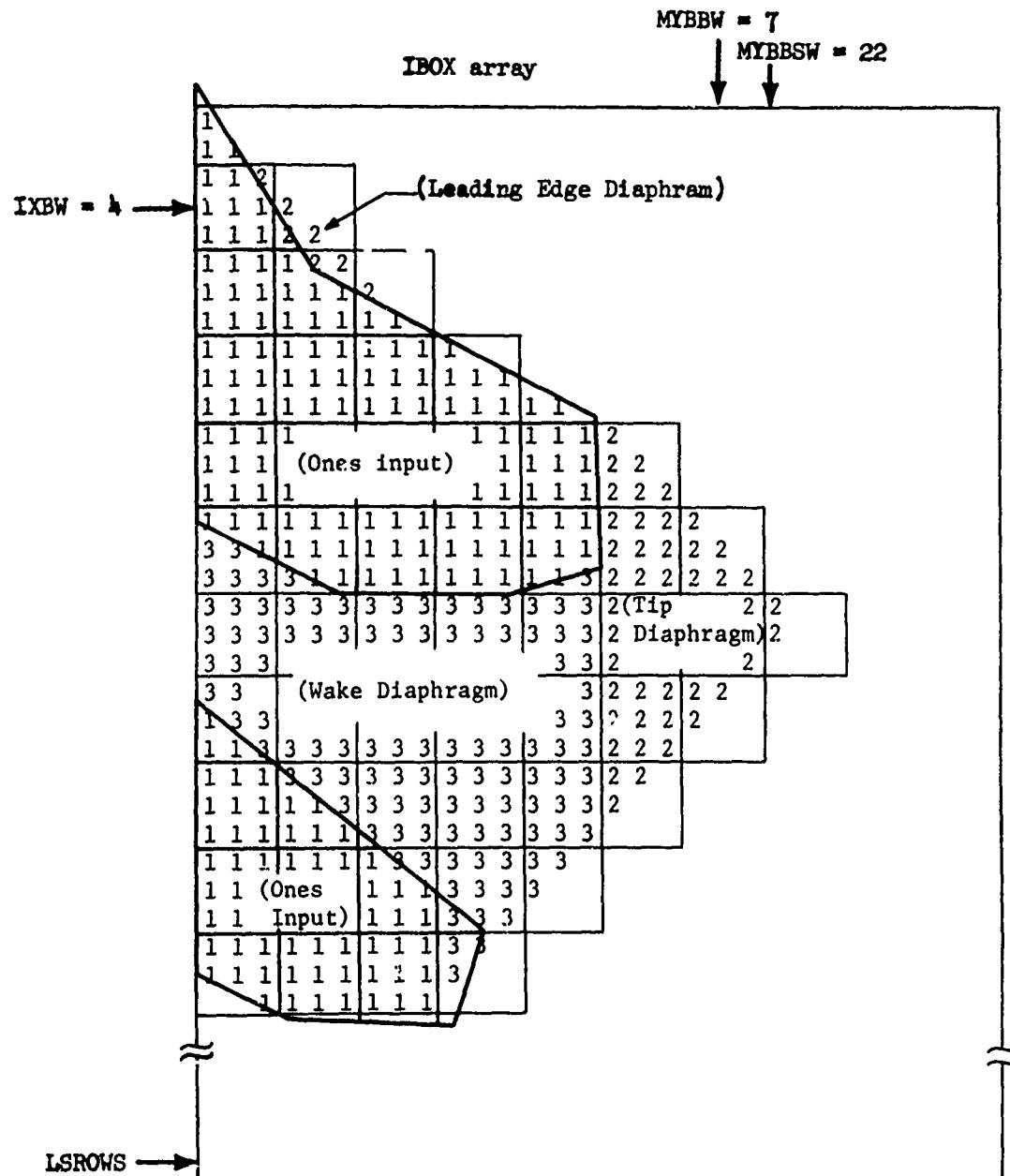


Figure 5 Output from BXCDI, called for a Coplanar Wing and Tail ($NSUBDV = 3$)

Fortran Subroutine PRNTBC

Author: G. D. Kramer, G. E. Keylon

Purpose: Print the array of box codes, either all values or only unsubdivided box-center values

Method: The compressed box code array is decompressed using subroutine DCODER, one row at a time, and printed. If unsubdivided codes have been requested, only the control point values are printed.

Usage: CALL PRNTBC (IBOX, LBXCD, IROW, MXB, MYB, SUBD)

IBOX - Box code array
LBXCD - Row size of box code array
IROW - First row to print
MXB - Last row to print
MYB - Number of chords to print
SUBD - .T., subdivided box codes desired
 .F., unsubdivided (control point) box codes desired

Fortran Subroutine PWWAIC

AUTHOR: G. D. Kramer

PURPOSE: Given the box pattern and dihedral angle of the surface, to determine a pointer array (MUAIC) for one chord on the right surface which indicates contributing regions (if any) of the left surface on the given chord.

METHOD: The geometric relationship of the sending surface to the receiving chord is first determined. Then for all rows, from the last receiving box forward to the forward edge of the box pattern, any sending boxes on the left surface are indicated in the MUAIC array.

USAGE: CALL PWWAIC(WING,IBOX,LBXCD,IWAKE,JCOL)

Input Parameters:

WING .T., wing is being considered.
 .F., tail is being considered.

IBOX Array of box codes (IBOXW or IBOXT).

LBXCD Length of array IBOX.

IWAKE Array of locations of aft-most box to be considered on the wing. Ignored if WING = .F.

JCOL Chord being considered (receiving).

Common Input:

PSI {_W
_T} Dihedral angle

NSUBDV Number of subdivisions

YSUBDV Number of subdivisions, real

NSUBD2 NSUBDV/2

IXBW Location of first unsubdivided box center.

Output Parameters:

The computed results are returned via common block MJAICS. They are:

SURF Logical indicator - true means a sending surface was encountered.

MUAIC(2,50) Unsubdivided row "map" of sending box locations, see Figure 7.

EL Normal offset of receiving chord from sending surface.

YBAR Parallel offset of receiving chord.

NROWS Number of rows considered.

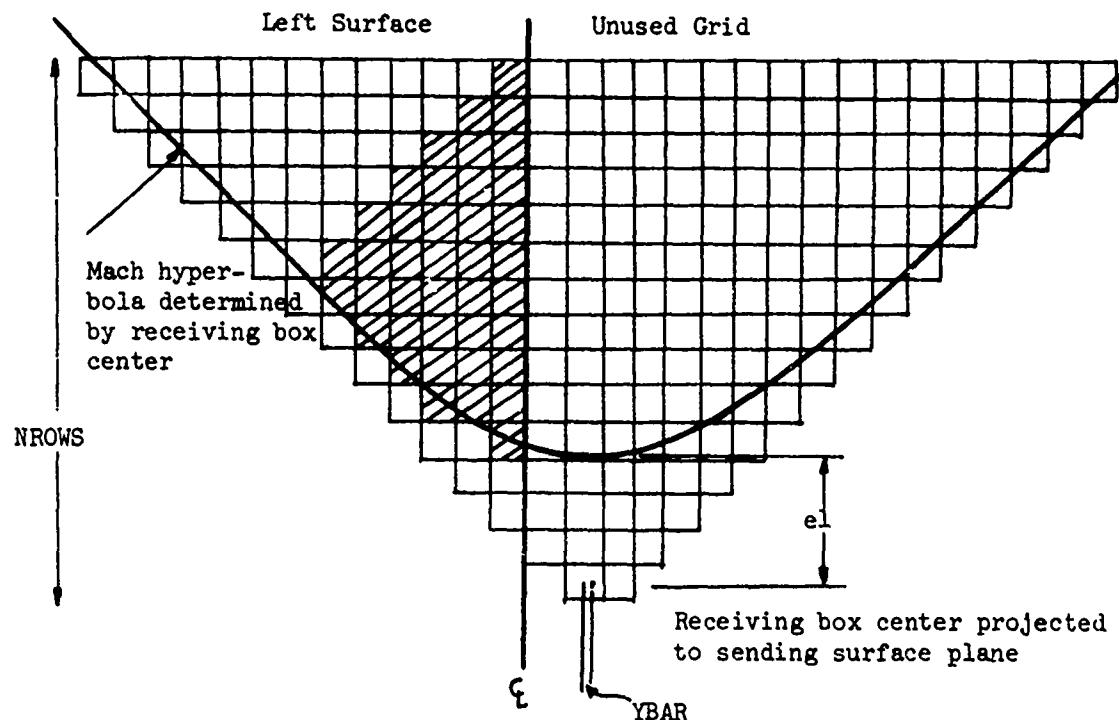


FIGURE 6. Possible Arrangement of Sending Boxes, Left Surface to Right
(Actual sending boxes shaded.)

0	0	0	0	3	2	2	2	3	5	7	9	11	13
0	0	0	0	3	4	5	6	7	8	9	10	11	12

NROWS = 15

FIGURE 7. MUAIC Array Generated by PWWAIC for Figure 4.

Fortran Subroutine PWTAIC

Author: G. D. Kramer

Purpose: Given the box patterns and dihedral angles of the two surfaces, to determine pointer arrays (MJAIC arrays) for the right wing and the left wing contributing regions to a desired tail chord.

Method:] The geometric relationship of the sending surfaces to the receiving chord is first determined. Then for all rows, from the last receiving box forward to the forward edge of the sending vox patterns, any sending boxes are indicated in the MJAIC arrays.

Usage: The subroutine is called, after suitable setup, by:

```
CALL PWTAIC (IBOXW, LBXCDW, IROW, JCOL, CAPLL, YMUVSP)
```

Input Parameters:

IBOXW	- Array of wing box codes
LBXCDW	- Row dimension of IBOXW
IROW	- Unsubdivided receiving row number
JCOL	- Unsubdivided receiving chord number
CAPLL	- Vertical Separation of sending center line receiving center line
YMUVSP	- $\bar{\alpha}$ contribution due to vertical separation = CAPLL* sin(ψ_w)

Input Common Variables

PSIDIF	= $\phi_r - \phi_w$	PSIT
NSUBDV		FEXLOC
PSIW		TEXLOC
MYBBW		SYM
MYBSW		
IXBW		

Output Parameters

The computed results are returned via common block MJAICS. They are:

{SURF}	= .T., Contributing boxes were found on the
{SURFL}	{right} wing
	{left}
	= .F., No contributions were found

$\{MJAIC\}$ Map of contributing boxes on the $\{\text{right}\}$
 $\{MJAICL\}$ wing, see Figure 7.

$\{\text{EL}\}$ The normal offset between the wing $\{\text{right}\}$
 $\{\text{ELL}\}$ plane and the receiving point

$\{\text{YBAR}\}$ The paralleled offset between the nearest
 $\{\text{YBARN}\}$ chord center on the wing $\{\text{right}\}$ box
pattern and the receiving point

$\{\text{NROWS}\}$ Number of rows covered by the MUAIC
 $\{\text{NROWSL}\}$ array for the $\{\text{right}\}$ wing contributions

Fortran Subroutine GMAREA

AUTHOR: G. E. Keylon, G. D. Kramer

PURPOSE: To compute the fractional on-planform portion of all planform boxes which are cut by a planform edge.

METHOD: For each chord, the X coordinates of the left side intercept, right side intercept, and any kinks within the box width are determined by subroutine NTRCEP for each planform edge cut by the chord (wing and/or tail). Then for each planform box on the chord, the routine determines whether any edge cuts the box or causes a contribution to the box area. For any affected box, subroutine ALPHAC is called to compute the fractional area, which is then stored in array ALPH, and its location is stored in array IJALPH as ($J * 512 + I$). The fraction may be greater than one, since it includes the planform area of any chordwise adjacent box whose center is off planform.

USAGE: The routine is called by:

```
CALL GMAREA (IBOX, LBVCD, WING, ALPHA, IJALPH, NALPH)
```

Input Parameters:

IBOX	Box code array
LBVCD	Size of box code array
WING	.T., Wing or coplanar case .F., Tail

Input Common Parameters:

COPLAN	MXBT	NSURP	XWLE	XTLE
FEXLOC	MXBW	NWLE	YWLE	YTLE
TEXLOC	MYBT	NWTE	XWTE	XTTE
IXBT	MYBW	NTLE	YWTE	YTTE
IXBW	NSUBDV	NTTE		

Output Parameters:

ALPHA	Array of area multipliers
IJALPH	Array of corresponding IJ locations, as ($J * 512 + I$)
NALPH	Number of fractions calculated

Fortran Subroutine NTRCEP

AUTHOR: G. E. Keylon

PURPOSE: To compute the X coordinates of the intersections of a planform edge with the sides and center of a chord.

METHOD: The routine determines in which interval of the edge the desired point lies. The x-coordinate of the point is then obtained from the standard two point equation of a line. This is done for all three points.

USAGE: The routine is called by:

CALL NTRCEP (J, YEDG, XEDG, L1, C1, R1, NBK1, K1, IDEX)

Input Parameters:

J = Chord number

YEDG} = Arrays of {Y locations of the edge definition points
XEDG} X

IDEK = 1, leading edge
2, trailing edge

Common Input Values (from local common block/LAREA/)

LEFT = Y-location of the left side of the chord
RIGHT = Y-location of the right side of the chord

Output Parameters

L1 = X coordinate of left side intersection

C1 = X coordinate of center line intersection

R1 = X coordinate of right side intersection

NBK1 = Number of edge definition points encountered between the left and right sides of the chord.

K1 = 0 if no edge definition point lies between the left and right sides of the chord.

= The first (leftmost) edge definition point number lying within the chord.

Subroutine ALPHAC

AUTHOR: G. E. Keylon, G. D. Kramer

PURPOSE: To compute the on-planform area of a box which is partially off the planform or which must include area from neighboring off-planform box(es) cut by a planform edge.

METHOD: If the box is the first box on the chord, or the last box on the chord, the box is divided spanwise into a series of trapezoids (or triangles) determined by planform edge definition points occurring within the chord. The areas of these trapezoids are then added, yielding α .

If the box is an interior box which is cut by one or more planform edge segments, the area is first set to one, then the area of the off-planform corner(s) determined as trapezoids or triangles is subtracted.

USAGE: The routine is called by

```
CALL ALPHAC (X, XLED, YLED, XTED, YTED, L1, C1, R1, NBK1, K1,  
L2, C2, R2, NBK2, K2, AREA)
```

Input Parameters:

X = X coordinate of box center

XLED } = Planform leading edge definition points
YLED }

XTED } = Planform trailing edge definition points
YTED }

L1 } = { Left } Chord edge intersections with the planform
C1 } = { Center } leading edge
R1 } = { Right }

NBK1 = Number of planform leading edge definition points within
the chord

K1 = First leading edge definition point within chord

L2 }
C2 }
R2 } = Same as above for trailing edge
NBK2 }
K2 }

Output Value:

AREA = The desired box area, α

6. MODAL DATA PROCESSOR

Fortran Program MODES

AUTHOR: G. E. Keylon

PURPOSE: To read the modal input data, compute it by a least squares surface fitting routine or evaluate a polynomial equation with coefficients supplied by input and store this information on a scratch file.

METHOD: The information needed to determine the mode shapes is read in. The planform information is read from a scratch file created in the geometry section. The program then computes or reads the modal data at control points, orders the data and writes the data on a scratch file for use in following sections.

Subroutine ROPER is used to compute row pointers for storing box center modal values row-wise. Modal input from tape is handled by TAPMOD. If modal input option 2 was specified, FITTER is called to compute the surface fit polynomial coefficients. PRECOF is called if the coefficients from option 1 or 2 are to be printed. The coefficients are saved on a scratch file for future cycles, and the polynomial is evaluated at box centers, with the results stored on scratch file MODESC.

The program also has an option to read an array of Thickness slope function values derived from "Piston Theory" calculations. These values are input to an equation that computes the thickness correction factor.

$$\bar{Z}_\tau(x,y) = 1 + \frac{\gamma+1}{2} \cdot M \cdot \frac{\partial z_\tau}{\partial x} \quad (1)$$

where γ is ratio of specific heats for a perfect gas (1.40)

M is Mach number.

$\frac{\partial z_\tau}{\partial x}$ is the thickness slope function values.

$\bar{Z}_\tau(x,y)$ is the thickness correction factor.

Input Methods:

(1) Polynomial Coefficient Input

The degree of a surface polynomial equation and the coefficients are read in. The deflection is then computed by the following polynomial equation:

$$\text{Deflection} = a_{00} + \sum_{i=1}^{\# \text{ of degrees}} (a_{i0} x^i y^0 + \dots + a_{0i} x^0 y^i) \quad (2)$$

where, point (X,Y) is the coordinates of a box center in the planform local coordinate system, and a is the array of polynomial coefficients read in. The slope is computed by taking the derivative of the deflection in the X direction.

$$\begin{aligned} \text{Slope} &= \sum_{i=0}^{\# \text{ of degrees}} d(a_{i0} x^i y^0 + \dots + a_{0i} x^0 y^i) / dx \\ &= \sum_{i=0}^{\# \text{ of degrees}} (i a_{i0} x^{i-1} y^0 + \dots + a_{0i} y^i) \end{aligned} \quad (3)$$

These equations are used to compute the deflection and slopes for all the planform boxes. The array of modal values is stored on a scratch file for use in the velocity potential and generalized forces sections of the program.

(2) Interpolation

The degree of a surface polynomial equation, the number of locations where deflections are to be given and the locations and deflections are read in. The deflections are perpendicular to the surface and the (X,Y) locations are input in the planform local coordinate system. The program uses this data to fit a surface polynomial expression in the least squares error approximation. The routine that performs the surface fit is subroutine FITTER. This routine sets up an upper triangular, augmented matrix that represents the set of simultaneous linear equations formed by taking the partial derivatives of each deviation equation squared and setting it to zero. It then solves the set of simultaneous linear equations by using the Choleski square root method given in Reference 1. The solution is an array of polynomial coefficients that are used to compute the modal values in the same manner as method (1).

(3) Modal Values at Box Centers

The values of the deflections and slopes are read in from cards or tape and stored on a scratch file for use in the velocity potential and generalized forces sections of the program. The values are stored in order of boxes within chord, and chords within planform. The order is fore to aft boxes, center most to tip chord and wing before tail. For card input, each chord begins on a new card. All of the mode shape for the wing will be read followed by all of the mode shapes for the tail.

USAGE: The MODES program is the main program of a secondary overlay of the Mach Box program. It is called as follows:

CALL OVERLAY (6HAFMBBX, 1, 3, 0)

Input:

Uses labeled common blocks:

/PROBLM/
/GEOMTY/
/GEOM2/
/FILES/
/IOCONT/
/TAPEIO/
/MODES/
/RWBUFF/

Uses the following files:

IGEOSC

Output:

Output is stored on file:

MODESC

Fortran Subprogram ROPER

AUTHOR: G. E. Keylon

PURPOSE: To compute the row pointers indicating location of planform boxes.

METHOD: The subprogram uses the column pointers and determines the row pointers. The subprogram will also calculate pointers for a tail surface with overlapped planform and store the pointers after the first planform pointers.

USAGE: CALL ROPER

General labeled common blocks used:

/GEOMTY/
/GEOM2/

LOCAL labeled common blocks used:

/INDEX/ IS(100), NOC(100), JS(50), JOC(50)

Common Input:

IS(J) - The ith index of the first planform box on chord J.

NOC(J) - The number of planform boxes on chord J.

Common Output:

JS(I) - The jth index of the first planform box on row I.

JOC(I) - The number of planform boxes on row I.

Fortran Subprogram FITTER

AUTHOR: G. E. Keylon

PURPOSE: To fit a surface in the least squares sense through a set of data points.

METHOD: The fitter routine is passed a set of ordered triplets and the degree of polynomial to fit. It is also given a scale factor if needed to scale the data to prevent the occurrence of arithmetic overflow or underflow. The program can fit real or complex data. The system of simultaneous linear equations that must be solved for employs the Choleski square root method (see Ref. 1). If the polynomial exceeds the maximum capability in either X or Y direction that degree is held and the other direction is allowed to use the full degree.

USAGE: CALL FITTER (M, N, X, Y, Z, C, CN, IDIM)

Input:

M - degree of polynomial equation
N - number of data points to fit curve through
X - Array of X coordinates
Y - Array of Y coordinates
Z - Array of Z coordinates
CN - scale factor
IDIM - Indicator of real or complex function
= 1, function to fit is real
2, function to fit is complex

Output:

C - Output polynomial coefficient array.

Fortran Subprogram MODOUT

AUTHOR: G. E. Keylon

PURPOSE: To print the mode shapes in a manner that the user can readily determine Mach box values of deflections and slopes.

METHOD: The mode shapes are rearranged in a print array so that one row or part of a row will be printed at a time. If there are more than 15 chords on the planform the program prints information for 15 chords, for all rows, and then prints for the next 15 chords until all information has been printed. The values may be scaled before printing to allow values to be printed under F mode Fortran format control. The scaling factor will be indicated in the title.

USAGE: CALL MODOUT (DEFSL, JS, JOC, NROWS, NM, IOVLAP)

Input:

DEFSL - Array of mode shapes
DEFSL(1,I) = deflection
DEFSL(2,I) = slope
JS - Array of pointers to first planform box on each row
JOC - Array of counters for the number of planform boxes on each row.
NROWS - number of rows
NM - Mode shape number
IOVLAP - Number of boxes of overlap between planforms for non-coplanar surfaces.

Output:

None

Fortran Subprogram PRECOF

AUTHOR: G. E. Keylon

PURPOSE: To print the polynomial coefficients used in evaluating mode shapes.

METHOD: The coefficients are printed with each coefficient having over it the corresponding powers of X and Y labeled. All the coefficients for a total power will on one line (i.e., line 1 - 0 power, line 2 - first power, line 3 - second power etc.).

USAGE: CALL PRECOF(IDEV, A, IFR)

Input:

Labeled common block /FILES/

IDEV - Degree of polynomial equation

A - Array of coefficients

IFR - Flag indicating how coefficients are obtained.

= 1, read from cards

= 2, computed by least squares surface fit.

Output:

None

7. AERODYNAMIC INFLUENCE COEFFICIENTS SECTION

Fortran Program VICMAIN

AUTHOR: G. E. Keylon

PURPOSE: To determine all aerodynamics influence coefficients (AIC's) that must be computed or retrieved for a specified reduced frequency.

METHOD: A parameter array is read from the geometry scratch file for each spatial AIC that is needed. The program then determines if an array already exists on permanent tape storage. If it exists the array is read in, expanded if necessary, and stored on scratch file IAICSC if spatial, or in blank common if planar. If calculation is necessary, subroutine KERNEL is called to control the actual computations. KERNEL in turn calls ROMBER to do the integrations of FUNCT and VFUNC.

USAGE: The VICMAIN program is the main program of a secondary overlay of the Mach Box program. It is called as follows:

```
CALL OVERLAY (6HAFMB0X 1, 4, 0)
```

Input:

Uses labeled common blocks

```
/KERN/  
/KVAL/  
/PROBLM/  
/FILES/  
/GEOMTY/  
/IOCONT/  
/ARRAYS/  
/RWBUFF/  
/TAPEIO/
```

Uses the following files

IGEOSC, OSPAIC (optional), OPLAIC (optional)

Output:

Output is stored on files:

NPLAIC, NSPAIC, IAICSC (all optional)

Fortran Subprogram KERNEL

AUTHOR: G. E. Keylon

PURPOSE: To determine the boxes to be integrated and the limits of integration prior to calling the integration routine.

METHOD: The program determines from a parameter array from the geometry scratch file, the intersection of the Mach cone with the planform boxes it is attempting to integrate. It determines what boxes on a row are to be integrated and breaks each box up into a set of integrable limits. It then passes the limits of integration to subroutine ROMBER for integration by the Romberg integration method described in Reference 2.

Box Patterns and Limits:

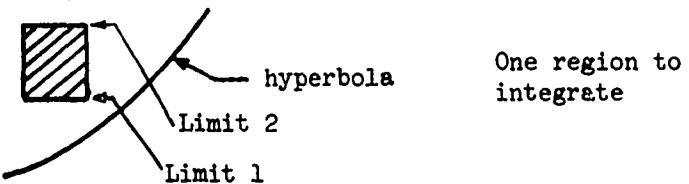


FIGURE 8 AIC Integration, Full Box

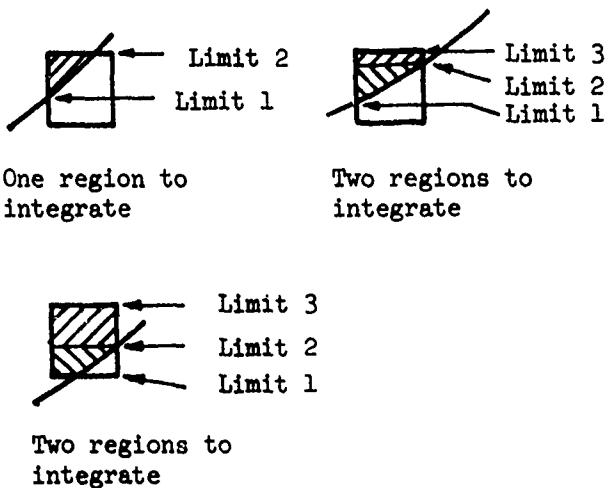


FIGURE 9 AIC Integration Edge Boxes

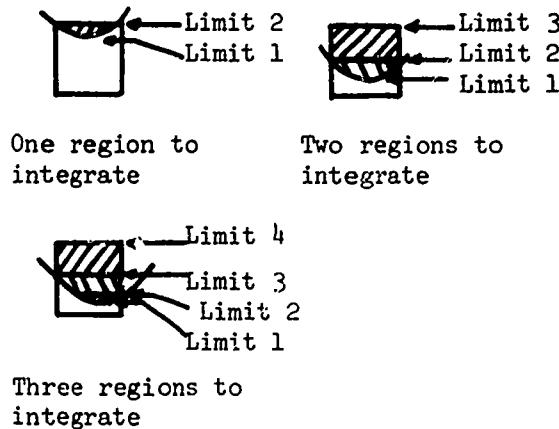


FIGURE 10 AIC Integration Apex Boxes

Box patterns and limits for boxes cut on the left side by the Mach hyperbola are computed in a like manner.

The functions integrated by subroutine ROMBER are those for the velocity potential aerodynamic influence coefficients ($C_{\bar{\nu}\bar{\mu}\lambda}$), the upwash aerodynamic influence coefficients ($W_{\bar{\nu}\bar{\mu}\lambda}$), and the sidewash aerodynamic influence coefficients ($V_{\bar{\nu},\bar{\mu},\lambda}$).

The equations for $C_{\bar{\nu}\bar{\mu}\lambda}$ and $W_{\bar{\nu}\bar{\mu}\lambda}$ are:

$$C_{\bar{\nu}\bar{\mu}\lambda} = -\frac{1}{\pi} \int_{\xi l}^{\bar{\xi} u} e^{-i\bar{k}_1 \bar{\xi}} \left\{ J_0 \left(\frac{\bar{k}_1}{M} \sqrt{\bar{\xi}^2 - l^2} \right) \left[\sin^{-1} \left(\frac{\bar{\eta}_u}{\sqrt{\bar{\xi}^2 - l^2}} \right) - \sin^{-1} \left(\frac{\bar{\eta}_L}{\sqrt{\bar{\xi}^2 - l^2}} \right) \right] + \sum_{r=1}^{\infty} \frac{(-1)^r}{r} J_{2r} \left(\frac{\bar{k}_1}{M} \sqrt{\bar{\xi}^2 - l^2} \right) \left[\sin \left(2r \sin^{-1} \left(\frac{\bar{\eta}_u}{\sqrt{\bar{\xi}^2 - l^2}} \right) \right) - \sin \left(2r \sin^{-1} \left(\frac{\bar{\eta}_L}{\sqrt{\bar{\xi}^2 - l^2}} \right) \right) \right] \right\} d\bar{\xi}$$

$$\begin{aligned}
w_{\bar{\mu} \bar{\lambda}} = & \frac{1}{\pi} \left[\int_{\bar{\xi}_L}^{\bar{\xi}_U} e^{-i\bar{k}_1 \bar{\xi}} \left\{ J_0 \left(\frac{\bar{k}_1}{M} \sqrt{\bar{\xi}^2 - \ell^2} \right) \left[\sin^{-1} \left(\frac{\bar{\eta}_u}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) - \right. \right. \right. \right. \\
& \left. \left. \left. \left. \sin^{-1} \left(\frac{\bar{\eta}_L}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) \right] + \sum_{r=1}^{\infty} \frac{(-1)^r}{r} J_{2r} \left(\frac{\bar{k}_1}{M} \sqrt{\bar{\xi}^2 - \ell^2} \right) \cdot \right. \right. \\
& \left. \left. \left. \left. \left[\sin \left(2r \sin^{-1} \left(\frac{\bar{\eta}_u}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) \right) - \sin \left(2r \sin^{-1} \left(\frac{\bar{\eta}_L}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) \right) \right] \right\} \frac{1+i\bar{k}_1 \bar{\xi}}{\bar{\xi}^2} d\bar{\xi} \right. \right. \\
& \left. \left. \left. \left. + \left| \frac{e^{-i\bar{k}_1 \bar{\xi}}}{\bar{\xi}} \left\{ J_0 \left(\frac{\bar{k}_1}{M} \sqrt{\bar{\xi}^2 - \ell^2} \right) \left[\sin^{-1} \left(\frac{\bar{\eta}_u}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) - \sin^{-1} \left(\frac{\bar{\eta}_L}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) \right] \right. \right. \right. \right. \right. \\
& \left. \left. \left. \left. \left. \left. + \sum_{r=1}^{\infty} \frac{(-1)^r}{r} J_{2r} \left(\frac{\bar{k}_1}{M} \sqrt{\bar{\xi}^2 - \ell^2} \right) \left[\sin \left(2r \sin^{-1} \left(\frac{\bar{\eta}_u}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) \right) \right. \right. \right. \right. \right. \right. \\
& \left. \left. \left. \left. \left. \left. - \sin \left(2r \sin^{-1} \left(\frac{\bar{\eta}_L}{\sqrt{\bar{\xi}^2 - \ell^2}} \right) \right) \right] \right\} \left[\begin{array}{l} \bar{\xi}_U \\ \bar{\xi}_L \end{array} + e^{-i\bar{k}_1 \bar{\xi}} \frac{\pi}{\ell} \right] \right] \right] \quad (5)
\end{aligned}$$

The $C_{\bar{\mu} \bar{\lambda}}$ equations is valid for all types of boxes, which are shown in Figures 8, 9, and 10. The $W_{\bar{\mu} \bar{\lambda}}$ equation, as written, is valid for the region of an apex box shown in Figure 10, that has the hyperbola as a boundary on both sides. The last term is zero for regions that have the hyperbola as a boundary for one side and the box edge as the other side boundary. The last term and the terms evaluated at the integration limits are zero for full boxes or regions bounded on both sides by the box edges. The values of the integrand used in ROMBER are computed by subroutine FUNCT which also calls subroutines RANGE and BESSEL to evaluate the Bessel functions.

The function $V_{\bar{\mu} \bar{\lambda}}$ is considerably different and holds for all regions to be integrated. The evaluation of the integral is done by subroutine VFUNC. The equation is:

$$\begin{aligned}
 v_{\bar{\nu}\bar{\mu}\lambda} = -\frac{M}{\pi k_1} & \left[\left| \frac{e^{-ik_1\bar{\xi}}}{\bar{\xi}} - \frac{1+ik_1\bar{\xi}}{\bar{\xi}^2} \left(\sin\left(\frac{k_1}{M}\sqrt{\bar{\xi}^2 - \bar{\eta}_u^2 - l^2}\right) \right. \right. \right. \\
 & \left. \left. \left. - \sin\left(\frac{k_1}{M}\sqrt{\bar{\xi}^2 - \bar{\eta}_L^2 - l^2}\right)\right) \right] \left[\frac{\bar{\xi}_u}{\bar{\xi}_L} + \int \frac{\bar{\xi}_u}{\bar{\xi}_L} \left[e^{-ik_1\bar{\xi}} \frac{1+ik_1\bar{\xi}}{\bar{\xi}^2} \right. \right. \\
 & \left. \left. \left(\sin\left(\frac{k_1}{M}\sqrt{\bar{\xi}^2 - \bar{\eta}_u^2 - l^2}\right) - \sin\left(\frac{k_1}{M}\sqrt{\bar{\xi}^2 - \bar{\eta}_L^2 - l^2}\right)\right) \right] d\bar{\xi} \quad (6)
 \end{aligned}$$

For cases where $\bar{k}_1 = 0$, the equation becomes

$$v_{\bar{\nu}\bar{\mu}\lambda} = -\frac{1}{\pi} \left[\left[\cosh^{-1} \frac{\bar{\xi}}{\sqrt{\bar{\eta}_u^2 + \bar{\lambda}^2}} - \cosh^{-1} \frac{\bar{\xi}}{\sqrt{\bar{\eta}_L^2 + \bar{\lambda}^2}} \right] \right] \left[\frac{\bar{\xi}_u}{\bar{\xi}_L} \right] \quad (7)$$

USAGE: CALL KERNEL(XMACH,K1,ERR,C,W,V)

Input:

XMACH Mach number
 K1 Reduced frequency
 ERR Convergence criteria (relative not absolute)
 Labeled Common Block /VICPAR/

Output:

C Velocity potential aerodynamic influence coefficients.
 W Upwash aerodynamic influence coefficients.
 V Sidewash aerodynamic influence coefficients.

Fortran Subprogram ROMBER

AUTHOR: G. E. Keylon

PURPOSE: To integrate the aerodynamic influence coefficient functions.

METHOD: The program uses the Romberg integration technique (Ref. 2). The technique is a modified trapezoidal area method with an extrapolation method added. For analytical cases the sidewash aerodynamic influence coefficient will be solved by an analytic equation, not by numerical approximation.

USAGE: CALL ROMBER (XILL, XILU, IUC, ERR, IFLAG, K1BAR, YMUBAR, EL, XMACH
C, W, V)

Input:

XILL - Lower limit of integration
XILU - Upper limit of integration
IUC - Flag indicating type of box or edge condition of interval
to be integrated = 0, full box
1, left side of box is edge of Mach hyperbola.
2, right side of box is edge of Mach hyperbola.
3, both sides of box are edges of Mach hyperbola.

ERR - Convergence criteria (relative, not absolute)
IFLAG - Indicator of real or imaginary parts
= 0, real part
= 1, imaginary part
K1BAR - Function of reduced frequency and Mach number, = $\frac{K_1 M^2}{(M^2 - 1)}$
YMUBAR - Parallel offset of pulse sending box.
EL - Normal offset of receiving point from sending plane.
XMACH - Mach number

Output:

C - Velocity potential aerodynamics influence coefficient, $C_{\bar{\nu} \bar{\mu} \lambda}$
W - Upwash aerodynamic influence coefficient, $W_{\bar{\nu} \bar{\mu} \lambda}$
V - Sidewash aerodynamic influence coefficient, $V_{\bar{\nu} \bar{\mu} \lambda}$

Fortran Subprogram FUNCT

AUTHOR: G. E. Keylon

PURPOSE: To evaluate the velocity potential and upwash aerodynamic influence coefficient functions for a set of independent variables.

METHOD: An array XI of independent variables is passed to the program through the calling sequence. The program evaluates the function at each point first checking for boundary conditions where the function approaches a singularity. Routines to find the range of and value of Bessel functions are called in the evaluation of the function.

USAGE: CALL FUNCT (K, XI, FXIC, FXIW, IFLAG, K1BAR, EL, YMUBAR, IUC, XMACH,
BESSY)

Input:

K - Number of functions to evaluate
XI - Array of independent variables
IFLAG - Indicator of real or imaginary parts
= 0, real part
= 1, imaginary part
K1BAR - Function of reduced frequency and Mach number, $K_1 M^2 / (M^2 - 1)$
EL - Normal offset of receiving point from sending plane.
YMUBAR - Parallel offset of pulse sending box.
IUC - Flag indicating type of box or edge condition of interval
to be integrated.
XMACH - Mach number

Output:

FXIC - Function values for Velocity Potential AIC.
FXIW - Function values for Upwash AIC.
BESSY - Evaluation at end points for upwash AIC.

Fortran Subprogram BESSEL

AUTHOR: G. E. Keylon

PURPOSE: To evaluate the Bessel functions for a given argument over a range of orders.

METHOD: The argument and range (# of terms or order) is passed to the routine. The routine then calculates the required terms and places them in an array and returns.

USAGE: CALL BESSEL (K12, AV, NA)

Input:

K12 - The argument, a function of independent variable, Mach number and reduced frequency.

NA - Highest order of the Bessel function to be evaluated.

Output:

AV - Array containing the Bessel functions.

Fortran Subprogram RANGE

AUTHOR: G. E. Keylon

PURPOSE: To determine the range (or order) of a Bessel function with a given argument.

METHOD: An order, or equation for an order, is given for various increments of arguments. This routine determines which interval the argument is in and computes the order.

USAGE: CALL RANGE (K12, NA)

Input:

K12 - The argument, function of independent variable, Mach number and reduced frequency.

Output:

NA - Highest order of the Bessel function to be evaluated.

Fortran Subprogram VFUNC

AUTHOR: G. E. Keylon

PURPOSE: To evaluate the sidewash aerodynamic influence coefficient function for a set of independent variables.

METHOD: An array of independent variables is passed to the program through the calling sequence. The program evaluates the function at each point, first checking for boundary conditions where the function approaches a singularity.

USAGE: CALL VFUNC (K, XI, FXIV, IFLAG, K1BAR, EL, YMUBAR, INC, XMACH, IND, VT)

Input:

K - Number of values to calculate
XI - Array of independent variables
IFLAG - Flag indicating real or complex part
= 0, real part
= 1, imaginary part
K1BAR - Function of reduced frequency and Mach number, $K_1 M^2 / (M^2 - 1)$
EL - Normal offset of receiving box above sending plane.
YMUBAR - Parallel offset of pulse sending box.
INC - Flag indicating type of box or edge condition of interval
to be integrated.
XMACH - Mach number
IND - Indicator to calculate VT terms
= 0, do not calculate
= 1, calculate

Output:

FXIV - Function values for sidewash AIC.
VT - Extra terms calculated at the limits of integration.

8. NORMAL-WASHES AND VELOCITY POTENTIALS

Fortran Program NWVLPT

AUTHOR: G. D. Kramer

PURPOSE: To compute normal washes and associated velocity potentials for each oscillatory mode shape at box centers. Wake sampling of upwash, sidewash and longitudinal wash is also provided.

METHOD: The necessary box patterns and other geometric items are first read in from the scratch file IGEOSC. The mode shape and velocity potential pointer array IPNTRM is read from scratch file MODESC, and a pointer array for normal-washes, IPNTDW, is generated by subroutine POINTR. These pointer arrays serve to associate a box location in a sparsely filled rectangular array with the corresponding mode, velocity potential or normal wash value in a singly dimensioned, densely filled array.

A loop on mode shapes is entered next. The box center deflections and shapes are read from MODESC into array DEFSL. Subroutine VELPOT is called for the wing to compute N_{RW} , N_{RUW} , and $\Delta\bar{\phi}$ at box centers, and trailing edge $\Delta\bar{\phi}$ values in array TVP. If a tail is being analyzed as well, the contributing wing normal-washes are determined and VELPOT is called again. Optional printing of N_{RUW} , etc. and $\Delta\bar{\phi}$ is done in routine PRINTR.

If sampling of wake washes is desired, subroutine SMPLW is called to compute and print these results.

The $\Delta\bar{\phi}$ array VELPOT and the TVP array are written on scratch file IVPSC for each mode shape.

USAGE: The DWVLPT program is the main program of a secondary overlay of the Mach box program. It is called as follows:

CALL OVERLAY (6HAFMB0X, 1, 5, 0)

Input: Uses labelled common blocks

/CTRL/	/FILES/
/PROBLM/	/IOCONT/
/GEOMTY/	/TAPEIO/
/GEOM2/	/MODES/
/KERN/	/ARRAYS/
/KVAL/	/SAMPLW/

Uses scratch files

IGEOSC
IMODESC

Fortran Subroutine POINTR

AUTHOR: G. E. Keylon, G. D. Kramer

PURPOSE: To generate part or all of a pointer array which indexes another array of box associated values (modes, normal-washes, etc.) stored compactly, row-wise.

METHOD: The box codes are scanned to determine the first box of interest and the number of boxes of interest on each row. From this, the pointer array is generated such that IPNTR(1,i) = the location of the first box value for row i, and IPNTR (2, i) = the chord number of the first box value for row i.

USAGE: The routine is called by:

```
DIMENSION IBOX (LBXCD, # chords/10), IPNTR (2, MXIR)
LOGICAL DIAPH, SUBD, WING
```

```
CALL POINTR (IX, MX, MYB, IOVLAP, SUBD, DIAPH, IBOX, LBXCD, MXIR,
IPOINT, IPNTIN, IPNTR)
```

Input Parameters:

IX = First row of the box pattern for which the pointer array is desired.
MX = Number of rows desired.
MYB = Maximum row length
IOVLAP = Number of rows to allow for overlap (tail only).
SUBD = .T., a pointer array for subdivided boxes is desired
= .F., only unsubdivided box information is desired.
DIAPH = .T., boxes in diaphragm areas are to be included.
= .F., only on-planform boxes are of interest though space may be left within a row if imbedded diaphragm areas occur.
IBOX = Array of subdivided box codes generated in the geometry section.
LBXCD = Length of box code array.
MXIR = Length of IPNTR array, used to control end-around buildup of the array.
IPOINT = Value to be used for first pointer; 1 if IX=1, else the next location available in the array "pointed to" for row IX.

In/out Parameters:

IPNTIN = Location of next available cell in the IPNTR array. This will be incremented for each row processed until MXIR is reached, when it is reset to 1.

IPNTR (2, MXIR) = The pointer array, see Method above.

FORTRAN Subroutine GETAIC

Author: G. D. Kramer

Purpose: To get the desired Aerodynamic Influence Coefficient (AIC) arrays from scratch file IAICSC.

Method: From the calling sequence, the location of the desired AIC array(s) is determined. If they are in core, the routine returns. If there are none, the error flag is set. Otherwise, the disk file is positioned, and the desired arrays are read into local common block AICS.

Usage: EL, YBAR, NROWS, MUAIC (2,50) are in a common block, MUAICS for output from GETAIC
NWWAIC, NTTAIC, NRWTAIC, NLWTAIC and PAIC (4,50) are in a common block /PAICS/, for use by the routine.
CALL GETAIC (JUCENT, ITYPE, ICODE, IR)

Input Parameters:

JUCENT = receiving chord number
ITPE = 1, 2, 3, 4 indicating wing-wing, tail-tail,
right-wing-tail, or left-wing-tail AIC's
desired
ICODE = 0, C,V,W desired
1, V,W desired
2, V desired

Common Input:

NWWK }
NTTK } = Number of AIC arrays avail- { wing-wing
NRWTK } able for influence. { tail-tail
NLWTK } { right wing-tail
left wing-tail

PAIC (4,50) = Table of contents for the AIC's.
PAIC (I,J) indicates where the AIC's
for the Ith form of influence (see
above) on the Jth chord are located.

Output Parameters:

IR = D, Success
1, C not found
2, C and W not found
3, Nothing found

Common Output:

C = C---
W = W_{Vμω}
V = V_{Vμω}

Fortran Subroutine VELPOT

AUTHOR: G. D. Kramer

PURPOSE: To compute normal wash and velocity potential values for one mode shape.

METHOD: This routine calculates the following equations:

$$\frac{D\bar{f}_j}{Dt}^{n,m} = \left[i k_i f_j^{n,m} + b_i \frac{\partial f_j}{\partial x}^{n,m} \right] \quad (8)$$

(1) For the wing:

$$N_{RUW}^{n,m} = \frac{D\bar{f}_i}{Dt} - \hat{N}_{LUW}^{n,m} \quad , \quad N_{RLW}^{n,m} = -\frac{D\bar{f}_i}{Dt} - \hat{N}_{LLW}^{n,m} \quad (9)$$

where

$$\hat{N}_{LUW}^{n,m} = \sum_{\substack{\text{left wing} \\ + diaphragm}} \left[\cos 2\psi_w W_{\bar{D}\bar{U}\bar{\lambda}}^{(RW)} - \sin 2\psi_w V_{\bar{D}\bar{U}\bar{\lambda}}^{(RW)} \right] N_{RUW}^{n,m} * \text{SYM} \quad (10)$$

and

$$\hat{N}_{LLW}^{n,m} = - \sum_{\substack{\text{left wing} \\ + diaphragm}} \left[\cos 2\psi_w W_{\bar{D}\bar{U}\bar{\lambda}}^{(RW)} - \sin 2\psi_w V_{\bar{D}\bar{U}\bar{\lambda}}^{(RW)} \right] N_{RLW}^{n,m} * \text{SYM} \quad (11)$$

$$\text{and SYM} = \begin{cases} +1.0, \text{ symmetric} \\ -1.0, \text{ antisymmetric} \end{cases}$$

(2) For leading edge or tip diaphragm boxes:

$$\begin{aligned} (N_{RUS} - N_{LUS})^{n,m} &= \frac{1}{C_{\text{box}}} \left[- \sum_{\substack{\text{right wing} \\ + diaphragm}} C_{\bar{D}\bar{U}\bar{\lambda}} (N_{RUS} - N_{LUS})^{n,m} \right. \\ &\quad \left. + \text{SYM} \times \sum_{\substack{\text{left wing} \\ + diaphragm}} C_{\bar{D}\bar{U}\bar{\lambda}} (N_{RUW}^{n,m} - N_{RLW}^{n,m}) \right] \quad (12) \end{aligned}$$

(3) For wake diaphragm boxes:

$$(N_{RUS}^{n,m} - N_{RLS}^{n,m})^{n,m} = \frac{1}{C_{cos}} \left[\Delta \bar{\phi}_j^{n,m} - \sum_{\substack{\text{right wing} \\ \text{+ diaphragm}}} C_{D,20} (N_{RUS}^{uu} - N_{RLS}^{uu})^{uu} + \text{SYM} + \sum_{\substack{\text{left wing} \\ \text{+ diaphragm}}} C_{D,20} (N_{RUS}^{uu} - N_{RLS}^{uu})^{uu} \right] \quad (13)$$

where

$$\Delta \bar{\phi}_j^{n,m} = \Delta \phi_{TE}^m e^{-i k_j (\frac{x_n - x_{TE}}{b_i})} \quad (14)$$

$$(N_{RUS}^{n,m} + N_{RLS}^{n,m}) = - (\hat{N}_{RLS}^{n,m} + \hat{N}_{RLS}^{n,m}) \quad (15)$$

(4) For the tail:

$$(N_{RUT}^{n,m} - N_{RLT}^{n,m})^{n,m} = 2 \left(\frac{Df^{n,m}}{Dt} \right) + (\hat{N}_{LLT}^{n,m} - \hat{N}_{LUT}^{n,m}) - 2 \hat{N}_{RW}^{n,m} - 2 \hat{N}_{LW}^{n,m} \quad (16)$$

where \hat{N}_{RUT} and \hat{N}_{RLT} are computed as in Equation (11) and (12).

$$\hat{N}_{RW}^{RUT} = \sum_{\substack{\text{rt wing} \\ \text{+ diaph.}}} [\cos(\psi_T - \psi_w) W_{\bar{D}\bar{U}\bar{\lambda}}^{(RW)} - \sin(\psi_T - \psi_w) V_{\bar{D}\bar{U}\bar{\lambda}}] N_{RW}^{uu} \quad (17)$$

$$\hat{N}_{LW}^{RUT} = \text{SYM} \sum_{\substack{\text{left wing} \\ \text{+ diaph.}}} [\cos(\psi_T + \psi_w) W_{\bar{D}\bar{U}\bar{\lambda}}^{(LW)} - \sin(\psi_T + \psi_w) V_{\bar{D}\bar{U}\bar{\lambda}}] N_{LW}^{uu} \quad (18)$$

(5) Velocity potentials:

$$\Delta \bar{\phi}_j^{n,m} = \sum_{\substack{\text{right surface} \\ \text{and diaphragm}}} C_{\bar{\nu}\bar{\mu}0} (N_{RUS}^{\bar{\nu}\bar{\mu}} - N_{RLS}^{\bar{\nu}\bar{\mu}}) + \sum_{\substack{\text{left surface} \\ \text{and diaphragm}}} C_{\bar{\nu}\bar{\mu}1} (N_{RUS}^{\bar{\nu}\bar{\mu}} - N_{RLS}^{\bar{\nu}\bar{\mu}}) * \text{SYM} \quad (19)$$

Because the equations involve summations over unknown values, the order of calculation is very critical. The routine computes normal washes and velocity potentials in parallel, one row at a time, inboard-most box first. If the subdivision option is on, each subdivided box must have a set of normal washes computed as well, using equations similar to those above.

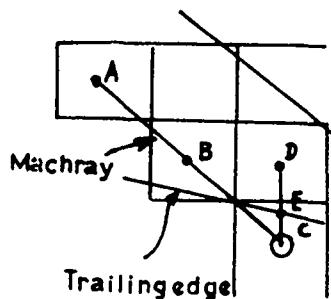
For each box, the \hat{N} terms are first zeroed out. If spatial contribution is present, subroutine GETAIC is called to get the necessary AIC arrays from scratch file IAICSC, and the proper summation is computed over the forward Mach hyperbola. This is first done for \hat{N}_{RUS} and \hat{N}_{RLS} and stored in variables ENRULU, ENRLL. LUS LLS

If the surface is a tail, a similar procedure of getting AIC arrays and computing the proper summation for $\hat{N}_{RUT}^{\text{RW}}$ and $\hat{N}_{RUT}^{\text{LW}}$ is followed. The results are stored in ENRURW and ENRULW.

If the box being considered is a planform box, the normal wash values are next computed from Equations (9) or (16).

Function B is called to compute the planar $\Delta \bar{\phi}$ contribution, except for the contribution of the box to itself. This is stored in variable DELPH. If the box is on planform, the out-of-plane contribution is added, yielding Equation (19). If the box is on a diaphragm, DELPH is used in Equation (12) or (13) to eventually yield the normal wash values at the diaphragm box center.

Trailing edge velocity potentials, array TVP, are computed whenever a trailing edge box is encountered. The computation is normally linear extrapolation from the last two box center values. In the event there is only one box on the tip chord, a Mach ray extrapolation is first done, followed by chord-wise linear interpolation. See Figure 11.



Values at A and B are extrapolated to C. Then the values at C and D are interpolated to give a value at E, the desired trailing edge value.

FIGURE 11 Tip Chord Trailing Edge Velocity Potential Calculation

The subdivision option causes the following:

- (1) All row and column loops are on subdivided boxes.
- (2) Any necessary \hat{N} terms are calculated once per control point, and stored in temporary arrays for use on all subdivided boxes within the unsubdivided box. \hat{N} terms and spatial contribution of left surface to $\Delta\phi$ are not calculated using subdivided values.
- (3) Function B and $\bar{\Delta}\phi$ are not computed for on-planform subdivided boxes which do not contain a control point.
- (4) Function B, when called, applies two equations - one within the "effective area" of subdivision, and the other outside this area. It is within function B that the subdivision refinement actually takes place.
- (5) Any unsubdivided box which has one or more off-planform subdivided boxes has its normal wash values computed as the average of all subdivided values within its bounds, i.e.

$$N_{RUS}^{n,m} = \left(\sum_{\substack{\text{all subdivided} \\ \text{boxes on box } n,m}} N_{RUS}^s \right) / NSUBDV$$

USAGE: The subroutine is called by:
CALL VELPOT(IBOX,LBXCD,PKERNL,SKERNL,WING,DIHS)

Input Parameters:

IBOX Array of box codes for the surface.
LBXCD Length of the box code array.
PKERNL Primary (unsubdivided) $C_{\bar{V}\bar{H}O}$ array.
SKERNL Subdivided $C_{\bar{V}\bar{H}O}$ array.
 K_1/N_s
WING .TRUE., the surface is a wing.
.FALSE., the surface is a tail.
DIHS .TRUE., any surface dihedral is to be accounted for.
.FALSE., any surface dihedral may be ignored.

Input Common Variables:

Global common blocks used:

/GEOMTY/
'GEOM2/
/MODES/
/FILES/
/CHECKPR/
Blank Common for C

Local common values:

/MUAICS/YBAR	Parallel offset
EL	Normal offset
MUAIC(2,50)	AIC pointer array determined in the geometry section.
NROWS	Number of rows defined for the AIC set.
/AICS/ XKVL	Current value of K_1
C	C
W	W
V	V
/DELTAP/{TEXLOC} {FEXLOC}	{leading } edge X-locations at chord centers {trailing}
IPNTRM	Pointer array for modes and velocity potentials.
DEFSL	Mode shape array - equivalenced to velocity potential array.
IOVLAP	Measure of tail overlap of wing, box mode shapes.
/NWASHES/ IPNTDW	Pointer array for normal wash values.
/BXCDES/ IBOXW	Wing box codes

Output Common Variables:

/DELTAP/	DELPHI	$\Delta \bar{\phi}$ array
	TVP	$\Delta \bar{\phi}_{TE}$ array
/NWASHES/	ENRUS	N_{RUW} or N_{RUT}
	ENRLS	N_{RLW} or N_{RLT}
	IOVLAPN	Measure of tail overlap of wing diaphragm, for normal washes.
/SNWASH/	IPNTSD	Pointer array for subdivided normal washes
	ENSUBD	$N_{RUS}^{(S)}$ and $N_{RLS}^{(S)}$
	IPNTIN IPNTOT IPNTLS	{ End-around pointers for array IPNTSD }

Fortran Function B

AUTHOR: G. D. Kramer

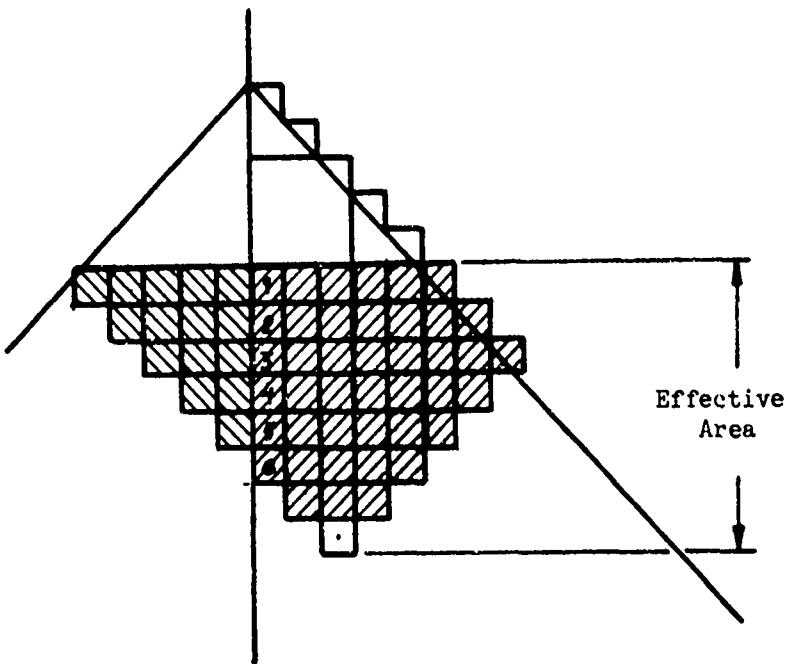
PURPOSE: Given the location of a Mach box, to compute the planar contribution of the rest of the surface to the velocity potential difference for the box.

METHOD: The routine has two sections, one for computing the subdivided contribution within the "effective area," and the other to compute the unsubdivided contribution from ahead of the "effective area". If the subdivision option is off, the second section is used for the full contribution.

In the first section, the summation performed is

$$B_s = \sum_{\bar{\nu}} \sum_{\bar{\mu}} C_{\bar{\nu}, \bar{\mu}, 0}^{k_s/N_s} (N_{RUS}^{(s)} - N_{RLS}^{(s)}) \quad (21)$$

where the summation limits are as shown in Figure 12.



Always contributes to the summation
 Contributes if the dihedral angle $\gamma = 0$

FIGURE 12 Subdivided "Effective Area"

The second section starts up where the first leaves off, and computes the second summation in Equation (22)

$$B = B_s + \sum \sum C_{\bar{\nu} \bar{\mu}_0} (N_{RUS} - N_{RLS}) \quad (22)$$

where the summation limits are as shown in Figure 13 and the AIC array and normal wash values are now unsubdivided, computed at control points.

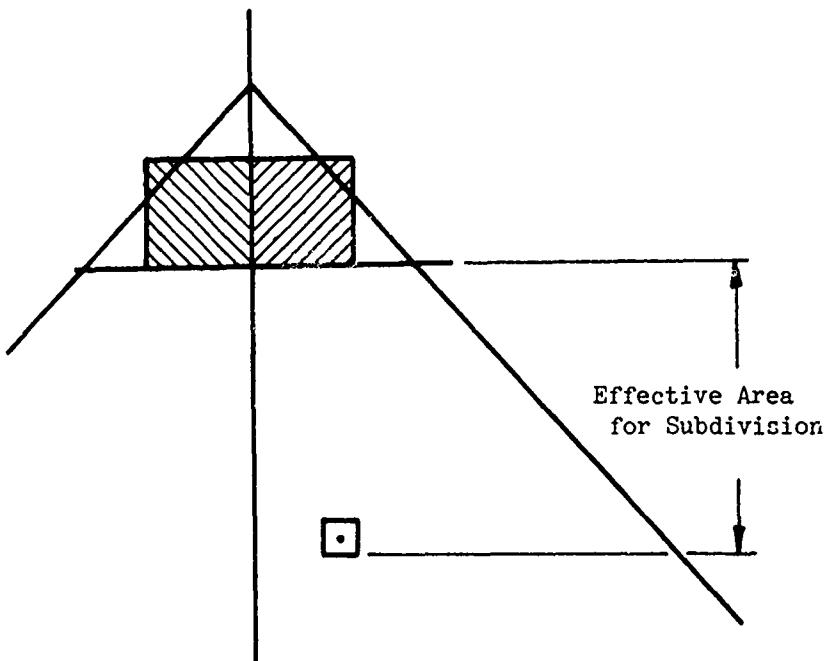


FIGURE 13. Unsubdivided Boxes Outside the "Effective Area"

USAGE: The function is called by:

DELPH = B(IROW,JCOL,PKERNL,SKERNL,IBOX,LBXCD,WING,DIH)

Input Parameters:

IROW } Location of receiving point, subdivided
JCOL }
PKERNL Primary (unsubdivided) $C_{\bar{\nu}\mu_0}$ array
SKERNL Subdivided $C_{\bar{\nu}\mu_0}^{k_i/N_i}$ array
IBOX Box code array
LBXCD Length of box/code array
WING .TRUE., the surface is the wing
.FALSE., the surface is the tail
DIH .TRUE., leftside is to be ignored
.FALSE., Include left side.

Input Common Parameters:

See subroutine VELPOT. Both subdivided and unsubdivided values are used.

Output:

The function value, B, in this case stored in DELPH, is the result of the summations described under METHOD.

Fortran Subroutine SMPLW

AUTHOR: G. D. Kramer

PURPOSE: To compute and print upwashes, sidewashes, and longitudinal washes at arbitrary chord locations in the wake of a wing.

METHOD: This routine is called once for each sampling chord. For each box on the chord, the right wing contribution is summed as

$$WSUM = \sum_{rt. wing} w_{\bar{\nu} \bar{\mu} \bar{\lambda}} * N_{RW}^{\nu \mu} \quad (23)$$

$$VSUM = \sum_{rt wing} v_{\bar{\nu} \bar{\mu} \bar{\lambda}} * N_{RW}^{\nu \mu} \quad (24)$$

$$\PHISUM = \sum_{rt wing} c_{\bar{\nu} \bar{\mu} \bar{\lambda}} * N_{RW}^{\nu \mu} \quad (25)$$

where $N_{RW} = \begin{cases} N_{RUW} & \text{if the chord is above the wing,} \\ N_{RLW} & \text{if the chord is below the wing.} \end{cases}$

These sums are then combined as:

$$UW_R = \frac{u}{U} = \left(\frac{1}{b_1} \right) * \left(\cos \psi_w * WSUM + \sin \psi_w * VSUM \right) \quad (26)$$

$$SW_R = \frac{v}{U} = \left(\frac{1}{b_1} \right) * \left(\cos \psi_w * VSUM - \sin \psi_w * WSUM \right) \quad (27)$$

$$\Phi_R = \PHISUM$$

The left wing contributing summations are identical to Equations (23), (24), and (25), with N_{RW} replaced by N_{LW} .

The results are then combined by

$$UW_{complete} = UW_R + \left(\frac{1}{b_1} \right) * \left(\cos \psi_w * WSUM - \sin \psi_w * VSUM \right) * SYM \quad (28)$$

$$SW_{complete} = SW_R + \left(\frac{1}{b_1} \right) * \left(\cos \psi_w * VSUM - \sin \psi_w * WSUM \right) * SYM \quad (29)$$

$$\text{PHI}_{\text{complete}} = \text{PHI}_R + \text{PHISUM} * \text{SYM} \quad (30)$$

The printed upwash and sidewash is given by Equations (28) and (29). For longitudinal wash, the PHI values computed in Equation (30) are used in

$$LW(I) = \frac{1}{2 b_I \beta} [\text{PHI}(I+1) - \text{PHI}(I-1)] \quad (31)$$

USAGE:

The routine is called by:

```
CALL SMPLW(IBOX,LBXCD,JCHRD,JT,IFRST,ILAST)
```

Input Parameters:

IBOX	Array of wing box codes
LBXCD	Length of box code array
JCHRD	Sample wash chord number, with reference to the order specified in the card data
JT	The y-location of the chord
IFRST	Number of the first sample box desired
ILAST	Number of the last sample box desired

9. VELOCITY POTENTIAL SMOOTHING SECTIONS

FORTRAN Program SMTH

Author: G. E. Keylon

Purpose: To smooth the velocity potentials by using a least squares surface fitting technique.

Method: The velocity potentials are read in from a disk file and smoothed with a least squares fit by subroutine FITTER, previously described. The polynomial equation derived from the fit is then used to compute an array of velocity potentials at planform box centers.

Usage: The SMTH program is the main program of a secondary overlay of the Mach box program. It is called as follows:

```
CALL OVERLAY (6HAFMB0X, 1, 6, 0)
```

Input:

USES LABELLED COMMON BLOCKS

```
/ARRAYS/  
/FILES/  
/IOCONT/  
/PROBLM/  
/KVAL/  
/GEOM1/  
/GEOM2/  
/TAPEIO/  
/RWBUFF/
```

Uses the following files
MODESC, IGEOSC, IVPSC

Output:

Output is stored on file IWTFSC which is changed to IVPSC.

FORTRAN Program CRDFIT

Author: G. E. Keylon

Purpose: To smooth the velocity potentials by using a least squares curve fit along each chord.

Method: The velocity potentials are read in from a disk file. The values for each chord are then separated into an array. The values are then changed to the numerical slope between the midpoint average values. Subroutine CURVE is then called to fit a least squares polynomial curve though these slopes. The polynomial equation is then integrated at each box on the chord and the integral value becomes the velocity potential at that box.

Usage: The CRDFIT program is the main program of a secondary overlay of the Mach box program. It is called as follows:

```
CALL OVERLAY (6HAFMBOX, 1, 7, 0)
```

Input:

```
USES LABELED COMMON BLOCKS
```

```
/ARRAYS/
/FILES /
/IOCONT/
/PROBLM/
/KVAL /
/GEMTY/
/GEM2 /
/TAPEIO/
/RWBUFF/
```

Uses the following files

```
MODESC, IGEOSC, IVPSC
```

Output:

```
Output is stored on file IWTFSC which is changed to IVPSC.
```

FORTRAN Subprogram CURVE

Author: G. E. Keylon

Purpose: To fit a curve in the least squares sense through a set of data points.

Method: The CURVE routine is passed a set of ordered complex pairs and the degree of polynomial to fit. The system of simultaneous linear equations is solved employing the Choleski square root method (see Ref. 1). If the polynomial degree exceeds the limits possible to fit the degree is reduced to a lower level.

Usage: CALL CURVE (M,N,X,Z,C)

Input:

M - degree of polynomial equation
N - number of data points to fit curve through
X - Array of X coordinates (independent variable)
Z - Array of Z coordinates (dependent variable, complex)

Output:

C - output polynomial coefficient array, complex

10. GENERALIZED AIR FORCES SECTION

Fortran Program FORCES

AUTHOR: G. E. Keylon, G. D. Kramer

PURPOSE: To calculate the boxlifts, section lifts, and generalized air forces for a problem.

METHOD: Planform information is first read from the geometry and modes scratch files. The outer-most loop on thickness slope functions is then entered. One set of thickness slope functions, defined at box centers by Equation (1), is read in from scratch file ITSLSC. Next a loop on mode shapes, used as weighting functions for the generalized forces calculations, is entered. One mode shape is read from scratch file MODESC.

The third loop entered is on velocity potentials. The $\bar{\Delta}\phi$ array is read into DELPHI and $\bar{\Delta}\phi_{TE}$ into array TVP from scratch file IVPSC. The box pattern for each surface is then passed over, one row at a time. For each box the following values are computed:

$$\bar{L}_j^{nm} = BXLIFT(1DC) = \frac{2}{\beta} [\bar{\Delta}\phi_{jTE}^{nm} - \bar{\Delta}\phi_{jLE}^{nm} + i\alpha^{nm} k_1 \bar{\Delta}\phi_j^{nm}] \bar{Z}_\tau^{nm} \quad (3.1)$$

$$\Delta c_{pj}^{nm} = DELCP(1DC) = \bar{L}_j^{nm} / (\alpha^{nm} * b_1) \quad (3.2)$$

$$\bar{Q}_{ij}^{nm} \beta_2 = [(f_{TE}^{nm} * \bar{\Delta}\phi_{TE}^{nm} - f_{LE}^{nm} * \bar{\Delta}\phi_{LE}^{nm}) - b_1 \alpha^{nm} \frac{\partial f}{\partial x}^{nm} \bar{\Delta}\phi^{nm} + i\alpha^{nm} k_1 f^{nm} \bar{\Delta}\phi_j^{nm}] \bar{Z}_\tau^{nm} \quad (3.4)$$

The $\bar{Q}_{ij}^{nm} \beta_2$ terms are summed as calculated, and stored as

$$AFROW(JVP) = \sum_m \sum_n Q_{ij}^{nm} \beta_2 \quad (3.5)$$

After all boxes have been processed, if boxlifts and section lifts are desired and this is the first mode shape, box lifts are printed, section lifts are computed and printed, and total lift is printed:

$$\bar{L}_j^m = SLIFT(JCOL) = \sum_n \bar{L}_j^{nm} \quad (3.6)$$

$$\bar{L}_j^m = TLIFT = \sum_m \bar{L}_j^{nm} \quad (3.7)$$

After this has been done for all velocity potentials, one row of the final generalized air forces arrays is computed as:

$$\bar{\bar{Q}}_{ij} = \text{GENAF}(ij) = 2/\beta \text{ AFROW}(JVP) \quad (38)$$

$$Q'_{ij} = -b_1/s^3\beta \text{ Re} [\bar{\bar{Q}}_{ij}] \quad (39)$$

$$Q''_{ij} = -b_1^2/(k_1 s^4 \beta) \text{ Im} [\bar{\bar{Q}}_{ij}] \quad (40)$$

The program does the above for all mode shapes, prints the results, optionally writes them on tape, then terminates. Printing is done in routines PRNTBL, PRNTSL, and PRNTAF.

For Equations (32) and (34) box leading and trailing edge values are needed. Several geometric conditions exist:

(1) Box leading or trailing edge is internal to the planform:
Linear interpolation is used,

$$f_{LE}^{nm} = \frac{1}{2} (f^{n-1,m} + f^{n,m}) \quad (41)$$

$$\Delta \bar{\phi}_{LE}^{nm} = \frac{1}{2} (\Delta \bar{\phi}^{n-1,m} + \Delta \phi^{nm}) \quad (42)$$

and similarly for the box trailing edge.

(2) Box is cut by the planform leading edge:

$$f_{LE}^{nm} = f^{nm} - (X_n - X_{LE}) * \frac{\partial f^{nm}}{\partial x} \quad (\text{point-slope}) \quad (43)$$

$$\Delta \phi_{LE}^{nm} = \begin{cases} 0 & \text{for wing or spatial tail} \\ \bar{\bar{\Delta \phi}}_{TE \text{ of wing}}^{nm} * e^{-i(x_{LE \text{ tail}}^m - x_{TE \text{ wing}}^m)/b_1} k_1 & \end{cases} \quad (44)$$

(3) Box is cut by the planform trailing edge

$$f_{TE}^{nm} = f^{nm} + (x_{TE} - x_n) \cdot \frac{\partial f^{nm}}{\partial x} \quad (45)$$

$\Delta\bar{\phi}_{TE}$ Computed planform trailing edge value, TVP, as described under subroutine VELPOT. This normally is a linear extrapolation using the forward adjacent box center and the current one for the two necessary $\Delta\bar{\phi}$ values.

USAGE: The FORCES program is the main program of a secondary level overlay of the Mach Box program. It is called as follows:

```
CALL OVERLAY( 6HAFMB0X, 1, 7, 0)
```

Input:

Uses labeled common blocks:

```
/ARRAYS/  
/FILES/  
/IOCONT/  
/KERN/  
/KVAL/  
/PROBLM/  
/MODES/  
/GEOMTY/  
/GEOM2/  
/TAPEIO/  
/RWBUFF/
```

Uses the following files:

```
MODESC  
IPNTRM  
IVPSC  
ITSLSC
```

Output:

Printer and tape NOUTP (optional).

11. COMMON BLOCK ORGANIZATION

The basic geometric and program control parameters are stored in a set of labeled common blocks which are loaded with the primary level overlay and thus are available to any secondary overlay. Some of the variables come directly from card input values (see Part I, Section III of this report), and others are internally computed.

	Default
/CTRL/ PREVEX, OMACH, TITLE(8), PRVGEOM, PRVMODE, DIHW, DIHT, DEFAULT	
PREVEX Tested for code word in the data preprocessor link to determine whether defaults should be set or prior status maintained (recycle)	--
OMACH Mach # from previous cycle, compared on recycle to determine whether planform geometry needs changing.	0.
TITLE(8) One-line title for all printed headings	blank
PRVGEOM .T. previous geometry is to be used this cycle .F. New geometry is to be read	.F.
PRVMODE .T. previous modes are to be used this cycle .F. new modes are to be read	.F.
DIHW} .T. {Wing} dihedral is to be used computing DIHT} {Tail} influence on itself	.F.
.F. The {wing} is to be considered flat in computing influence on itself, but dihedral will be used in wing/tail calculations	
DEFAULT .T. All parameters on Card C are to be set to their default values .F. Do not set parameters to default.	.F.
/PROBLM/ XMACH, NMODES, NTSLOP, NKVALS, SMOOTH, NDEG, CRDFIT, EXAIC, SUBDV, PLYWOOD	
XMACH = Mach number for current cycle	no default
NMODES = Number of input modes to use	no default
NTSLOP = Number of thickness slope functions to be used	0
NKVALS = Number of reduced frequencies to be used	0
SMOOTH = .T., Velocity potentials surface smoothing desired .F., No velocity potential surface smoothing desired	.F.
NDEG = Maximum order for smoothing polynomial	0, program will determine
CRDFIT = .T., Chordwise velocity potential smoothing desired .F., No chordwise smoothing desired	.F.
EXAIC = .T., Integration accuracy of 10^{-4} desired .F., Integration accuracy of 10^{-2} desired	.F.
SUBDV = .T., Subdivision is to be applied .F., No subdivision is desired	.F.
PLYWOOD = .T., Full box areas to be used in box lifts .F., Planform box areas to be used.	

Default

/GEOMTY/ COPLAN, NSUBDV, XSUBDV, NSUBD2, NSUBCN, NSURF,
 B1, B1BETA, BIS, B1BTAS, WLAX, WLAZ, PSIW, MXBW, MXBBW,
 MYBW, MYBBW, MXBSW, MYBSW, MYBBSW, IXBW, XCENTR

/GEOM2/ TLAX, TLAZ, PSIT, MXBT, MYBT, MYBBT, MXBST,
 MYBST, MYBBST, IXBT, IXBST, CAPL

COPLAN .T. PSIW = PSIT and CAPL = 0. One box array --
 is used
 .F. The 2 surfaces are not coplanar, or only
 one surface is defined

NSUBDV	Number of subdivided rows (columns) per box	1
XSUBDV	= Float (NSUBDV)	1.0
NSUBD2	= NSUBDV/2	0
NSUBCN	= NSUBD2 + 1 = center location of first chord	1
NSURF	Number of surfaces, 1 or 2	1
B1	Box length = ! BETA * $\sqrt{M^2 - 1}$	--
B1BETA	b_1/β , box width, = YWLE(NWLE)/MYBW	--
B1S } B1BTAS }	Subdivided box {length = B1/XSUBDV width = B1BETA/XSUBDV}	--
WLAX } TLAX }	{Wing} local axis location, in global X co- ordinate	0. 0.
WLAX } TLAX }	{Wing} local axis location, in global Z co- ordinate	0. 0.
PSIW } PSIT }	{Wing} dihedral angle, input in degrees but immediately changed to radians.	0. 0.
MXBW } MXBT }	Number of rows to aftmost portion of the {wing} measured in the n_c coordinate {tail}	--
MXBBW	Number of rows to aftmost wing diaphragm box, n_c coordinate	--
MYBW } MYBT }	Number of chords on the {wing}, m_c coordinate {tail} {=NCHRDS}	--
MYBBW } MYBBT }	Number of {wing} {tail} chords, including tip diaphragm	--
MXBSW } MXBST }	Subdivided MXB {W} {T} count	--

	Default
MYBSW } Subdivided MYB $\left\{ \begin{matrix} W \\ T \end{matrix} \right\}$ count	--
MYBST }	--
MYBBSW } Subdivided MYBB $\left\{ \begin{matrix} W \\ T \end{matrix} \right\}$ count	--
MYBBST }	--
IXBW } Subdivided grid X-location of the first	--
IXBT } unsubdivided {wing} box center	--
IXBT } {tail}	--
IXBST Subdivided grid X-location of the first	--
subdivided tail box	
XCENTR X _w location of the center of the first box	No default
on the wing	
CAPL Non-dimensionalized vertical distance be-	0.
tween centerlines of the wing and tail	
 /KERN/ ERR, MXSKRN, IPKERN, NPLKRN, NSPATK, NR0WEA	
ERR Integration accuracy in AIC calculations	.01
MXSKRN Size of the subdivided AIC, array (number of rows)	
IPKERN Location in array SKERNL where PKERNL(1)]	
would be if it were not overlaid by the	
subdivided C ₀ _μ array.	
NPLKRN Size of the planar AIC array (number of rows)	--
NSPATK Number of spatial AIC arrays necessary	0
NR0WEA Number of rows for the subdivided effective area	--
 /KVAL/ IKVAL, XKVAL(20), XKS (20)	
IKVAL Current k-value number being solved	
XKVAL Array of reduced frequencies, k ₁ , based on	
box length, b ₁	
XKS Array of reduced frequencies, k _s , based on	
semispan, s.	
 /FILES/ NT5, NT6, INTAPE, INFSP, NPLAIC, NSPAIC, NOUTP,	
IOUFSP, MODESC, IVPSC, IGEOSC, IWTFSC, IAICCC	
NT5 Card file (INPUT)	'
NT6 Print file (OUTPUT)	','
INTAPE Binary input tape number, If 0 or < card	''
input will be used	

Default

INFSP	Initial file spacing on the input tape	0
NPLAIC	Tape number for the { planar } AIC arrays	0
NSPAIC	{ spatial }	0
NOUTP	Binary output tape number. If 0, none written	0
IOUFSP	Initial file spacing on tape NOUTP	0
MODESC, IVPSC IGEOSC, IWTFSC IAICSC	{ Internal scratch files }	
 /IOCONT/ OPLAIC, OSPAIC, WTGEOM, WTGNNAF, WTSI, WTBL, PRBOX, PRPAIC, PRSAIC, PRMODS, PRCOEF, PRUW, PRSW, PRVP, PRBL, PRDCP, PRGNNAF, PRGNAC, PRSL, PRLW, PRNW		
OPLAIC}	.T., an old { planar }	AIC tape is being used
OSPAIC}	.F., a new { spatial }	AIC tape is being used
WTGEOM	Not used	.F.
WTGNNAF	.T., Write generalized air forces on tape	.T.
WTSI	.T., Write section lifts on tape	.F.
WTBL	.T., Write box lifts on tape	.F.
PRBOX	.T., Print the box code pattern(s)	.F.
PRPAIC}	.T., Print the { planar }	AIC arrays
PRSAIC}	.T., Print the { spatial }	
PRMODS	.T., Print modal deflections and slopes	.F.
PRCOEF	.T., Print modal polynomial coefficients, if available	.F.
PRUW}	.T., for wake wash sampling, print	{ upwashes side washes longitudinal washes }
PRSW}		.F.
PRLW}		.F.
PRVP	.T., Print velocity potential differences	.F.
PRBL	.T., Print box lifts, $\bar{L}_j^{n,m}$.F.
PRDCP	.T., Print change in pressure, $\Delta C_{p_j}^{n,m}$.F.
PRGNNAF	.T., Print generalized airforces, \bar{Q}_{ij}	.T.
PRGNAC	.T., Print generalized aerodynamic coefficients, Q' and Q"	.F.
PRCM	.T. Print sectional generalized airforces, \bar{Q}_{ij}^m	.F.

		Default
PRSL	.T., Print section lifts, \bar{L}_j^m	.F.
PRNW	.T., Print normal washes, N_{RUW} , N_{RWU} , etc.	.F.
/TAPEIO/	NFS, NMS, LS, NMR, ID(20), NID, ITYPE, LRS, LWS, M, N, PARM(10), IRR DIMENSION IPARM(10) EQUIVALENCE (PARM, IPARM)	
NFS } {File	spacing	0
NMS } {Matrix		0
LS }	Not used	
NMR }		
ID	ID array for the matrix	
NID	Number of words in the ID array on tape	1
ITYPE	Matrix type - MIXED, COMPLEX	
LRS }	Not used	
LWS }		
M }	Matrix dimensions	--
N }		--
PARM	Numerical parameters for the matrix	--
IRR	Error return	-0-
/MODES/	SYM, SYMT, MTYPEW, MTYPET	
SYM	1, Symmetric modes -1, Antisymmetric modes	1
SYMT	0, Left surface contribution will be ignored As above, for a non-planar tail. Differs only for vertical tail	SYM
MTYPEW	1, Polynomial coefficients will be read for the wing 2, Deflections at arbitrary locations will be read 3, Box center values will be read	2
MTYPET	1, Same as above for the tail 2, 3,	2
/ARRAYS/	KBXCDW, LBXCDW, LBOXC, KBXCDT, LBXCDT, KJALPH, LJALPH, KALPHA, KKERNL, LKERNL, KPNTRM, LPNTRM, KDEFSL, KELPHI, LMODES, KPNTSD, LPNTSD, KSDW, LSDW, KPNTDW, LPNTDW, KDW, LDW, KTPV, LTVP	

Locations and limits for arrays:

Variable	Array affected	Value
KBXCDW } LBXCDW }	IBOXW (LBXCDW, LBOXC)	{ Not used 150 8
LBOXC }		
KBXCDT } LBXCDT }	IBOXT (LBXCDT, LBOXC)	{ Not used 90
KJALPH } LJALPH }	IJALPH (LJALPH)	{ Not used 200
KALPHA	ALPHA (LJALPH)	Not used
KKERNL } LKERNL }	SKERNL (LKERNL), PKERNL	{ 1 1640
KPNTRM } LPNTRM }	IPNTRM (2, LPNTRM)	{ Not used 100
KDEFSL	DEFSL (2, LMODES)	Not used
KELPHI } LMODES }	DELPHI (LMODES), complex	{ Not used 500
KPNTSD } LPNTSD }	IPNTSD (2, LPNTSD)	{ Not used 50
KSDW } LSDW }	ENSUBD (2, LSDW)	{ Not used 600
KPNTDW } LPNTDW }	IPNTDW (2, LPNTDW)	{ Not used 100
KDW } LDW }	ENRUS (LDW), ENRLS (LDW)	{ Not used 1275
KTVP } LTVP }	TVP (LTVP), TEXLOC (LTVP), FEXLOC (LTVP)	{ Not used 250

Default

/RWBUFF/ BFCODE, IBFCNT, BUFF (3280)

BFCODE = Code word 8HBUFSIZE
IBFCNT = Size of buffer 3280

BUFF = Buffer array for use by READMX and WRTEMX

/SAMPLW/ ISMPLW, ICHORD(10), IBOXF(10), IBOXL(10), ZLOC(10)

ISMPLW Number of chords specified for wash sampling 0
ICHORD Chord number for sampling -
IBOXF First box on chord to be sampled -
IBOXL Last box on chord to be sampled -
ZLOC Z-location of sampling chord, transformed
internally to correspond to wing coordinates 0.

/PLANXY/ NWLE, NWTE, NTLE, NTTE, XWLE(10), YWLE(10),
XWTE(10), YWTE(10), XTLE(10), YTLE(10),
XTTE(10), YTTE(10)

NWLE } Number of wing { leading edge } definition
NWTE } trailing edge } points

NTLE } Number of tail { leading edge } definition
NTTE } trailing edge } points

XWLE } Wing leading edge definition points
YWLE }

XWTE } Wing trailing edge definition points
YWTE }

XTLE } Tail leading edge definition points
YTLE }

XTTE } Tail trailing edge definition points
YTTE }

COMMON/ CHECKPR/ DPPCPR, GEOCPR, AICCP, NWSCPR,
SMCPR, GAF CPR

These variables are all typed logical. They control whether or not internal checkout print statements will be executed. They will be read from Card C, default .FALSE.

DPPCPR Data preprocessor check-print
GEOCPR Geometry check-print
AICCP AIC section check-print
NWSCPR Normal wash & velocity potential check-print
SMCPR Velocity potential smoothing check-print
GAF CPR Generalized Airforces check-print

12. ARRAY STORAGE

In order to conserve storage, a number of arrays are used as pointers for sparse arrays. All unusual array usage is described below.

a. Arrays Generated in the Geometry Section

IBOXW - Subdivided box pattern
for first planform, or both if
"coplanar"

IBOXT - Subdivided box pattern
for 2nd planform, if non-
"coplanar"

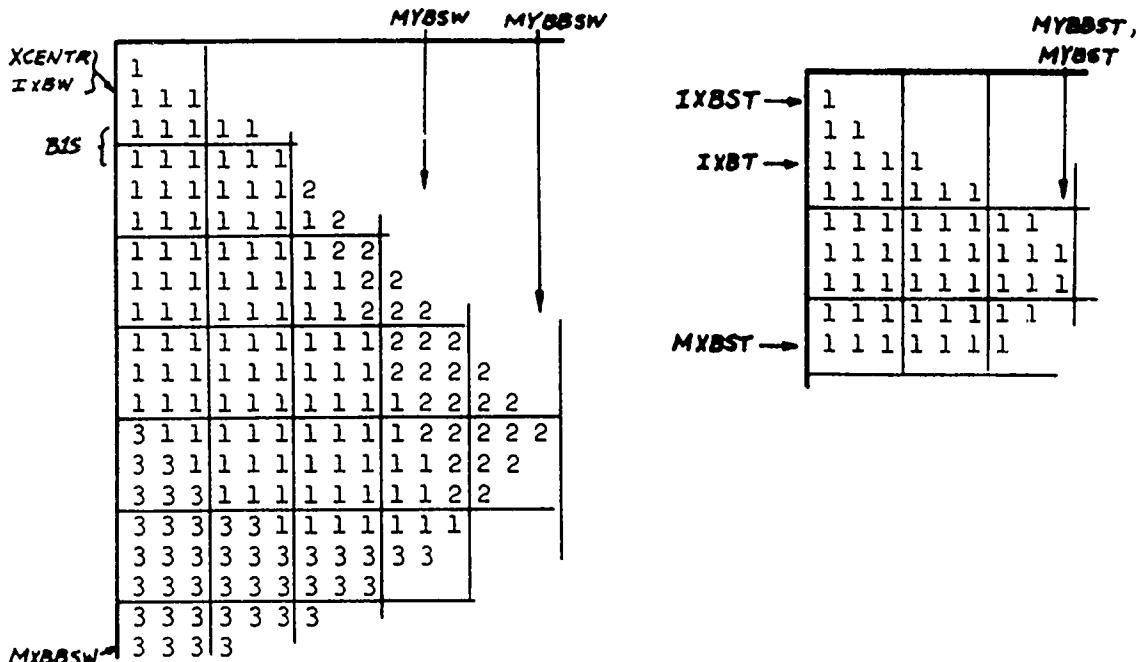


FIGURE 14. Box Code Arrays

The box code arrays are packed twenty numbers to a word, so IBOXW(1,1) contains codes for box (1,1) through box (1,20), IBOXW(2,1) contains codes for box (2,1) through box (2,20), etc.

	Chord -	MXBSW, 1	MXBT	250
FEXLOC				
TEXLOC				
IWAKE			150	

FEXLOC(I) = The location of the leading edge at chord I, normalized to BIS with 1.0 corresponding to the center of the 1st (subdivided) row.

TEXLOC(I) = Same for trailing edge.

IWAKE(I) = Aftmost subdivided wing wake box needed by the tail.

FIGURE 15. Leading and Trailing Edge Arrays

	NWLE	10
XWLE		
YWLE	NNTB	
XWTE		
YWTE		NTLE
XTLE		
YTLE	NTTB	
XTTE		
YTTE		

Planform edge definition, Non-dimensionalized
and shifted after reading.

FIGURE 16. Planform Edge Definitions

	1 2 3	NCHRDS
KPTWW		NWWK
KPTTT		NTTK
KPTRWT		NP.WTK
KPTLWT		NLWTK

For each AIC array needed:

		NROWS
MUAIC	1	
	2	

MUAIC(1,J) = first box needed in row j
 MUAIC(2,J) = last box needed in row J

FIGURE 17. AIC Array Pointers

The four KPT-- arrays indicate the location on scratch file 1AI'SC of the desired AIC array set. For example, KPTT(3) is the AIC set number (4 matrices per set) of the AIC's for the influence of the left tail on right tail chord 3.

b. Arrays generated in the Modes Section

IPNTRM Pointer array for planform boxes on a row.

J	1	2	3	4	N	N+1
IPNTRM(1,J)	1	2	5	10	81	89
IPNTRM(2,J)	1	1	1	1	3	0

J

Normally the row number for which the pointer value is being computed. If there are 2 surfaces that are noncoplanar, the value of J representing the first row of the second planform is MYBW+IOVLAP. IOVLAP is the number of rows on the tail planform that have same x coordinates as rows on the wing planform. If there are no rows with this condition IOVLAP is zero.

IPNTRM(1,J) The sequential count + 1 of all boxes, planform or wake region, that are on or between the first and last planform box of all rows forward of the one J represents

IPNTRM(2,J) The chord number of the first planform box on the row represented by J.

FIGURE 18. Row Pointers

c. Arrays Generated in the AIC Section

The $C_{\bar{\nu}\bar{\mu}\lambda}$, $W_{\bar{\nu}\bar{\mu}\lambda}$ and $V_{\bar{\nu}\bar{\mu}\lambda}$ arrays are stored in a one dimensional matrix. For planar AIC's the $W_{\bar{\nu}\bar{\mu}\lambda}$ and $V_{\bar{\nu}\bar{\mu}\lambda}$ are not computed and the $C_{\bar{\nu}\bar{\mu}\lambda}$ array is calculated for 1/2 of the Mach cone since it will be symmetrical. If subdivision is applied then 2 planar arrays are calculated with the subdivided array overlaying part of the un-subdivided array.

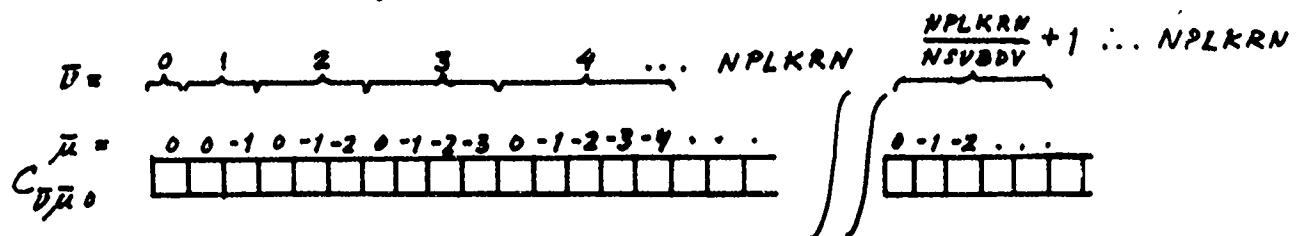


FIGURE 19. Planar AIC

For unsubdivided cases the array ends when $\bar{v} = \text{NPLKRN}$. For subdivided cases when $\bar{v} = \text{NPLKRN}$ the array contains the subdivided AIC calculated at $k_1 = k_1 / \text{NSUBDV}$. \bar{v} then is reduced to $\frac{\text{NPLKRN}}{\text{NSUBDV}} + 1$ and is allowed to increase again until it reaches NPLKRN or the number of rows to cover the planform.

Because of the possible condition where the receiving point of a planform may not be in alignment with boxes on other planforms the spatial AIC's must be calculated on both sides.

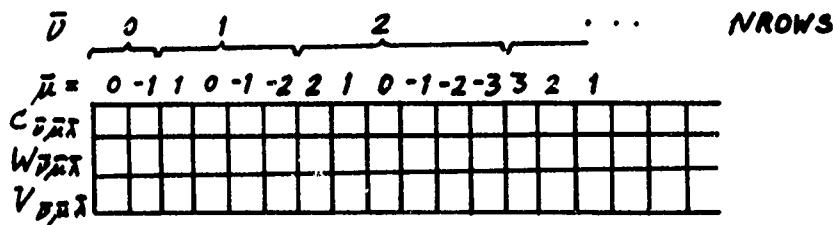


FIGURE 20. Spatial AIC's

13. INTERNAL SCRATCH FILES

a. Matrix Format

All arrays written on disk or tape files are formatted as variable sized matrices. Each matrix consists of two logical records of binary information, the first one being a 16-word matrix identification record, and the second containing the contents of the matrix. The reading/writing of these matrices is done by subroutines READMX and WRTEMX. In the following tape maps, each matrix is a separate box.

Matrix Identification Record - 16₁₀ words

Word	Contents
1	One-word ID label, an integer or label
2	M, the number of rows in the matrix
3	N, the number of columns in the matrix
4	Not used, = 0
5	Not used, = 0
6	Number of words in matrix record
7	K ₁
8	Mach Number
9	
10	
11	
to	
16	

} User parameters, array PARM

Matrix Record - variable length (word 6 above)

Ordered consecutively by row, left to right within each row.

b. Geometry Scratch File IGEOSC

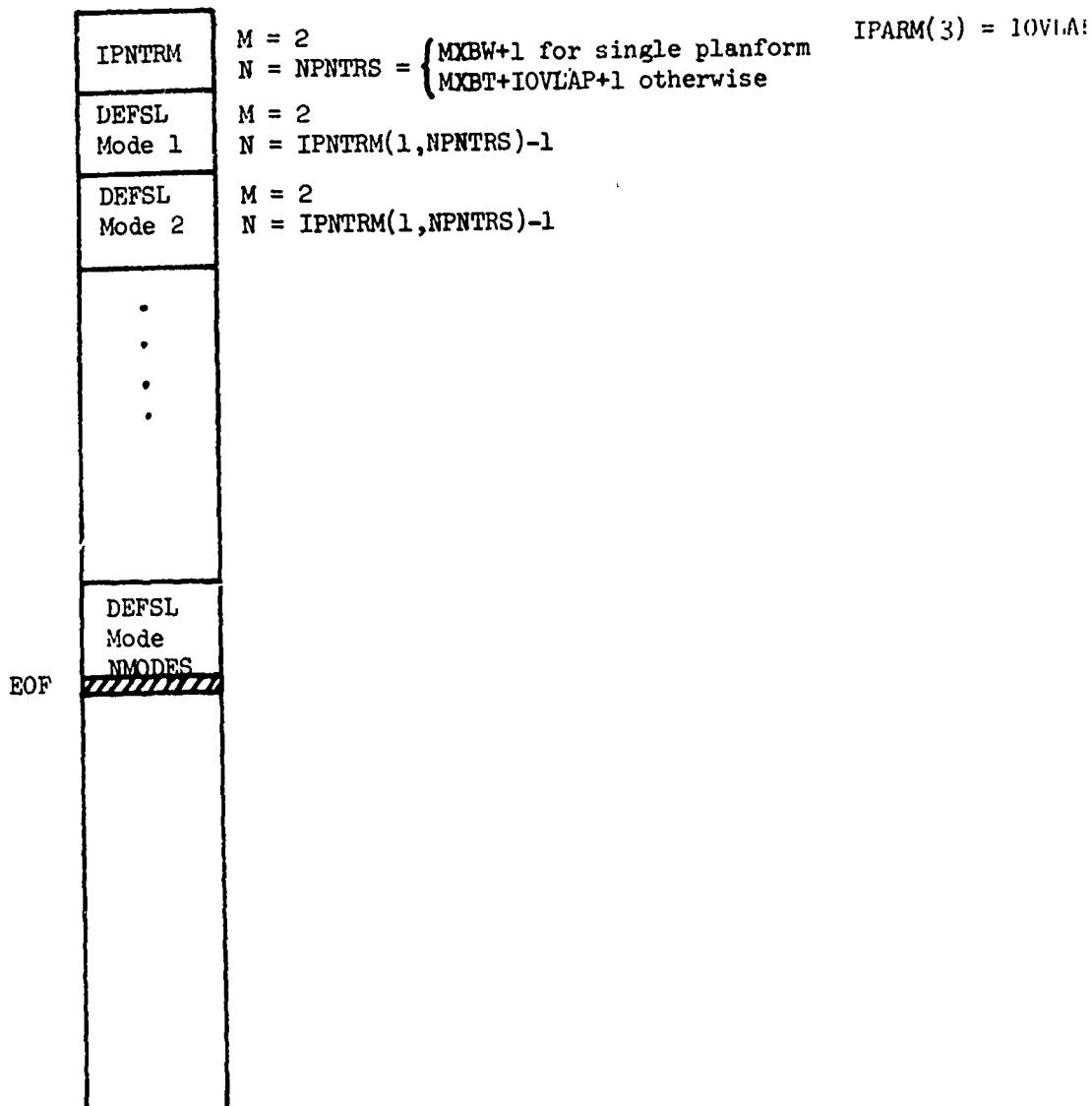
This file is generated in the geometry processor and contains all large geometry arrays. The space after the two geometry files is used for temporary scratch during mode shape processing.

	Matrix Dimensions	Parameter Array										
Present only if NSURF=2 & COPLAN=.F.	<table border="1"> <tr><td>IBOXW</td></tr> <tr><td>IBOXT</td></tr> <tr><td>FEXLOC</td></tr> <tr><td>TEXLOC</td></tr> <tr><td>ALPHA</td></tr> <tr><td>IJALPH</td></tr> <tr><td>KPT</td></tr> <tr><td>MUAIC</td></tr> <tr><td>MUAIC</td></tr> <tr><td>MUAIC</td></tr> </table>	IBOXW	IBOXT	FEXLOC	TEXLOC	ALPHA	IJALPH	KPT	MUAIC	MUAIC	MUAIC	PARM(1) = 0. PARM(2) = XMACH
IBOXW												
IBOXT												
FEXLOC												
TEXLOC												
ALPHA												
IJALPH												
KPT												
MUAIC												
MUAIC												
MUAIC												
EOF	M = MXEBW*NSUBDV N = (MYBBSW-1)/NBWRD+1											
EOF	M = MXEBST-IXBST+1 N = (MYBBST-1)/NBWRD+1											
EOF	M = 1 N = MYBSW+MYBST											
EOF	M = 1 N = MYBSW+MYBST											
EOF	M = 1 N = NAL	IPARM(3) = NALPHW										
EOF	M = 1 N = NAL											
EOF	M = 1,2,3 or 4 N = max. # of AIC's needed in the 4 categories	IPARM(3) = NWWK IPARM(4) = NTTK IPARM(5) = NRWTK IPARM(6) = NLWTK										
EOF	M = 2 N = NROWS	PARM(4) = YBAR PARM(5) = EL IPARM(6) = 0; C,W,V needed 1; W,V needed 2; V needed										
EOF	M = 2 N = NROWS											
EOF	M = 2 N = NROWS											
		(File 2 is first built on IVPSC by GEOMBX, then copied to IGEOSC.)										

c. Modes Scratch File MODESC

This file is generated in the modal data processor. The deflections and slopes are given at all box centers.

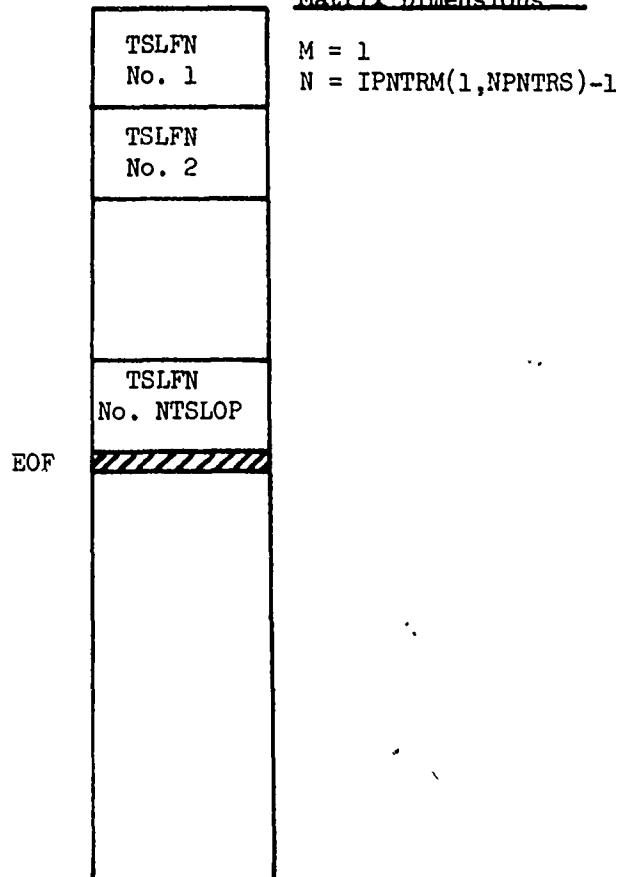
Matrix Dimensions



d. Thickness Slopes Scratch File ITSLSC

This file is equivalenced to IWTFSC, which is first used in GEOMBX for temporary scratch while building the MUAIC arrays. The thickness slope functions are then written on the file at the end of the modal data processor. If NTSLOP = 0, one matrix of ones will be written, corresponding to $\frac{\partial Z_t}{\partial x} = 0$.

Matrix Dimensions



e. Spatial AIC Scratch File IAICSC

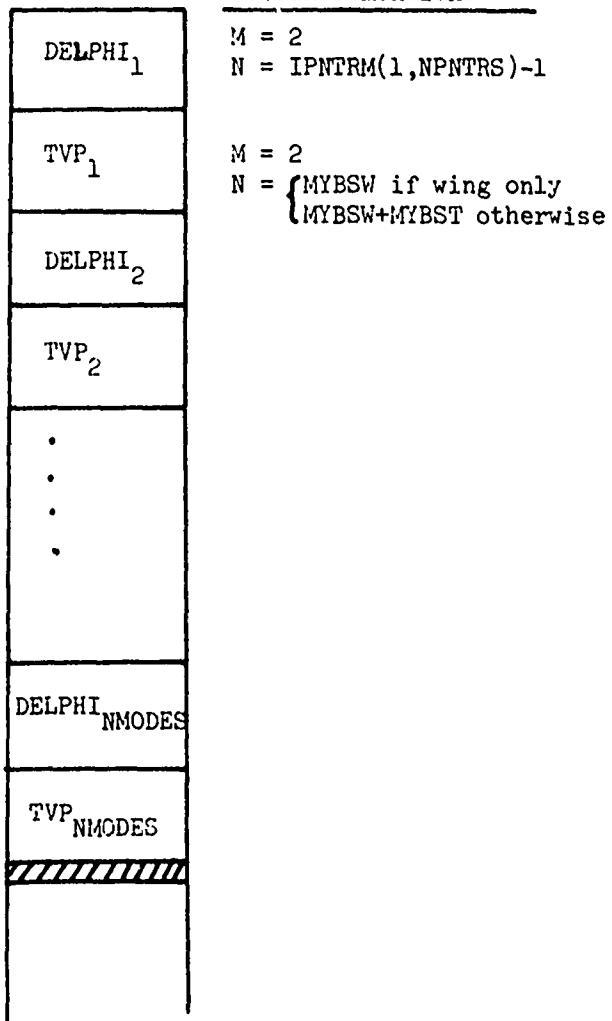
This file is first used for internal scratch during calculation of polynomial coefficients by the modal data processor. In the AIC section it is written with all spatial AIC's needed for one reduced frequency, as determined in the geometry section. IAICSC is re-written for each new reduced frequency.

	Matrix Dimensions	Parameters
MUAIC ₁	M = 2 N = NROWS ₁	PARM(1) = k ₁ PARM(2) = XMACH PARM(4) = YBAR ₁ PARM(5) = EL ₁
C _{̄v̄λ̄}	M = 2 N = (NROWS ₁ +1)(NROWS ₁ /2)	
W _{̄v̄λ̄}		
V _{̄v̄λ̄}		
MUAIC ₂	M = 2 N = NROWS ₂	PARM(4) = YBAR ₂ PARM(5) = EL ₂
C _{̄v̄λ̄}	M = 2 N = (NROWS ₂ +1)(NROWS ₂ /2)	
W _{̄v̄λ̄}		
V _{̄v̄λ̄}		
MUAIC _{NSPATK}		
C _{̄v̄λ̄}		
W _{̄v̄λ̄}		
V _{̄v̄λ̄}		
EOF		

f. Velocity Potentials Scratch File IVPSC

This file is first used for internal scratch by the geometry processor while assembling MUAIC arrays. It is later used in the modal data processor as temporary storage for the wing mode shapes to be merged with the tail modes, and again for the same purpose when working with thickness slope functions. In the normal wash and velocity potentials section it is written with the $\Delta\phi$ and $\Delta\psi_{ee}$ arrays for each mode.

Matrix Dimensions



14. OUTPUT FILES

The program generates three optional output files. Two of them, the AIC files, are designed for reuse with the program during subsequent executions. The program automatically searches these files and updates them with any new AIC's generated.

The optional final output file is designed to pass the generalized air-forces matrices on for flutter or dynamic loads analyses. It is written optionally in the forces section of the program.

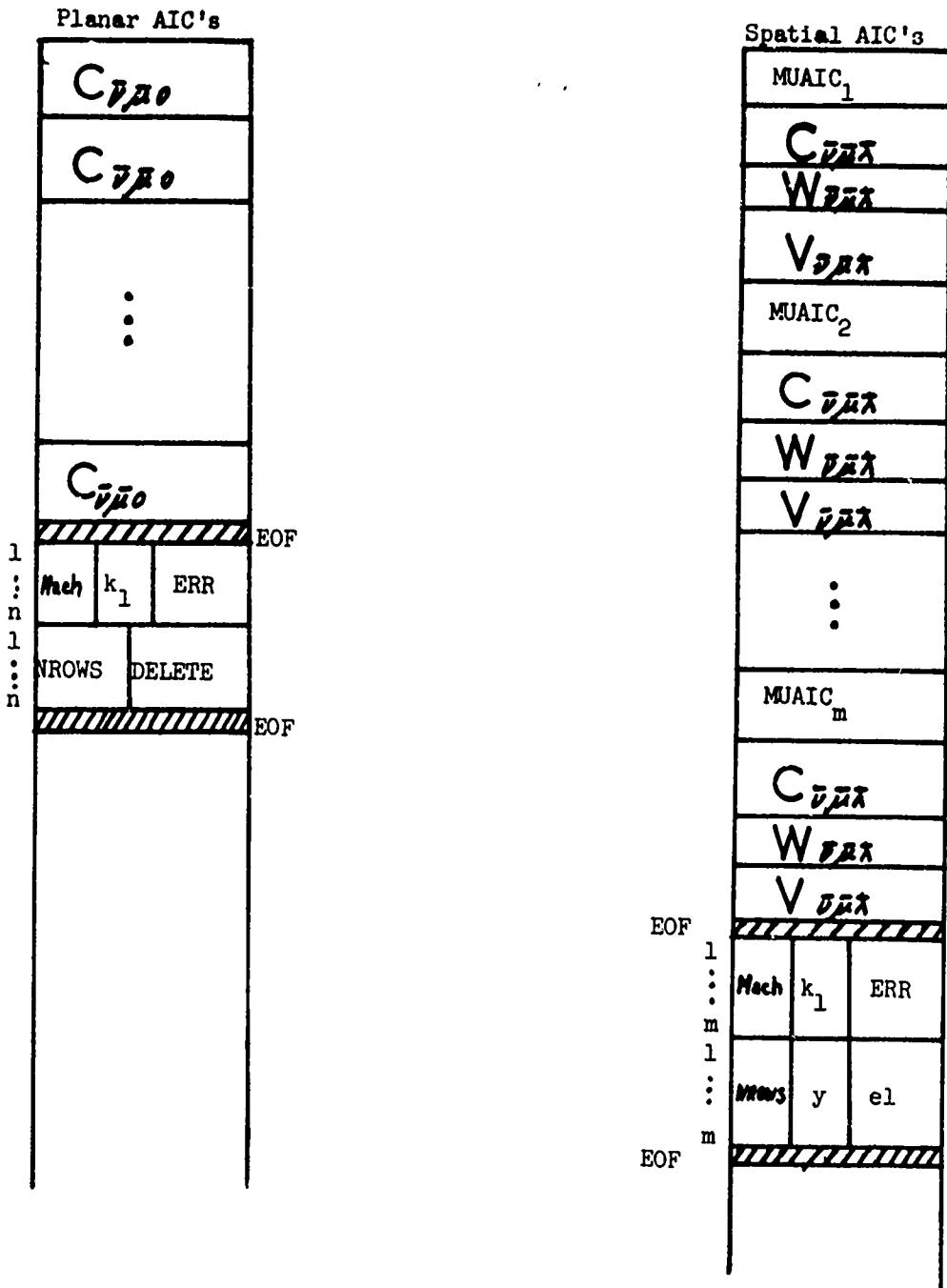


FIGURE 41. Tape Storage of AIC Arrays

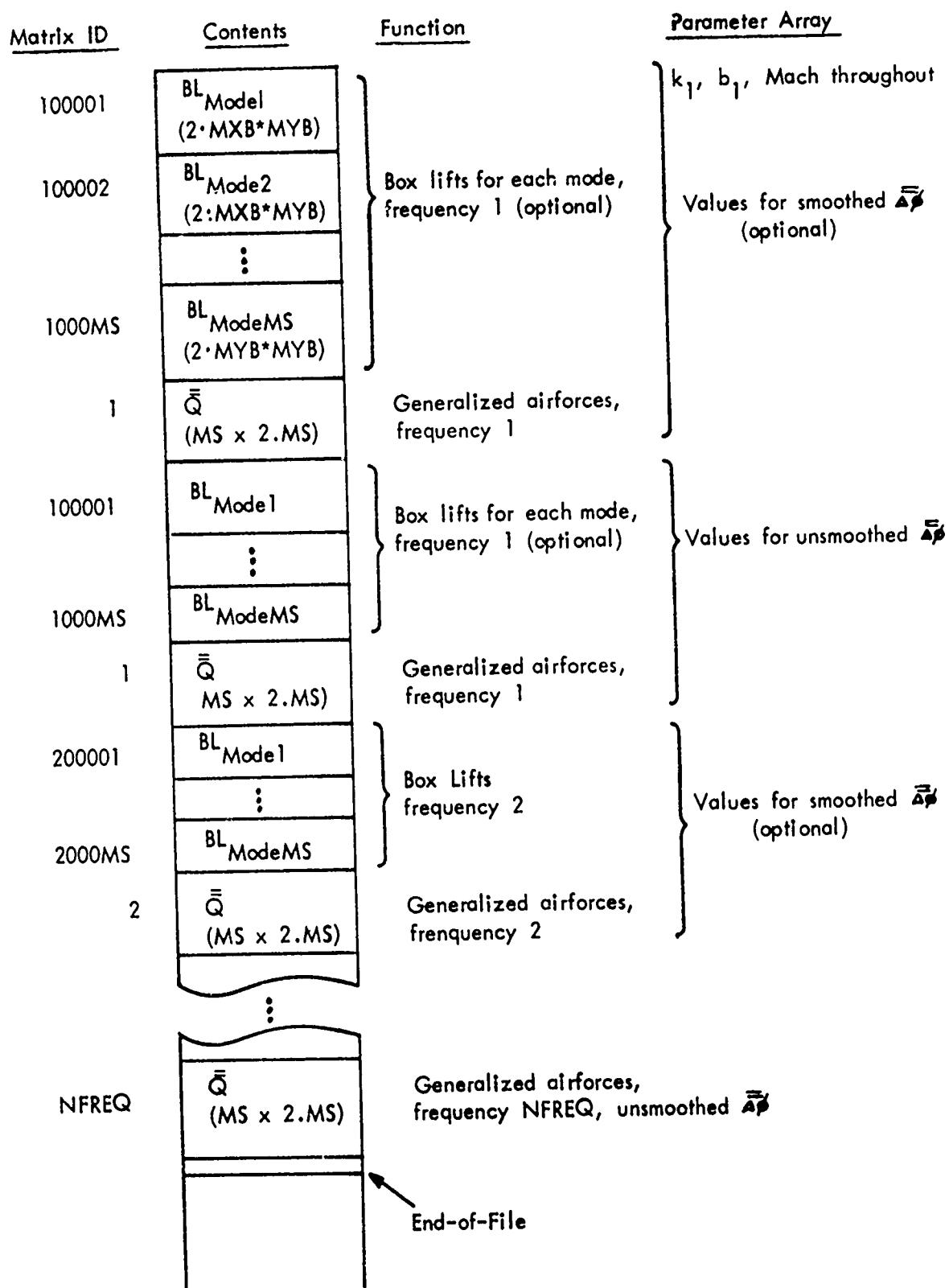


FIGURE 22. TAPE MAP OF FORCES OUTPUT TAPE

15. IMPLEMENTATION AND DEBUGGING

a. Update

The source program is written and maintained using the CDC 6600 SCOPE operating system UPDATE feature. Under UPDATE, all coding is either part of a *COMDECK or a *DECK. A *COMDECK may be replicated many times throughout the other decks. This feature is used for all global labeled common blocks and for most local common blocks, to insure that all routines needing them have identical common statements. A few subroutines which are needed in more than one overlay are also set up as *COMDECKs. The names of the *COMDECKs and *DECKs correspond as closely as possible to their Fortran identifiers - program name, subroutine name or common block name.

b. Open-ended Features

The writers of the program feel that most potential users probably have unique system features which may be utilized to optimize the execution of the program beyond its release status. With this in mind, numerous "hooks" have been coded in to make other features easy to implement.

1. All references to disk or tape files are by name, rather than by number. All file names are together in one common block, /FILES/. The internal scratch files are defined in one DATA statement in the zero overlay DRIVER, and the input, output and AIC files are defined via card input data.
2. All reading and writing of internal and external scratch files is handled by subroutines READMX and WRTEMX. These routines have several calling parameters which are unused, but available if it is desired to make use of labeling, random I.O., or level numbers. Because READMX and WRTEMX use BUFFERIN and BUFFEROUT, all files may share a common buffer area, allowing for a considerable savings in storage requirements.
3. Subroutine FLUSH is always called when a fatal error is encountered. This routine may be written to make use of any system error recovery procedure available. The release version prints a comment, flushes the OUTPUT file, and terminates with a Mode 1 error.
4. Subroutine DTIME is called between each secondary overlay. The release version returns CP time only; however, provision is made for PP time if the implementing system has that capability.

c. Debugging

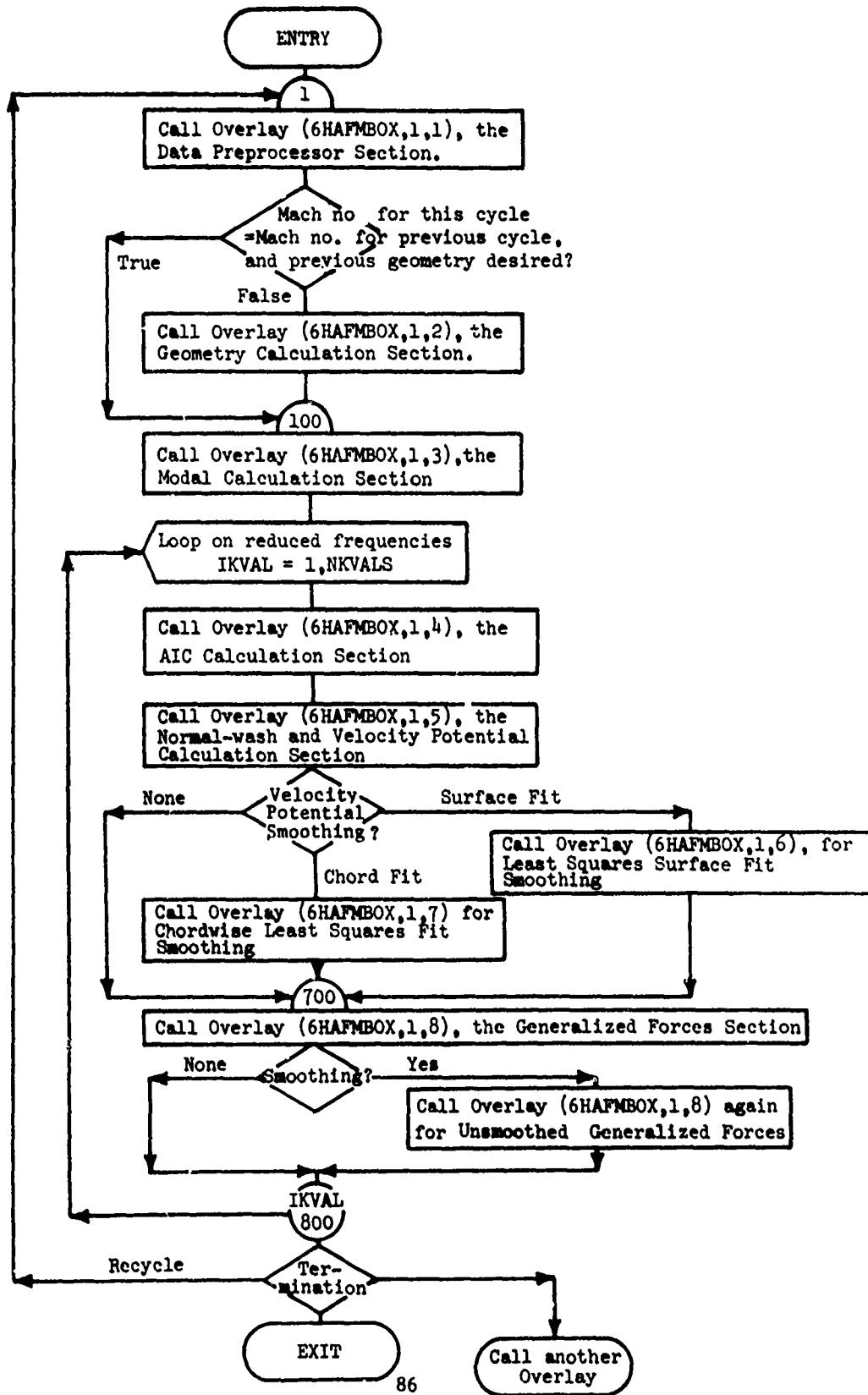
It is recommended that a new user first run one of the sample data cases, to familiarize himself with the program features and to insure

that the program gives correct answers at his installation.

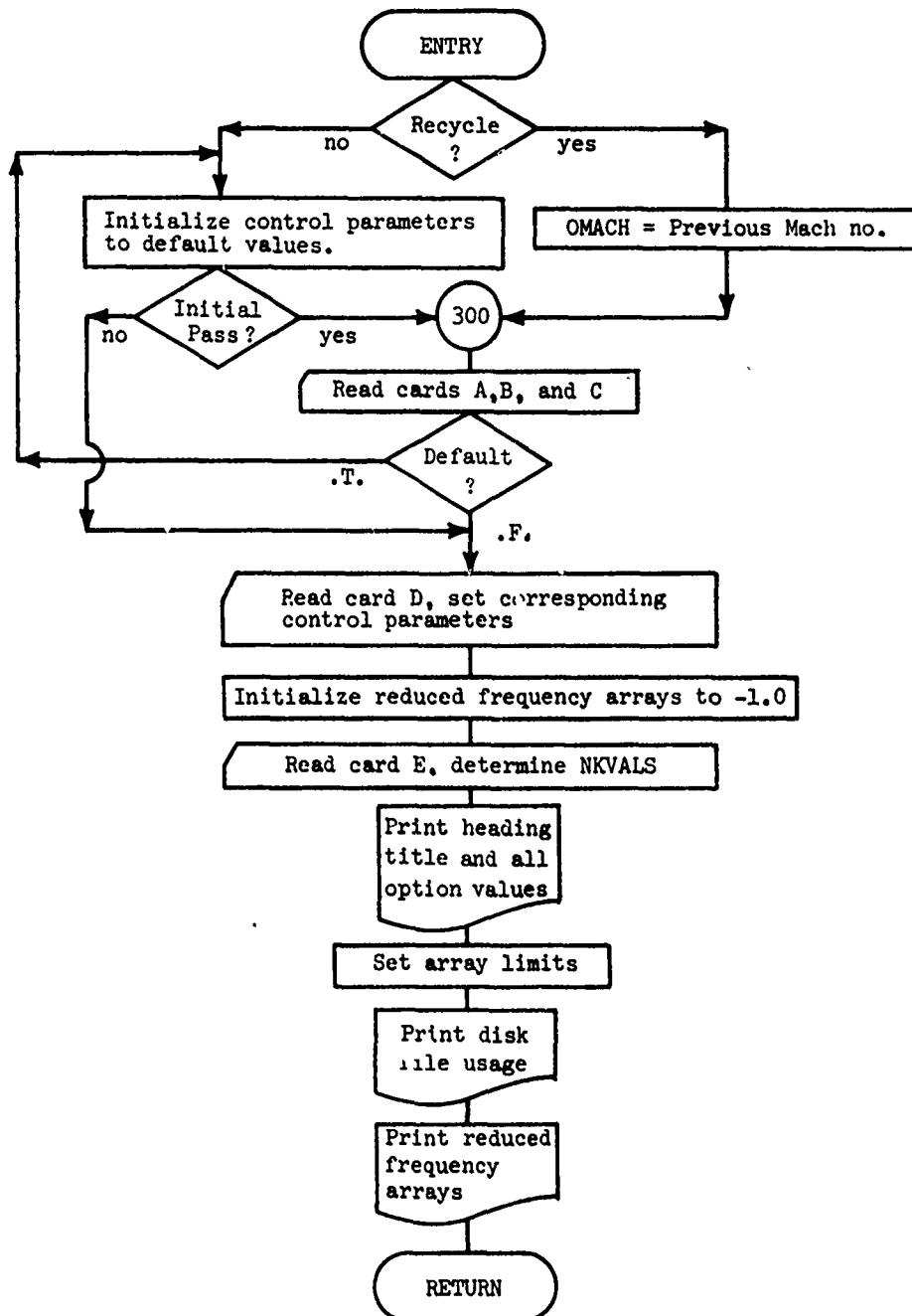
In the event the program fails "hard" (mode error, time limit, etc.), standard use of listings, load maps and dumps will usually pinpoint the cause.

If the program executes but seems to give bad numbers, additional intermediate printout may be helpful. The variables in common block /CHECKPR/ are designed to control the printing of additional check values. Each variable controls printing from one secondary overlay, so only the suspected area need be printed. The check prints provided are rudimentary, so for given problems additional prints would probably have to be written, but if they are made conditional on the common variables, they can be left in for future needs. The CHECKPR variables are all read from Card C of the data, or may be set in an executable statement after the call to DATAPP.

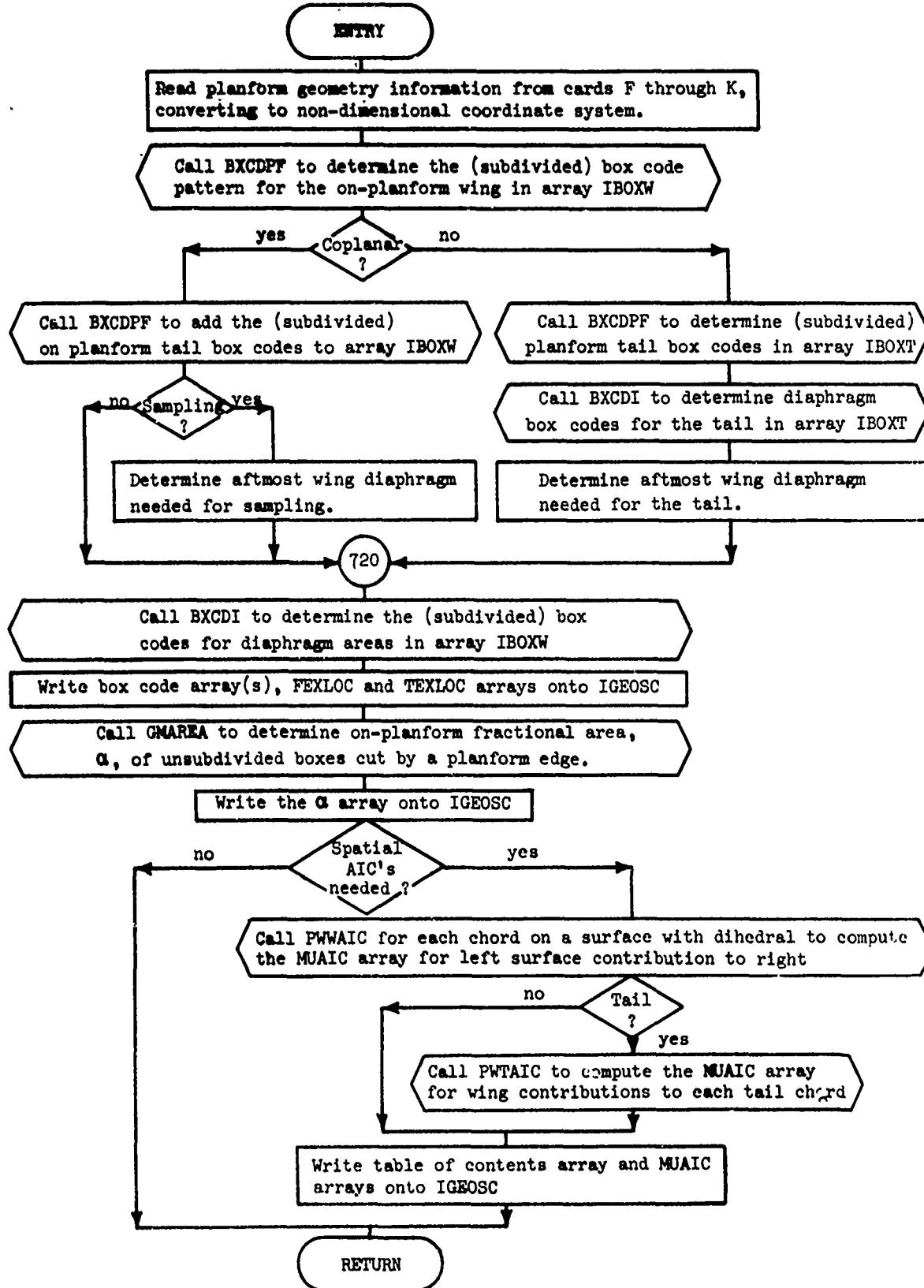
Program CONTROL — Primary level overlay which controls the program flow



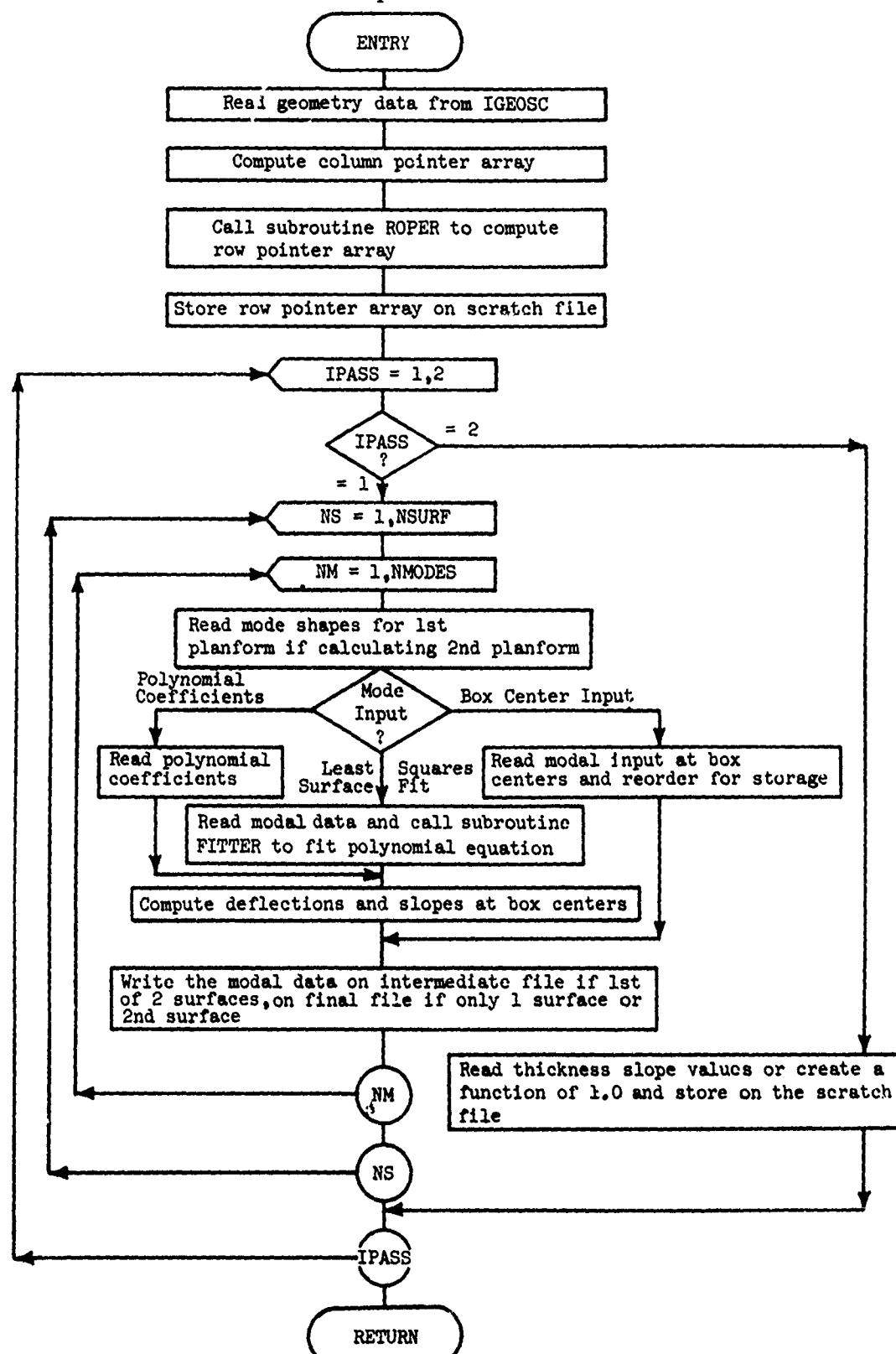
Program DATAPP — Secondary overlay which initiates control parameters as a function of defaults or card data.



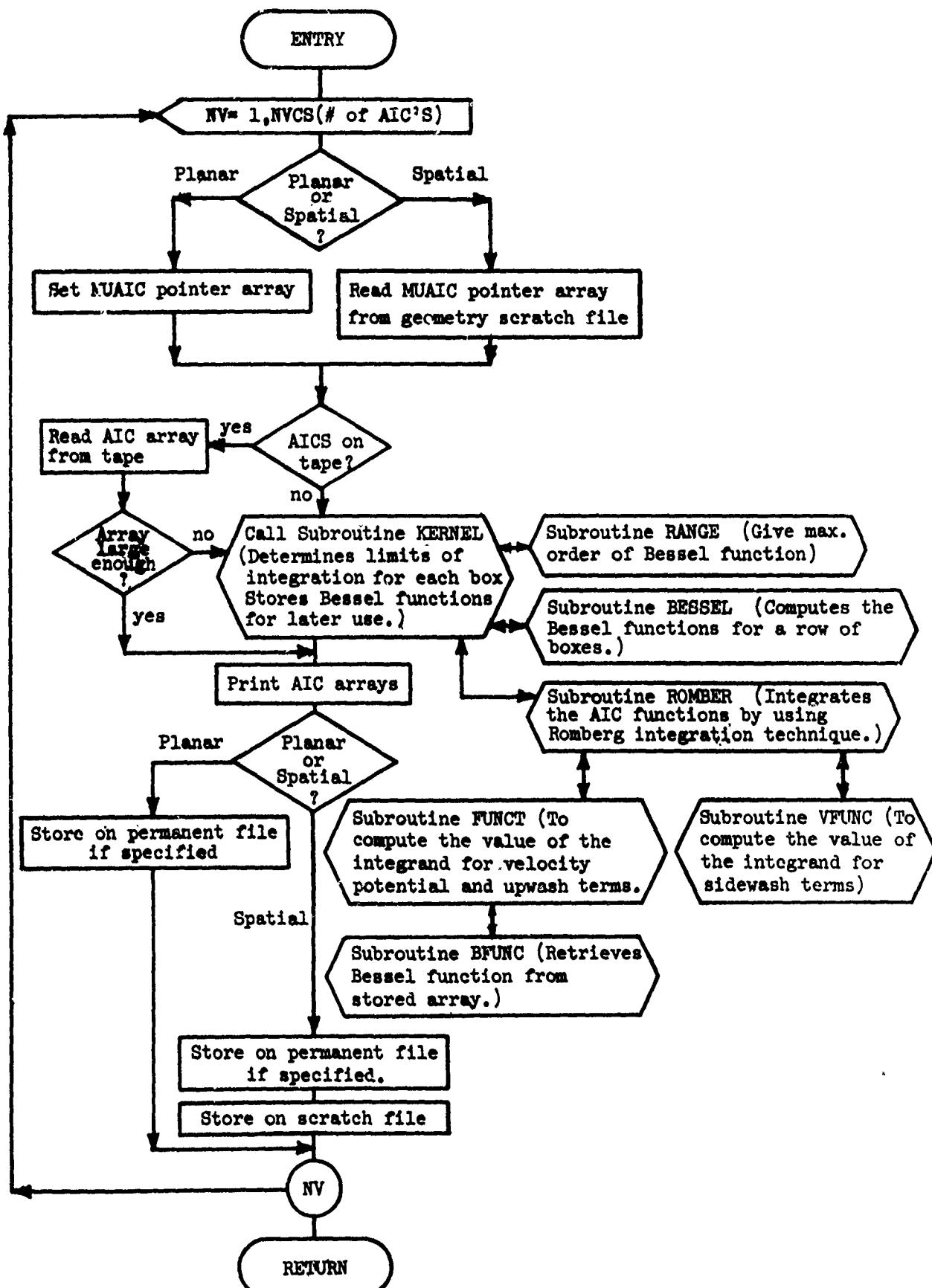
Program GEOSMK — Secondary level overlay which computes planform geometry.



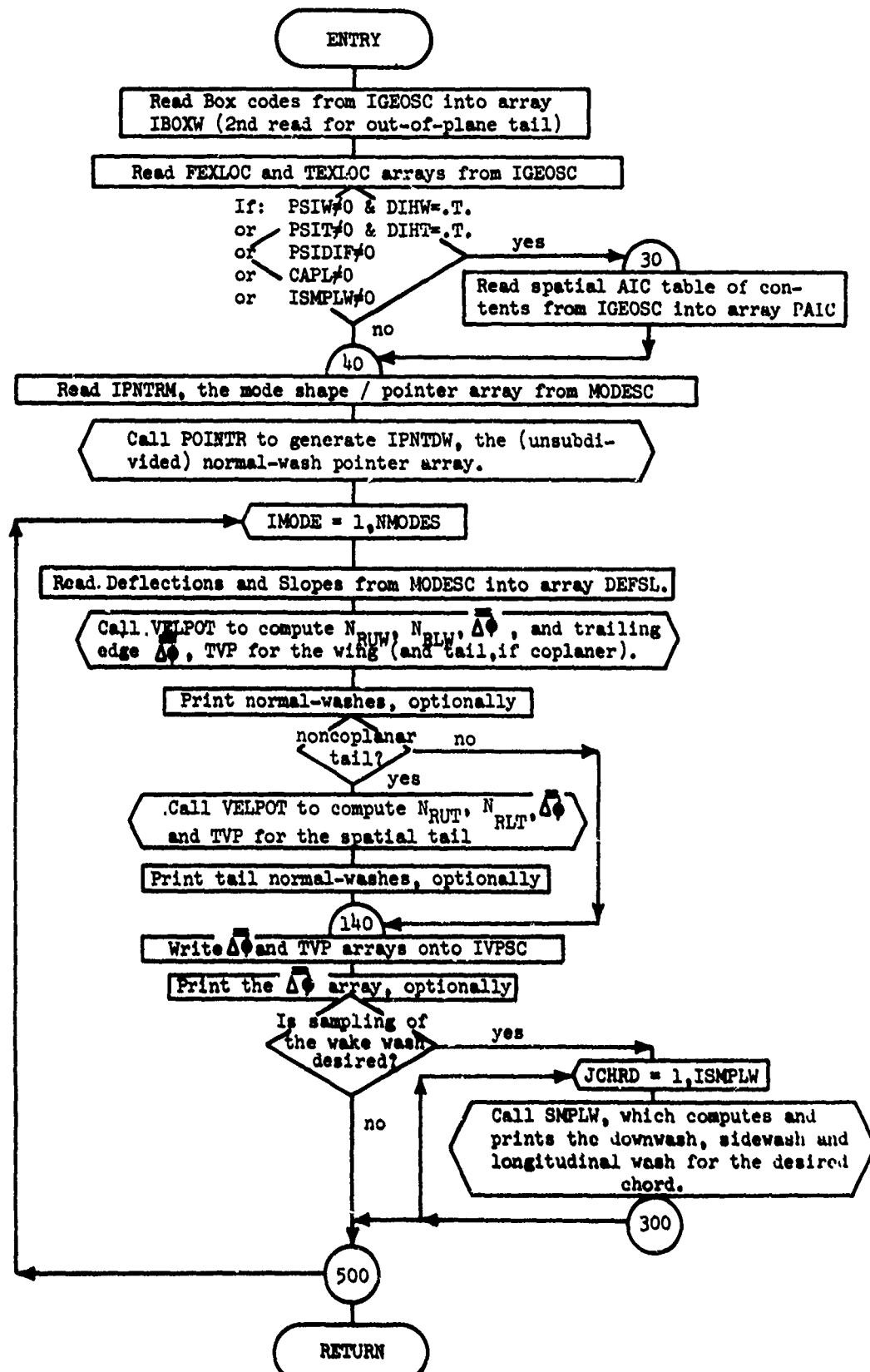
Program MODES— Secondary level overlay which reads mode shapes and thickness slope functions



Program VICMAIN — Secondary level overlay which calculates all Aerodynamic Influence Coefficients

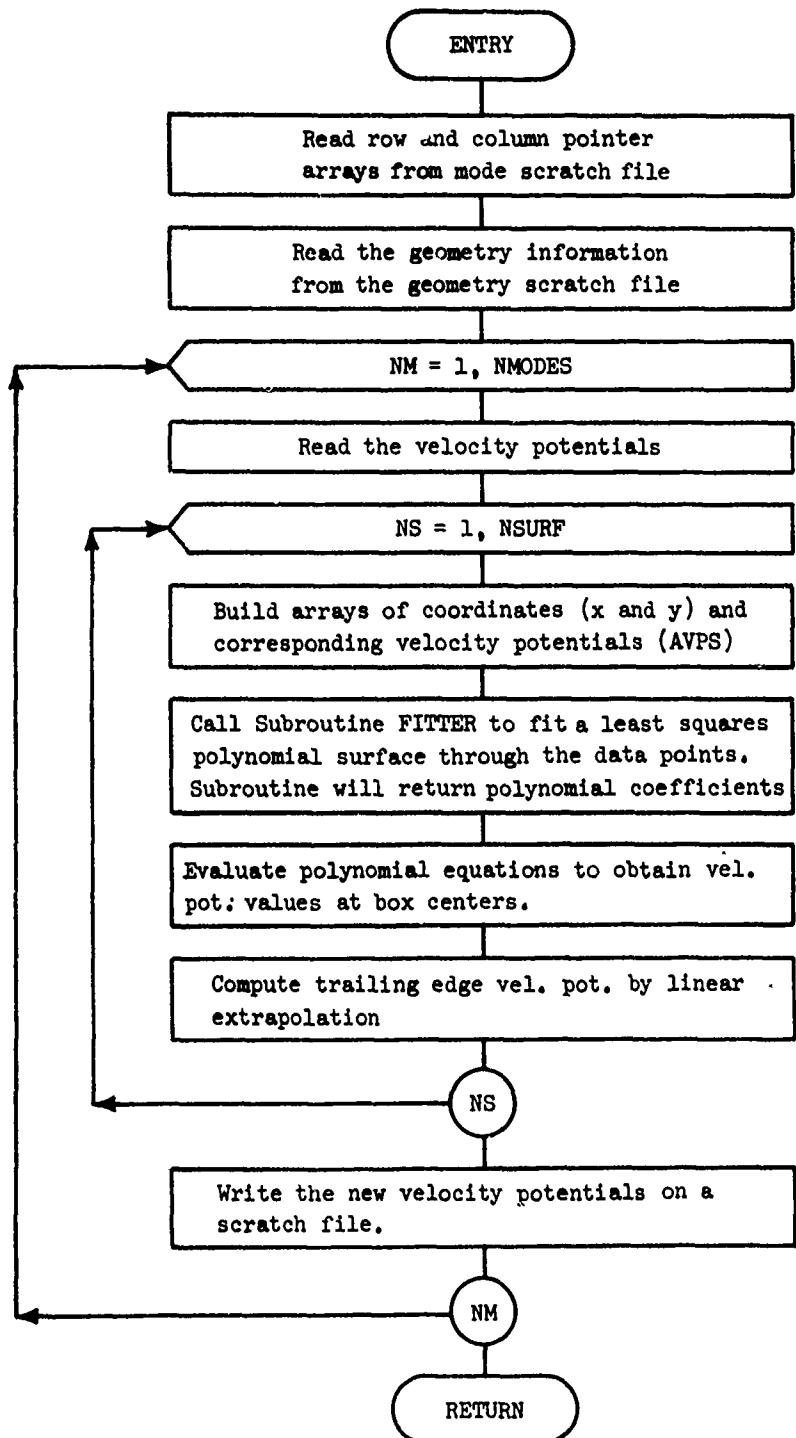


Program NWVLOC Secondary overlay which calculates normal washes and velocity potentials.

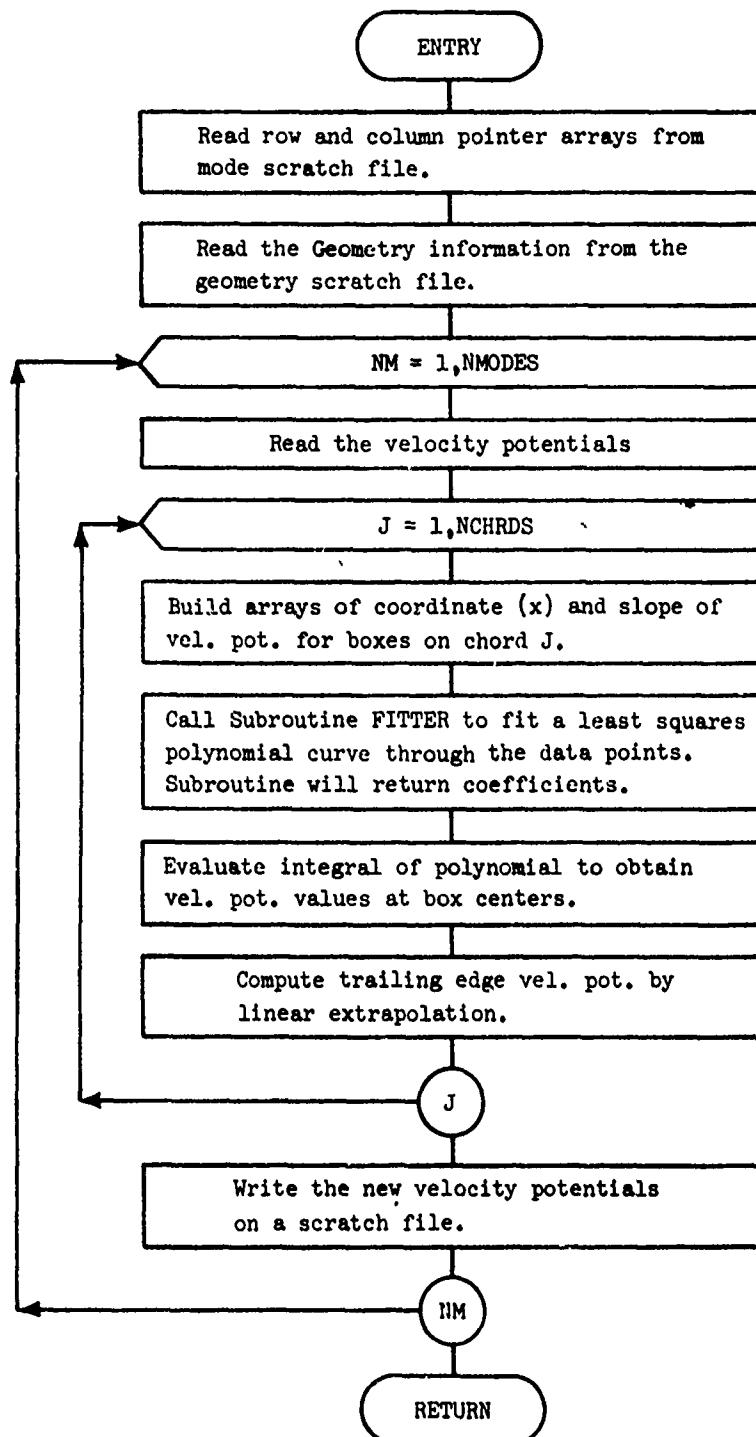


Program SMTS

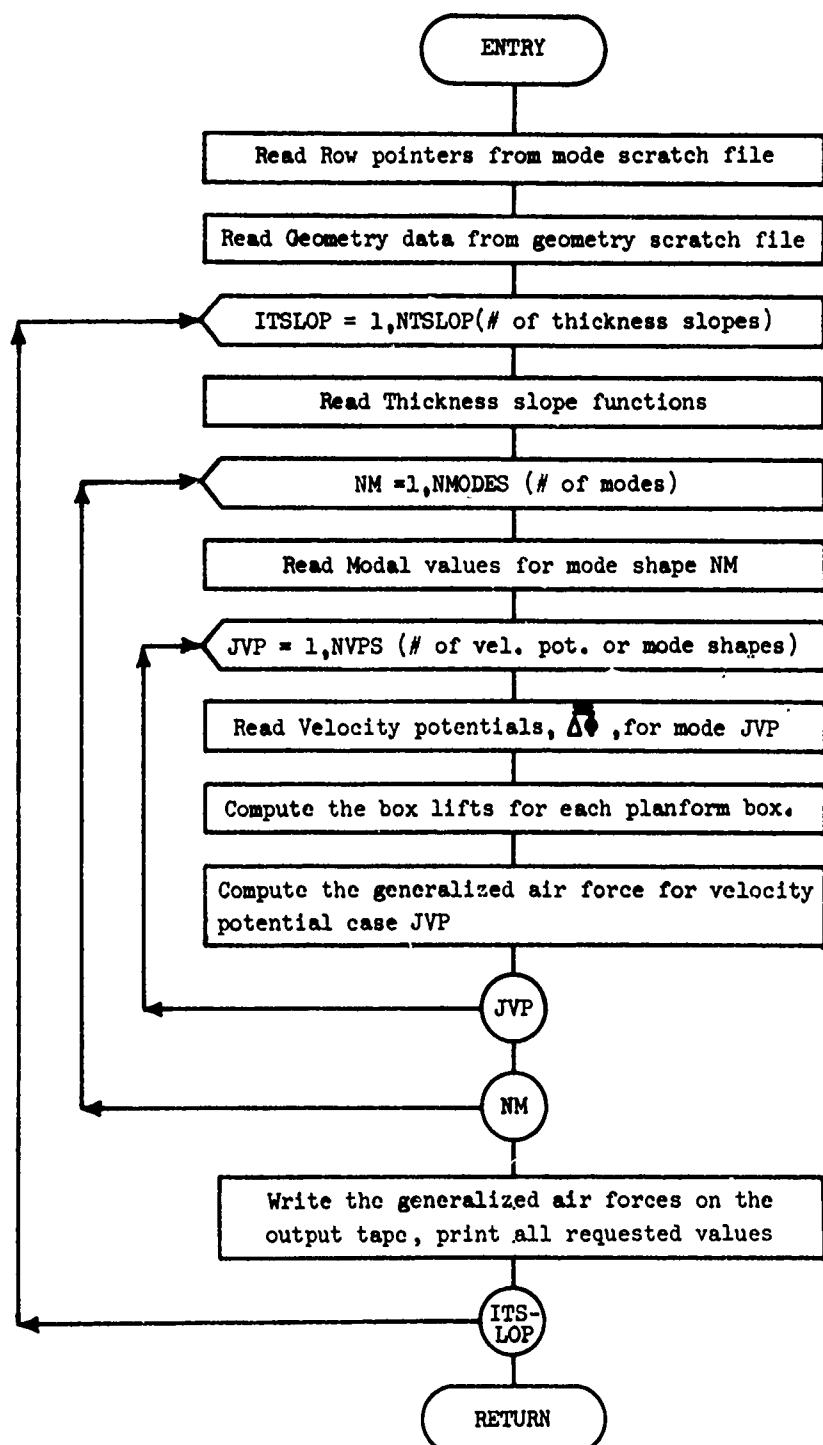
Secondary overlay which smooths velocity potentials
by fitting a least squares surface through them



Program CHORDF — Secondary overlay which smooths velocity potentials by fitting a least squares curve along each chord.



Program FORCES Secondary overlay which computes box lifts
section lifts, and generalized air forces



REFERENCES

1. Faddeeva, V. N., Computational Methods of Linear Algebra, Dover Publications, Inc., 1959.
2. Ralston, A., And Wilf, H., Mathematical Methods for Digital Computers, Vol. 2, New York, John Wiley and Sons, 1967, pp. 133-137.

APPENDIX A
SAMPLE INPUT AND OUTPUT DATA

A simple spatial configuration, shown in Figure 23, was chosen as a sample problem for the demonstration of the card data input and a selection of the printed output. The planform used is a pair of identical rectangular surfaces (wing and tail) with small horizontal and vertical separation.

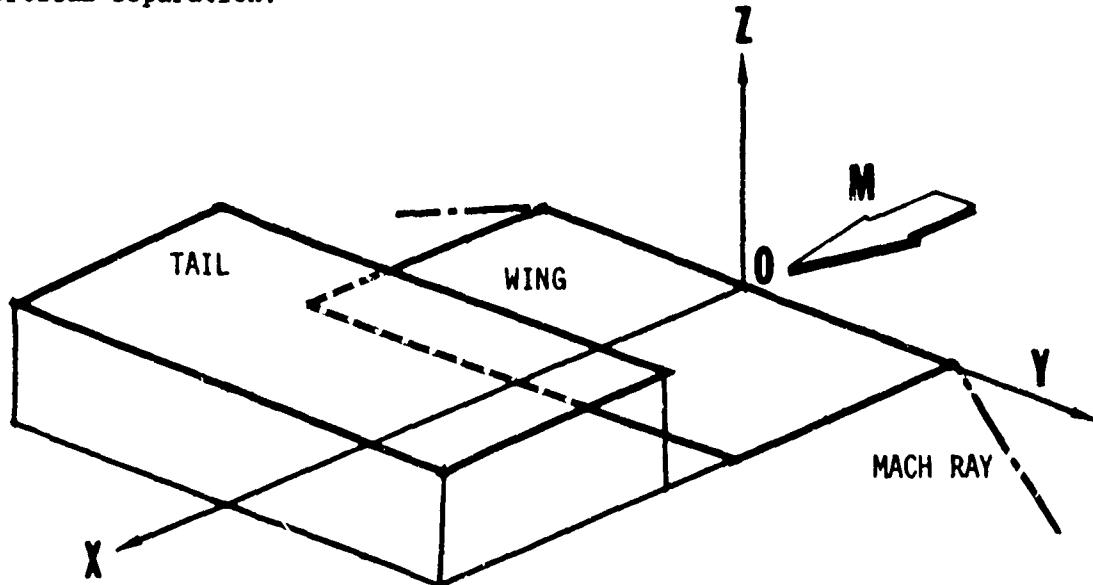


FIGURE 23 SAMPLE PROBLEM CONFIGURATION

The configuration was analyzed at Mach 1.2 for a reduced frequency (based on semi-span) of .5. Only the wing surface was allowed to oscillate, in plunging motion for mode 1 and in pitch about the wing leading edge for mode 2. These two modes were input on cards as polynomials. Chordwise velocity potential smoothing was requested.

In the interest of space the printout was edited to give samples only. A few pages of one spatial AIC array and the planar AIC are included, as well as most of the computations for mode 2 (wing pitch). Since for this configuration the upper and lower surface normal wash differs only in sign, only the upper normal washes are included. The generalized force calculations at the end are for smoothed velocity potentials.

Card Input Data

```

AFMDOX      1      0
SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
$CARDD XMACH=1.20
$CARDC SYM=1.0,MTYPEW=1,MTYFET=1,NSURF=2,WTGNAF=.T.,WTBL=.T.,
PRGNAC=.T., PRDCF=.T.,
PLYWOOD = .T.,
CRDFIT=.T.,NDEG=4,
FRGNAF=.T.,PRDL=.T.,PRSL=.T.,PRPAIC=.T.,PRSAIC=.T.,PRCOEF=.T.,
PRDOX=.T.,PRVF=.T.,EXAIC=.F., FRNW=.T., PRMOOS=.T. $
$CARDD
$CARDE XKS(1) =      .50
$CARDF TLAX=1.20, TLAZ=.40 $
$CARDG NCHRDS=10, XEDGE=0.00
      2    2    2    2
      0.      0.      0.      1.0          CARD H
      1.0     0.0     1.0      1.0          CARD I
      0.      0.      0.      1.0          CARD J
      1.0     0.0     1.0      1.0          CARD K
$CARDM NMODES=2 $
      2
      1.      0.      0.      0.      0.      0.          CARD L
      2
      0.      1.      0.      0.      0.      0.          WING-1
      1
      0.      0.      0.          TAIL-1
      1
      0.      0.          TAIL-2

```

* UNSTEADY AERODYNAMICS OF WING-HORIZONTAL TAIL
* CONFIGURATIONS IN SUPERSONIC FLOW
*
* PREPARED UNDER CONTRACT NO. AF 33615-70-C-1126
* PROJECT NO. 1370
*
* FOR DEPARTMENT OF THE AIR FORCE
* AERONAUTICAL SYSTEMS DIVISION
* AIR FORCE FLIGHT DYNAMICS LABORATORY
* WRIGHT-PATTERSON AIR FORCE BASE
*
* BY THE BOEING COMPANY
* COMMERCIAL AIRPLANE DIVISION
* SEATTLE, WASHINGTON

TITLE -

SAMPLE CASE --- TWO ARE2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

- TITLE

THE FOLLOWING OPTIONS ARE REQUESTED -

MACH NUMBER = 1.200000
SYMMETRIC ANALYSIS
FLYING OPTION IS USED. (FLAP/DEFLAP BOUNDARY DETERMINED BY DOX PATTERN.)
BASIC (UNSUBDIVIDED) ANALYSIS WILL BE USED
ANALYSIS FOR 2 FLAP/DEFLAPS
APPROXIMATE KERNELS WILL BE USED
VELOCITY POTENTIALS WILL BE SMOOTHED BY A LEAST SQUARES
POLYNOMIAL CHORDWISE FIT, OF ORDER 4.
(0 = PROGRAM DETERMINED.)
PRINT THE DOX PATTERN
PRINT WISE SHAPE POLYNOMIAL COEFFICIENTS, IF AVAILABLE
PRINT WISE SHAPES USED
PRINT THE FLAP/DEFLAP AIR ARRAYS USED
PRINT THE SEATL AIR ARRAYS USED
PRINT NORMAL WISE'S
PRINT THE VELOCITY POTENTIALS
PRINT THE DOX LIFTS
PRINT THE SECTION LIFTS
PRINT PRESSURE DIFFERENCE COEFFICIENTS

PRINT GENERALIZED AERODYNAMIC COEFFICIENTS
 PRINT GENERALIZED AIR FORCES
 WRITE BOX LIFTS ON TAPE
 WRITE GENERALIZED AIR FORCES ON TAPE
 MODAL INPUT FOR WING IS POLYNOMIAL COEFFICIENTS
 MODAL INPUT FOR TAIL IS POLYNOMIAL COEFFICIENTS
 DIHEDRAL WING INFLUENCE CALCULATED
 DIHEDRAL TAIL INFLUENCE CALCULATED

THE FOLLOWING TAPE SETUP IS REQUESTED -

OLD AIC TAPE =	0
NEW AIC TAPE =	0
OLD SPATIAL AIC TAPE =	0
NEW SPATIAL AIC TAPE =	0
INPUT DATA TAPE =	0 SPACED 0 FILES,
OUTPUT TAPE =	1 SPACED 0 FILES,

THE FOLLOWING IS THE REDUCED FREQUENCY ARRAY BASED ON WING SEMI-SPAN

.50000	ENTERING PROGRAM GEOM	CURRENT ELAPSED TIME IS	CF = .224, FF = 37.757
- - - - -	GEOMETRIC PARAMETERS	- - - - -	
CARDF -LOCAL AXES DEFINITION-	X-LOCATION	Z-LOCATION	DIHEDRAL ANGLE (FSI)
WING	0.000	0.000	0.00 DEGREES
TAIL	1.200	.400	0.00 DEGREES
CARDG -BOX PATTERN DEFINITION-	NCHEDS	XCENTR	OF XEDGE
10	*0000.0000	0.0000	0.0000
CARDH -PLANFORM DEFINITION POINT COUNTS-	LEADING EDGE	TRAILING EDGE	
WING	2	2	
TAIL	2	2	
CARDI TO CARDL -PLANFORM DEFINITIONS-	X	Y	(LOCAL AXES)
WING L.E.	0.000	0.000	
	0.000	1.000	

WING T.C. 1.000 0.000
 1.000 1.000
TAIL L.E. 0.000 0.000
 0.000 1.000
TAIL T.E. 1.000 0.000
 1.000 1.000

-DOX DIMENSIONS- D1 (LENGTH) = 6.63324958E-02 D1 / BETA (WIDTH) = 1.00000000E-01

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

DOX COCE PATTERN
MACH 1.2000000

2 4 6 8 10 12 14 16

19	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
20	1	1	1	1	1	1	1	1	1	1	1	1	1	1	2
21	1	1	1	1	1	1	1	1	1	1	1	1	1	1	2
22	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
23	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
24	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
25	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
26	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
27	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
28	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
29	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
30	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
31	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
32	1	1	1	1	1	1	1	1	1	1	1	1	1	2	2
33	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1

CODE - 1 = PLANFORM DOX
2 = DIAPHRAGM BOX
3 = WAKE BOX

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

BOX CODE PATTERN
MACH 1.2000000

CODE - 1 = PLANFORM BOX
2 = DIAPHRAGM BOX
3 = WAKE BOX

2 4 6 8 10 12 14 16 18 20 22 24 26

1 1 2 2 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 2
3 1 1 1 1 1 1 1 1 1 2 2
4 1 1 1 2 1 1 1 1 1 2 2
5 1 1 1 2 1 1 1 1 1 2 2 2
6 1 1 1 2 1 1 1 1 1 2 2 2 2
7 1 1 1 2 1 2 1 1 1 2 2 2 2 2
8 1 1 1 2 1 2 1 1 1 2 2 2 2 2 2
9 1 1 1 2 1 2 1 1 1 2 2 2 2 2 2
10 1 1 1 2 1 2 1 1 1 2 2 2 2 2 2
11 1 1 1 2 1 2 1 1 1 2 2 2 2 2 2
12 1 1 1 2 1 2 1 1 1 2 2 2 2 2 2 2
13 1 1 1 2 1 2 1 1 1 2 2 2 2 2 2 2
14 2 2 1 1 1 1 1 1 1 2 2 2 2 2 2 2
15 1 1 1 1 1 1 1 1 1 2 2 2 2 2 2 2
16 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
17 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
18 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
19 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
20 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
21 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
22 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
23 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
24 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
25 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
26 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
27 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
28 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2 2 2
29 3 3 3 3 3 3 3 3 3 3 3 2 2 2 2 2 2

ENTERING PROGRAM MODES CURRENT ELAPSED TIME IS CF = 1.894, FF = 45.123

SAMPLE CASE ---- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

MODE SHAPE NUMBER 2
MACH NUMBER = 1.200000

MODAL POLYNOMIAL COEFFICIENTS
FROM CARD INPUT

CONSTANT

0.

X **1 Y **0 X **0 Y **1
1.0000E+00

0.

X **2 Y **0 X **1 Y **1 X **0 Y **2
0.

0.

MODAL POLYNOMIAL COEFFICIENTS

FROM CARD INPUT

CONSTANT

0.

X **1 Y **0 X **0 Y **1
0.

0.

DEFLECTIONS X 1.0E 1

	1	2	3	4	5	6	7	8	9	10
1	.332	.332	.332	.332	.332	.332	.332	.332	.332	.332
2	.995	.995	.995	.995	.995	.995	.995	.995	.995	.995
3	1.658	1.658	1.658	1.658	1.658	1.658	1.658	1.658	1.658	1.658
4	2.322	2.322	2.322	2.322	2.322	2.322	2.322	2.322	2.322	2.322
5	2.985	2.985	2.985	2.985	2.985	2.985	2.985	2.985	2.985	2.985
6	3.648	3.648	3.648	3.648	3.648	3.648	3.648	3.648	3.648	3.648
7	4.312	4.312	4.312	4.312	4.312	4.312	4.312	4.312	4.312	4.312
8	4.975	4.975	4.975	4.975	4.975	4.975	4.975	4.975	4.975	4.975
9	5.638	5.638	5.638	5.638	5.638	5.638	5.638	5.638	5.638	5.638

10	6.302	6.302	6.302	6.302	6.302	6.302	6.302	6.302	6.302
11	6.965	6.965	6.965	6.965	6.965	6.965	6.965	6.965	6.965
12	7.628	7.628	7.628	7.628	7.628	7.628	7.628	7.628	7.628
13	8.292	8.292	8.292	8.292	8.292	8.292	8.292	8.292	8.292
14	8.955	8.955	8.955	8.955	8.955	8.955	8.955	8.955	8.955
15	9.618	9.618	9.618	9.618	9.618	9.618	9.618	9.618	9.618
16	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
20	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
21	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
22	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
23	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
24	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
25	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
26	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
27	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
28	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
29	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
30	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
31	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
32	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
33	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000

SLOPES X 1.0E 0

1	2	3	4	5	6	7	8	9	10
1	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
2	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
3	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
4	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
5	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
6	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
7	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
8	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
9	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
10	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
11	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
12	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
13	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
14	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
15	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
19	0.500	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000

20	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
21	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
22	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
23	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
24	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
25	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
26	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
27	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
28	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
29	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
30	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
31	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
32	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
33	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000

ENTERING PROGRAM AIC CURRENT ELAPSED TIME IS CF = 2.555, FP = 44.910

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

AIC CALCULATIONS

MACH =	1.20000	K1 =	.0331662	ERR = 1.00000E-02	EL = 4.00	YBAR = 0.00	
NU	MU	C		W		V	
		VELOCITY POTENTIAL COEFFICIENT		UPWASH COEFFICIENT		SIDEWASH COEFFICIENT	
0	-1	0.	0.	0.	0.	0.	0.
1	1	0.	0.	0.	0.	0.	0.
2	2	0.	0.	0.	0.	0.	0.
3	3	0.	0.	0.	0.	0.	0.
4	4	0.	0.	0.	0.	0.	0.
5	5	0.	0.	0.	0.	0.	0.
6	6	0.	0.	0.	0.	0.	0.
7	7	0.	0.	0.	0.	0.	0.
8	8	0.	0.	0.	0.	0.	0.
9	9	0.	0.	0.	0.	0.	0.
10	10	0.	0.	0.	0.	0.	0.
11	11	0.	0.	0.	0.	0.	0.
12	12	0.	0.	0.	0.	0.	0.
13	13	0.	0.	0.	0.	0.	0.
14	14	0.	0.	0.	0.	0.	0.
15	15	0.	0.	0.	0.	0.	0.
16	16	0.	0.	0.	0.	0.	0.
17	17	0.	0.	0.	0.	0.	0.
18	18	0.	0.	0.	0.	0.	0.
19	19	0.	0.	0.	0.	0.	0.
20	20	0.	0.	0.	0.	0.	0.
21	21	0.	0.	0.	0.	0.	0.
22	22	0.	0.	0.	0.	0.	0.
23	23	0.	0.	0.	0.	0.	0.
24	24	0.	0.	0.	0.	0.	0.
25	25	0.	0.	0.	0.	0.	0.
26	26	0.	0.	0.	0.	0.	0.
27	27	0.	0.	0.	0.	0.	0.
28	28	0.	0.	0.	0.	0.	0.
29	29	0.	0.	0.	0.	0.	0.
30	30	0.	0.	0.	0.	0.	0.
31	31	0.	0.	0.	0.	0.	0.
32	32	0.	0.	0.	0.	0.	0.
33	33	0.	0.	0.	0.	0.	0.
34	34	0.	0.	0.	0.	0.	0.
35	35	0.	0.	0.	0.	0.	0.
36	36	0.	0.	0.	0.	0.	0.
37	37	0.	0.	0.	0.	0.	0.
38	38	0.	0.	0.	0.	0.	0.
39	39	0.	0.	0.	0.	0.	0.
40	40	0.	0.	0.	0.	0.	0.
41	41	0.	0.	0.	0.	0.	0.
42	42	0.	0.	0.	0.	0.	0.
43	43	0.	0.	0.	0.	0.	0.
44	44	0.	0.	0.	0.	0.	0.
45	45	0.	0.	0.	0.	0.	0.
46	46	0.	0.	0.	0.	0.	0.
47	47	0.	0.	0.	0.	0.	0.
48	48	0.	0.	0.	0.	0.	0.
49	49	0.	0.	0.	0.	0.	0.
50	50	0.	0.	0.	0.	0.	0.
51	51	0.	0.	0.	0.	0.	0.
52	52	0.	0.	0.	0.	0.	0.
53	53	0.	0.	0.	0.	0.	0.
54	54	0.	0.	0.	0.	0.	0.
55	55	0.	0.	0.	0.	0.	0.
56	56	0.	0.	0.	0.	0.	0.
57	57	0.	0.	0.	0.	0.	0.
58	58	0.	0.	0.	0.	0.	0.
59	59	0.	0.	0.	0.	0.	0.
60	60	0.	0.	0.	0.	0.	0.
61	61	0.	0.	0.	0.	0.	0.
62	62	0.	0.	0.	0.	0.	0.
63	63	0.	0.	0.	0.	0.	0.
64	64	0.	0.	0.	0.	0.	0.
65	65	0.	0.	0.	0.	0.	0.
66	66	0.	0.	0.	0.	0.	0.
67	67	0.	0.	0.	0.	0.	0.
68	68	0.	0.	0.	0.	0.	0.
69	69	0.	0.	0.	0.	0.	0.
70	70	0.	0.	0.	0.	0.	0.
71	71	0.	0.	0.	0.	0.	0.
72	72	0.	0.	0.	0.	0.	0.
73	73	0.	0.	0.	0.	0.	0.
74	74	0.	0.	0.	0.	0.	0.
75	75	0.	0.	0.	0.	0.	0.
76	76	0.	0.	0.	0.	0.	0.
77	77	0.	0.	0.	0.	0.	0.
78	78	0.	0.	0.	0.	0.	0.
79	79	0.	0.	0.	0.	0.	0.
80	80	0.	0.	0.	0.	0.	0.
81	81	0.	0.	0.	0.	0.	0.
82	82	0.	0.	0.	0.	0.	0.
83	83	0.	0.	0.	0.	0.	0.
84	84	0.	0.	0.	0.	0.	0.
85	85	0.	0.	0.	0.	0.	0.
86	86	0.	0.	0.	0.	0.	0.
87	87	0.	0.	0.	0.	0.	0.
88	88	0.	0.	0.	0.	0.	0.
89	89	0.	0.	0.	0.	0.	0.
90	90	0.	0.	0.	0.	0.	0.
91	91	0.	0.	0.	0.	0.	0.
92	92	0.	0.	0.	0.	0.	0.
93	93	0.	0.	0.	0.	0.	0.
94	94	0.	0.	0.	0.	0.	0.
95	95	0.	0.	0.	0.	0.	0.
96	96	0.	0.	0.	0.	0.	0.
97	97	0.	0.	0.	0.	0.	0.
98	98	0.	0.	0.	0.	0.	0.
99	99	0.	0.	0.	0.	0.	0.
100	100	0.	0.	0.	0.	0.	0.
101	101	0.	0.	0.	0.	0.	0.
102	102	0.	0.	0.	0.	0.	0.
103	103	0.	0.	0.	0.	0.	0.
104	104	0.	0.	0.	0.	0.	0.
105	105	0.	0.	0.	0.	0.	0.
106	106	0.	0.	0.	0.	0.	0.
107	107	0.	0.	0.	0.	0.	0.
108	108	0.	0.	0.	0.	0.	0.
109	109	0.	0.	0.	0.	0.	0.
110	110	0.	0.	0.	0.	0.	0.
111	111	0.	0.	0.	0.	0.	0.
112	112	0.	0.	0.	0.	0.	0.
113	113	0.	0.	0.	0.	0.	0.
114	114	0.	0.	0.	0.	0.	0.
115	115	0.	0.	0.	0.	0.	0.
116	116	0.	0.	0.	0.	0.	0.
117	117	0.	0.	0.	0.	0.	0.
118	118	0.	0.	0.	0.	0.	0.
119	119	0.	0.	0.	0.	0.	0.
120	120	0.	0.	0.	0.	0.	0.
121	121	0.	0.	0.	0.	0.	0.
122	122	0.	0.	0.	0.	0.	0.
123	123	0.	0.	0.	0.	0.	0.
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12	10	-1.52506711E-02	5.43187266E-02	-3.38304633E-03	1.16880687E-02	-8.55318583E-03	2.95400203E-02
12	9	-1.031515072E-02	3.70290494E-02	-1.33820525E-03	4.74458707E-03	-3.02684968E-03	1.07305559E-02
12	8	-8.05661514E-03	2.89904861E-02	-8.46758430E-04	3.02522663E-03	-1.69955143E-03	6.07168542E-03
12	7	-4.0828451E-02	-6.30077270E-04	2.25950416E-03	-1.10566269E-03	3.96486125E-03	
12	6	-5.76180714E-03	2.07657237E-02	-5.11573400E-04	1.83850727E-03	-7.69132814E-04	2.76406283E-03
12	5	-5.10897688E-03	1.84179437E-02	-4.39299702E-04	1.58092128E-03	-5.0262143E-04	1.98021174E-03
12	4	-7.64105922E-03	1.67332808E-02	-3.92716140E-04	1.41454784E-03	-3.93483559E-04	1.41729147E-03
12	3	-4.31064695E-03	1.55428127E-02	-3.62239121E-04	1.30554646E-03	-2.72195355E-04	9.81011826E-04
12	2	-7.05014224E-03	1.47479566E-02	-3.42948235E-04	1.23648864E-03	-1.7179790E-04	6.19405321E-04
12	1	-3.96343727E-03	1.42911603E-02	-3.32229523E-04	1.19809594E-03	-8.32140693E-05	3.00086475E-04
12	0	-3.92213375E-03	1.41420631E-02	-3.28781103E-04	1.18576433E-03	0.	0.
12	-1	-3.96343727E-03	1.42911603E-02	-3.32229523E-04	1.19809594E-03	-8.32140693E-05	-3.00086475E-04
12	-2	-4.09014224E-03	1.47479566E-02	-3.42948235E-04	1.23648864E-03	-1.7179790E-04	-6.19405321E-04
12	-3	-4.31064695E-03	1.55428127E-02	-3.62239121E-04	1.30554646E-03	-2.72195355E-04	-9.81011826E-04
12	-4	-7.64105922E-03	1.67332808E-02	-3.92716140E-04	1.41454784E-03	-3.93483559E-04	-1.41729147E-03
12	-5	-4.31064695E-03	1.55428127E-02	-4.39299702E-04	1.58092128E-03	-5.0262143E-04	-9.8021174E-03
12	-6	-7.6180714E-03	2.07657237E-02	-5.11573400E-04	1.83850727E-03	-7.69132814E-04	-2.76406283E-03
12	-7	-6.68576228E-03	2.40828451E-02	-6.30077270E-04	2.25950416E-03	-1.0566569E-03	-3.96486125E-03

12	-8	-8.05661514E-03	2.89954861E-02	-8.46758430E-04	3.02522663E-03	1.69955143E-03	-6.07168542E-03
12	-9	-1.03153072E-02	3.70293494E-02	-1.33802523E-03	4.74456787E-03	3.02684968E-03	-1.07305559E-02
12	-10	-1.52506711E-02	5.43187266E-02	-3.38304633E-03	1.16860687E-02	6.55318683E-03	-2.95400203E-02
12	-11	-2.62917139E-02	9.55031131E-02	1.02580332E-03	4.09961357E-05	-3.50152909E-03	2.07692808E-03
12	-12	-3.72776120E-03	1.60534916E-02	5.802061141E-03	-2.40293415E-02	-1.70533830E-02	7.04581208E-02
13	13	-2.37982443E-03	1.95553051E-02	3.11216511E-03	-2.32796233E-02	9.91317750E-03	-7.42776169E-02
13	12	-1.52050711E-02	9.33755139E-02	3.40777951E-04	3.01418639E-03	1.40306475E-03	7.16239221E-03
13	11	-8.47075744E-03	5.068228967E-02	-1.67259381E-03	9.62191041E-03	-4.64421110E-03	2.67048472E-02
13	10	-5.71152866E-03	3.46555701E-02	-6.88665739E-04	4.1092874E-03	-1.72968514E-03	1.03199772E-02
13	9	-4.40961260E-03	2.68324055E-02	-4.37916073E-04	2.64354961E-03	-9.88367581E-04	5.96597701E-03
13	8	-3.60689733E-03	2.202191343E-02	-3.25424985E-04	1.975759521E-03	-6.52372674E-04	3.960633562E-03
13	7	-3.05762254E-03	1.865812756E-02	-2.63066930E-04	1.602317641E-03	-4.61255474E-04	2.80981791E-03
13	6	-2.66227705E-03	1.627572707E-02	-2.24440739E-04	1.37021537E-03	-3.37239926E-04	2.05880126E-03
13	5	-2.37139376E-03	1.44931720E-02	-1.98987432E-04	1.21657840E-03	-2.49140101E-04	1.52317359E-03
13	4	-2.15730557E-03	1.31636639E-02	-1.81720132E-04	1.11210912E-03	-1.82014008E-04	1.11389114E-03
13	3	-2.00343386E-03	1.22417056E-02	-1.700255667E-04	1.04124164E-03	-1.27726500E-04	7.6222C4177E-04
13	2	-1.69954413E-03	1.165373736E-02	-1.~2451508E-04	9.95292673E-04	-8.13619251E-05	4.98476564E-04
13	1	-1.53943247E-03	1.12370608E-02	-1.58183254E-04	9.69383543E-04	-3.96132471E-05	2.42757056E-04
13	0	-1.81974877E-03	1.11164365E-02	-1.56803425E-04	9.61002342E-04	0.	0.
13	-1	-1.63943247E-03	1.12370608E-02	-1.58183254E-04	9.69381543E-04	-2.42757056E-04	3.96132471E-05
13	-2	-1.69954413E-03	1.165373736E-02	-1.62451508E-04	9.95292673E-04	8.13619251E-05	-4.98476564E-04
13	-3	-2.00343386E-03	1.22417056E-02	-1.700255667E-04	1.04124164E-03	1.27726500E-04	-7.622204177E-04
13	-4	-2.15730557E-03	1.31636639E-02	-1.81720132E-04	1.11210912E-03	1.82014008E-04	-1.11389114E-03
13	-5	-2.37139376E-03	1.44931720E-02	-1.98987432E-04	1.21657840E-03	2.49140101E-04	-1.52317359E-03
13	-6	-2.66227705E-03	1.627572707E-02	-2.24440739E-04	1.37021537E-03	3.37239926E-04	-2.05880126E-03
13	-7	-3.05762254E-03	1.865812756E-02	-2.63066930E-04	1.60257641E-03	4.61255474E-04	-2.80981791E-03
13	-8	-3.60689733E-03	2.202191343E-02	-3.25424985E-04	1.975759521E-03	6.52372674E-04	-3.960633562E-03
13	-9	-4.40961260E-03	2.66224055E-02	-4.37916073E-04	2.64354961E-03	9.88367581E-04	-5.96597701E-03
13	-10	-5.71152866E-03	3.46555701E-02	-6.88665739E-04	4.1092874E-03	1.72968514E-03	-1.03199772E-02
13	-11	-6.47075744E-03	5.08828967E-02	-1.67259381E-03	9.62191041E-03	4.64421110E-03	-2.67048472E-02
13	-12	-1.52050771E-02	9.33755139E-02	3.40777951E-04	3.01418639E-03	-1.40306475E-03	-7.16239221E-03
13	-13	-2.37982443E-03	1.94553051E-02	3.11216511E-03	-2.32796233E-02	-9.91317750E-03	7.42776169E-02
14	14	-3.377197820E-04	2.21734022E-02	6.13390725E-04	-2.21130237E-02	2.09519971E-03	-7.62485125E-02
14	13	-4.88029980E-03	9.06444575E-02	4.24329426E-04	4.7996496E-03	1.53358715E-03	1.40154796E-02
14	12	-2.67813939E-03	4.75296373E-02	-5.08248412E-04	8.08657408E-03	-1.53853599E-03	2.4454889E-02
14	11	-1.76752479E-03	3.23150690E-02	-2.05463417E-04	3.58583402E-03	-5.67491190E-04	9.90055102E-03
14	10	-1.34114504E-03	2.417538405E-02	-1.28969852E-04	2.32330469E-03	-3.23336994E-04	5.82348819E-03
14	9	-1.07828293E-03	1.99791133E-02	-9.49001168E-05	1.73739101E-03	-2.13963810E-04	3.91659915E-03
14	8	-6.97576116E-04	1.66547183E-02	-7.60369332E-05	1.40549463E-03	-1.52321872E-04	2.81527999E-03
14	7	-7.66319627E-04	1.42210380E-02	-6.43051829E-05	1.19606537E-03	-1.12693768E-04	2.09592267E-03
14	6	-6.68289510E-04	1.23939742E-02	-5.64950716E-05	1.05526558E-03	-8.48566207E-05	1.58495940E-03
14	5	-5.94379382E-04	1.10115304E-02	-5.10936127E-05	9.57210891E-04	-6.39538869E-05	1.19809510E-03
14	4	-5.39068248E-04	9.97437849E-03	-4.73045206E-05	8.88083239E-04	-4.73719298E-05	8.8932316E-04
14	3	-4.986656938E-04	9.21954634E-03	-4.46777593E-05	8.39990809E-04	-3.35588745E-05	6.30935449E-04

SAMPLE CASE --- TWO ARE=2 SURF
ORI ZONTAL AND VERTICAL SEPARATIONS

AIC CALCULATIONS

NU	MU	--	MACH = 1.20000	K1 = .0331662	ERR = 1.00000E-02	EL = 0.00	YEAR = 0.00	V
			C	VELOCITY POTENTIAL COEFFICIENT	W	UFWASH COEFFICIENT	SIDEWASH COEFFICIENT	
0	0		-4.99669441E-01	1.35612131E-02	0.	0.	0.	
1	0		-3.83780388E-01	3.72653793E-02	0.	0.	0.	
1	-1		-3.02537716E-01	3.51774045E-02	0.	0.	0.	
2	0		-1.58224582E-01	3.41050490E-02	0.	0.	0.	
2	-1		-1.98775746E-01	4.21384433E-02	0.	0.	0.	
2	-2		-2.06053452E-01	4.75173856E-02	0.	0.	0.	
3	0		-9.83703386E-02	3.28543730E-02	0.	0.	0.	
3	-1		-1.05489007E-01	3.51833148E-02	0.	0.	0.	
3	-2		-1.47785247E-01	4.86337202E-02	0.	0.	0.	
3	-3		-1.62295376E-01	5.66261105E-02	0.	0.	0.	
4	0		-6.81525261E-02	3.13856404E-02	0.	0.	0.	
4	-1		-7.08290748E-02	3.26038898E-02	0.	0.	0.	
4	-2		-8.69774972E-02	3.72153085E-02	0.	0.	0.	
4	-3		-1.187733126E-01	5.39547333E-02	0.	0.	0.	
4	-4		-1.34046042E-01	6.38424648E-02	0.	0.	0.	
4	-5		-4.93566001E-02	2.96149940E-02	0.	0.	0.	
4	0		-5.56323716E-02	3.03790933E-02	0.	0.	0.	
4	-1		-5.502355263E-02	3.2992612E-02	0.	0.	0.	
4	-2		-6.55871198E-02	3.89671498E-02	0.	0.	0.	
4	-3		-9.815222109E-02	5.81981438E-02	0.	0.	0.	
4	-4		-1.12751838E-01	6.96207603E-02	0.	0.	0.	
4	-5		-3.63050688E-02	2.75563287E-02	0.	0.	0.	
4	0		-3.70563878E-02	2.85856988E-02	0.	0.	0.	
4	-1		-3.70563878E-02	2.85856988E-02	0.	0.	0.	
4	-2		-3.929595919E-02	2.98145968E-02	0.	0.	0.	
4	-3		-4.35152695E-02	3.32964611E-02	0.	0.	0.	
4	-4		-5.31898899E-02	4.02714174E-02	0.	0.	0.	
4	-5		-6.19476102E-02	6.14878343E-02	0.	0.	0.	
4	0		-9.52385400E-02	7.41890253E-02	0.	0.	0.	
4	-1		-2.66803249E-02	7.52527061E-02	0.	0.	0.	
4	-2		-2.75944000E-02	2.56432429E-02	0.	0.	0.	
4	-3		-2.84124374E-02	2.68859576E-02	0.	0.	0.	

-3.09101104E-02	2.92388864E-02
-3.52622140E-02	3.33509956E-02
-4.35461149E-02	4.11045928E-02
-6.82661173E-02	6.39065410E-02
-8.00574122E-02	7.76904920E-02
-1.93534078E-02	2.27776282E-02
-1.96065790E-02	2.35748510E-02
-2.04006205E-02	2.40068433E-02
-2.18546126E-02	2.57125717E-02
-2.42349520E-02	2.85024106E-02
-2.81504160E-02	3.30434407E-02
-3.53149521E-02	4.14460143E-02
-5.63226441E-02	6.55224972E-02
-6.64612250E-02	6.02531141E-02
-1.36536158E-02	2.01395719E-02
-1.38130948E-02	2.03743775E-02
-1.43061589E-02	2.11032178E-02
-1.51944494E-02	2.24073396E-02
-1.65676175E-02	2.44565463E-02
-1.87192445E-02	2.75883532E-02
-2.20996834E-02	3.25455198E-02
-2.81468341E-02	4.13922112E-02
-4.55880625E-02	6.64147805E-02
-5.40196170E-02	8.18676228E-02
-9.22935494E-03	1.74162143E-02
-9.32978337E-03	1.76055805E-02
-9.63939934E-03	1.61691284E-02
-1.01853736E-02	1.92178725E-02
-1.10215237E-02	2.079265504E-02
-1.22469657E-02	2.30969505E-02
-1.405217571E-02	2.64913718E-02
-1.68462984E-02	3.17355784E-02
-2.17694763E-02	4.09392936E-02
-3.57973866E-02	6.66196935E-02
-4.24590939E-02	8.26216235E-02
-5.63916773E-03	1.49664657E-02
-5.90078968E-03	1.73569420E-02
-7.62551603E-03	1.91435536E-02
-8.62053143E-03	1.52958242E-02
-8.119700687E-03	1.61239579E-02
-5.63916773E-03	1.49664657E-02
-6.916056609E-03	1.73569420E-02
-7.62551603E-03	1.91435536E-02
-8.62053143E-03	1.52958242E-02
-1.005333793E-02	2.522252670E-02
-1.223367151E-02	3.06771625E-02

14	-14	-3.16510175E-03	7.79147557E-02	0.
15	0	2.39657385E-04	4.51992876E-03	0.
15	-1	2.43895493E-04	4.59257037E-03	0.
15	-2	2.56161447E-04	4.81246030E-03	0.
15	-3	2.77124615E-04	5.18876595E-03	0.
15	-4	3.07616702E-04	5.73652946E-03	0.
15	-5	3.48932101E-04	6.47935363E-03	0.
15	-6	4.02957604E-04	7.12142277E-03	0.
15	-7	4.72467458E-04	8.70613507E-03	0.
15	-8	5.61571159E-04	1.03179566E-02	0.
15	-9	6.76935686E-04	1.24056678E-02	0.
15	-10	8.27843443E-04	1.51689756E-02	0.
15	-11	1.03363264E-03	1.89613281E-02	0.
15	-12	1.33109416E-03	2.45299598E-02	0.
15	-13	1.81457923E-03	3.38840598E-02	0.
15	-14	2.901054485E-03	5.88699629E-02	0.
15	-15	5.11245065E-03	7.50311841E-02	0.
16	0	3.90947265E-04	2.44086871E-03	0.
16	-1	4.00028484E-04	2.500347635E-03	0.
16	-2	4.30615044E-04	2.68137753E-03	0.
16	-3	4.81964054E-04	2.98997947E-03	0.
16	-4	5.56137462E-04	3.43757361E-03	0.
16	-5	6.56173024E-04	4.64136196E-03	0.
16	-6	7.86172916E-04	4.82632687E-03	0.
16	-7	9.51981429E-04	5.82796598E-03	0.
16	-8	1.16157726E-03	7.6974539E-03	0.
16	-9	1.4..855805E-03	8.7103139E-03	0.
16	-10	1.77..56494E-03	1.0762920E-02	0.
16	-11	2.2193525DE-03	1.35065064E-02	0.
16	-12	2.63027434E-03	1.7226170E-02	0.
16	-13	3.71636382E-03	2.26622338E-02	0.
16	-14	5.16722403E-03	3.17493622E-02	0.
16	-15	8.685666854E-03	5.56716323E-02	0.
16	-16	1.26955900E-02	7.15605086E-02	0.
17	0	1.58767531E-04	6.21654294E-04	0.
17	-1	1.7240938E-04	6.75254198E-04	0.
17	-2	2.13682504E-04	8.1778276E-04	0.
17	-3	2.83984684E-04	1.06903946E-03	0.
17	-4	3.856755563E-04	1.43250087E-03	0.
17	-5	5.22316763E-04	1.92594629E-03	0.
17	-6	6.96985929E-04	2.55254234E-03	0.
17	-7	9.22720363E-04	3.35251876E-03	0.
17	-8	1.20337675E-03	4.35652728E-03	0.
17	-9	1.55495316E-03	5.61442762E-03	0.

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 WING UPPER SURFACE NORMAL WASH
 (MACH 1.200 FED. FREQ.= .00000)

MODE SHAPE 2

ROW	CHORD 1		CHORD 2		CHORD 3		CHORD 4	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	6.63324958E-02	1.10000000E-03	6.63324958E-02	1.10000000E-03	6.63324958E-02	1.10000000E-03	6.63324958E-02	1.10000000E-03
2	6.63324958E-02	5.30000000E-03	6.63324958E-02	3.30000000E-03	6.63324958E-02	3.30000000E-03	6.63324958E-02	3.30000000E-03
3	6.63324958E-02	5.30000000E-03	6.63324958E-02	5.50000000E-03	6.63324958E-02	5.50000000E-03	6.63324958E-02	5.50000000E-03
4	6.63324958E-02	7.70000000E-03	6.63324958E-02	7.70000000E-03	6.63324958E-02	7.70000000E-03	6.63324958E-02	7.70000000E-03
5	6.63324958E-02	9.90000000E-03	6.63324958E-02	9.90000000E-03	6.63324958E-02	9.90000000E-03	6.63324958E-02	9.90000000E-03
6	6.63324958E-02	1.21000000E-02	6.63324958E-02	1.21000000E-02	6.63324958E-02	1.21000000E-02	6.63324958E-02	1.21000000E-02
7	6.63324958E-02	1.43000000E-02	6.63324958E-02	1.43000000E-02	6.63324958E-02	1.43000000E-02	6.63324958E-02	1.43000000E-02
8	6.63324958E-02	1.65000000E-02	6.63324958E-02	1.65000000E-02	6.63324958E-02	1.65000000E-02	6.63324958E-02	1.65000000E-02
9	6.63324958E-02	1.87000000E-02	6.63324958E-02	1.87000000E-02	6.63324958E-02	1.87000000E-02	6.63324958E-02	1.87000000E-02
10	6.63324958E-02	2.09000000E-02	6.63324958E-02	2.09000000E-02	6.63324958E-02	2.09000000E-02	6.63324958E-02	2.09000000E-02
11	6.63324958E-02	2.31000000E-02	6.63324958E-02	2.31000000E-02	6.63324958E-02	2.31000000E-02	6.63324958E-02	2.31000000E-02
12	6.63324958E-02	2.53000000E-02	6.63324958E-02	2.53000000E-02	6.63324958E-02	2.53000000E-02	6.63324958E-02	2.53000000E-02
13	6.63324958E-02	2.75000000E-02	6.63324958E-02	2.75000000E-02	6.63324958E-02	2.75000000E-02	6.63324958E-02	2.75000000E-02
14	6.63324958E-02	2.97000000E-02	6.63324958E-02	2.97000000E-02	6.63324958E-02	2.97000000E-02	6.63324958E-02	2.97000000E-02
15	6.63324958E-02	3.19000000E-02	6.63324958E-02	3.19000000E-02	6.63324958E-02	3.19000000E-02	6.63324958E-02	3.19000000E-02
16	4.69641560E-02	1.30749482E-02	4.93038417E-02	-1.22954610E-02	5.00166418E-02	-1.10011154E-02	5.13430373E-02	-1.06606568E-02
17	5.34053695E-02	-1.25650430E-02	5.35633434E-02	-1.1413468E-02	5.386617229E-02	-6.06011302E-03	5.404410732E-02	-9.10196770E-03
18	5.41381772E-02	-2.0217265E-02	5.43168889E-02	-1.92313635E-02	5.44424617E-02	-1.89244049E-02	5.46292480E-02	-2.1286790E-02
19	5.79352051E-02	-1.96826846E-02	5.78454023E-02	-1.8878027E-02	5.76530897E-02	-1.69356306E-02	5.69733823E-02	-1.68264435E-02
20	5.73734614E-02	-2.71512149E-02	5.71352697E-02	-2.71388376E-02	5.68168521E-02	-2.65568648E-02	5.62169272E-02	-2.6877261E-02
21	6.04599963E-02	-2.70321605E-02	6.02605942E-02	-2.62066415E-02	5.97677831E-02	-2.45199788E-02	5.84133933E-02	-2.65929915E-02
22	5.63434725C-G2	-3.42361761E-02	5.62811653E-02	-3.39618414E-02	5.77905313E-02	-3.30531375E-02	5.65157681E-02	-3.52168190E-02
23	6.11982717E-02	-3.39216002E-02	6.09598075E-02	-3.30714761E-02	6.03843690E-02	-3.12862399E-02	5.64167606E-02	-3.3057775E-02
24	5.82471646E-02	-4.04904736E-02	5.78703117E-02	-4.0106143E-02	5.71449832E-02	-3.90600117E-02	5.53246549E-02	-4.1097404E-02
25	6.03605679C-02	-4.00119142E-02	6.01769947E-02	-3.92066531C-02	5.93562001E-02	-3.75892667E-02	5.69321679E-02	-3.9567035E-02
26	5.66229102E-02	-4.59595304E-02	5.60746605E-02	-4.56291335E-02	5.52331335E-02	-4.44969255E-02	5.30926973E-02	-4.642125C-F-02
27	5.85342487E-02	-4.54800721E-02	5.81445607E-02	-4.47221366E-02	5.73012935E-02	-4.31999506E-02	5.46365132E-02	-4.51101665E-02
28	5.37129457E-02	-5.07605932E-02	5.32926932E-02	-5.03777050E-02	5.25696835E-02	-4.92341999E-02	5.02363327E-02	-5.10742794E-02
29	5.95032401E-02	-5.02673669E-02	5.51623216E-02	-4.961013166E-02	5.44977244E-02	-4.81416667E-02	5.17346319E-02	-4.99633399E-02

ROW	CHORD 5		CHORD 6		CHORD 7		CHORD 8	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	6.43324958E-02	1.10000000E-03	6.63324958E-02	1.10000000E-03	6.63324958E-02	1.10000000E-03	6.63324958E-02	1.10000000E-03
2	6.63324958E-02	3.30000000E-03	6.63324958E-02	3.30000000E-03	6.63324958E-02	3.30000000E-03	6.63324958E-02	3.30000000E-03
3	6.63324958E-02	5.30000000E-03	6.63324958E-02	5.50000000E-03	6.63324958E-02	5.90000000E-03	6.63324958E-02	5.30000000E-03
4	6.63324958E-02	7.70000000E-03	6.63324958E-02	7.70000000E-03	6.63324958E-02	7.70000000E-03	6.63324958E-02	7.70000000E-03

6	6.63324958E-02	9.90000000E-03	6.63324958E-02	9.90000000E-03
6	6.63324958E-02	1.21000000E-02	6.63324958E-02	1.21000000E-02
7	6.63324958E-02	1.43000000E-02	6.63324958E-02	1.43000000E-02
8	6.63324958E-02	1.65000000E-02	6.63324958E-02	1.65000000E-02
9	6.63324958E-02	1.87000000E-02	6.63324958E-02	1.87000000E-02
10	6.63324958E-02	2.09000000E-02	6.63324958E-02	2.09000000E-02
11	6.63324958E-02	2.31000000E-02	6.63324958E-02	2.31000000E-02
12	6.63324958E-02	2.53000000E-02	6.63324958E-02	2.53000000E-02
13	6.63324958E-02	2.75000000E-02	6.63324958E-02	2.75000000E-02
14	6.63324958E-02	2.97000000E-02	6.63324958E-02	2.97000000E-02
15	6.63324958E-02	3.19000000E-02	6.63324958E-02	3.19000000E-02
16	6.63324958E-02	3.41000000E-02	6.63324958E-02	3.41000000E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 WING UPFRONT SURFACE NORMAL WASH
 (MACH 1.200 FED. FREQ.= .500000)
 MODE SHAPE 2

PAGE CONTINUE

ROW	CHORD 5				CHORD 6				CHORD 7				CHORD 8				
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	
16	5.30176345E-02	-9.91932790E-04	5.50613145E-02	1.0453983E-02	5.69535875E-02	1.23276143E-02	5.88563377E-02	1.49505673E-02	5.32706146E-02	1.32164635E-02	6.03004025E-02	1.32164635E-02	6.20595209E-02	1.20595209E-02	6.03004025E-02	1.32164635E-02	
17	5.43792077E-02	-5.14658627E-03	5.64545668E-02	6.87301296E-03	5.86168279E-02	1.24004221E-02	7.2951221E-03	1.32706146E-02	6.20595209E-02	1.32706146E-02	6.03004025E-02	1.32706146E-02	6.20595209E-02	1.20595209E-02	6.03004025E-02	1.32706146E-02	
18	5.33270913E-02	-1.44606763E-02	5.75290195E-02	-1.0466120E-03	5.99711433E-02	5.32387866E-03	6.00786565E-02	5.32387866E-03	6.14715913E-02	4.29761401E-02	6.14715913E-02	4.29761401E-02	6.10546534E-02	1.06693474E-02	6.10546534E-02	1.06693474E-02	
19	5.67210304E-02	-1.47246991E-02	5.82865856E-02	-3.4050016E-03	6.01737174E-02	-1.99920037E-03	6.1737174E-02	-1.99920037E-03	6.14715913E-02	4.29761401E-02	6.14715913E-02	4.29761401E-02	6.10546534E-02	1.06693474E-02	6.10546534E-02	1.06693474E-02	
20	5.66734345E-02	-2.26377513E-02	5.86437176E-02	-9.5326289E-03	6.01737174E-02	-1.99920037E-03	6.1737174E-02	-1.99920037E-03	6.14715913E-02	4.29761401E-02	6.14715913E-02	4.29761401E-02	6.10546534E-02	1.06693474E-02	6.10546534E-02	1.06693474E-02	
21	5.77773567E-02	-2.21106111E-02	5.90086161E-02	-1.100866512E-02	5.99448603E-02	-4.79230965E-02	5.99448603E-02	-4.79230965E-02	6.10546534E-02	1.06693474E-02	6.10546534E-02	1.06693474E-02	6.05020749E-02	-3.49653951E-02	6.05020749E-02	-3.49653951E-02	
22	6.-62311153E-02	-2.-931615307E-02	5.-63963192E-02	-1.-68161852E-02	5.-9527644E-02	-9.-65812058E-02	5.-9527644E-02	-9.-65812058E-02	6.-01316667E-02	-6.-13267006E-02	6.-01316667E-02	-6.-13267006E-02	6.-01316667E-02	-6.-13267006E-02	6.-01316667E-02	-6.-13267006E-02	
23	5.-7376943E-02	-2.-87675354E-02	5.-86463618E-02	-1.-76121969E-02	5.-93340972E-02	-5.-93340972E-02	5.-94705656E-02	-5.-94705656E-02	5.-62468669E-02	-1.-62468669E-02	5.-62468669E-02	-1.-62468669E-02	5.-91504993E-02	-1.-23946730E-02	5.-91504993E-02	-1.-23946730E-02	
24	5.-52344972E-02	-3.-54524593E-02	5.-74446811E-02	-2.-32232346E-02	5.-85694276E-02	-5.-85694276E-02	5.-84255983E-02	-5.-84255983E-02	5.-79039299E-02	-1.-79039299E-02	5.-79039299E-02	-1.-79039299E-02	5.-829393476E-02	-1.-609393672E-02	5.-829393476E-02	-1.-609393672E-02	
25	5.-60449323E-02	-3.-46533143E-02	5.-77569677E-02	-2.-36946467E-02	5.-84255983E-02	-5.-84255983E-02	5.-73138244E-02	-2.-20966980E-02	5.-71513511E-02	-2.-348421761E-02	5.-8002525374E-02	-1.-80323749E-02	5.-68669856E-02	-2.-14771115E-02	5.-68669856E-02	-2.-14771115E-02	
26	5.-31226009E-02	-4.-92676561E-02	5.-595930567E-02	-2.-895930566E-02	5.-73138244E-02	-2.-20966980E-02	5.-556696660E-02	-2.-74147884E-02	5.-71513511E-02	-2.-348421761E-02	5.-8002525374E-02	-1.-80323749E-02	5.-64936414E-02	-2.-32117324E-02	5.-64936414E-02	-2.-32117324E-02	
27	5.-39440323E-02	-4.-03352086E-02	5.-62390169E-02	-2.-916060602E-02	5.-71513511E-02	-2.-348421761E-02	5.-68669856E-02	-2.-74147884E-02	5.-71513511E-02	-2.-348421761E-02	5.-8002525374E-02	-1.-80323749E-02	5.-68669856E-02	-2.-14771115E-02	5.-68669856E-02	-2.-14771115E-02	
28	5.-04532622E-02	-4.-37299653E-02	5.-39331747E-02	-3.-41017684E-02	5.-556696660E-02	-2.-556696660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	
29	5.-13339490E-02	-4.-91762067E-02	5.-42664754E-02	-3.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	5.-44715660E-02	-2.-44715660E-02	
ROW	CHORD 9				CHORD 10				CHORD 11				CHORD 12				
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	
	1	6.-63324958E-02	1.-10000000E-03	6.-63324958E-02	1.-10000000E-03	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.
	2	6.-63324958E-02	5.-30000000E-03	6.-63324958E-02	3.-30000000E-03	-4.-03191249E-02	2.-9095920E-03	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.
	3	6.-63324958E-02	5.-50000000E-03	6.-63324958E-02	5.-50000000E-03	-6.-37656304E-02	6.-71026764E-03	-3.-28286178E-03	1.-16913312E-03	1.-16913312E-03	1.-16913312E-03	1.-16913312E-03	1.-16913312E-03	1.-16913312E-03	1.-16913312E-03	1.-16913312E-03	1.-16913312E-03
	4	6.-63324958E-02	7.-70000000E-03	6.-63324958E-02	7.-70000000E-03	-8.-771610363E-02	1.-13970809E-02	-1.-2795218E-02	3.-64123429E-03	3.-64123429E-03	3.-64123429E-03	3.-64123429E-03	3.-64123429E-03	3.-64123429E-03	3.-64123429E-03	3.-64123429E-03	3.-64123429E-03
	5	6.-63324958E-02	9.-90000000E-03	6.-63324958E-02	9.-90000000E-03	-1.-03366260E-01	1.-57205176E-02	-1.-65889770E-02	5.-33256845E-03	5.-33256845E-03	5.-33256845E-03	5.-33256845E-03	5.-33256845E-03	5.-33256845E-03	5.-33256845E-03	5.-33256845E-03	5.-33256845E-03
	6	6.-63324958E-02	1.-21000000E-02	6.-63324958E-02	1.-21000000E-02	-1.-20361173E-01	2.-096524249E-02	-2.-42533793E-02	6.-62275973E-03	6.-62275973E-03	6.-62275973E-03	6.-62275973E-03	6.-62275973E-03	6.-62275973E-03	6.-62275973E-03	6.-62275973E-03	6.-62275973E-03
	7	6.-63324958E-02	1.-43000000E-02	6.-63324958E-02	1.-43000000E-02	-1.-43000000E-02	2.-5258249E-02	-2.-7230874E-02	1.-05007671E-02	1.-05007671E-02	1.-05007671E-02	1.-05007671E-02	1.-05007671E-02	1.-05007671E-02	1.-05007671E-02	1.-05007671E-02	1.-05007671E-02
	8	6.-63324958E-02	1.-63000000E-02	6.-63324958E-02	1.-63000000E-02	-1.-44169558E-01	3.-0298862E-02	-3.-313163394E-02	1.-39369867E-02	1.-39369867E-02	1.-39369867E-02	1.-39369867E-02	1.-39369867E-02	1.-39369867E-02	1.-39369867E-02	1.-39369867E-02	1.-39369867E-02
	9	6.-63324958E-02	2.-09000000E-02	6.-63324958E-02	2.-09000000E-02	-1.-52601945E-01	3.-40551597E-02	-3.-53305647E-02	1.-38599667E-02	1.-38599667E-02	1.-38599667E-02	1.-38599667E-02	1.-38599667E-02	1.-38599667E-02	1.-38599667E-02	1.-38599667E-02	1.-38599667E-02
	10	6.-63324958E-02	2.-31000000E-02	6.-63324958E-02	2.-31000000E-02	-1.-67666671E-01	3.-03100086E-02	-3.-9714186E-02	1.-9551100E-02	1.-9551100E-02	1.-9551100E-02	1.-9551100E-02	1.-9551100E-02	1.-9551100E-02	1.-9551100E-02	1.-9551100E-02	1.-9551100E-02
	11	6.-63324958E-02	2.-31000000E-02	6.-63324958E-02	2.-31000000E-02	-1.-74174867E-01	4.-10611266E-02	-4.-1200257E-02	2.-01096602E-02	2.-01096602E-02	2.-01096602E-02	2.-01096602E-02	2.-01096602E-02	2.-01096602E-02	2.-01096602E-02	2.-01096602E-02	2.-01096602E-02
	12	6.-63324958E-02	2.-53000000E-02	6.-63324958E-02	2.-53000000E-02	-1.-74174867E-01	4.-41086530E-02	-4.-45013661E-02	2.-32806971E-02	2.-32806971E-02	2.-32806971E-02	2.-32806971E-02	2.-32806971E-02	2.-32806971E-02	2.-32806971E-02	2.-32806971E-02	2.-32806971E-02
	13	6.-63324958E-02	2.-75000000E-02	6.-63324958E-02	2.-75000000E-02	-1.-78535748E-01	4.-55466611E-02	-4.-54914213E-02	2.-4202706E-02	2.-4202706E-02	2.-4202706E-02	2.-4202706E-02	2.-4202706E-02	2.-4202706E-02	2.-4202706E-02	2.-4202706E-02	2.-4202706E-02
	14	6.-63324958E-02	2.-97000000E-02	6.-63324958E-02	2.-97000000E-02	-1.-83251245E-01	4.-7054804E-02	-4.-7890367E-02	2.-63266740C-02	2.-63266740C-02	2.-63266740C-02	2.-63266740C-02	2.-63266740C-02	2.-63266740C-02	2.-63266740C-02	2.-63266740C-02	2.-63266740C-02
	15	6.-63324958E-02	3.-19000000E-02	6.-63324958E-02	3.-19000000E-02	-1.-86493052E-01	4.-69335537E-02	-4.-85373076E-02	2.-67945501E-02	2.-67945501E-02	2.-67945501E-02	2.-67945501E-02	2.-67945501E-02	2.-67945501E-02	2.-67945501E-02	2.-67945501E-02	2.-67945501E-02
16	6.-63324958E-02	1.-84679076E-02	6.-63324958E-02	1.-84679076E-02	-1.-90119958E-01	4.-68069160E-02	-5.-03659424E-02	2.-62128903E-02	2.-62128903E-02	2.-62128903E-02	2.-62128903E-02	2.-62128903E-02	2.-62128903E-02	2.-62128903E-02	2.-62128903E-02	2.-62128903E-02	
17	6.-23204471E-02	1.-85005614E-02	6.-33756342E-02	1.-85005614E-02	-2.-2620652E-02	5.-1093337E-02	-5.-09293631E-02	2.-80140906E-02	2.-80140906E-02	2.-80140906E-02	2.-80140906E-02	2.-80140906E-02	2.-80140906E-02	2.-80140906E-02	2.-80140906E-02	2.-80140906E-02	
18	6.-30009000E-02	1.-44866655E-02	6.-33756342E-02	1.-44866655E-02	-2.-0050602E-02	5.-68669856E-02	-5.-90092157E-02	1.-90092157E-02	1.-90092157E-02	1.-90092157E-02	1.-90092157E-02	1.-90092157E-02	1.-90092157E-02	1.-90092157E-02	1.-90092157E-02	1.-90092157E-02	
19	6.-34131501E-02	1.-47000510E-02	6.-375684671E-02	1.-47000510E-02	-1.-80003960E-02	5.-10003960E-02	-5.-14202272E-02	1.-14202272E-02	1.-14202272E-02	1.-14202272E-02	1.-						

20	6.31e432416E-02	1.03561105E-02	6.39996152E-02	1.556041861E-02	-1.66602713E-01	6.90761780E-02	-5.1409006676E-02	3.2233917635E-02
21	6.2153267E-02	6.2935286E-02	6.36310963E-03	6.363109638E-02	1.21686638E-02	-1.84303575E-01	7.50734456E-02	-5.02523410E-02
22	6.14292608E-02	2.07668036E-02	6.24568453E-03	6.245684535E-02	7.7766235E-03	-1.63175413E-01	7.9887033E-02	3.33336056CE-02
23	6.0964776E-02	-9.99126031E-04	6.19944156E-02	6.1994415643E-03	1.81492997E-01	6.41661910E-02	-4.92946749E-02	3.52649950E-02
24	6.02610246E-02	-4.7123641E-03	6.1197429E-02	9.56613651E-04	-1.79664154E-01	6.9052969E-02	-4.98551296E-02	3.61486938E-02
25	5.99461612E-02	-7.32674556E-03	6.07745349E-03	6.077453493E-03	-1.9475443E-03	-1.7735G702E-01	6.3614949E-02	5.24861702E-02
26	5.91174285E-02	-1.06877451E-02	5.99667379E-02	5.99667379E-02	-5.04681649E-03	-1.75067863E-01	9.68635518E-02	3.75344595C-02
27	5.8756932E-02	-1.30221590E-02	5.96653320E-02	5.96653320E-02	-7.686603120E-03	-1.71938660E-01	1.03716705E-01	-4.03770463E-02
28	5.7469399E-02	-1.61133529E-02	5.98362503E-02	5.98362503E-02	-1.05127273E-02	-1.69066777E-01	1.08791922E-01	-4.765776953E-02
29	5.74447492E-02	-1.62503798E-02	5.95207798E-02	5.95207798E-02	-1.29660019E-02	-1.65445011E-01	1.13589766E-01	-4.67697655E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 WING UPPER SURFACE NORMAL WASH

(MACH 1.200 RED. FREQ.= .90000)

MODE SHAPE 2

PAGE CONTINUED

ROW	CHORD 13		CHORD 14		CHORD 15		CHORD 16	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
4	-3.44373466E-03	1.09582333E-03	0.	0.	0.	0.	0.	0.
5	-7.24405673E-03	2.7697302E-03	-1.94322794E-03	9.14503279E-04	0.	0.	0.	0.
6	-1.1463490E-02	4.75141796E-03	-4.63556705E-03	2.37317247E-03	-1.33621092E-03	0.053635569E-04	0.	0.
7	-1.44401398E-02	6.72365208E-03	-6.94868535E-03	3.93003987E-03	-3.12042302E-03	2.05351456E-03	-9.40314423E-04	7.17394071E-04
8	-1.81664905E-02	9.05484645E-03	-9.47261486E-03	5.782023922E-03	-4.80292119E-03	3.43996838E-03	-2.20337043E-03	1.61624070E-03
9	-2.02131926E-02	1.09509815E-02	-1.12706190E-02	7.47664451E-03	-6.3960541E-03	4.99731558E-03	-3.37792331E-03	3.03931286E-03
10	-2.32317936E-02	1.33351588E-02	-1.33298884E-02	9.41936548E-03	-7.82872197E-03	6.54581970E-03	-4.53056056E-03	4.41260790E-03
11	-2.45529999E-02	1.49533110E-02	-1.45765125E-02	1.10071844E-02	-9.00767772E-03	8.09634408E-03	-3.48306080E-03	5.75757719E-03
12	-2.68635747E-02	1.71581298E-02	-1.61628975E-02	1.28415526E-02	-1.01029286E-02	9.62019276E-03	-6.35119173E-03	7.14663044E-03
13	-2.76730137E-02	1.83620450E-02	-1.69400700E-02	1.41529826E-02	-1.08807889E-02	1.03959437E-02	-7.01319218E-03	6.42304653E-03
14	-2.943235337E-02	2.022433594E-02	-1.812835837E-02	1.57496171E-02	-1.16728375E-02	1.28521301E-02	-7.60542406E-03	9.68264618E-03
15	-2.96641640E-02	2.093553591E-02	-1.657123134E-02	1.668331635E-02	-1.21382507E-02	1.3415927E-02	-6.01383648E-03	1.07675525E-02
16	-3.12326537E-02	2.23541012E-02	-1.94720907E-02	1.79584355E-02	-1.27059111E-02	1.45444494E-02	-8.39943021E-03	1.16067750E-02
17	-3.15438555E-02	2.25455495E-02	-1.97418835E-02	1.84647869E-02	-1.29896767E-02	1.52873042E-02	-8.64480352E-03	1.26291324E-02
18	-3.26370046E-02	2.34817603E-02	-2.04616947E-02	1.93748727E-02	-1.342123265E-02	1.69361994E-02	-8.89608598E-03	1.34087037E-02
19	-3.26359469E-02	2.36464018E-02	-2.07060665E-02	1.94660691E-02	-1.36409376E-02	1.6825469E-02	-9.07545007E-03	1.39392964E-02
20	-3.26643249E-02	2.50550314E-02	-2.11469786E-02	2.02693319E-02	-1.40545600E-02	1.69978187E-02	-9.30568956E-03	1.4453D753E-02
21	-3.22014537E-02	2.55999263E-02	-2.08702972E-02	2.05794628E-02	-1.41270485E-02	1.72172288E-02	-9.50493914E-03	1.47044530C-02
22	-3.19347106E-02	2.70052876E-02	-2.07607923E-02	2.15422036E-02	-1.41230280E-02	1.78460528E-02	-9.64058979E-03	1.50660003E-02
23	-3.10061574E-02	2.75224811E-02	-2.01632499E-02	2.16353212E-02	-1.38141986E-02	1.81072346E-02	-9.52991494E-03	1.535059958E-02
24	-3.14155437E-02	2.806935849E-02	-1.98879084E-02	2.27253158E-02	-1.354384222E-02	1.69978187E-02	-9.35848176E-03	1.57512844E-02
25	-3.17087568E-02	2.76185339E-02	-2.00256666E-02	2.23575239E-02	-1.30835092E-02	1.86058713E-02	-9.07068221E-03	1.59323004E-02
26	-3.19525492E-02	2.84245844E-02	-2.07941214E-02	2.24824962E-02	-1.35689014E-02	1.89658924E-02	-8.76404766E-03	1.62611932E-02
27	-3.17216686E-02	2.81994459E-02	-2.09429577E-02	2.22170399E-02	-1.42786537E-02	1.85661245E-02	0.	0.
28	-3.15969916E-02	2.89866994E-02	-2.11453559E-02	2.26414863E-02	0.	0.	0.	0.
ROW	CHORD 17		CHORD 18		CHORD 19		CHORD 20	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
8	-6.80367896E-04	6.46577963E-04	0.	0.	0.	0.	0.	0.
9	-1.36609326E-03	1.63111099E-03	-5.00370367E-03	5.90566475E-04	5.90566475E-04	0.	0.	0.
10	-2.43016460E-03	2.72071361E-03	-1.25025148E-03	1.47036362E-03	-3.58996464E-04	5.31656260E-04	0.	0.
11	-3.23673723E-03	5.93930114E-03	-1.75196522E-03	2.45469521E-03	-8.22442031E-04	1.32663563E-03	-2.56037860E-04	4.85044908E-04
12	-3.90666612E-03	5.13412743E-03	-2.31126680E-03	3.53736006E-03	-1.23591954E-03	2.20743583E-03	-5.7493281E-04	1.20177793E-03
13	-4.17622989E-03	6.34627040E-03	-2.75948579E-03	4.60117377E-03	-1.60958492E-03	3.17036589E-03	-8.47535689E-04	1.99452066E-03
14	-4.92396936E-03	7.48156522E-03	-3.12763024E-03	5.67425163E-03	-1.06873132E-03	4.12041401E-03	-1.07977463E-03	2.65952186E-03
15	-5.27247416E-03	8.55905455E-03	-3.39420425E-03	6.66791793E-03	-2.10191067E-03	5.06697526E-03	-1.22603278E-03	3.69967164E-03

16	-9.5362195CE-03	9.52779216E-03	-3.58969376E-03	7.61387747E-03	-2.23425160E-03	5.94106762E-03	-1.32058094E-03	4.03333333E-03
17	-3.7237293G18E-03	1.03907150C-02	-3.723729222E-03	6.4487514C-03	-2.32040128E-03	6.76224043E-03	-1.36217350C-03	5.302244163E-03
18	-5.66996431C-03	1.11394753C-03	-3.78611679E-03	9.20509CE-03	-2.32693800E-03	7.492831E-03	-1.32992370E-03	5.0203722E-03
19	-3.9549000C-03	1.17565319E-02	-3.65606277E-03	9.6357359E-03	-2.34520198E-03	6.14194766E-03	-1.30371396E-03	6.65024192E-03
20	-6.10416C-03	1.22699031C-02	-3.66300303C-03	1.0379604E-02	-2.3181928E-03	6.6825021E-03	-1.221757205E-03	7.20615631E-03
21	-6.21629381E-03	1.26426121E-02	-3.94833663E-03	1.07939106E-02	-2.31551503E-03	9.14635370E-03	-1.17650479E-03	7.675030305E-03
22	-6.35669605E-03	1.29393270E-02	-4.01009098E-03	1.11376211E-02	-2.3056547E-03	9.5065000E-03	-1.10604250E-03	8.0620447E-03
23	-6.4674110CE-03	1.3174623E-02	-4.13444493E-03	1.13766915E-02	-2.35466728E-03	9.79886666E-03	-1.09550927E-03	8.37657866E-03
24	-6.39760843E-03	1.34638659E-02	-4.17073026E-03	1.15987044E-02	-2.42197068E-03	1.00047719E-02	0.	0.
25	-6.29331690E-03	1.365266040E-02	0.	0.	0.	0.	0.	0.

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 WING UPPER SURFACE NORMAL WASH
 (MACH 1.200 RED. FREQ.= .50000)

PAGE CONTINUED

ROW	CHORD 21			CHORD 22			CHORD 23			CHORD 24		
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
12	-1.74717044E-04	4.41134319E-04	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.
13	-3.62337016E-04	1.06787407E-03	-1.10360272E-04	4.00858990E-04	0.	0.	0.	0.	0.	0.	0.	0.
14	-5.42669730E-04	1.60198221E-03	-2.29793159E-04	9.63631432E-04	-5.67005649E-05	3.63373162E-04	0.	0.	0.	0.	0.	0.
15	-6.59348064E-04	2.57422456E-03	-3.02209597E-04	1.62445034E-03	-1.07912744E-04	6.86690371E-04	-2.04470693E-05	3.20139270E-05	0.	0.	0.	0.
16	-7.10695662E-04	3.32035316E-03	-3.30306758E-04	2.31228605E-03	-1.07	1.2261E-04	1.46660629E-03	-1.267171C2E-03	7.97961650C-04	0.	0.	0.
17	-1.21670133E-04	4.53775333E-03	-3.115242567C-04	2.97251953C-03	-7.646334981E-05	2.07686864E-03	2.69335621E-03	1.30636654C-03	1.30636654C-03	0.	0.	0.
18	-1.5662744652C-04	4.72743576E-03	-2.2645465C-03	3.619115666C-03	3.07807933E-05	2.65147365E-03	1.43172565E-04	1.6463531EC-03	1.6463531EC-03	0.	0.	0.
19	-5.95249172C-04	5.3544239C-03	-1.36563385E-03	4.20650277E-03	1.36125841E-04	3.22025310E-03	2.56141846C-04	2.35903639C-03	2.35903639C-03	0.	0.	0.
20	-4.7775207CUC-04	5.892223467C-03	1.085C33442C-05	4.74381945E-05	3.05080543E-04	3.72424214C-03	0.	0.	0.	0.	0.	0.
21	-3.88257274C-04	6.37C59696C-03	1.35043501E-04	5.212626666C-03	0.	0.	0.	0.	0.	0.	0.	0.
22	-2.72659739C-04	6.76165449C-03	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.

ROW	CHORD 25			CHORD 26			CHORD 27			CHORD 28		
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
15	2.11855982C-05	2.95121951C-04	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.
17	6.25336556C-05	7.145944672C-04	3.34829454E-05	2.63852131E-04	0.	0.	0.	0.	0.	0.	0.	0.
18	1.622643667C-04	2.16460652C-03	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

TAIL OFFER SURFACE NORMAL WASH
(MACH 1.200 RCD. FREQ.= .30000)

MODE SHAPE 2

ROW	CHORD 1		CHORD 2		CHORD 3		CHORD 4	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
19	-5.91117621E-02	-2.47284538E-02	-5.64920791E-02	-2.46367453E-02	-5.725C4692E-02	-2.44095793E-02	-5.51534693E-02	-2.42748930E-02
20	-2.46461090E-02	7.62604131E-03	-2.46604922E-02	7.66129327E-03	-2.53310010E-02	5.95447931E-03	-2.6393979CE-02	2.21401631E-03
21	-3.06626114E-02	6.03011666E-03	-3.13032001E-02	4.98595339E-03	-3.08401114E-02	3.69355450E-03	-3.0020266AE-02	2.03175787E-03
22	-2.96066597E-02	1.20060762E-02	-2.86532315E-02	1.26432369E-02	-2.85690760E-02	1.16241815E-02	-2.812646226E-02	9.75105104E-03
23	-3.32291786E-02	1.222602512E-02	-3.40362409E-02	1.19307078E-02	-3.32165604E-02	9.95689165E-03	-3.21064546E-02	8.10691405E-03
24	-3.1102725E-02	1.77250011E-02	-3.05313275E-02	1.76149357E-02	-3.04158143E-02	1.66354085E-02	-2.91035322E-02	1.47214772E-02
25	-3.33730719E-02	1.7661909CE-02	-3.5376430CE-02	1.66124153E-02	-3.45626225E-02	1.51036125E-02	-3.33954026E-02	1.293606666E-02
26	-3.18115445E-02	2.23968962E-02	-3.12358641E-02	2.19958548E-02	-3.03487004E-02	2.09051624E-02	-2.93686729E-02	1.869435649E-02
27	-3.36790652E-02	2.212303691E-02	-3.53209339E-02	2.11576034E-02	-3.42314106E-02	1.96584395E-02	-2.26952760E-02	1.75115526E-02
28	-3.11340922E-02	2.60697632E-02	-3.04566532E-02	2.56364876E-02	-2.943493835E-02	2.45211349E-02	-2.86291956E-02	2.24151502E-02
29	-3.45987728E-02	2.58209947E-02	-3.42202788E-02	2.4951715E-02	-3.31852087E-02	2.35320405E-02	-3.18147717E-02	2.13563401E-02
30	-2.93767437E-02	2.91053778E-02	-2.66783488E-02	2.85903569E-02	-2.79622051E-02	2.74651740E-02	-2.70761231E-02	2.53743520E-02
31	-3.26652244E-02	2.89605176E-02	-3.23820666E-02	2.81642294E-02	-3.15036603E-02	2.67441611E-02	-3.03207664E-02	2.45637678E-02
32	-2.72162488E-02	2.15949597E-02	-2.68237195E-02	3.03103101E-02	-2.60196742E-02	2.90457003E-02	-2.33350361E-02	2.77613574E-02
33	-3.022563394E-02	3.16924432E-02	-3.024689669E-02	3.09226013E-02	-2.94557784E-02	2.94746310E-02	-2.6195966E-02	2.72679793E-02
ROW	CHORD 5		CHORD 6		CHORD 7		CHORD 8	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
19	-3.21239316E-02	-2.39566663E-02	-4.79293886E-02	-2.343556027E-02	-4.23363163E-02	-2.26869769E-02	-3.4967769CE-02	-2.122G51594E-02
20	-2.63862529E-02	-1.01561359E-03	-2.58952780E-02	-4.16937027E-03	-2.45644304E-02	-7.96759266E-03	-2.16729163E-02	-1.07745919E-02
21	-2.9223321E-02	-5.47973131E-04	-2.73445926E-02	-2.77955362E-03	-2.46483538E-02	-5.32290548E-03	-2.1606426E-02	-6.80976012E-03
22	-2.73426725E-02	7.101277312E-03	-2.670025056E-02	2.89857579E-03	-2.49032017E-02	-1.75247646E-03	-2.03645508E-02	-5.26908722E-03
23	-3.0492510E-02	5.00160599E-03	-2.80354566E-02	3.08316269E-03	-2.45566608E-02	-3.51199937E-06	-2.00119223E-02	-3.50518960E-03
24	-2.83361607E-02	1.16844286E-02	-2.64446779E-02	7.89227473E-03	-2.16806037E-02	3.49825337E-03	-1.32051360E-02	-6.75563461E-04
25	-3.10766332E-02	1.05926293E-02	-2.79477886E-02	7.71316302E-03	-2.386255776E-02	4.36741255E-02	-4.66927359E-04	-1.61356939E-02
26	-2.81170795E-02	1.57413325E-02	-2.61020249E-02	1.1835678E-02	-2.29706639E-02	7.38917661E-03	-1.8025267E-02	2.98956983E-03
27	-3.05765678E-02	1.46862223E-02	-2.75293651E-02	1.16519883E-02	-2.34166344E-02	7.92646070E-03	-1.61323607E-02	5.84939446E-03
28	-2.7130877E-02	1.93245717E-02	-2.51600203E-02	1.53230418E-02	-2.21926048E-02	1.06830386E-02	-1.73781630E-02	5.90809642E-03
29	-2.9378211E-02	1.85767642E-02	-2.66038766E-02	1.31261304E-02	-2.26697574E-02	1.10511199E-02	-1.73755408E-02	6.57607034E-03
30	-2.36558739E-02	2.229355932E-02	-2.39751045E-02	1.822512376E-02	-2.11095075E-02	1.349700537E-02	-1.63410037E-02	6.40865099E-03
31	-2.62661733E-02	2.17120168E-02	-2.54641924E-02	1.80637886E-02	-2.17556449E-02	1.37261493E-02	-1.65379133E-02	6.91770117E-03
32	-2.42776195E-02	2.47232123E-02	-2.26292002E-02	2.06877060E-02	-1.99585437E-02	1.56337227E-02	-1.56330476E-02	1.04966373E-02
33	-2.45563394E-02	2.43114603E-02	-2.41465343E-02	2.03405601E-02	-2.07147208E-02	1.60001375E-02	-1.59561423E-02	1.06696999E-02

ROW

CHORD 9

REAL

IMAGINARY

REAL

IMAGINARY

REAL

IMAGINARY

19	-2.55259243E-02	-1.97606723E-02	-1.631726275E-02	-1.69298755E-02	0.	0.	0.	0.
20	-1.61412634E-02	-1.22913595E-02	-9.90033148E-03	-1.36362670E-02	1.1074454E-02	1.0595349E-02	0.	0.
21	-1.74744322E-02	-1.32599861E-02	-1.02335505E-02	-1.41645819E-02	1.784711225E-02	1.25946556E-02	1.16311944E-03	6.26503715E-04
22	-1.37567273E-02	-9.-31559556E-03	-9.-84001651E-03	-1.-35098295E-02	2.-81171846E-02	1.-46408357E-02	4.-61711667E-03	2.-28636453E-03
23	-1.-9390138E-02	-6.-09035379C-03	-6.-39649826E-03	-1.-1206069E-03	3.-60612125E-02	1.-27059992E-02	6.-58320169E-03	1.-5810658E-03
24	-1.-39520516E-02	-5.-76332640E-03	-7.-17765922E-03	-9.-56339179E-03	4.-17276699E-02	7.-21638329E-03	1.-04482331E-02	1.-5942211E-03
25	-1.-35229599E-02	-4.-42223203E-03	-6.-76934638E-03	-6.-26180001E-03	4.-41397703E-02	-7.-34915422E-04	1.-15766873E-02	-6.-45210037E-04
26	-1.-26624036E-02	-2.-53040020E-03	-5.-8772163E-03	-6.-93507076E-03	4.-69572737E-02	-8.-37461989E-03	1.-33856910E-02	-5.-82981572E-03
27	-1.-25910566E-02	-1.-34149650E-03	-5.-62035202E-03	-5.-900032583E-03	4.-63529653E-02	-8.-6435010E-03	1.-39169239E-02	-8.-6435010E-03
28	-1.-20677130E-02	-2.-56646955C-03	-5.-06859491E-03	-4.-90644161E-03	4.-51072967E-02	-2.-74816020E-02	1.-24991795E-02	-1.-18614964E-02
29	-1.-21329590E-02	7.74236549E-04	-5.-20730714E-03	-4.-1.0311505E-03	4.-27370943E-02	-3.-542539760E-02	1.-18614964E-02	-1.-18614964E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 TAIL UPPER SURFACE NORMAL WASH

1 MACH 1.200
 RED. FREQ.= .50000 !

MODE SHAPE 2

PAGE CONTINUED

ROW	CHORD 9		CHORD 10		CHORD 11		CHORD 12	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
20	-1.15607068C-02	2.09G57433E-03	-4.75938727E-03	-3.29669287E-03	4.099175C-02	-4.4305773E-02	1.20979134E-02	-1.4684504C-02
21	-1.16434588C-02	2.75438604E-03	-4.91097138C-03	-2.60795656E-03	3.69399556E-02	-5.27032714E-02	1.04671532E-02	-1.7565544E-02
32	-1.0709377E-02	3.67724276E-03	-4.46767084E-03	-1.96746554E-03	3.39315537E-02	-5.99028115E-02	0.	0.
33	-1.11439713E-02	4.41252218E-03	-4.63870147E-03	-1.39306731E-03	0.	0.	0.	0.
ROW	CHORD 13		CHORD 14		CHORD 15		CHORD 16	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
22	1.18454473E-03	6.89342320E-04	0.	0.	0.	0.	0.	0.
23	2.719265923E-03	9.26935130E-04	7.51679787E-04	3.136D9382E-04	0.	0.	0.	0.
24	4.93033116E-03	8.93452950E-04	1.94020972E-03	4.52327671E-04	5.66205863E-04	1.70718127E-04	0.	0.
25	6.59107323E-03	1.00952742E-04	3.19889254E-03	1.39905863E-04	1.43961766E-03	1.59598195E-04	4.41015286E-04	6.17972168E-05
26	6.036444677C-03	-1.13509997E-03	4.65598576E-03	-4.09409200E-04	2.38709161E-03	-1.50703789E-04	1.09673289E-03	2.234143C5E-06
27	6.40911069E-03	-3.03584970E-03	5.33423333E-03	-1.53639139E-03	3.36667695E-03	-7.25777346E-04	1.82619153E-03	-3.29137268E-04
28	9.03934072E-03	-4.89632446E-03	5.84424444E-03	-2.02022265E-03	3.92396929E-03	-1.60534939E-03	0.	0.
29	6.44787231E-03	-7.1273630E-03	5.613996097E-03	-4.34614532E-03	0.	0.	0.	0.
30	6.02746330E-03	-9.13226562E-03	0.	0.	0.	0.	0.	0.
ROW	CHORD 17		CHORD 18		CHORD 19		CHORD 20	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
26	3.34947168C-04	2.6734339E-05						

SAMPLE CASE --- TWO AIR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
WING VELOCITY POTENTIALS
(MACH 1.200 RED. FREQ.= .00000)

ROW	CHORD 1			CHORD 2			CHORD 3			CHORD 4		
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04
2	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02
3	-3.27215061E-01	3.02371871E-02	-3.27215061E-01	3.02371871E-02	-3.27215061E-01	3.02371871E-02	-3.27215061E-01	3.02371871E-02	-3.27215061E-01	3.02371871E-02	-3.27215061E-01	3.02371871E-02
4	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02
5	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02
6	-6.86349473E-01	1.40572461E-01	-6.86349473E-01	1.40572461E-01	-6.86349473E-01	1.40572461E-01	-6.86349473E-01	1.40572461E-01	-6.86349473E-01	1.40572461E-01	-6.86349473E-01	1.40572461E-01
7	-7.92220306E-01	1.91153573E-01	-7.92220306E-01	1.91153573E-01	-7.92220306E-01	1.91153573E-01	-7.92220306E-01	1.91153573E-01	-7.92220306E-01	1.91153573E-01	-7.92220306E-01	1.91153573E-01
8	-6.95601234E-01	2.16507223E-01	-6.89601234E-01	2.46507223E-01	-6.89601234E-01	2.46507223E-01	-6.89601234E-01	2.46507223E-01	-6.89601234E-01	2.46507223E-01	-6.89601234E-01	2.46507223E-01
9	-9.77902331E-01	3.05162316E-01	-9.77902331E-01	3.05162316E-01	-9.68906048E-01	2.96662062E-01	-9.44152668E-01	2.96662062E-01	-9.44152668E-01	2.96584203E-01	-9.44152668E-01	2.96584203E-01
10	-1.05679945E+00	3.65561916E-01	-1.04942693E+00	3.54639767E-01	-1.02901545E+00	3.29053251E-01	-9.97376643E-01	3.29053251E-01	-9.97376643E-01	2.96151534E-01	-9.97376643E-01	2.96151534E-01
11	-1.12041939E+00	4.15341562E-01	-1.1039329E+00	3.89161759E-01	-1.07795893E+00	3.53321572E-01	-1.04147231E+00	3.53321572E-01	-1.04147231E+00	3.16694320E-01	-1.04147231E+00	3.16694320E-01
12	-1.16462969E+00	4.37437154E-01	-1.1483520E+00	4.14327181E-01	-1.11856163E+00	3.7503558E-01	-1.07037117E+00	3.7503558E-01	-1.07037117E+00	3.31909232E-01	-1.07037117E+00	3.31909232E-01
13	-1.195663799E+00	4.34075393E-01	-1.16244992E+00	4.20746755E-01	-1.15273322E+00	3.87680205E-01	-1.10961148E+00	3.87680205E-01	-1.10961148E+00	3.41277931E-01	-1.10961148E+00	3.41277931E-01
14	-1.22347731E+00	4.14653563E-01	-1.2091051E+00	4.04609431E-01	-1.18025693E+00	3.8365452E-01	-1.13651259E+00	3.8365452E-01	-1.13651259E+00	3.450505063E-01	-1.13651259E+00	3.450505063E-01
15	-1.24801429E+00	3.83210227E-01	-1.23333603E+00	3.75190194E-01	-1.20393559E+00	3.58650505E-01	-1.15961275E+00	3.58650505E-01	-1.15961275E+00	3.32953755E-01	-1.15961275E+00	3.32953755E-01
ROW	CHORD 5			CHORD 6			CHORD 7			CHORD 8		
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04	-6.63184769E-02	6.99825447E-04
2	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02	-1.97801522E-01	1.06032274E-02
3	-3.27215061E-01	5.02371871E-02	-3.27215061E-01	5.02371871E-02	-3.27215061E-01	5.02371871E-02	-3.27215061E-01	5.02371871E-02	-3.27215061E-01	5.02371871E-02	-3.27215061E-01	5.02371871E-02
4	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.90062499E-02	-4.52690913E-01	5.16509973E-02	-4.52690913E-01	5.16509973E-02
5	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	9.61220392E-02	-5.72620366E-01	6.90987110E-02	-5.72620366E-01	6.90987110E-02
6	-6.86349473E-01	1.40572461E-01	-6.86349473E-01	1.40572461E-01	-6.71233086E-01	1.315844297E-01	-6.29446362E-01	1.10566235E-01	-5.63371006GE-01	6.49252329E-02	-5.63371006GE-01	6.49252329E-02
7	-7.79422474E-01	1.61520612E-01	-7.44311654E-01	1.58761501E-01	-6.90146356E-01	1.30252859E-01	-6.13076893E-01	9.94721747E-02	-6.13076893E-01	9.94721747E-02	-6.13076893E-01	9.94721747E-02
8	-8.18256846E-01	2.32296217E-01	-8.39321915E-01	1.82037984E-01	-7.40682396E-01	1.48268497E-01	-6.54641651E-01	1.12474107E-01	-6.54641651E-01	1.12474107E-01	-6.54641651E-01	1.12474107E-01
9	-9.05616295E-01	2.37761541E-01	-9.53471496E-01	2.01621332E-01	-7.83107124E-01	1.63319440E-01	-6.69725341E-01	1.23635293E-01	-6.69725341E-01	1.23635293E-01	-6.69725341E-01	1.23635293E-01
10	-9.635303295E-01	2.58611438E-01	-9.94953395E-01	2.17991041E-01	-8.18628469E-01	1.75651617E-01	-7.14525266E-01	1.32370286E-01	-7.14525266E-01	1.32370286E-01	-7.14525266E-01	1.32370286E-01
11	-9.928050345E-01	2.74750360E-01	-9.29862763E-01	2.30396337E-01	-8.49054421E-01	1.49554421E-01	-7.44581289E-01	1.39355789E-01	-7.44581289E-01	1.39355789E-01	-7.44581289E-01	1.39355789E-01
12	-1.02607161E-00	2.86153595E-01	-9.59320766E-01	2.366466G79E-01	-7.45784679E-01	1.39965435E-01	-7.65941648E-01	1.43432165E-01	-7.65941648E-01	1.43432165E-01	-7.65941648E-01	1.43432165E-01
13	-1.04326707E-00	2.92734607E-01	-9.64435369E-01	2.43170154E-01	-8.96410264E-01	1.39533520E-01	-7.84144167E-01	1.44633803E-01	-7.84144167E-01	1.44633803E-01	-7.84144167E-01	1.44633803E-01
14	-1.07669460E-00	2.94424260E-01	-9.00615700E-00	2.43293291E-01	-8.15257652E-01	1.39505303E-01	-7.998695776E-01	1.42924930E-01	-7.998695776E-01	1.42924930E-01	-7.998695776E-01	1.42924930E-01
15	-1.10020318E-00	2.91253437E-01	-9.31691691E-01	2.39162874E-01	-8.31691691E-01	1.87866023E-01	-8.13618207E-01	1.3618207E-01	-8.13618207E-01	1.3618207E-01	-8.13618207E-01	1.3618207E-01

A 30

1	-c-6.5164769E-02	6.-99625447C-04	-6.-63164769E-02	6.-29862447E-04
2	-1.-9.601924E-01	1.-0.6032274E-02	-1.-57567972E-01	6.-60255268E-03
3	-2.-9.9774444E-01	2.-4.9866111E-02	-2.-0.6751114E-01	1.-19663897E-02
4	-3.-17.5335927E-01	3.-57494077E-02	-2.-4.6534137E-01	1.-76262658E-02
5	-4.-21240426E-01	4.-67293688E-02	-2.-6.0706427E-01	2.-3.3186776E-02
6	-4.-16.629557E-01	5.-73101529E-02	-3.-6.0695766E-01	2.-9229546E-02
7	-5.-0.6374426E-01	6.-7.642266E-02	-3.-3.0696476E-01	3.-4.623432E-02
8	-5.-3.5232659E-01	7.-6.16053423E-02	-3.-5.0743423E-01	3.-9.6982473E-02
9	-5.-6.23231350E-01	8.-3.8784754E-02	-3.-6.7349192E-01	4.-3.9528200E-02
10	-5.-8.2231425E-01	9.-0.0363538E-02	-3.-6.1661317E-01	4.-7.3447109E-02
11	-6.-0.0717036E-01	9.-4.4563009E-02	-3.-9.95389296E-01	4.-9.60212124E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 WING VELOCITY POTENTIALS
 { MACH 1.200 RED. FREQ. .50000 }

ROW	CHORD 9	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
12	-6.21203699E-01	9.66617335E-02	-4.03949368E-02	-4.03949368E-02	5.07493370E-02		
13	-6.35321633E-01	9.71596346E-02	-4.12613392E-01	-4.12613392E-01	5.04689462E-02		
14	-6.47451737E-01	9.51715199E-02	-4.20176166E-01	-4.20176166E-01	4.89128886E-02		
15	-6.36142035E-01	9.09024242E-02	-4.26709994E-01	-4.26709994E-01	4.582742291E-02		

PAGE CONTINUED

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 TAIL VELOCITY POTENTIALS
 (MACH 1.200 RED. FREQ.= .50000)
 MODE SHAPE 2

ROW	CHORD 1		CHORD 2		CHORD 3		CHORD 4	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
19	5.97433780E-02	2.30868510E-02	5.91216173E-02	2.3040104E-02	5.78606629E-02	2.2805074E-02	5.5775420E-02	2.27629510E-02
20	1.4647000E-01	2.26757565E-02	1.45057517E-01	2.77702938E-02	1.43125201E-01	2.9306397E-02	1.40045621C-01	3.31633513E-02
21	2.01552935E-01	-5.37579613E-03	2.01274466E-01	-2.3296529E-03	1.99487290E-01	2.6229453E-03	1.96206358E-01	1.04770671E-02
22	2.36580431E-01	-4.70793609E-02	2.55802371E-01	-4.30567946E-02	2.54021631E-01	-3.50364370E-02	2.49149730E-01	-2.41029791E-02
23	3.06446622E-01	-9.74763524E-02	3.06674214E-01	-9.26065968E-02	3.03440111E-01	-8.27094581E-02	2.97336669E-01	-6.72252971E-02
24	3.54545105E-01	-1.55595559E-01	3.52155685E-01	-1.4957367E-01	3.47013540E-01	-1.37057780E-01	3.38513665E-01	-1.1763929E-01
25	3.93230814E-01	-2.89259555E-01	3.90080594E-01	-2.11507473E-01	3.83344452E-01	-1.95719526E-01	3.72601632E-01	-1.7411769E-01
26	4.23409786E-01	-2.65771694E-01	4.1951492E-01	-2.75976075E-01	4.11430946E-01	-2.5630588E-01	3.94204272E-01	-2.26690102E-01
27	4.44295615E-01	-3.52016374E-01	4.39753714E-01	-3.40349537E-01	4.25972110E-01	-3.17693554E-01	4.02342643E-01	-2.9754662E-01
28	4.35780167E-01	-4.14691134E-01	4.46456305E-01	-4.01960064E-01	4.27108516E-01	-3.76336436E-01	4.01364477E-01	-3.45336762E-01
29	4.54332386E-01	-4.22205637E-01	4.4018505E-01	-4.5974499E-01	4.19265624E-01	-4.3215577E-01	3.9159452E-01	-4.00029068E-01
30	4.37143390E-01	-3.22134107E-01	4.256763532E-01	-5.1012302E-01	4.03243902E-01	-4.8627529E-01	3.7476354E-01	-4.5026651E-01
31	4.09093227E-01	-5.6011702E-01	4.0473676E-01	-5.53005807E-01	3.81123365E-01	-5.29899932E-01	3.5372298E-01	-4.94311337E-01
32	3.746986221E-01	-5.97697356E-01	3.66908706E-01	-5.86174744E-01	3.52562793E-01	-5.64599112E-01	3.298666116E-01	-5.30358671E-01
33	3.36640700E-01	-6.22766219E-01	3.31036615E-01	-6.12047491E-01	3.19023216E-01	-5.90246446E-01	3.01479551E-01	-5.57447765E-01
ROW	CHORD 5		CHORD 6		CHORD 7		CHORD 8	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
19	5.27392107E-02	2.25273225E-02	4.65333333E-02	2.212021277E-02	4.292565515E-02	2.15236602E-02	3.55402099E-02	2.02215324E-02
20	1.34111663E-01	3.641347763E-02	1.24965066E-01	3.95574962E-02	1.12265330E-01	4.2709130E-02	9.4437074E-02	4.47594904E-02
21	1.90130935E-01	2.035935159E-02	1.768977856E-01	3.007151568E-02	1.620902635E-01	4.03214375E-02	1.39069447E-01	5.11601962E-02
22	2.4104741E-01	-9.58081912E-03	2.682470655E-01	6.29327055E-03	2.08666999E-01	2.76731701E-02	1.75224443E-01	4.26959115E-02
23	2.66599371E-01	-4.77504760E-02	2.70773363E-01	-2.2649413E-02	2.43073815E-01	2.25471556E-03	2.00846176E-01	2.11764993E-02
24	3.236265622E-01	-9.13774307E-02	3.01546347E-01	-6.2412939E-02	2.65034587E-01	-3.42666084E-02	2.19522663E-01	-6.53362719E-03
25	3.522305935E-01	-1.41516212E-01	3.20329212E-01	-1.09505164E-01	2.79453219E-01	-7.58956766E-02	2.29167812E-01	-4.4457259E-02
26	3.66271073E-01	-1.95777859E-01	3.30605361E-01	-1.58664353E-01	2.86127406E-01	-1.20475046E-01	2.32058896E-01	-8.2814432E-02
27	3.71593593E-01	-2.1224008E-01	3.32654184E-01	-2.10339671E-01	2.8530D4635E-01	-1.66449224E-01	2.31124392E-01	-1.2125542E-01
28	3.66089911E-01	-3.0952223E-01	3.27263865E-01	-2.60388692E-01	2.80059837E-01	-2.10677344E-01	2.2590592E-01	-1.3863456E-01
29	3.57093773E-01	-3.57939545E-01	3.16924298E-01	-3.0807460E-01	2.70517198E-01	-2.53217199E-01	2.1756436E-01	-1.94026037E-01
30	3.41239763E-01	-4.05593333E-01	3.02268759E-01	-3.52236664E-01	2.57429565E-01	-2.92560919E-01	2.06580637E-01	-2.28346195E-01
31	3.21539231E-01	-4.7771530E-01	2.84265562E-01	-3.91774332E-01	2.41727585E-01	-3.28557770E-01	1.93798663E-01	-2.56994292E-01
32	2.99267426E-01	-4.93436670E-01	2.6420726E-01	-4.2535666E-01	2.24422900E-01	-3.59639340E-01	1.79326366E-01	-2.86202224E-01
33	2.79979171E-01	-5.12056126E-01	2.43261619E-01	-4.5415447E-01	2.06002668E-01	-3.86107239E-01	1.63776445E-01	-3.09499650E-01

19	$2 \cdot 60450000E-62$	$1 \cdot 96152632E-02$	$1 \cdot 7G750942E-02$	$1 \cdot 64678707E-02$
20	$7 \cdot 15521769E-02$	$4 \cdot 53770154E-02$	$4 \cdot 11306969E-02$	$3 \cdot 68189604E-02$
21	$1 \cdot 04603119E-01$	$5 \cdot 50523982E-02$	$5 \cdot 71184533E-02$	$4 \cdot 02993116E-02$
22	$1 \cdot 30132624E-01$	$4 \cdot 93593270E-02$	$7 \cdot 50743746E-02$	$4 \cdot 05107605E-02$
23	$1 \cdot 50055431E-01$	$3 \cdot 39156233E-02$	$8 \cdot 70721134E-02$	$3 \cdot 16234622E-02$
24	$1 \cdot 61251455E-01$	$1 \cdot 09206750E-02$	$9 \cdot 34423749E-02$	$1 \cdot 67225050E-02$
25	$1 \cdot 69662335E-01$	$-1 \cdot 69196835E-02$	$9 \cdot 76416707E-02$	$2 \cdot 79335586E-03$
26	$1 \cdot 71529728E-01$	$-4 \cdot 6625435E-02$	$9 \cdot 94733429E-02$	$-1 \cdot 60265241E-02$
27	$1 \cdot 69633588E-1$	$7 \cdot 68031329E-02$	$9 \cdot 75119620E-02$	$-3 \cdot 66526302E-02$
28	$1 \cdot 65623585E-01$	$-1 \cdot 06030355E-01$	$9 \cdot 42648411E-02$	$-5 \cdot 49432734E-02$
29	$1 \cdot 56666699E-01$	$-1 \cdot 33988080E-01$	$9 \cdot 073699591E-02$	$-7 \cdot 21029427E-02$

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 TAIL VELOCITY POTENTIALS
 (MACH 1.200 RED. FREQ.z .50000)
 MODE SHAPE 2

PAGE CONTINUED

ROW	CHORD 9	REAL	IMAGINARY	CHORD 10	REAL	IMAGINARY
30	1.50820379E-01	-1.60200176E-01	6.55962221E-02	-6.84275986E-02		
31	1.40996919E-01	-1.84255674E-01	7.94166208E-02	-1.03457324E-01		
32	1.29763770E-01	-2.0596326E-01	7.26917087E-02	-1.16666695E-01		
33	1.17931938E-01	-2.24555247E-01	6.52551056E-02	-1.26768131E-01		

ENTERING PROGRAM CHORDF CURRENT ELAPSED TIME IS CP = 122.634, PP = 56.058

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 SMOOTHED WING VELOCITY POTENTIALS
 (MACH 1.200 RED. FREQ. = .50000)

MODE SHAPE 2

ROW	CHORD 1		CHORD 2		CHORD 3		CHORD 4	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	-6.63164769E-02	6.99625447E-04	-6.63164769E-02	6.99625447E-04	-6.63164769E-02	6.99625447E-04	-6.63164769E-02	6.99625447E-04
2	-1.97118157E-01	1.27157593E-02	-1.97063941E-01	1.27223759E-02	-1.971181415E-01	1.28063146E-02	-1.97463806E-01	1.308661618E-02
3	-3.24356639E-01	2.76274330E-02	-3.25551934E-01	2.96614675E-02	-3.27860349E-01	3.30453819E-02	-3.29647122E-01	3.364443604E-02
4	-4.46830232E-01	5.23722716E-02	-4.50871795E-01	5.63349495E-02	-4.54319688E-01	6.23455157E-02	-4.57678050E-01	6.67555393E-02
5	-5.69164048E-01	8.86695931E-02	-5.71621530E-01	9.34466986E-02	-5.75448680E-01	1.05118235E-01	-5.78361484E-01	1.04244901E-01
6	-6.44089498E-01	1.36746220E-01	-6.46020340E-01	1.40147900E-01	-6.6836640E-01	1.44566246E-01	-6.68319682E-01	1.45640959E-01
7	-7.91721792E-01	1.93055618E-01	-7.92141143E-01	1.93598327E-01	-7.91286080E-01	1.93003593E-01	-7.05936163E-01	1.86314743E-01
8	-8.90037439E-01	2.53205022E-01	-8.66123104E-01	2.49727622E-01	-8.62260909E-01	2.42173613E-01	-8.70275106E-01	2.29638682E-01
9	-9.77151748E-01	3.12676568E-01	-9.72364162E-01	3.03796575E-01	-9.61345816E-01	2.66568094E-01	-9.41206748E-01	2.67146750E-01
10	-1.05159632E+00	3.62752835E-01	-1.04363355E+00	3.50956405E-01	-1.02174544E+00	3.22743423E-01	-9.99328279E-01	2.98691636E-01
11	-1.11259656E+00	4.01236152E-01	-1.10208434E+00	3.668202042E-01	-1.0802556E+00	3.59640745E-01	-1.04586654E+00	3.22605676E-01
12	-1.16034915E+00	4.23175879E-01	-1.14766593E+00	4.06003403E-01	-1.12230226E+00	3.76903316E-01	-1.0827043E+00	3.3766004E-01
13	-1.19629595E+00	4.26369966E-01	-1.162233663E+00	4.12706675E-01	-1.15417112E+00	3.85193656E-01	-1.11208329E+00	3.44221897E-01
14	-1.22341963E+00	4.12187232E-01	-1.20679613E+00	4.012653594E-01	-1.197796339E+00	3.76514702E-01	-1.13676531E+00	3.42415923E-01
15	-1.24646491E+00	3.64093495E-01	-1.23189805E+00	3.76714729E-01	-1.20296018E+00	3.60523966E-01	-1.15981590E+00	3.34262086E-01
ROW	CHORD 5		CHORD 6		CHORD 7		CHORD 8	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	-6.63164769E-02	6.99825447E-04	-6.63164769E-02	6.99825447E-04	-6.63164769E-02	6.99825447E-04	-6.63164769E-02	6.99825447E-04
2	-1.96219650E-01	1.35986591E-02	-1.99178143E-01	1.42478878E-02	-1.9934854E-01	1.452867950E-02	-1.97359615E-01	1.36503623E-02
3	-3.31648431E-01	3.71963629E-02	-3.31782358E-01	3.75643401E-02	-3.27178757E-01	3.55904975E-02	-3.1335077E-01	3.07418372E-02
4	-4.59639160E-01	6.82277631E-02	-4.56277017E-01	6.65536327E-02	-4.42775937E-01	6.02536010E-02	-4.12652518E-01	4.92501208E-02
5	-5.77356162E-01	1.03565655E-01	-5.67662406E-01	9.79173967E-02	-5.42778938E-01	8.57678355E-02	-4.95038964E-01	6.77768178E-02
6	-6.61620779E-01	1.40273786E-01	-6.63357301E-01	1.29082285E-01	-6.25897784E-01	1.1U151661E-01	-5.61656328E-01	6.51798730E-02
7	-7.71638657E-01	2.76996969E-01	-7.42763071E-01	1.58127810E-01	-6.92270670E-01	1.32085035E-01	-6.14222480E-01	1.00704298E-01
8	-8.46727012E-01	2.1041897E-01	-8.06827782E-01	1.63714187E-01	-7.4499650E-01	1.50193840E-01	-6.55242880E-01	1.13912676E-01
9	-9.08041773E-01	2.39357113E-01	-8.57610295E-01	2.05D10171E-01	-7.85563331E-01	1.55942494E-01	-6.87464203E-01	1.24616871E-01
10	-9.57304805E-01	2.626532162E-01	-8.97044381E-01	2.21620898E-01	-8.17862559E-01	1.77521739E-01	-7.13635236E-01	1.32605731E-01
11	-9.26731103E-01	2.796667193E-01	-9.305028607E-01	2.335157226E-01	-8.44771107E-01	1.85738019E-01	-7.36200623E-01	1.38562793E-01
12	-1.02675604E+00	2.90336672E-01	-9.56361449E-01	2.40956070E-01	-8.69921374E-01	1.90902619E-01	-7.57036476E-01	1.42085996E-01
13	-1.0376254C+00	2.95215416E-01	-9.35563396E-01	2.4423251E-01	-8.9225063E-01	1.33320817E-01	-7.77128768E-01	1.43426578E-01
14	-1.07980829E+00	2.95527578E-01	-1.00716305E+00	2.44546327E-01	-9.14656681E-01	1.93181031E-01	-7.96296387E-01	1.42624789E-01
15	-1.10233293E+00	2.93215636E-01	-1.02879021E+00	2.42029937E-01	-9.34565222E-01	1.90443967E-01	-8.12694929E-01	1.39521593E-01

ROW CHORD 9 REAL IMAGINARY ROW CHORD 10 REAL IMAGINARY

1	-6.-C318476C-02	6.-99625447C-04	-6.-C3164769C-02	6.-99825447E-04
2	-1.-87C18875E-01	1.-15968923E-02	-1.-43345422E-01	6.-43275503C-03
3	-2.-82C95229E-01	2.-35677992E-02	-2.-03047262E-01	1.-20415075E-02
4	-3.-37193181C-01	3.-47372587E-02	-2.-45862286E-01	1.-76860369E-02
5	-4.-15793190C-01	4.-02349894E-02	-2.-78644672E-01	2.-334653676E-02
6	-4.-62303019C-01	5.-12043569E-02	-3.-04869615E-01	2.-90472125E-02
7	-4.-99735000C-01	6.-73110300E-02	-3.-26954662E-01	3.-44972678E-02
8	-3.-30491397E-01	7.-62316302E-02	-3.-45293044E-01	3.-95090314E-02
9	-3.-36265493E-01	8.-37223920E-02	-3.-62964003E-01	4.-3630190E-02
10	-3.-79331092E-01	9.-93277668E-02	-3.-77955130E-01	4.-72265012E-02
11	-3.-97326589E-01	9.-356991743E-02	-3.-906682692E-01	4.-9469200E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 SMOOTHED WING VELOCITY POTENTIALS
 (MACH 1.200 RED. FREQ. = .50000)

MODE SHAPE 2

PAGE CONTINUED

ROW	CHORD 9		CHORD 10	
	REAL	IMAGINARY	REAL	IMAGINARY
12	-6.14416104E-01	9.38533425E-02	-4.03612967E-01	5.04614257E-02
13	-6.292926610E-01	9.61020248E-02	-4.10160572E-01	5.01163937E-02
14	-6.42332339E-01	9.44963311E-02	-4.11210600E-01	4.65259444E-02
15	-6.53756024E-01	9.11966615E-02	-4.23736945E-01	4.59264449E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 SMOOTHED TAIL VELOCITY POTENTIALS
 (MACH 1.200 RED. FREQ. = .50000)

MODE SHAPE 2

ROW	CHORD 1	REAL	IMAGINARY	CHORD 2	REAL	IMAGINARY	CHORD 3	REAL	IMAGINARY	CHORD 4	REAL	IMAGINARY
19	5.97433760E-02	2.35066510E-02	5.91216173E-02	2.30340104E-02	5.76856629E-02	2.26405074E-02	5.37754206E-02	2.2629510E-02	5.37754206E-02	2.2629510E-02	2.2629510E-02	5.37754206E-02
20	1.37139179E-01	1.70146754E-02	1.36236961E-01	1.76367243E-02	1.3671664E-01	1.96073137E-02	1.31929613E-01	1.97212464E-01	1.31929613E-01	1.97212464E-01	1.31929613E-01	1.97212464E-01
21	2.01505110E-01	-7.60417390E-03	2.01017365E-01	-5.85563356E-03	2.00669169E-01	-1.51755304E-03	1.97212464E-01	5.96041923E-03	1.97212464E-01	5.96041923E-03	1.97212464E-01	5.96041923E-03
22	2.3729536E-01	-4.73661693E-02	2.57146395E-01	-4.41446737E-02	2.56433384E-01	-3.71176455E-02	2.52884636E-01	-2.57274212E-02	2.52884636E-01	-2.57274212E-02	2.52884636E-01	-2.57274212E-02
23	3.06926959E-01	-9.46213134E-02	3.06545565E-01	-9.38770000E-02	3.04984542E-01	-8.36668310E-02	2.99613792E-01	-6.84432937E-02	2.99613792E-01	-6.84432937E-02	2.99613792E-01	-6.84432937E-02
24	3.51022130E-01	-1.58445810E-01	3.49737711E-01	-1.51735595E-01	3.46533979E-01	-1.36584073E-01	3.37625229E-01	-1.19020063E-01	3.37625229E-01	-1.19020063E-01	3.37625229E-01	-1.19020063E-01
25	3.89184169E-01	-2.22915939E-01	3.86217079E-01	-2.14519155E-01	3.79352155E-01	-1.98266139E-01	3.66852495E-01	-1.7487863E-01	3.66852495E-01	-1.7487863E-01	3.66852495E-01	-1.7487863E-01
26	4.192305679E-01	-2.891016222E-01	4.14616298E-01	-2.7923G274E-01	4.04135036E-01	-2.60151086E-01	3.670886015E-01	-2.35047435E-01	3.670886015E-01	-2.35047435E-01	3.670886015E-01	-2.35047435E-01
27	4.41221665E-01	-3.51813222E-01	4.34085351E-01	-3.43159556E-01	4.19624570E-01	-3.21756255E-01	3.96133717E-01	-2.91491434E-01	3.96133717E-01	-2.91491434E-01	3.96133717E-01	-2.91491434E-01
28	4.51986649E-01	-4.15733910E-01	4.42640543E-01	-4.03950425E-01	4.25090365E-01	-3.80929597E-01	3.99931651E-01	-3.49179012E-01	3.99931651E-01	-3.49179012E-01	3.99931651E-01	-3.49179012E-01
29	4.43666769E-01	-4.71034549E-01	4.39553477E-01	-4.59727650E-01	4.2085475E-01	-4.35896146E-01	3.92816621E-01	-4.0177207E-01	3.92816621E-01	-4.0177207E-01	3.92816621E-01	-4.0177207E-01
30	4.35050969E-01	-5.21147517E-01	4.24710021E-01	-5.09135864E-01	4.05160532E-01	-4.85305869E-01	3.77456800E-01	-4.50114203E-01	3.77456800E-01	-4.50114203E-01	3.77456800E-01	-4.50114203E-01
31	4.08706386E-01	-5.632020373E-01	3.99161280E-01	-5.51446654E-01	3.81214832E-01	-5.28281265E-01	3.552244362E-01	-4.93201984E-01	3.552244362E-01	-4.93201984E-01	3.552244362E-01	-4.93201984E-01
32	3.7375767E-01	-5.97577737E-01	3.69592568E-01	-5.86663813E-01	3.50547499E-01	-5.64464989E-01	3.20226121E-01	-5.30295066E-01	3.20226121E-01	-5.30295066E-01	3.20226121E-01	-5.30295066E-01
33	3.33955561E-01	-6.25813209E-01	3.26492378E-01	-6.15473988E-01	3.16806117E-01	-5.94067443E-01	3.16806117E-01	-5.94067443E-01	3.16806117E-01	-5.94067443E-01	3.16806117E-01	-5.94067443E-01
ROW	CHORD 5	REAL	IMAGINARY	CHORD 6	REAL	IMAGINARY	CHORD 7	REAL	IMAGINARY	CHORD 8	REAL	IMAGINARY
19	5.27392407E-02	2.25273225E-02	4.85333323E-02	2.21202277E-02	4.29256915E-02	2.15236602E-02	3.55402098E-02	2.02373245E-02	3.55402098E-02	2.02373245E-02	3.55402098E-02	2.02373245E-02
20	1.22981660E-01	2.73878696E-02	1.19055130E-01	3.1719941E-02	1.07386336E-01	3.62411181E-02	9.05652650E-02	3.99355150E-02	9.05652650E-02	3.99355150E-02	9.05652650E-02	3.99355150E-02
21	1.90709405E-01	1.52077630E-02	1.79352322E-01	2.55213535E-02	1.61652194E-01	3.59942943E-02	1.35933077E-01	4.41633653E-02	1.35933077E-01	4.41633653E-02	1.35933077E-01	4.41633653E-02
22	2.41310136E-01	-1.10354228E-02	2.29147688E-01	5.6587336E-03	2.05466065E-01	2.23719269E-02	1.71718108E-01	3.53106518E-02	1.71718108E-01	3.53106518E-02	1.71718108E-01	3.53106518E-02
23	2.86046902E-01	-4.03032674E-02	2.66525844E-01	-2.54842551E-02	2.36994491E-01	-2.59391333E-03	1.96340599E-01	1.66806498E-02	1.96340599E-01	1.66806498E-02	1.96340599E-01	1.66806498E-02
24	3.22226451E-01	-9.39973708E-02	2.97771442E-01	-6.5373169E-02	2.62756365E-01	-3.65656542E-02	2.16498077E-01	-1.1205619E-02	2.16498077E-01	-1.1205619E-02	2.16498077E-01	-1.1205619E-02
25	3.66749677E-01	-1.45159921E-01	3.17407723E-01	-1.11442642E-01	2.77551627E-01	-7.70377523E-02	2.27096676E-01	-4.53452198E-02	2.27096676E-01	-4.53452198E-02	2.27096676E-01	-4.53452198E-02
26	3.6217203CE-01	-1.99349659E-01	3.26153529E-01	-1.61111522E-01	2.84390547E-01	-1.21474977E-01	2.31162925E-01	-8.32997251E-02	2.31162925E-01	-8.32997251E-02	2.31162925E-01	-8.32997251E-02
27	3.60773935E-01	-2.54267549E-01	3.30894603E-01	-2.12070002E-01	2.84421943E-01	-1.67449671E-01	2.29674905E-01	-1.22759653E-01	2.29674905E-01	-1.22759653E-01	2.29674905E-01	-1.22759653E-01
28	3.27113213E-01	-3.07987563E-01	3.26654569E-01	-2.62165002E-01	2.78962268E-01	-1.2779832E-01	2.1675472E-01	-1.61675472E-01	2.1675472E-01	-1.61675472E-01	2.1675472E-01	-1.61675472E-01
29	3.17987495E-01	-3.56607447E-01	3.16566656E-01	-3.0965641E-01	2.69224344E-01	-2.55666679E-01	2.15493769E-01	-1.93950653E-01	2.15493769E-01	-1.93950653E-01	2.15493769E-01	-1.93950653E-01
30	3.12494641E-01	-4.05499490E-01	3.01643554E-01	-3.53235887E-01	2.55746150E-01	-2.94632999E-01	2.04402619E-01	-2.31775339E-01	2.04402619E-01	-2.31775339E-01	2.04402619E-01	-2.31775339E-01
31	3.22093156E-01	-4.4726129CE-01	2.63749763E-01	-3.9216164E-01	2.40319615E-01	-3.29659831E-01	1.91733964E-01	-2.61361067E-01	1.91733964E-01	-2.61361067E-01	1.91733964E-01	-2.61361067E-01
32	2.986662610E-01	-4.937786521E-01	2.635372069E-01	-4.2636497E-01	2.23419408E-01	-3.603253313E-01	1.779353549E-01	-2.6474061E-01	1.779353549E-01	-2.6474061E-01	1.779353549E-01	-2.6474061E-01
33	2.74964094E-01	-5.153156999E-01	2.42590774E-01	-4.365550636E-01	2.05531727E-01	-3.67942162E-01	1.63123515E-01	-3.1369708E-01	1.63123515E-01	-3.1369708E-01	1.63123515E-01	-3.1369708E-01

19	$2.00450060C-02$	$1.90352632E-02$	$1.70750952E-02$	$1.84678707E-02$
20	$6.807672746C-02$	$4.0466865C-02$	$3.94010563E-02$	$3.2652066E-02$
21	$1.02114556E-01$	$4.7994951C-02$	$5.84053396E-02$	$3.84330205E-02$
22	$1.26915746E-01$	$4.36974129E-02$	$7.37037680E-02$	$2.65091120E-02$
23	$1.47625763E-01$	$2.97647422E-02$	$8.51676260E-02$	$2.6456594E-02$
24	$1.60727357E-01$	$8.43668629E-03$	$9.29746121C-02$	$1.56689316E-02$
25	$1.60001596E-01$	$-1.80701031E-02$	$9.73609899E-02$	$-4.186324E-04$
26	$1.70466964E-01$	$-4.7663537E-02$	$9.87707399E-02$	$-1.85296066E-02$
27	$1.69036365E-01$	$-7.8439202E-02$	$9.77087114E-02$	$-3.74706227E-02$
28	$1.64466310E-01$	$-1.06767221E-01$	$9.47107739E-02$	$-5.62142021E-02$
29	$1.57595767E-01$	$-1.37392106E-01$	$9.02851685E-02$	$-7.39627211E-02$

SAMPLE CASE --- TWO ARE=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
SMOOTHED TAIL VELOCITY POTENTIALS
(MACH 1.200 RED. FREQ. = .50000)

MODE SHAPE 2

PACF CONTINUEC

ROW	CHORD 9	CHORD 10		
	REAL	IMAGINARY	REAL	IMAGINARY
30	1.49025423E-01	-1.63454692E-01	6.4913605E-02	-5.01922222E-02
31	1.39286652E-01	-1.66674961E-01	7.89026902E-02	-1.04702691E-01
32	1.247029376E-01	-2.07333039E-01	7.24345246E-02	-1.17669622E-01
33	1.17367196E-01	-2.26460991E-01	6.546681096E-02	-1.23692531E-01

ENTERING PROGRAM FORCES CURRENT ELAPSED TIME IS CP = 123.887, PP = 60.520

SAMPLE CASE --- TWO ARIZ SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

WING BOX LIFTS

(MACH 1.200 REC.FREQ. = .50000)
MODE SHAPE 2

ROW	CHORD 1		CHORD 2		CHORD 3		CHORD 4	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	-3.97215651E-01	1.35929082E-02	-3.97164069E-01	1.35606822E-02	-3.97293729E-01	1.37321396E-02	-3.9776688E-01	1.41211589E-02
2	-3.90560554E-01	2.14846215E-02	-3.92061378E-01	2.42544355E-02	-3.94846441E-01	2.90481196E-02	-3.98289469E-01	3.29325302E-02
3	-3.42232377E-01	2.73287559E-02	-3.05585731E-01	3.31932352E-02	-3.91012651E-01	4.19421057E-02	-3.96123532E-01	4.79751000E-02
4	-3.73999770E-01	4.71420761E-02	-3.76597369E-01	5.07711631E-02	-3.80166406E-01	5.5684131E-02	-3.81626565E-01	5.76313589E-02
5	-3.63553667E-01	7.02621740E-02	-3.63644440E-01	6.91906240E-02	-3.62883753E-01	6.64105472E-02	-3.57628774E-01	6.10680666E-02
6	-3.49193047E-01	8.86667527E-02	-3.46460615E-01	8.25822568E-02	-3.59888974E-01	7.1191080E-02	-3.27494699E-01	5.79060607E-02
7	-3.29783566E-01	9.63960259E-02	-3.24041214E-01	6.598353319E-02	-3.12686371E-01	6.80203355E-02	-2.93139145E-01	4.80380167E-02
8	-3.04666679E-01	6.98266685E-02	-2.96699338E-01	7.73171985E-02	-2.80592080E-01	5.8608130E-02	-2.57043102E-01	3.18162067E-02
9	-2.74726897E-01	6.74353668E-02	-2.65121991E-01	5.53721320E-02	-2.46765890E-01	3.43740137E-02	-2.21269653E-01	9.98026936E-03
10	-2.40460015E-01	<-9.86159160E-02	-2.3062626214E-01	2.07792311E-02	-2.12243053E-01	4.43150999E-03	-1.67779892E-01	-1.63250497E-02
11	-2.04074374E-01	-2.01684892E-02	-1.95215208E-01	-2.42098643E-02	-1.78966105E-01	-3.24139722E-02	-1.37951564E-01	-1.53401160E-02
12	-1.68504595E-01	-7.76130246E-02	-1.61634466E-01	-7.57410653E-02	-1.49216679E-01	-7.36774623E-02	-1.33551355E-01	-7.36826654E-02
13	-1.37741321E-01	-1.95967E-01	-1.33427898E-01	-1.2038291E-01	-1.25547310E-01	-1.16002663E-01	-1.15920402E-01	-1.04340022E-01
14	-1.16975956E-01	-1.86407452E-01	-1.14993963E-01	-1.75139513E-01	-1.11313294E-01	-1.55170897E-01	-1.06201211E-01	-1.26661360E-01
15	-1.079353723E-01	-2.09334434E-01	-1.07326390E-01	-1.97213450E-01	-1.05712941E-01	-1.74534127E-01	-1.02926376E-01	-1.40506035E-01
ROW	CHORD 5		CHORD 6		CHORD 7		CHORD 8	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	-3.96676214E-01	1.49240793E-02	-4.00321197E-01	1.59026779E-02	-4.00919262E-01	1.63261614E-02	-3.97379662E-01	1.53034156E-02
2	-4.01359833E-01	3.51986354E-02	-4.01626647E-01	3.565753226E-02	-3.97114546E-01	3.26421618E-02	-3.73799187E-01	2.55540753E-02
3	-3.97324419E-01	4.9191791E-02	-3.91347569E-01	4.5675615E-02	-3.70198477E-01	3.62142631E-02	-3.27639754E-01	2.20321061E-02
4	-3.77241146E-01	5.22685899E-02	-3.62257916E-01	4.53579545E-02	-3.31056862E-01	3.13672096E-02	-2.7909724E-01	1.49671281E-02
5	-3.45317007E-01	5.15562666E-02	-3.21977014E-01	5.74992493E-02	-2.84643330E-01	2.09465939E-02	-2.31410071E-01	4.65717477E-03
6	-3.06964229E-01	4.23862223E-02	-2.76882413E-01	2.44348834E-02	-2.36985476E-01	7.23640692E-03	-1.88117491E-01	-6.52556877E-03
7	-2.66303195E-01	2.79053132E-02	-2.32102668E-01	6.68438473E-03	-1.92753785E-01	-7.99700767E-03	-1.51157465E-01	-1.61056155E-02
8	-2.266773329E-01	9.33876777E-03	-1.91510125E-01	-1.00049534E-02	-1.55560544E-01	-2.34574235E-02	-1.2180340E-01	-2.94747270E-02
9	-1.90637999E-01	-1.20596437E-02	-1.57713704E-01	-2.6145123E-02	-1.26446754E-01	-3.8218095E-02	-1.00487389E-01	-4.02628635E-02
10	-1.59969413E-01	-3.49607463E-02	-1.32051684E-01	-4.68106932E-02	-1.06871052E-01	-5.19433781E-02	-6.67535918E-02	-5.0304332E-02
11	-1.39660319E-01	-5.79371412E-02	-1.14584465E-01	-6.39014123E-02	-9.36986178E-02	-6.43046753E-02	-7.92921967E-02	-5.96310642E-02
12	-1.16026905E-01	-7.94337765E-02	-1.04067455E-01	-7.93924315E-02	-9.0900945E-02	-7.54706392E-02	-7.59097972E-02	-6.39863531E-02
13	-1.06465694E-01	-9.77506678E-02	-9.80436516E-02	-9.29438233E-02	-8.6505000E-02	-8.57916709E-02	-7.3529000E-02	-6.69006154E-02
14	-9.97904345E-02	-1.10993609E-01	-9.26566271E-02	-1.04326361E-01	-8.318161435E-02	-9.56067036E-02	-8.55196361E-02	-8.1819961E-02
15	-9.729622954E-02	-1.172056063E-01	-8.93510578E-02	-1.10466222E-01	-7.90100018E-02	-1.01711083E-01	-6.39986677E-02	-9.06459757E-02

ROW

CHORD 9

REAL

CHORD 10

REAL

IMAGINARY

1	-3.82694994E-01	1.19061519E-02	-3.19165317E-01	4.12092196E-03
2	-3.27319916E-01	1.49391002E-02	-2.06769675E-01	2.56368644E-03
3	-2.97949C63E-01	6.6159196E-03	-1.52743548E-01	-3.33976552E-03
4	-2.04126492E-01	-7.9346293E-04	-1.15740509E-01	-7.46498669E-03
5	-1.C3063742E-01	-7.7068395E-03	-9.12912121E-02	-1.07371504E-02
6	-1.32297626E-01	-1.44370347E-02	-7.55794936E-02	-1.37331619E-02
7	-1.09327690E-01	-2.12605952E-02	-6.54453266E-02	-1.69136809E-02
8	-9.29179767E-02	-2.82477379E-02	-5.15677625E-02	-2.05251609E-02
9	-6.004973521E-02	-3.54612375E-02	-5.25379600E-02	-2.46618767E-02
10	-7.11613710E-02	-4.26679028E-02	-4.68119595E-02	-2.92779666E-02
11	-6.37721413E-02	-5.03679390E-02	-4.096138234E-02	-3.42114371E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

WING BOX LIFTS

(MACH 1.200 RED.FREQ. = .90000)

MODE SHAPE 2

PAGE CONTINUED

ROW	CHORD 9			CHORD 10		
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
12	-5.74536692E-02	-5.78043016E-02	-3.41364900E-02	-3.92062991E-02	-5.92062991E-02	
13	-5.17227246E-02	-6.-43716698E-02	-2.82247455E-02	-4.39344020E-02		
14	-4.-63293056E-02	-7.15270061E-02	-2.43981753E-02	-4.80176636E-02		
15	-4.-35031367E-02	-7.-53247223E-02	-2.30396938E-02	-5.-01516667E-02		

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

TAIL BOX LIFTS

1 MACH 1.200 RED.FREQ. = .50000

MODE SHAPE 2

ROW	CHORD 1			CHORD 2			CHORD 3			CHORD 4		
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
19	2.94930884E-01	6.64932624E-02	2.922212425E-01	6.72252601E-02	2.87599504E-01	6.976035645E-02	2.80599690E-01	7.50377029E-02	2.80599690E-01	7.50377029E-02	2.12010354E-01	-3.75857093E-02
20	2.12010354E-01	-3.75857093E-02	2.12152216E-01	-2.93500274E-02	2.12396507E-01	-2.323539372E-02	2.10894522E-01	-1.21378065E-02	2.10894522E-01	-1.21378065E-02	1.61676420E-01	-1.616800214E-02
21	1.61676420E-01	-1.696769582E-02	1.62651775E-01	-7.30372254E-02	1.6314453E-01	-6.55091766E-02	1.63750516E-01	-5.416800214E-02	1.63750516E-01	-5.416800214E-02	1.63671051E-01	-1.06961511E-01
22	1.63671051E-01	-1.117650586E-01	1.6350215E-01	-1.06961511E-01	1.61077640E-01	-1.65050133E-02	1.56948552E-01	-6.68793571E-02	1.56948552E-01	-6.68793571E-02	1.51264711E-01	-1.36735782E-01
23	1.51264711E-01	-1.36735782E-01	1.48974362E-01	-1.31544559E-01	1.4394961E-01	-1.22467944E-01	1.34595561E-01	-8.30786099E-01	1.34595561E-01	-8.30786099E-01	1.39846976E-01	-1.46901507E-01
24	1.39846976E-01	-1.51971474E-01	1.35282666E-01	-1.46901507E-01	1.259562659E-01	-1.37656144E-01	1.13275165E-01	-1.26603322E-01	1.13275165E-01	-1.26603322E-01	1.293339332E-01	-1.58052932E-01
25	1.293339332E-01	-1.58052932E-01	1.19566161E-01	-1.535840535E-01	1.07367125E-01	-1.45335564E-01	9.20497520E-02	-1.35511625E-01	9.20497520E-02	-1.35511625E-01	1.07339962E-01	-1.55906944E-01
26	1.07339962E-01	-1.55906944E-01	1.00007162E-01	-1.52443362E-01	6.6735949E-02	-1.45754539E-01	7.04629508E-02	-1.37103075E-01	7.04629508E-02	-1.37103075E-01	6.33071320E-02	-1.46762870E-01
27	6.33071320E-02	-1.46762870E-01	7.6259066E-02	-1.44614168E-01	5.37669719E-02	-1.4011800DE-01	4.85416042E-02	-1.33754139E-01	4.85416042E-02	-1.33754139E-01	5.43045608E-02	-1.32209235E-01
28	5.43045608E-02	-1.32209235E-01	4.8635529E-02	-1.31475876E-01	3.89895561E-02	-1.29563324E-01	2.67990625E-02	-8.25814625E-01	2.67990625E-02	-8.25814625E-01	2.22399040E-02	-1.13950313E-01
29	2.22399040E-02	-1.13950313E-01	1.69444856E-02	-1.14617667E-01	1.35744127E-02	-1.15334602E-01	6.23545661E-03	-1.14391620E-01	6.23545661E-03	-1.14391620E-01	9.63145674E-03	-9.39627166E-02
30	9.63145674E-03	-9.39627166E-02	-9.94979052E-03	-9.58005999E-02	-1.0208671E-02	-9.07577539E-02	-1.16580268E-02	-1.00534624E-01	-1.16580268E-02	-1.00534624E-01	-2.66766273E-02	-7.43519948E-02
31	-2.66766273E-02	-7.43519948E-02	-3.3972479E-02	-7.69209091E-02	-2.9334056E-02	-6.12153800E-02	-2.49976739E-02	-8.93004907E-02	-2.49976739E-02	-8.93004907E-02	-5.26769134E-02	-5.99731193E-02
32	-5.26769134E-02	-5.73292000E-02	-4.79037183E-02	-5.99731193E-02	-4.06502774E-02	-6.41216544E-02	-3.10057661E-02	-6.97363545E-02	-3.10057661E-02	-6.97363545E-02	-5.61542084E-02	-5.17396212E-02
33	-5.61542084E-02	-5.17396212E-02	-5.03139639E-02	-5.41092721E-02	-4.23210481E-02	-5.75739501E-02	-3.03372041E-02	-6.34422365E-02	-3.03372041E-02	-6.34422365E-02		
ROW	CHORD 5	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL
ROW	CHORD 6	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL
ROW	CHORD 7	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL
ROW	CHORD 8	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL
19	2.68366751E-01	8.05239074E-02	2.50437100E-01	8.601946661E-02	2.2451482E-01	9.13762451E-02	1.88055403E-01	9.36321154E-02	1.88055403E-01	9.36321154E-02	2.05259062E-01	1.663351703E-03
20	2.05259062E-01	1.663351703E-03	1.94025376E-01	1.70330234E-02	1.75352886E-01	3.25539353E-02	1.47396590E-01	4.53215842E-02	1.47396590E-01	4.53215842E-02	1.75356526E-01	1.51264711E-03
21	1.75356526E-01	-3.98468145E-02	1.63446943E-01	-2.13540356E-02	1.44261356E-01	-4.74337293E-03	1.17924476E-01	6.19225354E-03	1.17924476E-01	6.19225354E-03	1.47647554E-01	-1.46762870E-01
22	1.47647554E-01	-1.1436102UE-01	1.33886003E-01	-5.39790289E-02	1.14360707E-01	-3.76273033E-02	9.049179356E-02	-2.427232383E-02	9.049179356E-02	-2.427232383E-02	1.22152652E-01	-9.62724900E-02
23	1.22152652E-01	-9.62724900E-02	1.06022320E-01	-6.0232328620E-02	6.62126489E-02	-6.49522222E-02	6.58391130E-02	-5.11951439E-02	6.58391130E-02	-5.11951439E-02	9.76944860E-02	-1.136633419E-01
24	9.76944860E-02	-1.136633419E-01	8.02297878E-02	-9.97673670UE-02	6.173836301E-02	-6.595267793E-02	4.44716777E-02	-7.16700740E-02	4.44716777E-02	-7.16700740E-02	7.48839501E-02	-1.24149582E-01
25	7.48839501E-02	-1.24149582E-01	5.69454639E-02	-1.12598442E-01	4.03168206E-02	-1.00250529E-01	2.66726087E-02	-6.5977851E-02	2.66726087E-02	-6.5977851E-02	9.31277641E-02	-1.26298867E-01
26	9.31277641E-02	-1.26298867E-01	3.644338864E-02	-1.16908713E-01	2.25048874E-02	-1.0762050E-01	1.25163069E-02	-9.55883565E-02	1.25163069E-02	-9.55883565E-02	3.20776695E-02	-1.26905409E-01
27	3.20776695E-02	-1.26905409E-01	1.89474200E-02	-1.19250620E-01	8.41077335E-03	-1.09265068E-01	1.89055303E-02	-9.51683935E-02	1.89055303E-02	-9.51683935E-02	1.45375664E-02	-1.20658353C-03
28	1.45375664E-02	-1.20658353C-03	4.61655732E-03	-1.144492211E-03	-2.08552511E-03	-1.0516077E-03	-5.30916532E-03	-9.15865271E-02	-5.30916532E-03	-9.15865271E-02	-1.23314691E-03	-1.12060122E-01
29	-1.23314691E-03	-1.12060122E-01	-6.4387601E-03	-1.05625048E-01	-9.28817953E-03	-9.66073552E-02	-1.01616041E-02	-8.41299473E-02	-1.01616041E-02	-8.41299473E-02	-1.35626041E-02	-9.09971224E-02
30	-1.35626041E-02	-9.09971224E-02	-1.4150200E-02	-9.41970986E-02	-1.3639516E-02	-8.59740934E-02	-1.26447343E-02	-7.4491317E-02	-1.26447343E-02	-7.4491317E-02	-2.15531442E-02	-9.58126246E-02
31	-2.15531442E-02	-9.58126246E-02	-1.8460193E-02	-8.18656275E-02	-1.5768149E-02	-7.47015105E-02	-1.375980613E-02	-6.47956139E-02	-1.375980613E-02	-6.47956139E-02	-2.27741650E-02	-7.27295204E-02
32	-2.27741650E-02	-7.27295204E-02	-1.94125597E-02	-7.07150628E-02	-1.64121772E-02	-6.55219762E-02	-3.79970G9E-02	-1.4381621E-02	-3.79970G9E-02	-1.4381621E-02	-2.12655100E-02	-6.76076427E-02
33	-2.12655100E-02	-6.76076427E-02	-1.76059104E-02	-6.67404018E-02	-1.51391694E-02	-6.27145607E-02	-1.33260067E-02	-3.27354759E-02	-1.33260067E-02	-3.27354759E-02		

19	1.39991339E-01	9.23373999E-02	9.32942147E-02	7.90751510E-02
20	1.10632463E-01	5.04360547E-02	5.90224602E-02	3.40387006E-02
21	6.45127364E-02	1.50161051E-02	4.70699810E-02	1.131610123E-02
22	6.45424992E-02	-1.46314300E-02	3.67249027E-02	-7.66663250E-03
23	4.55643523E-02	-5.63719377E-02	2.620633323E-02	-2.28682566E-02
24	2.29572344E-02	-5.60410657E-02	1.6731047E-02	-3.42663744E-02
25	1.63201716E-02	-6.77771167E-02	6.77667621E-03	-4.16808766E-02
26	0.32836778E-03	-7.39596410E-02	2.37717033E-03	-4.5991155E-02
27	-1.20250608E-03	-7.52134345E-02	-2.37356650E-03	-4.7007695E-02
28	-6.37367263E-03	-7.24126447E-02	-5.55491668E-03	-4.53628279E-02
29	-9.26995029E-03	-6.66648729E-02	-7.37343156E-03	-4.21642758E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

TAIL BOX LIFTS

(MACH 1.200 RED.FREQ. 2 .50000)

PAGE CONTINUED

ROW	CHORD 0		CHORD 10	
	REAL	IMAGINARY	REAL	IMAGINARY
50	-1.12165920E-02	-5.94192631E-02	-6.15558508E-03	-3.76511794E-02
51	-1.19699468E-02	-5.22505983E-02	-6.34271615E-03	-3.35334692E-02
52	-1.23093763E-02	-4.71033928E-02	-6.46597774E-03	-3.04299476E-02
53	-1.15314634E-02	-4.386738861E-02	-6.03437375E-03	-2.97035954E-02

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS
 WING PRESS. DIFFERENCE
 (MACH 1.200 AED.FREQ. = .30000)
 NOSE SHAPE 2

ROW	CHORD 1	REAL	IMAGINARY	CHORD 2	REAL	IMAGINARY	CHORD 3	REAL	IMAGINARY	CHORD 4	REAL	IMAGINARY
1	-5.96025125E+00	2.04920001E-01	-3.96747362E+00	2.05071161E-01	-5.98882530E+00	2.07023316E-01	-5.99556146E+00	2.12684481E-01				
2	-1.86621697E+00	3.1937C166E-01	-5.91034918E+00	3.-63649373E-01	-5.-95253362E+00	4.-37923102E-01	-6.-0043990E+00	4.-96476574E-01				
3	-5.-76266045E+00	4.-11966197E-01	-5.-812923561E+00	5.-00406849E-01	-5.-89473751E+00	6.-32301034E-01	-5.-97176693E+00	7.-23251845E-01				
4	-5.-63621343E+00	7.-10893537E-01	-5.-67741693E+00	7.-63404012E-01	-5.-7312C121E+00	6.-39470G3E-01	-5.-75323694E+00	6.-66025426E-01				
5	-5.-46077604E+00	1.-05954364E+00	-5.-48516131E+00	1.-04308790E+00	-5.-47067841E+00	1.-00115667E+00	-5.-39447174E+00	9.-20937256E-01				
6	-5.-26426326E+00	1.-33555129E+00	-5.-22309029E+00	1.-24195800E+00	-5.-12401908E+00	1.-07324592E+00	-4.-9311E-035E+00	8.-729966662E-01				
7	-4.-97167734E+00	1.-45322477E+00	-4.-88510511E+00	1.-29625052E+00	-4.-70467906E+00	1.-02544514E+00	-4.-41523889E+00	7.-24200350E-01				
8	-4.-59603612E+00	1.-35451383E+00	-4.-47291082E+00	1.-16360967E+00	-4.-23008477E+00	6.-4134942E+00	-3.-8597056E+00	4.-79641-93E-01				
9	-4.-14166346E+00	1.-01661133E+00	-3.-99686440E+00	6.-34775343E-01	-3.-72013576E+00	5.-16207785E-01	-3.-337653E+00	1.-50458222E-01				
10	-5.-62516158E+00	4.-5004999E-01	-3.-47662099E+00	3.-13266957E-01	-3.-19968441E+00	6.-68075265E-02	-2.-63750093E+00	-2.-46109385E-01				
11	-3.-07654146E+00	-3.-04051415E-01	-2.-94297743E+00	-3.-64977437E-01	-2.-59801553E+00	-4.-886593022E-01	-2.-36116633E+00	-6.-65431105E-01				
12	-2.-54030322E+00	-1.-17307546E+00	-2.-43373126E+00	-1.-14163693E+00	-2.-24952607E+00	-1.-11072954E+00	-2.-01381470E+00	-1.-14095911E+00				
13	-2.-07652063E+00	-2.-0532345E+00	-2.-01150125E+00	-1.-93552079E+00	-1.-89420458E+00	-1.-74880603E+00	-1.-7168113E+00	-1.-37298502E+00				
14	-1.-76197116E+00	-2.-61019807E+00	-1.-73359922E+00	-2.-64032750E+00	-1.-67611105E+00	-2.-35928829E+00	-1.-60104350E+00	-1.-93964296E+00				
15	-1.-6274-361E+00	-3.-1561365E+00	-1.-61600620E+00	-2.-97310463E+00	-1.-59366255E+00	-1.-551702097E+00	-1.-55170365E+00	-2.-11020814E+00				
ROW	CHORD 5	REAL	IMAGINARY	CHORD 6	REAL	IMAGINARY	CHORD 7	REAL	IMAGINARY	CHORD 8	REAL	IMAGINARY
1	-6.01326517E+00	2.24986364E-01	-6.0350618913E+00	2.3974189E-01	-6.04085288E+00	2.44126143E-01	-5.99375893E+00	2.30707674E-01				
2	-6.-05072714E+00	5.-30639494E-01	-6.-05474952E+00	5.-37757530E-01	-5.-950541E-01	4.-92099406E-01	-5.-63523478E+00	3.-05242100E-01				
3	-5.-99742878E+00	7.-41539151E-01	-5.-69978659E+00	6.-88566205E-01	-5.-58095202E+00	5.-43959603E-01	-4.-93355513E+00	3.-32146497E-01				
4	-5.-66712429E+00	6.-17526753E-01	-5.-46124337E+00	6.-63796893E-01	-4.-99083984E+00	4.-72878477E-01	-4.-2069214E+00	2.-19607720E-01				
5	-5.-20584975E+00	7.-77239592E-01	-4.-85398613E+00	5.-6532453E-01	-4.-29115966E+00	3.-15781780E-01	-3.-48633809E+00	7.-02695513E-02				
6	-4.-62765987E+00	6.-36943580E-01	-4.-17415944E+00	3.-68363728E-01	-3.-57269048E+00	1.-09092939E-01	-2.-83597788E+00	-9.-03769523E-02				
7	-4.-01475167E+00	4.-20691992E-01	-3.-49907938E+00	1.-21876686E-01	-2.-90587265E+00	1.-20539227E-01	-2.-27878453E+00	-2.-27952424E-01				
8	-3.-41278932E+00	1.-40767233E-01	-2.-68712377E-00	-1.-50803438E-01	-2.-34064076E-00	-3.-55333997E-01	-1.-83631474E+00	-4.-44346229E-01				
9	-2.-87397597E+00	-1.-61603969E-01	-2.-37762355E+00	-1.-61603969E-01	-1.-90625655E+00	-5.-76962537E-01	-1.-51190439E+00	-6.-06965503E-01				
10	-2.-41162964E+00	-5.-27035122E-01	-1.-99075403E+00	-7.-0567827E-01	-1.-61114173E+00	-7.-63015869E-01	-1.-30186413E+00	-7.-58434197E-01				
11	-2.-0455602E+00	-8.-73435268E-01	-1.-72742610E+00	-9.-63330037E-01	-1.-44271095E+00	-9.-69429456E-01	-1.-1937484E+00	-6.-98972118E-01				
12	-1.-77932254E+00	-1.-1975399E+00	-1.-56917742E+00	-1.-19686594E+00	-1.-36720462E+00	-1.-13776270E+00	-1.-1433245E+00	-1.-03114397E+00				
13	-1.-60533224E+00	-1.-4735676E+00	-1.-47606366E+00	-1.-4018007E+00	-1.-32952181E+00	-1.-29335910E+00	-1.-10849741E+00	-1.-15932040E+00				
14	-1.-30439736E+00	-1.-67332177E+00	-1.-39654970E+00	-1.-57277907E+00	-1.-25305316E+00	-1.-4443040E+00	-1.-02786227E+00	-1.-26923702E+00				
15	-1.-46679669E+00	-1.-76664788E+00	-1.-34701788E+00	-1.-66534095E+00	-1.-19112059E+00	-1.-53333226E+00	-1.-64615917E-01	-1.-36653947E+00				

ROW CHORD 9 REAL IMAGINARY ROW CHORD 10 REAL IMAGINARY

1	-9.77255647E+00	1.7941993E-01	-4.01159196E+00	6.22252350E-02
2	-4.9313C42E+00	2.25916921E-01	-3.11717D1E+00	3.669033E-02
3	-3.68893752C+00	2.97363213E-02	-2.30269562E+00	-5.03466596E-02
4	-3.0732205C+00	-1.15619092E-02	-1.74463393E+00	-1.1264D405E-01
5	-2.4587992C+00	-1.16215948E-01	-1.3762668E+00	-61698633E-01
6	-1.6946176E+00	-2.17946301E-01	-1.7394034E+00	-2.010665357E-01
7	-1.55119205E+00	-3.20516436E-01	-9.666225155E-01	-2.598333E-01
8	-1.39922214E+00	-4.25951426E-01	-6.80228915E-01	-3.C94284295E-01
9	-1.21954324E+00	-5.34596571E-01	-7.92491816E-01	-3.71791761E-01
10	-1.0730954C+00	-6.46256032E-01	-7.05716813E-01	-4.-1381656E-01
11	-9.61401203E-01	-7.99325461E-01	-6.1227642E-01	-5.15757121E-01

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

WING PRESS. DIFFERENCE

{ MACH 1.200 RED.FREQ. = .50000 ,

MODE SHAPE 2

PAGE CONTINUED

ROW	CHORD 9			CHORD 10		
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1.2	-6.6623740E-01	-8.7143263E-01	-5.14657102E-01	-5.91057198E-01		
1.3	-7.79749415E-01	-9.79466093E-01	-4.235504049E-01	-6.62336031E-01		
1.4	-6.98443591E-01	-1.07961773E+00	-3.67665577E-01	-7.23893546E-01		
1.5	-6.55634462E-01	-1.13596292E+00	-3.47788765E-01	-7.56110352E-01		

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

TAIL PRESS. DIFFERENCE

(MACH 1.200 RED.FREQ. = .000000)

MODE SHAPE 2

ROW	CHORD 1		CHORD 2		CHORD 3		CHORD 4	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
19	4.44022021E+00	1.00242395E+00	4.405224807E+00	1.01346407E+00	4.34155589E+00	1.05198159E+00	4.23170705E+00	1.13093442E+00
20	3.19617630E+00	-14.91246051E-01	3.19831500E+00	-1.51212146E-01	3.20198762E+00	-3.50566593E-01	3.1793459E+00	-1.82964348E-01
21	2.74192239E+00	-1.16033472E+00	2.75671499E+00	-1.10107766E+00	2.7699959E+00	-9.67587996E-01	2.7599212E+00	-6.16915159E-01
22	2.46743394E+00	-1.66522316E+00	2.46492679E+00	-1.61280695E+00	2.4403925E+00	-1.46502076E+00	2.36608845E+00	-1.30873559E+00
23	2.26040132E+00	-2.08136948E+00	2.24387300E+00	-1.96310885E+00	2.16326792E+00	-1.64627367E+00	2.02910774E+00	-1.67019344E+00
24	2.10830263E+00	-2.29105617E+00	2.03946594E+00	-2.21462355E+00	1.89871138E+00	-2.07025964E+00	1.7076937E+00	-1.90861669E+00
25	1.89652343E+00	-2.36223760E+00	1.80250444E+00	-2.31536672E+00	1.61692162E+00	-2.19101722E+00	1.58770222E+00	-2.02666874E+00
26	1.61651232E+00	-2.350305661E+00	1.50887073E+00	-2.29817317E+00	1.30781965E+00	-2.19733205E+00	1.06226685E+00	-2.0681267E+00
27	1.25590227E+00	-2.21263502E+00	1.14961932E+00	-2.18014061E+00	9.61523271E-01	-2.11235840E+00	7.31795235E-01	-2.01641951E+00
28	6.18672339E-01	-1.99332921E+00	7.33255374E-01	-1.9607341E+00	5.67020620E+00	-1.95324062E+00	4.04011056E-01	-1.89672664E+00
29	5.35279169E-01	-1.71786564E+00	2.60553642E-01	-2.72792635E+00	2.04641972E-01	-1.73073454E+00	9.40036431E-02	-1.7251056E+00
30	-1.45199674E-01	-1.41654128E+00	-1.499986356E-01	-1.44424637E+00	-1.54070293E-01	-1.46682916E+00	-1.75751356E-01	-1.5561648E+00
31	-9.52922975E-01	-1.12089650E+00	-5.12241361E-01	-1.15962634E+00	-4.45424917E-01	-1.22436792E+00	-3.75349511E-01	-1.2659326E+00
32	-7.97149463E-01	-6.6270209E-01	-7.22175725E-01	-9.0126795E-01	-6.12825999E-01	-9.66670311E-01	-4.67429511E-01	-1.05134540E+00
33	-8.46556544E-01	-7.80004139E-01	-7.98502795E-01	-8.15727969E-01	-6.36013866E-01	-6.67959566E-01	-4.60365673E-01	-9.56427701E-01
ROW	CHORD 5		CHORD 6		CHORD 7		CHORD 8	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
19	4.05060517E+00	1.21354358E+00	3.777546133E+00	1.29679224E+00	3.36373340E+00	1.377549414E+00	2.83549414E+00	1.41155725E+00
20	3.09439679E+00	2.92058462E-02	2.56782495E-01	2.64356949E+00	4.90769040E-01	2.22211736E+00	6.60233376E-01	2.23502695E-01
21	2.44362925E+00	-5.856537763E-01	2.463633327E+00	-3.21923746E-01	2.17482177E+00	-7.15090375E-02	1.77777636E+00	1.36421512E+00
22	2.22889027E+00	-2.01843739E+00	2.01843739E+00	-6.13764468E-01	1.72405253E+00	-5.67252940E-01	-6.55932203E-01	-6.55932203E-01
23	1.84153661E+00	-1.45135224E+00	1.59804519E+00	-1.20955591E+00	1.30596093E+00	-9.79191592E-01	9.92566498E-01	-7.71795634E-01
24	1.47581491E+00	-1.71364202E+00	1.20950560E+00	-1.50355146E+00	9.31423270E-01	-1.29376539E+00	6.70436782E-01	-1.06348213E+00
25	1.12691614E+00	-1.67162538E+00	8.58465470E-01	-1.69746539E+00	6.07626076E-01	-1.51133559E+00	4.02107721E-01	-1.29615466E+00
26	6.01062236E-01	-1.93417650E+00	5.49413791E-01	-1.72961630E+00	3.39273944E-01	-1.62606159E+00	1.66720577E-01	-1.41069716E+00
27	4.95652531E-01	-1.91309564E+00	2.655643317E-01	-1.79777074E+00	1.26797173E-01	-1.64632635E+00	2.64961562E-02	-1.4347175E+00
28	2.19162361E-01	-1.02220623E+00	6.95967270E-02	-1.72238677E+00	-3.14404740E-02	-1.58653372E+00	-8.30631390E-02	-1.38074905E+00
29	-1.05904343E-02	-1.6764371E+00	-9.70672961E-02	-1.59233751E+00	-1.39954122E-01	-1.4592579E+00	-1.52969275E-01	-1.26835566E+00
30	-2.04469644E-01	-1.4939645E+00	-2.13222891E-01	-1.42007469E+00	-2.05630007E-01	-1.29610823E-01	-1.90626541E-01	-1.12229968E+00
31	-3.21910740E-01	-1.29367701E+00	-2.78602603E-01	-1.23447228E+00	-2.37717800E-01	-1.12616764E+00	-2.07439512E-01	-7.6830631E-01
32	-3.50871239E-01	-1.09644028E+00	-2.92655349E-01	-1.066606999E+00	-2.474222880E-01	-1.66608001E-01	-6.68307379E-01	-1.02439122E-01
33	-3.16321258E-01	-1.01922356E+00	-2.65419206E-01	-1.00615062E+00	-2.262315666E-01	-9.4557877E-01	-2.03912223C-01	-6.40243912E-01

ROW CHORD 9 REAL IMAGINARY ROW CHORD 10 REAL IMAGINARY

ROW CHORD 11 REAL IMAGINARY

19	2.1104645E+00	1.39203864E+00	1.25570753E+00	1.19210276E+00
20	1.56764743E+00	7.60352131E-01	6.99797459E-01	5.1315219E-01
21	1.30121349E+00	2.27366766E-01	7.21667127E-01	1.71161591E-01
22	9.7301478E-01	-2.2057710E-01	5.5364877E-01	-1.1537834E-01
23	6.49209967E-01	-5.76479028E-01	3.95075328E-01	-3.4475190E-01
24	4.45419867E-01	-6.44850451E-01	2.553014623E-01	-5.15982006E-01
25	2.44905095E-01	-1.02177648E+00	1.32246539E-01	-6.30925704E-01
26	0.34168094E-02	-1.11496355E+00	3.5831987E-02	-6.93342154E-01
27	-1.81246912E-02	-1.153685119E+00	-3.5728613E-02	-7.0316655E-01
28	-6.60887352E-02	-1.09166766E+00	-6.37451536E-02	-6.86563984E-01
29	-1.44422428E-01	-1.00531228E+00	-1.11156663E-01	-6.36102642E-01

SAMPLE CASE --- TWO AR=2 SURFACES WITH HORIZONTAL AND VERTICAL SEPARATIONS

TAIL PRESS. DIFFERENCE

(MACH 1.200 RED.FREQ. = .50000)

PAGE CONTINUED

CHORD	CHORD 9		CHORD 10		CHORD 11		CHORD 12		CHORD 13		CHORD 14	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
30	-1.69695037E-01	-6.95718793E-01	-1.282950071E-01	-5.70527996E-01	-1.14178802E+00	-8.78372846E-02	-3.98859565E+00	-1.1115716E-01	-3.03484957E+00	-1.49584478E-01	-1.38101091E+00	-1.4047D8452E+00
31	-1.60433766E-01	-7.07707407E-01	-1.257771178E-01	-5.05336069E-01	-2.23649421E+00	-3.38649421E+00	-3.03653545E+00	-4.00020376E-01	-2.61657201E+00	-4.43659370E-01	-1.12096165E+00	-1.03667023E-01
32	-1.69372879E-01	-7.1010365E-01	-1.27930928E-01	-5.58747213E-01	-4.37798549E-01	-1.2122742E-01	-4.47798549E-01	-4.00020376E-01	-2.61657201E+00	-4.43659370E-01	-1.12096165E+00	-1.03667023E-01
33	-1.72842835E-01	-6.9169159E-01	-1.2122742E-01	-4.47798549E-01							-3.21233904E+01	-2.39287395E+00
SECTION LIFTS												
(MACH 1.200 RED. FREQ. = .50000)												
	WING	SHAPE	2									

CHORD	CHORD 9		CHORD 10		CHORD 11		CHORD 12		CHORD 13		CHORD 14	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	-4.14178802E+00	-7.6241066E-02	-4.080682028E+00	-8.78372846E-02	-3.98859565E+00	-1.1115716E-01	-3.03484957E+00	-1.49584478E-01	-1.38101091E+00	-1.4047D8452E+00	-1.35538682E+00	-1.28054991E+00
5	-5.62645533E+00	-2.23649421E+00	-3.38649421E+00	-3.38649421E+00	-3.03653545E+00	-4.00020376E-01	-2.61657201E+00	-4.43659370E-01	-1.12096165E+00	-1.03667023E-01	-6.2532264537E-01	-7.03464991E-01
9	-2.08432395E+00	-4.3141790E-01	-1.35493772E+00	-3.35315695E-01								
SECTION LIFTS												
TAIL												

CHORD	CHORD 9		CHORD 10		CHORD 11		CHORD 12		CHORD 13		CHORD 14	
	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY	REAL	IMAGINARY
1	1.38101091E+00	-1.42763847E+00	1.35633365E+00	-1.4047D8452E+00	1.30312439E+00	-1.35538682E+00	1.22415418E+00	-1.28054991E+00	1.35538682E+00	-1.4047D8452E+00	1.38101091E+00	-1.42763847E+00
5	1.12096165E+00	-1.1747445E+00	9.66667023E-01	-1.03667023E-01	8.05754573E-01	-8.7746537E-01	6.2532264537E-01	-7.03464991E-01	8.05754573E-01	-8.7746537E-01	6.2532264537E-01	-7.03464991E-01
9	4.39271394E-01	-3.11662055E-01	2.32736505E-01	-2.94431576E-01								
TOTAL LIFT - TAIL												
TOTAL LIFT - TAIL												

9.44499199E+00 -1.006776363E+01

TOTAL LIFT

-2.260092929E+01 -1.26605102E+01

GENERALIZED FORCES
(MACH 1.200 AED. FREQ. = .90000)

WT.	FUNCT	VELOCITY POTENTIAL MODE 1	VELOCITY POTENTIAL MODE 2
		REAL	IMAGINARY
		REAL	IMAGINARY
1	-3.63284231E+00	-1.49246649E+01	-3.21255904E+01
2	-1.21502372E+00	-4.98711150E+00	-1.14917114E+01

GENERALIZED AERODYNAMIC COEFFICIENTS
(MACH 1.200 RED.FREQ. = .50000)
REAL PART

WT.
FUNC#

	1	2
1	3.652842231E-01	3.21255904E+00
2	1.31502372E-01	1.1497114E+00

VELOCITY POTENTIAL MODES

GENERALIZED AERODYNAMIC COEFFICIENTS
MACH 1.200 RED.FREQ. 2 .50000)
IMAGINARY PART

PAGE CONTINUE

VELOCITIY POTENTIAL MODES

1 2-9849697E+00 3-1657479E-01
2 9-9742222E-01 6-9179279E-01
3 6-9849697E+00 3-1657479E-01
4 2-9849697E+00 3-1657479E-01

CONTINUING PROGRAM FORCES CURRENT ELAPSED TIME IS CP = 125.982; FP .. 66.297

PROGRAM FORCES IS BEING RECALLED TO COMPUTE AIR FORCES WITHOUT SHOOTING.

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APPENDIX B
PROGRAM LISTINGS

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OVERLAY (AFMBOX,0,0)                                DRIVER 00002
PROGRAM DRIVER (INPUT,OUTPUT,TAPES=INPUT,TAPEG=OUTPUT,TAPE1=1000,    DRIVER 00003
1           TAPE2=1000,TAPE3=1000,MODESC =110,IVPSC=110,IGEOSC=110,  DRIVER 00004
2           IWFSC=110,IAICSC=110)                           DRIVER 00005
C                                                     DRIVER 00006
C   THIS IS A DUMMY (0,0) OVERLAY DRIVING PROGRAM      DRIVER 00007
C                                                     DRIVER 00008
C   COMMON PKERNL(1640)                                DRIVER 00009
C   COMPLEX PKERNL                                     DRIVER 00010
C                                                     DRIVER 00011
COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP, FILES 00002
1           IQFSP,MODESC,IVPSC,IGEOSC,IWFSC,IAICSC      FILES 00003
COMMON /ARRAYS/ KBXCDW,LBXCDW,LBXXC,KBXCDT,LBXCDT,KJALPH,LJALPH, ARRAYS 00002
1           KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFSL,KELPHI,  ARRAYS 00003
2           LMODES,KPNTSD,LPNTSD,KSDW,LSDW,KPNTDW,LPNTDW,    ARRAYS 00004
3           KDW,LDW,KTVP,LTVP                           ARRAYS 00005
DATA ITPE1,MODESC,IVPSC,IGEOSC,IWFSC,IAICSC/          DRIVER 00014
1           SLTPE1,SLMODESC,SLIVPSC,SLIGEOSC,SLIWFSC,SLIAICSC/  DRIVER 00015
NT5 = 5                                              DRIVER 00016
NT6 = 6                                              DRIVER 00017
READ(5,5) LINK,L1,L2                               DRIVER 00018
5 FORMAT(A6,4X,2I10)                                DRIVER 00019
WRITE(6,6) LINK,L1,L2                               DRIVER 00020
6 FORMAT(*1 PROGRAM BEGINS *,A6,2I5)                DRIVER 00021
CALL OVERLAY(LINK,L1,L2,0)                          DRIVER 00022
WRITE(6,7)                                         DRIVER 00023
7 FORMAT(*0 PROGRAM TERMINATES*)                   DRIVER 00024
CALL EXIT                                         DRIVER 00025
END                                                 DRIVER 00026

```

OVERLAY (AFMBOX,1,0) CONTROL 00002
 PROGRAM CONTROL CONTROL 00003
 C CONTROL 00004
 C CONTROL 00005
 C CONTROL 00006
 C CONTROL 00007
 C THIS PRIMARY OVERLAY CONTROLS THE CALLING OF THE COMPUTATIONAL CONTROL 00008
 C SECTIONS OF THE PROGRAM CONTROL 00009
 C CONTROL 00010
 C THIS IS THE TOTAL COMMON FOR ALL THE OVERLAY STRUCTURE CONTROL 00014
 COMMON /CONTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM, PRVMODE, DIHW, DIHT, CONTRL 00002
 1 DEFAULT CONTRL 00003
 LOGICAL PRVGEOM, PRVMODE, DIHW, DIHT, DEFAULT CONTRL 00004
 COMMON /PRBLM/ XMACH, NNODES, NTSLOP, NKVALS, SMOOTH, NDEG, CRDFIT, PRBLM 00002
 1 EXAIC, SUBDV, PLYWOOD PRBLM 00003
 LOGICAL SMOOTH, CRDFIT, EXAIC, SUBDV, PLYWOOD PRBLM 00004
 COMMON /GEOMT/ COPLAN, NSUBDV, XSUBDV, NSUBD2, NSUBCN, NSURF, GEOMT 00002
 1 B1,B1BETA,B13,B1BTAS,W1AX,W1AZ,PSI1, GEOMT 00003
 2 MBBW,MXBBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW, GEOMT 00004
 3 IXBW,XCENTR GEOMT 00005
 LOGICAL COPLAN GEOMT 00006
 COMMON /GEOM2 / TLAX,TLAZ,PSIT,MYBT,MYBT,MYBST,MYBST, GEOM2 00002
 1 MYBST,IXBT,IXBT,CAPL GEOM2 00003
 COMMON /KERN / ERR,MDSKRN,IPKERN,NPLKRN,NSPATK,NRNEA KERN 00002
 COMMON /KVAL / IKVAL,IKVAL(20), XKS(20) KVAL 00002
 COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NFAIC,NOUTP, FILES 00002
 1 IOUFS,MODESC,IVPSC,IGEOM,WTCNAF,WTSI,WTEL,PRBOX, FILES 00003
 COMMON /IOCONT/ OPLAIC,OSFAIC,WTCNAF,WTCNAF,WTSI,WTEL,PRBOX, IOCONT 00002
 1 PRFAIC,PSAIC,PRMDS,PRCOEF,PRDW,PRSW,PRVP, IOCONT 00003
 2 PRBL,PRDCP,PRGNAF,PRGNAF,PRSL,PRLW,PRNW,PRCM BCSFRB 00001
 EQUIVALENCE (PRUM,PRDW) IOCONT 00005
 LOGICAL OPLAIC,OSFAIC,WTCNAF,WTSI,WTEL,PRBOX,PRPAIC, IOCONT 00006
 1 PSAIC,PRMDS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAF, IOCONT 00007
 2 PRDCP,PRGNAF,PRUM,PRLW,PRNW,PRCM BCSFRB 00002
 COMMON /TAPEIO/ NFS,NMS,LS,NMR, ID(20), NID, ITYPE,LRS,LWS,M,N, TAPEIO 00002
 1 PARM(10),IRR TAPEIO 00003
 DIMENSION IPARM(10) TAPEIO 00004
 EQUIVALENCE (PARM,IPARM) TAPEIO 00005
 COMMON / MODES/ SYM,SYMT,MTYPEW,MTYPEP MODCOM 00002
 COMMON / ARRAYS/ KBXCDW,LBXCDW,LBCAC,KBXCDT,LBXCDT,KJALPH,LJALPH, ARRAYS 00002
 1 KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFSL,KELPHI, ARRAYS 00003
 2 LMOCES,KPNTSD,LPTNSD,KSDW,LSDW,KPNTDW,LPTNDW, ARRAYS 00004
 3 KDW,LDW,KTVP,LTVP ARRAYS 00005
 COMMON /SAMPLW/ ISAMPLW,ICHORD(10),IBOXF(10),IBOXL(10),ZLOC(10) SAMPLW 00002
 COMMON /PLANXY/ NALE,NATE,NTLE,NTTE, XALE(10),YALE(10), PLANXY 00002
 1 XATE(10),YATE(10), XTE(10),YTE(10), PLANXY 00003
 2 XTE(10),YTE(10), PLANXY 00004
 COMMON /CHECKPR/ DPPCPR,GEOPCR,MODCPR,AICCPR,NMSCPR,SMCPR,GAFPCR CHECKPR 00002
 LOGICAL DPPCPR, GEOPCR, MODCPR, AICCPR, NMSCPR, SMCPR, GAFPCR CHECKPR 00003
 COMMON /RWDUFF/ BFPCODE,IBFCNT, BUFF(3280) RWDUFF 00002
 DATA BFPCODE,IBFCNT / 8HBUFSIZE,3280 / BCSCNA 00001
 DATA TEV1<7 /SHAPMBOX / FTNX1 00002
 DATA PREVEX /10HNEVER EXEC / FTNX1 00003
 DATA EXEC /10HAFMBOX EXC / FTNX1 00004
 CALL RDINIT CONTROL 00030
 1 CONTINU CONTROL 00031
 PROG = SHAPMAPP CONTROL 00032

```

CALL DTIME(CPTIME,PPTIME) CONTROL 00033
WRITE (NT6,8005) PROG,CPTIME,PPTIME CONTROL 00034
8005 FORMAT(1HD,10X,*ENTERING PROGRAM *,A6,* CURRENT ELAPSED TIME IS * CONTROL 00035
1 * CP =*,F8.3,*, PP =*,F8.3 ) CONTROL 00036
CALL OVERLAY (TEV147,1,1,0) CONTROL 00037
IPARM(3) = XMACH CONTROL 00038
IPARM(5)= NKVALS CONTROL 00039
IF (OMACH .EQ. XMACH .AND. PRVGEOM) GO TO 100 CONTROL 00040
C CONTROL 00041
PROG = GMGEOM CONTROL 00042
CALL DTIME(CPTIME,PPTIME) CONTROL 00043
WRITE (NT6,8005) PROG,CPTIME,PPTIME CONTROL 00044
C COMPUTE GEOMETRY SECTION CONTROL 00045
CALL OVERLAY (TEV147,1,2,0) CONTROL 00046
C CONTROL 00047
C READ MODE SHAPES, PLACE IN INTERNAL STORAGE CONVENTION, CONTROL 00048
C STORE ON SCRATCH FILE. COMPUTE AND STORE OPTIONAL CONTROL 00049
C THICKNESS SLOPE FUNCTIONS CONTROL 00050
C CONTROL 00051
100 CONTINUE CONTROL 00052
IF (NKVALS .LE. 0) GO TO 810 CONTROL 00053
PROG = GMODES CONTROL 00054
CALL DTIME(CPTIME,PPTIME) CONTROL 00055
WRITE (NT6,8005) PROG,CPTIME,PPTIME CONTROL 00056
CALL OVERLAY (TEV147,1,3,0) CONTROL 00057
C CONTROL 00058
C SPACE OUTPUT TAPE IF DESIRED CONTROL 00059
IF (NOUTP .LE. 0) GO TO 200 CONTROL 00060
IF (PREVEX .NE. EXEC) REWIND NOUTP CONTROL 00061
C FILE SPACING A FUNCTION OF INSTALLATION CAPABILITIES CONTROL 00062
200 CONTINUE CONTROL 00063
C CONTROL 00064
C LOOP ON NUMBER OF K1 VALUES THRU KERNELS, DOWNWASHES AND CONTROL 00065
C AIR FORCES CONTROL 00066
DO 800 IKVAL = 1, NKVALS CONTROL 00067
C CONTROL 00068
C CALL KERNEL ROUTINES CONTROL 00069
PROG = GHAIC CONTROL 00070
CALL DTIME(CPTIME,PPTIME) CONTROL 00071
WRITE (NT6,8005) PROG,CPTIME,PPTIME CONTROL 00072
CALL OVERLAY (TEV147,1,4,0) CONTROL 00073
C CONTROL 00074
PROG = GHVELPOT CONTROL 00075
CALL DTIME(CPTIME,PPTIME) CONTROL 00076
WRITE (NT6,8005) PROG,CPTIME,PPTIME CONTROL 00077
C CALL DOWNWASH AND VELOCITY POTENTIAL ROUTINES. CONTROL 00078
CALL OVERLAY (TEV147,1,5,0) CONTROL 00079
C CONTROL 00080
IF(.NOT.SMOOTH) GO TO 600 CONTROL 00081
IF(CRDFIT) GO TO 600 CONTROL 00082
C CONTROL 00083
PROG = GMISMOOTH CONTROL 00084
CALL DTIME(CPTIME,PPTIME) CONTROL 00085
WRITE (NT6,8005) PROG,CPTIME,PPTIME CONTROL 00086
CALL OVERLAY (TEV147,1,6,0) CONTROL 00087
GO TO 700 CONTROL 00088
C CONTROL 00089

```

```

600 CONTINUE
  IF(.NOT.CRDIT) GO TO 700
C
  PROG = SHCHORDF
  CALL DTIME(CPTIME,PPTIME)
  WRITE (NT6,6005) PROG,CPTIME,PPTIME
  CALL OVERLAY (TEV147,1,7,0)
C
  700 CONTINUE
  PROG = SHFORCES
  CALL DTIME(CPTIME,PPTIME)
  WRITE (NT6,6005) PROG,CPTIME,PPTIME
  CALL OVERLAY (TEV147,1,8,0)
  IF(.NOT.(SMOOTH.OR.CRDIT)) GO TO 800
  NIVPSC = IAICSC
  IAICSC = IVPSC
  IVPSC = NIVPSC
  CALL DTIME(CPTIME,PPTIME)
  WRITE (NT6,6005) PROG,CPTIME,PPTIME
  WRITE (NT6,6010)
0010 FORMAT(1H0,5X,90(1H*)) // 6X,*PROGRAM FORCES IS BEING RECALLED TO C CONTROL 00110
1COMPUTE AIR FORCES WITHOUT SMOOTHING.*// 6X,90(1H*) )
  CALL OVERLAY (TEV147,1,8,SHRECALL)
C
  800 CONTINUE
C      END OF LOOP ON REDUCED FREQUENCIES
C
  810 CONTINUE
    IF (NOUTP .GT. 0) REWIND NOUTP
    IF (NPLAIC .GT. 0) REWIND NPLAIC
    IF (NSPAIC .GT. 0) REWIND NSPAIC
    CALL DTIME(CPTIME,PPTIME)
    WRITE (NT6,6006) CPTIME,PPTIME
0006 FORMAT(1H0,10X*PROGRAM COMPLETED *,6X,* CURRENT ELAPSED TIME IS * CONTROL 00123
    1 * CP =*,F8.3,*, PP =*,F8.3 )
    READ(5,8005) LINK,L1,L2
  8005 FORMAT(A6,4X,?I10)
C
C      DETERMINE IF ANOTHER CYCLE IS WANTED.
C      IF L1 = -1, RECYCLE
  IF(L1.EQ.-1) GO TO 1
C      IF L1 = -2, RETURN TO CALLING PROGRAM
  IF(L1.EQ.-2) RETURN
C      IF L1 = 0, CALL EXIT
  IF(L1.EQ.0) CALL EXIT
C      IF L1 = POS. CALL OVERLAY
  IF(L1.GT.0) CALL OVERLAY(LINK,L1,L2,0)
END

```

CONTROL 00090
 CONTROL 00091
 CONTROL 00092
 CONTROL 00093
 CONTROL 00094
 CONTROL 00095
 CONTROL 00096
 CONTROL 00097
 CONTROL 00098
 CONTROL 00099
 CONTROL 00100
 CONTROL 00101
 CONTROL 00102
 CONTROL 00103
 CONTROL 00104
 CONTROL 00105
 CONTROL 00106
 CONTROL 00107
 CONTROL 00108
 CONTROL 00109
 CONTROL 00110
 CONTROL 00111
 CONTROL 00112
 CONTROL 00113
 CONTROL 00114
 CONTROL 00115
 CONTROL 00116
 CONTROL 00117
 CONTROL 00118
 CONTROL 00119
 CONTROL 00120
 CONTROL 00121
 CONTROL 00122
 CONTROL 00123
 CONTROL 00124
 CONTROL 00125
 CONTROL 00126
 CONTROL 00127
 CONTROL 00128
 CONTROL 00129
 CONTROL 00130
 CONTROL 00131
 CONTROL 00132
 CONTROL 00133
 CONTROL 00134
 CONTROL 00135
 CONTROL 00136
 CONTROL 00137

C	SUBROUTINE FLUSH(I)	FLUSH 00002
	ROUTINE TO FORCE AN ERROR EXIT	FLUSH 00003
	DIMENSION MESSAGE(4)	FLUSH 00004
	DATA MESSAGE /10H PROGRAM F,10HFLUSHED VIA,10H MODE 1 , 0 /	FLUSH 00005
	DATA NT5 /6L OUTPUT/	FLUSH 00006
	WRITE (NT5,8000) (MESSAGE(I),I=1,3)	FLUSH 00007
	ENDFILE NT5	FLUSH 00008
	CALL REMARK(MESSAGE)	FLUSH 00009
	CALL FLSHXXX	FLUSH 00010
	8000 FORMAT(5H0*** , 3A10, 4H ***)	FLUSH 00011
	END	FLUSH 00012

```
SUBROUTINE RDINIT
COMMON /TAPEIO/ NFS,NMS,L8,NMR,LD(20),NID,ITYPE,LRS,LWS,M,N,
1          PARM(10), IRR
DIMENSION IA(1)
EQUIVALENCE (IA,NFS)
DO 10 I= 1,41
IA(I) = 0
10 CONTINUE
NID = 20
RETURN
END
```

RDINIT	00002
RDINIT	00003
RDINIT	00004
RDINIT	00005
RDINIT	00006
RDINIT	00007
RDINIT	00008
RDINIT	00009
RDINIT	00010
RDINIT	00011
RDINIT	00012

C
SUBROUTINE DTIME(CPTIME,PPTIME)
ROUTINE TO INTERROGATE THE SYSTEM CLOCKS
PPTIME = 0
CALL SECOND(CPTIME)
RETURN
END

DTIME 00002
DTIME 00003
DTIME 00004
DTIME 00005
DTIME 00006
DTIME 00007

```

SUBROUTINE READMX(INFILE,MREAD,RANDIN, NFS, NMS, LS, NMR, K, NID, READMX 00002
1           ID, ITYPE, LRS, A, M, N, PARM, IRR )
C
C      ROUTINE TO READ A MATRIX ON TAPE OR DISK FILE.
C      THIS VERSION WILL WORK WITH SEQUENTIAL FILES ONLY.
C      SOME VARIABLES ARE PASSED FOR RANDOM OPERATION BUT
C      ARE NOT CURRENTLY USED.
C
C      INPUT -
C          INFILE - TAPE NUMBER OR LEFT ADJUSTED FILE NAME
C          MREAD - .T. SMART FORMAT      (NOT USED)
C                    .F. TEL001 FORMAT
C          RANDIN - .T. RANDOM FILE    (NOT USED)
C                    .F. SEQUENTIAL FILE
C          NFS  - NUMBER OF FILES TO SPACE
C          NMS  - NUMBER OF MATRICES TO SPACE
C          LS   - LEVEL NUMBER TO SPACE (NOT USED)
C          NMR  - IDENTIFIER (NAME OR NUMBER) (NOT USED)
C          K    - ROW DIMENSION OF ARRAY A
C                  (IF K=0, MATRIX WILL BE LEFT IN /RWBUFF/. IT WILL
C                  BE STORED AS A ROW-WISE MATRIX, NOT AS A FORTRAN
C                  COLUMN-WISE MATRIX. M-ROWS AND N-COLUMNS )
C          NID  - NUMBER OF WORDS AVAILABLE IN ID ARRAY
C
C      IN/OUT
C          ID   - IDENTIFICATION ARRAY
C          ITYPE - REAL,DIAGONAL,NULL,MIXED,COMPLEX
C
C      OUTPUT -
C          LRS  - LEVEL NUMBER OF MATRIX READ    (NOT USED)
C          A   - ARRAY CONTAINING MATRIX
C          M   - ROW DIMENSION OF MATRIX
C          N   - COLUMN DIMENSION OF MATRIX
C          PARM - ARRAY OF NUMERICAL PARAMETERS STORED WITH THE MATRIX
C          IRR  -
C                  0, NO ERROR
C                  1, MATRIX SPACING IS NEGATIVE
C                  2, FILE SPACING IS NEGATIVE
C                  4, MATRIX DIMENSIONS ILLEGAL
C                  5, M .GT. K
C                  1500 + I, ENCOUNTERED ' ' ' AFTER MATRIX I WHILE
C                          SKIPPING MATRICES.
C
C      DIMENSION ID(1), A(K,1), PARM(10), B(16)
C
C      COMMON /RWBUFF/ BFCODE,IBFCNT,    BUFF(3280)
C
C      DIMENSION IBUFF(2500),  IPARM(10),IB(16)
C      EQUIVALENCE (BUFF,IBUFF),(B,IB)
C
C      LOGICAL  MREAD,RANDIN
C      IRR = 0
C
C      DO FILE SPACING
C
C      IF(NFS) 215,230,220
215  CONTINUE
C      IRR = 2
C      GO TO 1000

```

```

220 CONTINUE          READMX 00060
    DO 225 I=1,NFS   READMX 00061
222 CONTINUE          READMX 00062
    BUFFER IN (INFILE,1) (BUFF(1),BUFF(IBFCNT))  READMX 00063
221 CONTINUE          READMX 00064
    IF(UNIT,INFILE) 221,222,225  READMX 00065
225 CONTINUE          READMX 00066
230 CONTINUE          READMX 00067
C
C      DO MATRIX SPACING
C
    IF(NMS) 235,250,240  READMX 00068
235 CONTINUE          READMX 00069
    IRR = 1             READMX 00070
    GO TO 1000          READMX 00071
240 CONTINUE          READMX 00072
    NM2 = NMS + NMS   READMX 00073
    DO 245 I=1,NM2    READMX 00074
    BUFFER IN (INFILE,1) (BUFF(1),BUFF(IBFCNT))  READMX 00075
241 CONTINUE          READMX 00076
    IF(UNIT,INFILE) 241,242,243  READMX 00077
242 CONTINUE          READMX 00078
    GO TO 245          READMX 00079
243 CONTINUE          READMX 00080
    IRR = 1500 +(I+1)/2  READMX 00081
    GO TO 1000          READMX 00082
245 CONTINUE          READMX 00083
250 CONTINUE          READMX 00084
C
C      READ B HEADER CARD
C
    BUFFER IN (INFILE,1) (B(1),B(16))  READMX 00085
300 CONTINUE          READMX 00086
    IF (UNIT,INFILE) 300,310,305  READMX 00087
305 CONTINUE          READMX 00088
    IRR = 1500 + NMS +1  READMX 00089
    GO TO 1000          READMX 00090
310 CONTINUE          READMX 00091
C
C      SET PARAMETERS AND SIZES
C
    ID(2) =IB(1)        READMX 00092
    M     =IB(2)        READMX 00093
    N     =IB(3)        READMX 00094
    MTN   =IB(6)        READMX 00095
    DO 325 I=7,16       READMX 00096
    PARM(I-6) = B(I)    READMX 00097
325 CONTINUE          READMX 00098
C
C      TEST FOR PROPER SIZES
C
    IF(M.GT.0.AND.N.GT.0.AND.MTN.LE.IBFCNT) GO TO 350  READMX 00099
    IRR = 4             READMX 00100
    GO TO 1000          READMX 00101
350 CONTINUE          READMX 00102
C
C      READ THE ARRAY
C

```

```

C
400 CONTINUE
    BUFFER IN (INFILE,1) (BUFF(1),BUFF(MTN))
410 CONTINUE
    IF (UNIT,INFILE) 410,420,415
415 CONTINUE
    IRR = 1500 +NMS +1
    GO TO 1000
420 CONTINUE
C
C      IF K=0 LEAVE THE MATRIX IN THE BUFF AREA AND EXIT
C      IF K.GT.0 TRANSFER BUFF TO ARRAY A
C
C      IF(K.LE.0) GO TO 1000
C
C      TRANSFORM BUFF TO ARRAY A
C
C      IF(IYPE.EQ.7HCOMPLEX) GO TO 475
IX = 0
DO 450 I=1,M
DO 450 J=1,N
IX = IX +1
A(I,J) = BUFF(IX)
450 CONTINUE
GO TO 500
475 CONTINUE
K2 = K+K
CALL CBUFFR(A,K2,M,N,BUFF)
C
500 CONTINUE
C
1000 CONTINUE
RETURN
END

```

```

READMX 00117
READMX 00118
READMX 00119
READMX 00120
READMX 00121
READMX 00122
READMX 00123
READMX 00124
READMX 00125
READMX 00126
READMX 00127
READMX 00128
READMX 00129
READMX 00130
READMX 00131
READMX 00132
READMX 00133
READMX 00134
READMX 00135
READMX 00136
READMX 00137
READMX 00138
READMX 00139
READMX 00140
READMX 00141
READMX 00142
READMX 00143
READMX 00144
READMX 00145
READMX 00146
READMX 00147
READMX 00148
READMX 00149
READMX 00150

```

```

SUBROUTINE CBUFFR(A,K2,M,N,BUFF)
DIMENSION A(K2,1),BUFF(1)

C
C      PUTS A COMPLEX ARRAY STORED IN BUFF INTO FORTRAN ARRAY A
C
IX = 0
IX2 = MM
MM = MM-1
DO 100 I=1,MM,2
DO 100 J=1,N
IX = IX +1
IX2 = IX2 + 1
A(I,J) = BUFF(IX)
A(I+1,J) = BUFF(IX2)
100 CONTINUE
RETURN
END

```

READMX	00151
READMX	00152
READMX	00153
READMX	00154
READMX	00155
READMX	00156
READMX	00157
READMX	00158
READMX	00159
READMX	00160
READMX	00161
READMX	00162
READMX	00163
READMX	00164
READMX	00165
READMX	00166
READMX	00167

```

SUBROUTINE WRTEMX(IOUTFL, MXWRIT, RANDOU, NFS, NMS, LS, NMR, LWS, WRTEMX 00002
1      K, ID, A, ITYPE, M, N, PARM, IRR ) WRTEMX 00003
C
C      ROUTINE TO WRITE A MATRIX ON TAPE OR DISK FILE.
C      THIS VERSION WILL ONLY WORK WITH SEQUENTIAL FILES.
C      SOME VARIABLES ARE PASSED FOR RANDOM OPERATION BUT
C      ARE NOT CURRENTLY USED.
C
C      IOUTFL - TAPE NUMBER OR LEFT-JUSTIFIED FILE NAME
C      MXWRIT - .T. SNARK FORMAT (NOT USED)
C                  .F. TEL001 FORMAT (NOT USED).
C      RANDOU - .T. RANDOM FILE (NOT USED)
C                  .F. SEQUENTIAL FILE (NOT USED)
C      NFS - NUMBER OF FILES TO SPACE (SEQ. ONLY)
C      NMS - NUMBER OF MATRICES TO SPACE
C      LS - LEVEL NUMBER TO SPACE (NOT USED )
C      NMR - IDENTIFIER (NAME OR NUMBER) (NOT USED )
C      LWS - LEVEL NUMBER OF THIS MATRIX (NOT USED )
C      K - ROW DIMENSION OF A
C          - (IF 0, MATRIX IS ALREADY IN /RWBUFF/ )
C      ID - ARRAY CONTAINING MATRIX NAME
C      A - ARRAY CONTAINING MATRIX
C      ITYPE - REAL,DIAGONAL,NULL,MIXED,COMPLEX
C      M - ROW DIMENSION OF MATRIX
C      N - COLUMN DIMENSION OF MATRIX
C      PARM - 10 WORD PARAMETER ARRAY
C      IRR - ERROR RETURN
C          = 0, NO ERROR
C          1, MATRIX SPACING IS NEGATIVE
C          2, FILE SPACING IS NEGATIVE
C          4, MN Dimensions ARE .GT. IBGCNT
C          1500 + I, ENCOUNTERED EOF AFTER MATRIX I/2 WHILE
C          SKIPPING MATRICES.
C
C
C      DIMENSION ID(1), A(K,1), PARM(10), B(16) WRTEMX 00036
C      DIMENSION IB(16) WRTEMX 00037
C      EQUIVALENCE (B,IB) WRTEMX 00038
C      LOGICAL MXWRIT,RANDOU WRTEMX 00039
C
C      COMMON /RWBUFF/ BFCODE,IBFCNT,    BUFF(3280) WRTEMX 00040
C      DATA           BFCODE,IBFCNT /8MBUFFSIZE, 3280 /
C
C
C      TEST FOR PROPER SIZE
C
C
ISIZ = MNM WRTEMX 00041
IF(ITYPE.EQ.7HCOMPLEX) ISIZ = ISIZ+ISIZ WRTEMX 00042
IF(ISIZ.LE.IBFCNT) GO TO 205 WRTEMX 00043
IRR = 4 WRTEMX 00044
GO TO 1000 WRTEMX 00045
C
C      205 CONTINUE
IF(K.LE.0) GO TO 300 WRTEMX 00046
IF(K.GE.M) GO TO 210 WRTEMX 00047
IRR = 5 WRTEMX 00048
GO TO 1000 WRTEMX 00049
C
C      WRTEMX 00050
IRR = 4 WRTEMX 00051
GO TO 1000 WRTEMX 00052
C
C      WRTEMX 00053
IRR = 5 WRTEMX 00054
GO TO 1000 WRTEMX 00055
C
C      WRTEMX 00056
IRR = 5 WRTEMX 00057
GO TO 1000 WRTEMX 00058

```

210 CONTINUE	WRTEMX 00059
C	WRTEMX 00060
C DO FILE SPACING	WRTEMX 00061
C	WRTEMX 00062
IF(NFS) 215,230,220	WRTEMX 00063
215 CONTINUE	WRTEMX 00064
IRR = 2	WRTEMX 00065
GO TO 1000	WRTEMX 00066
220 CONTINUE	WRTEMX 00067
DO 225 I=1,NFS	WRTEMX 00068
222 CONTINUE	WRTEMX 00069
BUFFER IN (ICUTFL,1) (BUFF(1),BUFF(IBFCNT))	WRTEMX 00070
221 CONTINUE	WRTEMX 00071
IF(UNIT,ICUTFL) 221,222,225	WRTEMX 00072
223 CONTINUE	WRTEMX 00073
230 CONTINUE	WRTEMX 00074
C	WRTEMX 00075
C DO MATRIX SPACING	WRTEMX 00076
C	WRTEMX 00077
IF(NMS) 235,250,240	WRTEMX 00078
235 CONTINUE	WRTEMX 00079
IRR = 1	WRTEMX 00080
GO TO 1000	WRTEMX 00081
240 CONTINUE	WRTEMX 00082
NM2 = NMS + NMS	WRTEMX 00083
DO 245 I=1,NM2	WRTEMX 00084
BUFFER IN (ICUTFL,1) (BUFF(1),BUFF(IBFCNT))	WRTEMX 00085
241 CONTINUE	WRTEMX 00086
IF(UNIT,ICUTFL) 241,242,243	WRTEMX 00087
242 CONTINUE	WRTEMX 00088
GO TO 245	WRTEMX 00099
243 CONTINUE	WRTEMX 00090
IRR = 1500 +(I+1)/2	WRTEMX 00091
GO TO 1000	WRTEMX 00092
245 CONTINUE	WRTEMX 00093
250 CONTINUE	WRTEMX 00094
C	WRTEMX 00095
C CREATE B HEADER RECORD	WRTEMX 00096
C	WRTEMX 00097
300 CONTINUE	WRTEMX 00098
IB(1) = ID(2)	WRTEMX 00099
IB(2) = M	WRTEMX 00100
IB(3) = N	WRTEMX 00101
IB(4) = 0	WRTEMX 00102
IB(5) = 0	WRTEMX 00103
IB(6) = ISIZ	WRTEMX 00104
DO 325 I=7,16	WRTEMX 00105
B(I) = PARM(I-6)	WRTEMX 00106
325 CONTINUE	WRTEMX 00107
C	WRTEMX 00108
IF(K.LE.0) GO TO 400	WRTEMX 00109
C	WRTEMX 00110
C PUT ARRAY A INTO BUFFER	WRTEMX 00111
C	WRTEMX 00112
IF(ITYPE.EQ.7)COMPLEX) GO TO 375	WRTEMX 00113
C	WRTEMX 00114
C NOT COMPLEX PUT INTO BUFFER.	WRTEMX 00115

C	WRTEMX 00116
IX = 0	WRTEMX 00117
DO 350 I=1,M	WRTEMX 00118
DO 350 J=1,N	WRTEMX 00119
IX = IX + 1	WRTEMX 00120
BUFF(IX) = A(I,J)	WRTEMX 00121
350 CONTINUE	WRTEMX 00122
GO TO 400	WRTEMX 00123
C	WRTEMX 00124
C COMPLEX, CALL ROUTINE TO STORE INTO BUFFER.	WRTEMX 00125
C	WRTEMX 00126
375 CONTINUE	WRTEMX 00127
K2 = K+K	WRTEMX 00128
CALL COMBUF(A,K2,M,N,BUFF)	WRTEMX 00129
IX = 2*NNN	WRTEMX 00130
C	WRTEMX 00131
400 CONTINUE	WRTEMX 00132
C	WRTEMX 00133
C WRITE THE B HEADER RECORD AND THE BUFFER ARRAY RECORD	WRTEMX 00134
C	WRTEMX 00135
BUFFER OUT (IOUTFL,1) (B(1),B(16))	WRTEMX 00136
500 CONTINUE	WRTEMX 00137
IF (UNIT,IOUTFL) 500,510,510	WRTEMX 00138
510 CONTINUE	WRTEMX 00139
C	WRTEMX 00140
BUFFER OUT (IOUTFL,1) (BUFF(1),BUFF(IX))	WRTEMX 00141
520 CONTINUE	WRTEMX 00142
IF (UNIT,IOUTFL) 520,530,530	WRTEMX 00143
530 CONTINUE	WRTEMX 00144
C	WRTEMX 00145
1000 CONTINUE	WRTEMX 00146
RETURN	WRTEMX 00147
END	WRTEMX 00148

```

SUBROUTINE COMBUF(A,K2,M,N,BUFF)
DIMENSION A(K2,1),BUFF(1)

C
C      PUTS COMPLEX ARRAY A INTO BUFFER BUFF
C

IX = 0
IX2 = M*M
MM = M*M-1
DO 100 I=1,MM,2
DO 100 J=1,N
IX = IX + 1
IX2 = IX2 + 1
BUFF(IX) = A(I,J)
BUFF(IX2)= A(I+1,J)
100 CONTINUE
RETURN
END

WRTEMX 00149
WRTEMX 00150
WRTEMX 00151
WRTEMX 00152
WRTEMX 00153
WRTEMX 00154
WRTEMX 00155
WRTEMX 00156
WRTEMX 00157
WRTEMX 00158
WRTEMX 00159
WRTEMX 00160
WRTEMX 00161
WRTEMX 00162
WRTEMX 00163
WRTEMX 00164
WRTEMX 00165

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```

OVERLAY (AFMBOX,1,1) DATAPP 00002
PROGRAM DATAPP DATAPP 00003
COMMON /CONTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT, CONTRL 00002
1           DEFAULT CONTRL 00003
LOGICAL     PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT CONTRL 00004
COMMON /PROBLM/ XMACH,NMOCES,NTSLOF,NKVALS,SMOOTH,NDEG,CRDFIT, PROBLM 00002
1           EXAIC,SUBDV,PLYWOOD PROBLM 00003
LOGICAL     SMOOTH,CRDFIT,EXAIC,SUBDV,PLYWOOD PROBLM 00004
COMMON /FILES / NTS,NT6,INTAPE,INFSP,NPLAIC,NSFAIC,NOUTP, FILES 00002
1           IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC FILES 00003
COMMON /KERN / ERR,MXSKRN,IPKERN,NPLKRN,NSPATK,NRWEA KERN 00002
COMMON /KVAL / IKVAL,XKVAL(20), XKS(20) KVAL 00002
DIMENSION XK1(20) DATAPP 00009
EQUIVALENCE (XK1,XKVAL) DATAPP 00010
COMMON /IOCONT/ OPLAIC,OSPAIC,WTGEOM,WTGNAF,WTSI,Wtbl,PRBOX, IOCONT 00002
1           PRAIIC,PSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP, IOCONT 00003
2           PRBL,PRDCP,PRGNAF,PRGNAC,PRSL,PRLW,PRNW,PRCM BCSFRB 00001
EQUIVALENCE (PRUW,PRDW) IOCONT 00005
LOGICAL     OPLAIC,OSPAIC,WTGEOM,WTGNAF,WTSI,Wtbl,PRBOX,PRPAIC, IOCONT 00006
1           PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAF, IOCONT 00007
2           PRDCP,PRGNAC,PRUW,PRLW,PRNW,PRCM BCSFRB 00002
COMMON /MOCES/ SYM,SYMT,MTYPEW,MTYPET MOCOM 00002
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NUBCN,NSURF, GEOMTY 00002
1           B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW, GEOMTY 00003
2           MXBW,MXBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW, GEOMTY 00004
3           IXBW,XCENTR GEOMTY 00005
LOGICAL COPLAN GEOMTY 00006
COMMON /SAMPLW/ ISAMPLW,ICHORD(10),IBOXF(10),IBOOL(10),ZLOC(10) SAMPLW 00002
COMMON /CHECKPR/ DPPCPR,GEOPR,MDCPR,AICCP,RNSCPR,SMCPR,GFCPR CHECKPR 00002
LOGICAL DPPCPR,GEOPR,MDCPR,AICCP,RNSCPR,SMCPR,GFCPR CHECKPR 00003
EQUIVALENCE (CHECKPR,DPPCPR) DATAPP 00016
LOGICAL CHECKPR DATAPP 00017
COMMON /ARRAYS/ KBXCDW,LBXCDW,LBOXC,KBXCDT,LBXCDT,KJALPH,LJALPH, ARRAYS 00002
1           KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFSL,KELPHI, ARRAYS 00003
2           LMODES,KPNTSD,LPNTRD,KSDW,LSDW,KPNTDW,LPTNDW, ARRAYS 00004
3           KDW,LDW,KTVP,LTVP ARRAYS 00005
LOGICAL MXWRIT,RANDOU,MREAD,RANDIN FTN01 00005
INTEGER OAIC,OSAIC FTN01 00006
EQUIVALENCE (MACH,XMACH) DATAPP 00020
REAL MACH DATAPP 00021
DATA EXEC  /1CHAFMBOX EXC /
NAMelist /CARD8 / XMACH DATAPP 00022
NAMelist /CARD8 / DEFAULT,PRVGEOM,PRVMODE,SYM,MTYPEW,MTYPET, DATAPP 00023
1           NSURF,DIHW,DIHT,ISAMPLW,WTGNAF,Wtbl, DATAPP 00024
2           PRGNAF,PRBL,PRSL,PRPAIC,PRSAIC,PRCOEF,PRMODS, DATAPP 00025
X           PRDCP,PRGNAC,PRLW,PRNW,PRUW,PRCM, BCSFRB 00003
3           PRBOX,PRDW,PRSW,PRVP,SUBDV,EXAIC,SMOOTH,NDEG, DATAPP 00027
4           DPPCPR,GEOPR,MDCPR,AICCP,RNSCPR,SMCPR,GFCPR, DATAPP 00028
5           NRWEA,CRDFIT,PLYWOOD DATAPP 00029
NAMelist /CARD8 / OAIC,NAIC,OSAIC,NSAIC,INTAPE,NOUTP,INFSP,IOUFSP DATAPP 00030
NAMelist /CARD8 / XK1,XKS,XKVAL DATAPP 00032
MXWRIT = .FALSE. DATAPP 00033
RANDOU = .FALSE. DATAPP 00034
MREAD = .FALSE. DATAPP 00035
RANDIN = .FALSE. DATAPP 00036
IF(PREVEX.EQ.EXEC) GO TO 200 DATAPP 00037
OMACH = 0.0 DATAPP 00038

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C	DATAPP	00039
C	DATAPP	00040
SET CONTROL PARAMETERS TO DEFAULT OPTIONS	DATAPP	00041
DEFAULT = .FALSE.	DATAPP	00042
100 CONTINUE	DATAPP	00043
PRVGEOM = .FALSE.	DATAPP	00044
PRVMODE = .FALSE.	DATAPP	00045
SYM = 1.0	DATAPP	00046
WTPEW = 2	DATAPP	00047
WTPET = 2	DATAPP	00048
NBURF = 1	DATAPP	00049
DIHW = .TRUE.	DATAPP	00050
DIHT = .TRUE.	DATAPP	00051
ISMPLW = 0	DATAPP	00052
WTGNAF = .TRUE.	DATAPP	00053
WTBL = .FALSE.	DATAPP	00054
PRGNAF = .TRUE.	DATAPP	00055
PRBL = .FALSE.	DATAPP	00056
PRSL = .FALSE.	DATAPP	00057
PRPAIC = .FALSE.	DATAPP	00058
PRSAIC = .FALSE.	DATAPP	00059
PRCOEF = .FALSE.	DATAPP	00060
PRMDS = .FALSE.	DATAPP	00061
PRBOX = .FALSE.	DATAPP	00062
PRDW = .FALSE.	DATAPP	00063
PRSW = .FALSE.	DATAPP	00064
PRDCP = .FALSE.	DATAPP	00065
PRGNAAC = .FALSE.	DATAPP	00066
PRLW = .FALSE.	DATAPP	00067
PRNW = .FALSE.	DATAPP	00068
PRUW = .FALSE.	DATAPP	00069
PRVP = .FALSE.	BCSFRB	00004
PRCM = .FALSE.	DATAPP	00070
SUBDV = .FALSE.	DATAPP	00071
EXAIC = .FALSE.	DATAPP	00072
SMOOTH = .FALSE.	DATAPP	00073
ORDFIT = .FALSE.	DATAPP	00074
NDEG = 0	DATAPP	00075
NROWEA = 0	DATAPP	00076
PLYWOOD = .FALSE.	DATAPP	00077
DPPCPR = .FALSE.	DATAPP	00078
GEOCPR = .FALSE.	DATAPP	00079
MODCPR = .FALSE.	DATAPP	00080
AICCPR = .FALSE.	DATAPP	00081
NMSCPR = .FALSE.	DATAPP	00082
SMCPR = .FALSE.	DATAPP	00083
GAFCPR = .FALSE.	DATAPP	00084
PREVEX = EXEC	DATAPP	00085
IF(DEFAULT) 400,300	DATAPP	00086
200 CONTINUE	DATAPP	00087
XMACH = XMACH	DATAPP	00088
300 CONTINUE	DATAPP	00089
READ (NT5,9005) TITLE	DATAPP	00090
9005 FORMAT(8A10)	DATAPP	00091
READ(NT5,CARDB)	DATAPP	00092
IF(XMACH.GT.1.0) GO TO 310	DATAPP	00093
WRITE (NT6,8005) XMACH	DATAPP	00094
CALL FLUSH(1)	DATAPP	

```

310 CONTINUE          DATAPP 00095
    IF(XMACH.GE.1.2) GO TO 320
    WRITE (NT6,8010)
    GO TO 350          DATAPP 00096
320 CONTINUE          DATAPP 00097
    IF(XMACH.LE.3.0) GO TO 350
C
C      MACH NO. GREATER THAN 3.0          DATAPP 00098
    IF(XMACH.LT.5.0) GO TO 340          DATAPP 00099
    WRITE (NT6,8015) XMACH
    CALL FLUSH(1)                    DATAPP 00100
340 CONTINUE          DATAPP 00101
    WRITE (NT6,8020)
350 CONTINUE          DATAPP 00102
8005 FORMAT(52H0*** MACH NUMBER OF LESS THAN 1.0 CAN NOT BE USED.
     1     14HMACH NUMBER = E15.6, SH *** ) DATAPP 00103
8010 FORMAT(62H0*** WARNING -- MACH NUMBER LESS THAN 1.2 IS BEING USED.
     1 *** ) DATAPP 00104
8015 FORMAT(43H0*** MACH NUMBER GREATER THAN 5.0. XMACH = E15.6,
     1     25H PROGRAM TERMINATED. *** ) DATAPP 00105
8020 FORMAT(64H0*** WARNING -- MACH NUMBER GREATER THAN 3.0 IS BEING US
     1ED. *** ) DATAPP 00106
     READ (NT5,CARDC)
     IF(SUBDV)500,510
500 NSUBDV =3          DATAPP 00107
     GO TO 515
510 NSUBDV =1          DATAPP 00108
515 CONTINUE          DATAPP 00109
C
C      IF(DEFAULT) 100,400          DATAPP 00110
C
C      CARD D          DATAPP 00111
400 CONTINUE          DATAPP 00112
    OAIC = 0          DATAPP 00113
    NAIC = 0          DATAPP 00114
    OSAIC = 0          DATAPP 00115
    NSAIC = 0          DATAPP 00116
    INTAPE = 0          DATAPP 00117
    NOUTP = 1          DATAPP 00118
    INFSP = 0          DATAPP 00119
    IUFSP = 0          DATAPP 00120
    READ (NT5,CARDD)
C
C      IF(OAIC.EQ.0) GO TO 520          DATAPP 00121
    NPLAIC = OAIC
    OPLAIC = .TRUE.
    GO TO 530          DATAPP 00122
520 CONTINUE          DATAPP 00123
    NPLAIC = 0
    OPLAIC = .FALSE.
530 CONTINUE          DATAPP 00124
    IF(NAIC.EQ.0) GO TO 540
    NPLAIC = NAIC
    OPLAIC = .FALSE.
540 CONTINUE          DATAPP 00125
C
C      DETERMINE OPTIONS OF SPATIAL KERNELS          DATAPP 00126

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IF(O8AIC.EQ.0) GO TO 560                               DATAPP 00152
N8PAIC = O8AIC                                         DATAPP 00153
O8PAIC = .TRUE.                                         DATAPP 00154
GO TO 570                                              DATAPP 00155
580 CONTINUE                                             DATAPP 00156
    N8PAIC = 0                                           DATAPP 00157
    O8PAIC = .FALSE.                                     DATAPP 00158
570 CONTINUE                                             DATAPP 00159
    IF(NSAIC.EQ.0) GO TO 580                           DATAPP 00160
    N8PAIC = NSAIC                                       DATAPP 00161
    O8PAIC = .FALSE.                                     DATAPP 00162
580 CONTINUE                                             DATAPP 00163
C
    IF (NOUTP.NE.0) GO TO 600                           DATAPP 00164
    IF(WTGNMF) WRITE (NT6,9041)                         DATAPP 00165
    IF(WTBL ) WRITE (NT6,9042)                           DATAPP 00166
    WTGNMF = .FALSE.                                     DATAPP 00167
    WTBL  = .FALSE.                                     DATAPP 00168
600 CONTINUE                                             DATAPP 00169
C
    CARD E                                              DATAPP 00170
    DO 610 I=1,20                                       DATAPP 00171
    XKS(I) = -1.                                         DATAPP 00172
    XK1(I) = -1.                                         DATAPP 00173
610 CONTINUE                                             DATAPP 00174
    READ(NT5,CARDE)                                     DATAPP 00175
    DO 620 I=1,20                                       DATAPP 00176
    IF(XKS(I).NE.-1.0.OR .XK1(I).NE.-1.0) GO TO 620
    NKVALS = I-1                                         DATAPP 00177
    GO TO 625                                           DATAPP 00178
620 CONTINUE                                             DATAPP 00179
    NKVALS = 20                                         DATAPP 00180
625 CONTINUE                                             DATAPP 00181
C
    WRITE (NT6,9500)                                     DATAPP 00182
    WRITE (NT6,9501)                                     DATAPP 00183
    WRITE (NT6,9551) TITLE                             DATAPP 00184
    WRITE (NT6,9580) XMACH                            DATAPP 00185
    IF(DEFAULT)  WRITE (NT6,9575)                         DATAPP 00186
    IF(SYM.EQ.1.0) WRITE (NT6,9552)                         DATAPP 00187
    IF(SYM.EQ.-1.) WRITE (NT6,9553)                        DATAPP 00188
    IF (PLYWOOD)  WRITE (NT6,9554)                         DATAPP 00189
    IF (DPPCPY.AND.PLYWOOD) WRITE (NT6,9558)             DATAPP 00190
    IF(.NOT.SUBDV) WRITE (NT6,9572)                        DATAPP 00191
    IF(SUBDV)    WRITE (NT6,9573)                         DATAPP 00192
    IF (SUBDV .AND. NROWEA .NE. 0) WRITE (NT6,9546) NROWEA
    IF(NBSURF.EQ.1) WRITE(NT6,9556)                      DATAPP 00193
    IF(NBSURF.EQ.2) WRITE(NT6,9557)                      DATAPP 00194
    IF(.NOT.EXAIC) WRITE (NT6,9576)                         DATAPP 00195
    IF(EXAIC)     WRITE (NT6,9577)                         DATAPP 00196
    IF(CRDFIT) SMOOTH = .FALSE.                           DATAPP 00197
    IF(SMOOTH)   WRITE (NT6,9581) NDEG                  DATAPP 00198
    IF(CRDFIT)   WRITE (NT6,9585) NDEG                  DATAPP 00199
    IF (.NOT. (SMOOTH .OR. CRDFIT) .OR. NDEG .LE. 10) GO TO 630
    NDEG = 10                                            DATAPP 00200
    WRITE (NT6,9043) NDEG                                DATAPP 00201
630 CONTINUE                                             DATAPP 00202
    IF(PRBOX)    WRITE (NT6,9569)                         DATAPP 00203

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IF(PRCOEF)	WRITE (NT6,9582)	DATAPP 00209
IF(PRMOCS)	WRITE (NT6,9568)	DATAPP 00210
IF(PRPAIC)	WRITE (NT6,9583)	DATAPP 00211
IF(PRSAIC)	WRITE (NT6,9584)	DATAPP 00212
IF(PRCDW)	WRITE (NT6,9570)	DATAPP 00213
IF(PRSW)	WRITE (NT6,9578)	DATAPP 00214
IF (PRLW)	WRITE (NT6,9544)	DATAPP 00215
IF (PRNM)	WRITE (NT6,9545)	DATAPP 00216
IF(PRVP)	WRITE (NT6,9571)	DATAPP 00217
IF(PRBL)	WRITE (NT6,9565)	DATAPP 00218
IF(PRLS)	WRITE (NT6,9566)	DATAPP 00219
IF(PRCM)	WRITE (NT6,9567)	BCSFRB 00005
IF(PRDCP)	WRITE (NT6,9542)	DATAPP 00220
IF(PRGNAF)	WRITE (NT6,9543)	DATAPP 00221
IF(PRGNAF)	WRITE (NT6,9564)	DATAPP 00222
IF(WTBL)	WRITE (NT6,9562)	DATAPP 00223
IF(WTCNAF)	WRITE (NT6,9561)	DATAPP 00224
IF(PRVGEOM)	WRITE (NT6,9531)	DATAPP 00225
IF(PRVMODE)	WRITE (NT6,9532)	DATAPP 00226
IF(MTYPEW.EQ.1) WRITE (NT6,9533)		DATAPP 00227
IF(MTYPEW.EQ.2) WRITE (NT6,9534)		DATAPP 00228
IF(MTYPEW.EQ.3) WRITE (NT6,9535)		DATAPP 00229
IF(NSURF.EQ.1) GO TO 650		DATAPP 00230
IF(MTYPET.EQ.1) WRITE (NT6,9536)		DATAPP 00231
IF(MTYPET.EQ.2) WRITE (NT6,9537)		DATAPP 00232
IF(MTYPET.EQ.3) WRITE (NT6,9538)		DATAPP 00233
650 CONTINUE		DATAPP 00234
IF(CIHW)	WRITE (NT6,9539)	DATAPP 00235
IF (NSURF.EQ.1) GO TO 680		DATAPP 00236
IF(CINT)	WRITE (NT6,9540)	DATAPP 00237
680 CONTINUE		DATAPP 00238
ERR = 0.01		DATAPP 00239
IF(EXAIC) ERR = 0.0001		DATAPP 00240
C		DATAPP 00241
C THIS SET OF VARIABLES ARE DIMENSION SIZES FOR ARRAYS.		DATAPP 00242
C THE NUMBER IS THE DIMENSION OF THE ARRAY.		DATAPP 00243
C FOR DOUBLE DIMENSIONED ARRAYS IT IS THE LARGEST NUMBER,		DATAPP 00244
C NOT THE PRODUCT OF THE TWO DIMENSIONS.		DATAPP 00245
C		DATAPP 00246
KKERNEL = 1		DATAPP 00247
LKERNEL = 1640		DATAPP 00248
LBXCWD = 150		DATAPP 00249
LBXCDT = 90		DATAPP 00250
LBOKC = 8		DATAPP 00251
LJALPH = 200		DATAPP 00252
LPMTRM = 100		DATAPP 00253
LNODES = 1000		DATAPP 00254
LPMTSD = 90		DATAPP 00255
LSDW = 800		DATAPP 00256
LPMTDW = 100		DATAPP 00257
LDW = 1275		DATAPP 00258
LYVP = 250		DATAPP 00259
C		DATAPP 00260
WRITE (NT6,6001)		DATAPP 00261
WRITE (NT6,6002) OAIC,NAIC,OBAIC,NSAIC,INTAPE,INFSP,NOUTP,IOUFSP		DATAPP 00262
C		DATAPP 00263
6001 FORMAT(1HO/45X, 39HTHE FOLLOWING TAPE SETUP IS REQUESTED - /)		DATAPP 00264

6002 FORMAT(51X,*OLD AIC TAPE =*,I3,/51X,*NEW AIC TAPE =*,I3,/	DATAPP 00265
1 51X,*OLD SPATIAL AIC TAPE =*,I3,/51X,*NEW SPATIAL AIC TAPE =*,I3,	DATAPP 00266
2 / 51X,*INPUT DATA TAPE =*,I3,* SPACED*,I3,* FILES,*	DATAPP 00267
3 / 51X,*OUTPUT TAPE =*,I3,* SPACED*,I3,* FILES,* //)	DATAPP 00268
C	DATAPP 00269
C PRINT THE XKVAL OR XKS ARRAY.	DATAPP 00270
C	DATAPP 00271
IF(XK1(1).EQ.-1.0) GO TO 700	DATAPP 00272
WRITE(NTS,6003)	DATAPP 00273
WRITE(NTS,6004) (XK1(I),I=1,NKVALS)	DATAPP 00274
GO TO 900	DATAPP 00275
C	DATAPP 00276
700 CONTINUE	DATAPP 00277
IF(XKS(1).EQ.-1.0) GO TO 800	DATAPP 00278
WRITE(NTS,6005)	DATAPP 00279
WRITE(NTS,6004) (XKS(I),I=1,NKVALS)	DATAPP 00280
GO TO 900	DATAPP 00281
C	DATAPP 00282
800 CONTINUE	DATAPP 00283
WRITE(NTS,6006)	DATAPP 00284
C	DATAPP 00285
900 CONTINUE	DATAPP 00286
C	DATAPP 00287
6003 FORMAT(1H0,29X, *THE FOLLOWING IS THE REDUCED FREQUENCY ARRAY BASE	DATAPP 00288
1D ON BOX LENGTH* /)	DATAPP 00289
6005 FORMAT(1H0,29X, *THE FOLLOWING IS THE REDUCED FREQUENCY ARRAY BASE	DATAPP 00290
1D ON WING SEMI-SPAN* /)	DATAPP 00291
6004 FORMAT(1H / (31X,6F11.5))	DATAPP 00292
6006 FORMAT(4SHD*** WARNING -- NO REDUCED FREQUENCIES SPECIFIED.	DATAPP 00293
1 51H PROGRAM WILL TERMINATE AFTER GEOMETRY SECTION ***)	DATAPP 00294
C	DATAPP 00295
C	DATAPP 00296
1000 RETURN	DATAPP 00297
9500 FORMAT(1H1,29X,58(1H*),/30X,1H*,56X,1H*,/30X,58H* UNSTEADY AE	DATAPP 00299
1RDYNAMICS OF WING-HORIZONTAL TAIL *,/30X,1H*,12X,*CONFIGURATI	DATAPP 00300
20NS IN SUPERSONIC FLOW*11X,1H*,/30X,1H*,56X,1H*,/30X,58H* PREP	DATAPP 00301
3ARED UNDER CONTRACT NO. AF 33615-70-C-1126 *,/30X,1H*,20X,*PRO	DATAPP 00302
4JECT NO. 1370*, 20X,1H*,/30X,1H*,56X,1H*)	DATAPP 00303
9501 FORMAT(30X,1H*,5X,*FOR DEPARTMENT OF THE AIR FORCE*,19X,1H*, /	DATAPP 00304
1 30X,1H*,10X,*AERONAUTICAL SYSTEMS DIVISION*, 17X,1H*, /	DATAPP 00305
2 30X,1H*,10X,*AIR FORCE FLIGHT DYNAMICS LABORATORY*,10X,1H*, /	DATAPP 00306
3 30X,1H*,10X,*RIGHT-FATTERSON AIR FORCE BASE*,15X,1H*, /	DATAPP 00307
4 30X,1H*,56X,1H*, /	DATAPP 00308
5 30X,1H*,5X,*BY THE BOEING COMPANY*,28X,1H*, /	DATAPP 00309
6 30X,1H*,10X,*COMMERCIAL AIRPLANE DIVISION*,18X,1H*, /	DATAPP 00310
7 30X,1H*,10X,*SEATTLE, WASHINGTON*, 27X,1H*, /	DATAPP 00311
8 30X,1H*,56X,1H*,/ 30X,58(1H*),/)	DATAPP 00312
9041 FORMAT(72H0*** WARNING -- NO OUTPUT TAPE WAS REQUESTED FOR GENERAL	DATAPP 00313
1IZED FORCES. ***)	DATAPP 00314
9042 FORMAT(63H0*** WARNING -- NO OUTPUT TAPE WAS REQUESTED FOR BOX LIF	DATAPP 00315
1TS. ***)	DATAPP 00316
9043 FORMAT(54H0*** WARNING -- ORDER FOR VELOCITY POTENTIAL SMOOTHING	DATAPP 00317
1 36H TOO LARGE. IT HAS BEEN REDUCED TO ,12, 4H ***)	DATAPP 00318
9551 FORMAT(1H0,5X,7HTITLE -,13X,8A10,13X,7H- TITLE /1H0/45X,	DATAPP 00319
1 37HTHE FOLLOWING OPTIONS ARE REQUESTED - /)	DATAPP 00320
9531 FORMAT(51X,*GEOMETRY FROM PREVIOUS CYCLE*)	DATAPP 00321
9532 FORMAT(51X,*MCCE SHAPES FROM PREVIOUS CYCLE*)	DATAPP 00322

9533 FORMAT(51X,*MODAL INPUT FOR WING IS POLYNOMIAL COEFFICIENTS*)	DATAPP	00323
9534 FORMAT(51X,*MODAL INPUT FOR WING IS ARBITRARY LOCATIONS FOR SURFAC 1E FITTING *)	DATAPP	00324
9535 FORMAT(51X,*MODAL INPUT FOR WING IS BOX CENTER VALUES*)	DATAPP	00325
9536 FORMAT(51X,*MODAL INPUT FOR TAIL IS POLYNOMIAL COEFFICIENTS*)	DATAPP	00327
9537 FORMAT(51X,*MODAL INPUT FOR TAIL IS ARBITRARY LOCATIONS FOR SURFAC 1E FITTING *)	DATAPP	00328
9538 FORMAT(51X,*MODAL INPUT FOR TAIL IS BOX CENTER VALUES*)	DATAPP	00329
9539 FORMAT(51X,*DIHEDRAL WING INFLUENCE CALCULATED*)	DATAPP	00330
9540 FORMAT(51X,*DIHEDRAL TAIL INFLUENCE CALCULATED*)	DATAPP	00331
9542 FORMAT(51X,*PRINT PRESSURE DIFFERENCE COEFFICIENTS*)	DATAPP	00332
9543 FORMAT(51X,*PRINT GENERALIZED AERODYNAMIC COEFFICIENTS*)	DATAPP	00333
9544 FORMAT(51X,*PRINT LONGITUDINAL WASHES ALONG SAMPLING CHORDS*)	DATAPP	00334
9545 FORMAT(51X,*PRINT NORMAL WASHES*)	DATAPP	00335
9546 FORMAT(51X,*EFFECTIVE SUBDIVIDED AREA OF*,I3,* ROWS REQUESTED*)	DATAPP	00337
9552 FORMAT(51X,*SYMMETRIC ANALYSIS*)	DATAPP	00338
9553 FORMAT(51X,*ANTI-SYMMETRIC ANALYSIS*)	DATAPP	00339
9554 FORMAT(51X,*PLYWOOD OPTION IS USED. (PLANFORM BOUNDARY DETERMINED 1BY BOX PATTERN.) *)	DATAPP	00340
9556 FORMAT(51X,*SINGLE PLANFORM ANALYSIS*)	DATAPP	00341
9557 FORMAT(51X,*ANALYSIS FOR 2 PLANFORMS*)	DATAPP	00342
9558 FORMAT(1HO,100(1H\$)///* THE SPRUCE GOOSE IS LOOSE * //1HO, 1 100(1H\$))	DATAPP	00343
9561 FORMAT(51X,*WRITE GENERALIZED AIR FORCES ON TAPE*)	DATAPP	00344
9562 FORMAT(51X,*WRITE BOX LIFTS ON TAPE*)	DATAPP	00345
9564 FORMAT(51X,*PRINT GENERALIZED AIR FORCES*)	DATAPP	00346
9565 FORMAT(51X,*PRINT THE BOX LIFTS*)	DATAPP	00347
9566 FORMAT(51X,*PRINT THE SECTION LIFTS*)	DATAPP	00348
9567 FORMAT(51X,*SECTION MOMENTS WILL BE COMPUTED WITH MODE SHAPE ONE*/ 1 51X,* ASSUMED FOR THE PITCH MODE.*)	BCSFRB	00006
9568 FORMAT(51X,*PRINT MODE SHAPES USED*)	BCSFRB	00007
9569 FORMAT(51X,*PRINT THE BOX PATTERN*)	DATAPP	00349
9570 FORMAT(51X,*PRINT THE UPWASHES ALONG SAMPLING CHORDS*)	DATAPP	00350
9571 FORMAT(51X,*PRINT THE VELOCITY POTENTIALS*)	DATAPP	00351
9572 FORMAT(51X,*BASIC (UNSUBDIVIDED) ANALYSIS WILL BE USED*)	DATAPP	00352
9573 FORMAT(51X,*SUBDIVISION WILL BE APPLIED*)	DATAPP	00353
9575 FORMAT(51X,*ALL PARAMETERS SET TO "FAULT VALUES*)	DATAPP	00354
9576 FORMAT(51X,*APPROXIMATE KERNELS WILL BE USED*)	DATAPP	00355
9577 FORMAT(51X,*EXACT KERNELS WILL BE USED*)	DATAPP	00356
9578 FORMAT(51X,*PRINT THE SIDEWASHES ALONG SAMPLING CHORDS*)	DATAPP	00357
9580 FORMAT(51X,*MACH NUMBER = *, F8.6)	DATAPP	00358
9581 FORMAT(51X,*VELOCITY POTENTIALS WILL BE SMOOTHED BY A LEAST-SQUA*, 1 *RES* / 61X,*POLYNOMIAL SURFACE FIT, OF ORDER*,I2,1H./	DATAPP	00359
2 61X,* (O = PROGRAM DETERMINED.) * ;	DATAPP	00360
9582 FORMAT(51X,*PRINT MODE SHAPE POLYNOMIAL COEFFICIENTS, IF AVAILAB*, 1 *LE *)	DATAPP	00361
9583 FORMAT(51X,*PRINT THE PLANAR AIC ARRAYS USED *)	DATAPP	00362
9584 FORMAT(51X,*PRINT THE SPATIAL AIC ARRAYS USED *)	DATAPP	00363
9585 FORMAT(51X,*VELOCITY POTENTIALS WILL BE SMOOTHED BY A LEAST SQUA*, 1 *RES* / 61X,*POLYNOMIAL CHORDWISE FIT, OF ORDER*,I2,1H./	DATAPP	00364
2 61X,* (O = PROGRAM DETERMINED.) *)	DATAPP	00365
END	DATAPP	00366
	DATAPP	00367
	DATAPP	00368
	DATAPP	00369
	DATAPP	00370
	DATAPP	00371
	DATAPP	00372

OVERLAY (AFMBOX,1,2) GEOMBX 00002
 PROGRAM GEOMBX GEOMBX 00003
 C THIS OVERLAY READS ALL GEOMETRIC INFORMATION (CARDS G TO L, OR GEOMBX 00004
 C FROM TAPE) AND COMPUTES THE INTERNAL GEOMETRY NEEDED GEOMBX 00005
 C ERRORS IN GEOMETRIC DEFINITIONS ARE CAUGHT GEOMBX 00006
 C ALL GEOMETRY IS NON-DIMENSIONALIZED BY BOX WIDTH (LENGTH) GEOMBX 00007
 C BOX CODES ARE DEFINED - GEOMBX 00008
 C 0 = NOT USED GEOMBX 00009
 C 1 = ON-PLANFORM GEOMBX 00010
 C 2 = DIAPHRAGM GEOMBX 00011
 C 3 = WAKE GEOMBX 00012
 C ALPHA ARRAY, FRACTIONAL PART OF EDGE BOXES, IS COMPUTED GEOMBX 00013
 C MAXIMUM PLANAR AIC ARRAY SIZE IS DETERMINED GEOMBX 00014
 C FOR EACH CHORD REQUIRING A SPATIAL AIC ARRAY, DETERMINE GEOMBX 00015
 C WHICH AIC ARRAY TO USE (KPTWW, KPTTT,KPTWRT,DPTWL) GEOMBX 00016
 C EL, THE VERTICAL DISTANCE SEPARATING THE SURFACES GEOMBX 00017
 C YBAR, THE HORIZONTAL OFFSET GEOMBX 00018
 C MUAC ARRAY, A MAP OF NEEDED AIC VALUES GEOMBX 00019
 C GEOMBX 00020
 COMMON /CONTRL/ PREVEX,CWACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT, CONTRL 00002
 1 DEFAULT CONTRL 00003
 LOGICAL PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT CONTRL 00004
 COMMON /PROBLM/ XMAHC,NMOCES,NTSLOP,NKVALS,SMOOTH,NDEG,CRDFIT, PROBLM 00002
 1 EXAIC,SUBDV,PLYWOOD PROBLM 00003
 LOGICAL SMOOTH,CRDFIT,EXAIC,SUBDV,PLYWOOD PROBLM 00004
 COMMON /KVAL / IKVAL,XKVAL (20), XKS (20) KVAL 00002
 COMMON /GEOHTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF, GEOHTY 00002
 1 B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,
 2 MXBW,MYBBW,MYBW,MYBBW,MXBW,MYSBW,MYBSW,MYBBSW,
 3 IXBW,XCENTR GEOHTY 00005
 LOGICAL COPLAN GEOHTY 00006
 COMMON /GEOH2 / TLAX,TLAZ,PSIT,MXB1,MYBT,MYBBT,MXBST,MYBST, GEOH2 00002
 1 MYBBST,IXBT,IXBST,CAPL GEOH2 00003
 COMMON / KERN / ERR,MXSKRN,IPKERN,NPLKRN,NSPATK,NROWEA KERN 00002
 COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP, FILES 00002
 1 IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC FILES 00003
 COMMON /IICONT/ OPLAIC,OSPAIC,WTGEOM,WTGNAF,WTSL,WTBL,PRBOX, IICONT 00002
 1 PRPAIC,PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,
 2 PRBL,PRDCP,PRGNAC,PRGNAC,PRSL,PRLW,PRNW,PRCM BCSFRB 00001
 EQUIVALENCE (PRUW,PRDW) IICONT 00005
 LOGICAL OPLAIC,OSPAIC,WTGEOM,WTGNAF,WTSL,WTBL,PRBOX,PRFAIC,
 1 PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAC,
 2 PRDCP,PRGNAC,PRUW,PRLW,PRNW,PRCM IICONT 00007
 COMMON /TAPEIO/ NFS,NMS,LS,NMR, ID(20),NID,ITYPE,LRS,LWS,M,N, TAPEIO J 00002
 1 PARM(10),IRR TAPEIO 00003
 DIMENSION IPARM(10) TAPEIO 00004
 EQUIVALENCE (PARM,IPARM) TAPEIO 00005
 COMMON / MODES/ SYM,SYMT,MTYPEW,MTYPE^T MDCOM 00002
 COMMON /ARRAYS/ KBXCDW,LBXCDW,LBOXC^L,BXCDT,LBXCDT,KJALPH,LJALPH, ARRAYS 00002
 1 KALPHA,KKERNL,LKER^L,KPNTRM,LPNTRM,KDEFSL,KELPHI, ARRAYS 00003
 2 LHODES,KPNTSD,LPTNSD,KSDW,LSDW,KPNTDW,LPTDW, ARRAYS 00004
 3 KDW,LDW,KTVP,LTV^P ARRAYS 00005
 COMMON /SAMPLW/ ISAMPLW,ICHORD(10),IBOXF(10),IBOXL(10),ZLOC(10) SAMPLW 00002
 COMMON /MUACIS/ YBAR,EL,MUAC(2,50),NROWS,SURF, MUACIS 00002
 1 YBARL,ELL, MUACL(2,50),NROWSL,SURFL,PSIDIF MUACIS 00003
 LOGICAL SURF,SURFL MUACIS 00004
 COMMON /EDGES / FEXLOC(250), TEXLOC(250),JDIAG EDGES 00002

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COMMON /PLANXY/ NALE,NATE,NTLE,NTTE, X'LE(10),Y'LE(10),
1           X'WE(10),Y'WE(10), X'LE(10),Y'TLE(10),
2           X'TE(10),Y'TE(10)
LOGICAL MXWRIT,RANDOU, MXREAD,RANDIN
COMMON /CHECKPR/ DPPCPR,GEOPCR,MDCPR,AICCP,NSCPR,SMCPR,GFCPR
LOGICAL DPPCPR, GEOPCR, MDCPR, AICCP,NSCPR, SMCPR, GFCPR
EQUIVALENCE (CHECKPR,GEOPCR)
LOGICAL CHECKPR
C
C
DIMENSION IBOXW(150,8),IBOXT(90,8),
1   IWAKE(160), ICODE(160)
DIMENSION KPTW(50),KPTTT(50),KPTLWT(50),KPTRWT(50)
DIMENSION ALPHA(200), IJALPH(200)
DIMENSION KPT(4,50)
EQUIVALENCE (KPT,ALPHA)
DATA MXWRIT,RANDOU, MXREAD,RANDIN / 4*.F. /
DATA NBWRD /20/
DATA INIT,XINIT / 37767DB. 376543217777777777B/
DATA EPS / 1.0E-4 /
C
C      NAMELIST PARAMETERS FOR CARDS TO BE READ IN THIS SECTION
NAMELIST /CARDF/ WLAX,WLAZ,PSIW, TLAX,TLAZ, PSIT, CHECKPR
1   /CARDG / NCARDS, XCENTR, XEDGE, ICHORD,IBOXF,IBOXL,ZLOC
C         CARDH NALE, NATE, NTLE, NTTE (4I5)
C         CARDS I TO L (6E10.0)
C
5001 FORMAT(4I5)
5002 F'FORMAT(6E10.0)
C
C
LSCHDS = LBOXC * 20
XSUBDV = NSUBDV
NSUBD2 = NSUBDV/2
NSUBCN = NSUBD2 + 1
HALFBX = XSUBDV/2.0
C
C      IS PREVIOUS GEOMETRY TO BE USED -
IF (.NOT. PRVGEOM) GO TO 15
C      YES. HAS THE MACH NUMBER CHANGED -
IF ( XMACH .EQ. OMACH) GO TO 2000
C      YES. SKIP THE GEOMETRY READS, BUT RECO THE REST FOR THE
C      NEW BOX PATTERN
B10 = 31
XCENTR = XCENTR
GO TO 272
C
C      READ CARDS F AND G
15 CONTINUE
WLAX = 0.
WLAZ = 0.
PSIW = 0.
TLAX = 0.
TLAZ = 0.
PSIT = 0.
READ (NT5,CARDF)
WRITE (NT5,6010) WLAX,WLAZ,PSIW

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PLANXY	00002
PLANXY	00003
PLANXY	00004
GEOBX	00036
CHECKPR	00002
CHECKPR	00003
GEOBX	00039
GEOBX	00040
GEOBX	00042
GEOBX	00051
GEOBX	00052
GEOBX	00053
GEOBX	00054
GEOBX	00055
GEOBX	00056
GEOBX	00057
FTNX1	00008
FTNX1	00009
GEOBX	00058
GEOBX	00059
FTNX1	00010
FTNX1	00011
FTNX1	00012
FTNX1	00013
FTNX1	00014
FTNX1	00015
FTNX1	00016
FTNX1	00017
FTNX1	00018
FTNX1	00019
GEOBX	00060
GEOBX	00061
GEOBX	00062
GEOBX	00063
GEOBX	00064
GEOBX	00065
GEOBX	00066
GEOBX	00067
GEOBX	00068
GEOBX	00069
GEOBX	00070
GEOBX	00071
GEOBX	00072
GEOBX	00073
GEOBX	00074
GEOBX	00075
GEOBX	00076
GEOBX	00077
GEOBX	00078
GEOBX	00079
GEOBX	00080
GEOBX	00081
GEOBX	00082
GEOBX	00083
GEOBX	00084
GEOBX	00085
GEOBX	00086

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IF (NSURF .EQ. 2) WRITE (NT6,6012) TLAX,TLAZ,PSIT           GEOMBX 00087
DEGREE = .01745329251943                                GEOMBX 00088
C      CONVERT DEGREES TO RADIANS                         GEOMBX 00089
IVAL = 4HPSIW                                           GEOMBX 00090
IF (PSIW .GT. 45. .OR. PSIW .LT. -45.) GO TO 8030        GEOMBX 00091
PSIW = PSIW * DEGREE                                     GEOMBX 00092
IF (NSURF .EQ. 1) GO TO 30                               GEOMBX 00093
IVAL = 4HPSIT                                           GEOMBX 00094
C      SPECIAL CHECK FOR VERTICAL TAIL                   GEOMBX 00095
IF (PSIT .EQ. 90. .AND. SYM .LE. 0) GO TO 25            GEOMBX 00096
IF (PSIT .GT. 45. .OR. PSIT .LT. -45.) GO TO 8030        GEOMBX 00097
C      SYMT = SYMMETRY INDICATOR FOR THE TAIL, IDENTICAL TO GEOMBX 00098
C      THE WING EXCEPT FOR A VERTICAL TAIL                GEOMBX 00099
SYMT = SYM                                              GEOMBX 00100
GO TO 28                                                 GEOMBX 00101
25 CONTINUE                                             GEOMBX 00102
SYMT = 0.                                                 GEOMBX 00103
26 CONTINUE                                             GEOMBX 00104
PSIT = PSIT * DEGREE                                     GEOMBX 00105
C
30 CONTINUE                                             GEOMBX 00106
NCHRDS = INIT                                           GEOMBX 00107
XCENTR = XINIT                                         GEOMBX 00108
XEDGE = XINIT                                         GEOMBX 00109
DO 50 I=1,10                                           GEOMBX 00110
ICHORD(I) = INIT                                       GEOMBX 00111
IBOXF(I) = INIT                                       GEOMBX 00112
IBOXL(I) = INIT                                       GEOMBX 00113
50 ZLOC(I) = XINIT                                     GEOMBX 00114
READ(NT5,CARDG)                                         GEOMBX 00115
C
C      CHECK AND PRINT PARAMETERS READ                  GEOMBX 00116
WRITE (NT6,6015) NCHRDS,XCENTR, XEDGE                 GEOMBX 00117
C
IVAL = 8NCHRDS                                         GEOMBX 00118
IF (NCHRDS .EQ. INIT) GO TO 8010                      GEOMBX 00119
IF (NCHRDS .LE. 0 .OR. NCHRDS .GE. LSCHDS/NSUBDV) GO TO 8015 GEOMBX 00120
MYBW = NCHRDS                                         GEOMBX 00121
IVAL = 8XCENTR                                         GEOMBX 00122
IF (XCENTR .EQ. XINIT) GO TO 120                      GEOMBX 00123
C      USE XCENTR DIRECTLY, IGNORE XEDGE               GEOMBX 00124
IF (XEDGE .NE. XINIT) WRITE (NT6,9010)                 GEOMBX 00125
GO TO 125                                              GEOMBX 00126
C      GET XCENTR FROM XEDGE                           GEOMBX 00127
120 CONTINUE                                            GEOMBX 00128
IF (XEDGE .EQ. XINIT) GO TO 8020                      GEOMBX 00129
125 CONTINUE                                            GEOMBX 00130
IF (ISMPLW .EQ. 0) GO TO 200                          GEOMBX 00131
IF (NSURF .EQ. 2) GO TO 170                          GEOMBX 00132
DO 150 I = 1,ISMPLW                                    GEOMBX 00133
IF (ICHORD(I) .GT. MYBW .OR. ICHORD(I) .LE. 0) GO TO 130 GEOMBX 00134
IF (IBOXF(I) .EQ. INIT .OR. IBOXL(I) .EQ. INIT) GO TO 140 GEOMBX 00135
IF (IBOXF(I) .LT. 1 .OR. IBOXL(I) .GT. LBXCDF/NSUBDV) GO TO 130 GEOMBX 00136
IF (IBOXF(I) .LE. IBOXL(I) ) GO TO 140              GEOMBX 00137
130 WRITE (NT6,9020) I                                 GEOMBX 00138
ISMPLW = I - 1                                         GEOMBX 00139
GO TO 160                                              GEOMBX 00140
                                                GEOMBX 00141
                                                GEOMBX 00142
                                                GEOMBX 00143

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140 CONTINUE           GEOMBX 00144
    IF (ZLOC(I) .EQ. XINIT) ZLOC(I) = 0.
150 CONTINUE           GEOMBX 00145
160 CONTINUE           GEOMBX 00146
    WRITE (NT6,6017) ISMPLW, (ICHORD(I), IBOXF(I), IBOXL(I), ZLOC(I),
1   I = 1,ISMPLW)
    GO TO 200           GEOMBX 00147
C      SAMPLING OF SWASHES ILLEGAL IF TAIL DEFINED
170 CONTINUE           GEOMBX 00148
    WRITE (NT6,9030) ISMPLW           GEOMBX 00149
    ISMPLW = 0            GEOMBX 00150
200 CONTINUE           GEOMBX 00151
C
C      OBTAIN THE LEADING AND TRAILING EDGE VALUES
C      CARD INPUT OF PLANFORMS IS REQUIRED
    READ (NT5,5001) NMLE,NWTE,NTLE,NTTE           GEOMBX 00152
210 WRITE(NT6,6021) NMLE,NWTE           GEOMBX 00153
    GO TO (214,212),NSURF           GEOMBX 00154
212 WRITE(NT6,6022) NTLE,NTTE           GEOMBX 00155
214 IVAL = 4*NMLE           GEOMBX 00156
    IF (NMLE .LT. 2 .OR. NMLE .GT. 10) GO TO 8030
    IVAL = 4*NWTE           GEOMBX 00157
    IF (NWTE .LT. 2 .OR. NWTE .GT. 10) GO TO 8030
    IF (NSURF .EQ. 1) GO TO 220           GEOMBX 00158
    IVAL = 4*NTLE           GEOMBX 00159
    IF (NTLE .LT. 2 .OR. NTLE .GT. 10) GO TO 8030
    IVAL = 4*NTTE           GEOMBX 00160
    IF (NTTE .LT. 2 .OR. NTTE .GT. 10) GO TO 8030
220 CONTINUE           GEOMBX 00161
C
C      CARDS I AND J - WING DEFINITION POINTS
    WRITE (NT6,6029)           GEOMBX 00162
    IVAL = 9*WING L.E.           GEOMBX 00163
    READ (NT5,5002) (XMLE(I),YMLE(I),I=1,NMLE)
    WRITE (NT6,6030) IVAL, (XMLE(I),YMLE(I),I=1,NMLE)           GEOMBX 00164
    CALL EDGCHK(XMLE,YMLE,NMLE,1,IRR)
    IF (IRR .NE. 0) GO TO 8050           GEOMBX 00165
    IVAL = 9*WING T.E.           GEOMBX 00166
    READ (NT5,5002) (XWTE(I),YWTE(I),I=1,NWTE)
    WRITE (NT6,6030) IVAL, (XWTE(I),YWTE(I),I=1,NWTE)           GEOMBX 00167
    CALL EDGCHK(XWTE,YWTE,NWTE,2,IRR)
    IF (IRR .NE. 0) GO TO 8050           GEOMBX 00168
    IF (NSURF .EQ. 1) GO TO 270           GEOMBX 00169
C
C      CARDS K AND L - TAIL DEFINITION POINTS
    IVAL = 9*TAIL L.E.           GEOMBX 00170
    READ (NT5,5002) (XTLE(I),YTLE(I),I=1,NTLE)
    WRITE (NT6,6030) IVAL, (XTLE(I),YTLE(I),I=1,NTLE)           GEOMBX 00171
    CALL EDGCHK(XTLE,YTLE,NTLE,1,IRR)
    IF (IRR .NE. 0) GO TO 8050           GEOMBX 00172
    IVAL = 9*TAIL T.E.           GEOMBX 00173
    READ (NT5,5002) (XTTE(I),YTTE(I),I=1,NTTE)
    WRITE (NT6,6030) IVAL, (XTTE(I),YTTE(I),I=1,NTTE)           GEOMBX 00174
    CALL EDGCHK(XTTE,YTTE,NTTE,2,IRR)
    IF (IRR .NE. 0) GO TO 8050           GEOMBX 00175
    GO TO 270           GEOMBX 00176
C

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C      PLANFORM DEFINITIONS TO BE READ FROM TAPE           GEOMBX 00201
230 CONTINUE                                     GEOMBX 00202
C                                         GEOMBX 00203
270 CONTINUE                                     GEOMBX 00204
    B1BETA = YMLE(NMLE)/MYBW                     GEOMBX 00205
272 CONTINUE                                     GEOMBX 00206
    B1 = B1BETA * SQRT((XKACH*XKACH-1.0))        GEOMBX 00207
    B1BTAZ = B1BETA* 0.5                          GEOMBX 00208
    B12 = B1 + 0.5                                GEOMBX 00209
    IF (NSUBDV .NE. 1) GO TO 275                  GEOMBX 00210
    BIS = B1                                       GEOMBX 00211
    BIBTAS = B1BETA                                GEOMBX 00212
    GO TO 280                                      GEOMBX 00213
275 BIS = B1/XSUBDV                            GEOMBX 00214
    BIBTAS = B1BETA/XSUBDV                         GEOMBX 00215
280 CONTINUE                                     GEOMBX 00216
    WRITE (NTS,6040) B1,B1BETA                   GEOMBX 00217
C                                         GEOMBX 00218
C      SET THE XVAL ARRAY IF XS WAS INPUT          GEOMBX 00219
C                                         GEOMBX 00220
    IF(XKS(1).EQ.-1.0) GO TO 295                 GEOMBX 00221
    DO 290 I=1,NKVALS                           GEOMBX 00222
    XVAL(I) = XKS(I) * (B1/YMLE(NMLE))          GEOMBX 00223
290 CONTINUE                                     GEOMBX 00224
295 CONTINUE                                     GEOMBX 00225
C                                         GEOMBX 00226
C      DETERMINE THE GLOBAL COORDINATE LOCATION OF THE FIRST UN-   GEOMBX 00227
C      SUBDIVIDED PLANFORM BOX CENTER, XCENTR             GEOMBX 00228
    IVAL = @XCENTR                                 GEOMBX 00229
    IF (XCENTR .EQ. XINIT) XCENTR = XEDGE + B12       GEOMBX 00230
    XEDGEW = XMLE(1) + (XMLE(2)-XMLE(1)) * B1BTAZ / YMLE(2)  GEOMBX 00231
    IF (PRVGEOM) XEDGEW = B10* XMLE(1) - B10 + XCNTR0 +
    1     B10 *(XMLE(2) - XMLE(1)) * .5 / (YMLE(2) -.5)  GEOMBX 00232
    IF (XCENTR-XEDGEW) 310,330,320                  GEOMBX 00233
310 DO 315 I = 1,51                           GEOMBX 00234
    XCENTR = XCENTR + B1                         GEOMBX 00235
    IF (XCENTR .GE. XEDGEW) GO TO 330            GEOMBX 00236
315 CONTINUE                                     GEOMBX 00237
    GO TO 8060                                    GEOMBX 00238
320 DO 325 I = 1,51                           GEOMBX 00239
    IF (XCENTR-B1 .LT. XEDGEW) GO TO 330          GEOMBX 00240
    XCENTR = XCENTR - B1                         GEOMBX 00241
325 CONTINUE                                     GEOMBX 00242
    GO TO 8060                                    GEOMBX 00243
330 CONTINUE                                     GEOMBX 00244
C      IS PREVIOUS GEOMETRY BEGIN USED -          GEOMBX 00245
    IF (.NOT. PRVGEOM) GO TO 355                GEOMBX 00246
C                                         GEOMBX 00247
C      YES. CONVERT X-COORDINATE VALUES TO NEW BOX LENGTH          GEOMBX 00248
    PSLIDIF = PSIT - PSTW                         GEOMBX 00249
    SLIDE = -B10 + XCNTR0 - XCENTR               GEOMBX 00250
    DO 335 I = 1,NMLE                           GEOMBX 00251
335 XMLE(I) = (B10*XMLE(I) + SLIDE)/B1 + 1.0  GEOMBX 00252
    DO 340 I = 1,NMLE                           GEOMBX 00253
340 XMLE(I) = (B10*XMLE(I) + SLIDE)/B1 + 1.0  GEOMBX 00254
    IF (NSURF .EQ. 1) GO TO 390                GEOMBX 00255
    DO 345 I = 1,NMLE                           GEOMBX 00256
345 XMLE(I) = (B10*XMLE(I) + SLIDE)/B1 + 1.0  GEOMBX 00257

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345 XITLE(I) = (B10*XITLE(I) + SLIDE)/B1 + 1.0           GEOMBX 00258
DO 350 I = 1,NTTE                                         GEOMBX 00259
350 XTTTE(I) = (B10*XTTTE(I) + SLIDE)/B1 + 1.0           GEOMBX 00260
GO TO 390                                                 GEOMBX 00261
C
C      CONVERT GEOMETRIC INFORMATION TO THE NON-DIMENSIONAL   GEOMBX 00262
C      NC, MC, LC COORDINATE SYSTEM                           GEOMBX 00253
C
345 CONTINUE                                              GEOMBX 00264
DO 360 I = 1,NMLE                                         GEOMBX 00265
XMLE(I) = (XMLE(I)-XCENTR)/B1 + 1.0                      GEOMBX 00266
360 YMLE(I) = YMLE(I)/B1BETA + 0.5                         GEOMBX 00267
DO 365 I = 1,NTTE                                         GEOMBX 00268
XTTTE(I) = (XTTTE(I)-XCENTR)/B1 + 1.0                     GEOMBX 00269
365 YTTTE(I) = YTTTE(I)/B1BETA + 0.5                       GEOMBX 00270
GO TO (370,375),NSURF                                     GEOMBX 00271
370 CAPL = 0.                                                 GEOMBX 00272
PSIT = 0.                                                 GEOMBX 00273
PSIDIF = -PSIW                                           GEOMBX 00274
MWBT = 0.                                                 GEOMBX 00275
MWBBT = 0.                                               GEOMBX 00276
MWBST = 0.                                               GEOMBX 00277
MWBBST = 0.                                              GEOMBX 00278
IF (ISMPLW.EQ.0) GO TO 390                               GEOMBX 00279
C      TRANSFORM ZLOC FOR THE SAMPLE WASH CHORDS TO A NON-DIMENSIONAL   GEOMBX 00281
C      UNROTATED LC COORDINATE HAVING ITS ZERO ON THE WING CENTER     GEOMBX 00282
C      LINE                                                       GEOMBX 00283
DO 372 I = 1,ISMPLW                                       GEOMBX 00284
ZLOC(I) = (ZLOC(I) - WLAZ) /B1BETA                      GEOMBX 00285
372 CONTINUE                                              GEOMBX 00286
GO TO 390                                                 GEOMBX 00287
375 XDIFF = WLAX + XCENTR - TLAX                         GEOMBX 00288
DO 380 I = 1,NTLE                                         GEOMBX 00289
XITLE(I) = (XITLE(I)-XDIFF)/B1 + 1.0                     GEOMBX 00290
380 YTLE(I) = YTLE(I)/B1BETA + 0.5                         GEOMBX 00291
DO 385 I = 1,NTTE                                         GEOMBX 00292
XTTTE(I) = (XTTTE(I)-XDIFF)/B1 + 1.0                     GEOMBX 00293
385 YTTTE(I) = YTTTE(I)/B1BETA + 0.5                       GEOMBX 00294
CAPL = (TLAZ-WLAZ)/B1BETA                                GEOMBX 00295
PSIDIF = PSIT - PSIW                                      GEOMBX 00296
C
C      CHECK FOR TAIL CROSSING WING                           GEOMBX 00297
IF (PSIDIF) 386,389,387                                    GEOMBX 00298
386 IF (CAPL .LE. 0) GO TO 389                           GEOMBX 00299
GO TO 388                                                 GEOMBX 00300
387 IF (CAPL .GE. 0) GO TO 389                           GEOMBX 00301
388 YCROSS = CAPL/(SIN(PSIW)*SIN(PSIT)) + .5            GEOMBX 00302
IF (YMLE(NMLE)*COS(PSIW) .LT. YCROSS) GO TO 390          GEOMBX 00303
IF (YTLE(NTLE)*COS(PSIT) .GE. YCROSS) GO TO 3800         GEOMBX 00304
GO TO 390                                                 GEOMBX 00305
389 YCROSS = .5                                           GEOMBX 00306
C
C      ZERO OUT THE BOX CODE ARRAYS                          GEOMBX 00307
390 CONTINUE                                              GEOMBX 00308
DO 430 J = 1,LBONC                                         GEOMBX 00309
DO 420 I = 1,LBXCDW                                         GEOMBX 00310
420 IBORW(I,J) = 0                                         GEOMBX 00311
DO 430 I = 1,LBXCT                                         GEOMBX 00312
                                         GEOMBX 00313
                                         GEOMBX 00314

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430 IBORT(I,J) = 0                                GEOMBX 00315
C
C      GET THE (SUBDIVIDED) BOX CODE ARRAY FOR THE ON-PLANFORM WING    GEOMBX 00316
C      BOXES
C      IXBW = 0                                GEOMBX 00317
C      IXBT = 0                                GEOMBX 00318
C      IXBT = 0                                GEOMBX 00319
C      IXBT = 0                                GEOMBX 00320
C
C      CALL BXCDPF( XMLE,YMLE,MYLE,XMTE,YMTE,MYTE, LBXCDW,IBOWW)    GEOMBX 00321
C          RETURNS- IBOWW , ONES FOR ON PLANFORM BOXES
C          IXBW = LOCATION OF FIRST UNSUBDIVIDED BOX CENTER    GEOMBX 00322
C          MXBSW = NUMBER OF SUBDIVIDED ROWS ON THE WING    GEOMBX 00323
C          MYBSW = NUMBER OF SUBDIVIDED CHORDS ON THE WING    GEOMBX 00324
C          MXBW = NUMBER OF UNSUBDIVIDED ROWS    GEOMBX 00325
C          MYBW = NUMBER OF UNSUBDIVIDED CHORDS    GEOMBX 00326
C          FEXLOC = ARRAY OF LEADING EDGE LOCATIONS    GEOMBX 00327
C          TEXLOC = ARRAY OF TRAILING EDGE LOCATIONS    GEOMBX 00328
C
C          MYBBSW = MYBSW
C          MXBBSW = MXBW
C          MXBBSW = MXBSW
C          IF (.NOT. CHECKPR) GO TO 440
C          CALL PRNTBC(IBOWW,LBXCDW, 1, MXBSW, MYBSW, .T. )
C          WRITE (NT6,7040) ( FEXLOC(I), I = 1,MYBSW)
C          WRITE (NT6,7045) (TEXLOC(I), I = 1,MYBSW)
440 CONTINUE
C
C      SEARCH THE WING FOR THE FORWARD MOST DIAGONAL INTERSECTING    GEOMBX 00329
C      AN ON-PLANFORM BOX. THIS DEFINES THE LIMIT FOR ANY TIP    GEOMBX 00330
C      DIAPHRAGM.
C      JDIAG = THE J-LOCATION (SUBDIVIDED) OF THE DIAGONAL AT    GEOMBX 00331
C          THE FIRST ROW OF THE PATTERN.    GEOMBX 00332
C
C          JDIAG = 1                                GEOMBX 00333
C          PREV = 0.                                GEOMBX 00334
C          DO 530 J = 2,MYBSW
C          PREV = PREV + 1.0
C          IF (FEXLOC(J) .GT. PREV) GO TO 530
C          PREV = FLOAT(IFIX(FEXLOC(J)))
C          JDIAG = J - PREV
530 CONTINUE
C
C      INITIALIZE THE IWAKE ARRAY
C      DO 540 J = 1,MYBSW
C          IWAKE(J) = TEXLOC(J)
C
540 CONTINUE
C      IF (MYBSW .EQ. LSCHDS) GO TO 548
C      MYBSW1 = MYBSW + 1
C      DO 544 J = MYBSW1,LSCHDS
C          IWAKE(J) = 0
C
544 CONTINUE
C      IF (INSURF .NE. 2) GO TO 705
C          THERE ARE 2 SURFACES. DETERMINE THE FIRST PLANFORM BOX OF THE GEOMBX 00367
C          SECOND SURFACE
C          YM1N = .5*(1.0 + 1.0/XSUBDV)
C          DELE = (XTLE(2)-XTLE(1)) / (YTLE(2)-YTLE(1))
C          XM1NS = XTLE(1)+ (YM1N-YTLE(1)) * DELE

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IXBST = XSUBDV*(XMIN-1.0) + IXBW + 1           GEOMBX 00372
IF (AINT(XMIN)) .EQ. XMIN) IXBST = IXBST - 1   GEOMBX 00373
XMIN = XTLE(1)+ (1.0 - YTLE(1)) * DELE        GEOMBX 00374
IXBT = XMIN                                     GEOMBX 00375
IF (FLOAT(IXBT) .EQ. XMIN) IXBT = IXBT - 1     GEOMBX 00376
IXBT = NSUBDV * IXBT + IXBW                     GEOMBX 00377
C          IXBT = LOCATION OF FIRST SUBDIVIDED TAIL BOX    GEOMBX 00378
C          IXBT = LOCATION OF FIRST UNSUBDIVIDED TAIL BOX CENTER  GEOMBX 00379
ISUBT = 2 - IXBST                                GEOMBX 00380
C          ISUBT = THE SUBSCRIPT FOR ARRAY IBOXT WHICH WILL KEEP TAIL  GEOMBX 00381
C          ROWS WITHIN THE BOUNDS OF IBOXT                         GEOMBX 00382
IF (CARL .NE. 0) GO TO 510                        GEOMBX 00383
IF (PSIDIF .EQ. 0) GO TO 700                      GEOMBX 00384
C
C          THE TWO SURFACES ARE NOT COPLANAR               GEOMBX 00385
C
510 CONTINUE                                         GEOMBX 00386
COPLAN = .F.                                       GEOMBX 00387
C          DETERMINE THE BOX CODES FOR THE SECOND PLANFORM      GEOMBX 00388
CALL BXCDPF(XTLE,YTLE,NTLE,XTTE,YTTE,NTTE,LBXCDT,IBOXT(ISUBT,1))  GEOMBX 00389
C          RETURNS - IBOXT, ONES FOR ON-PLANFORM BOXES          GEOMBX 00390
C          MXBST = NUMBER OF SUBDIVIDED ROWS TO END OF TAIL      GEOMBX 00391
C          MYBST = NUMBER OF SUBDIVIDED CHORDS ON TAIL            GEOMBX 00392
C          MXBT = NUMBER OF UNSUBDIVIDED ROWS, BOTH PLANFORMS     GEOMBX 00393
C          MYBT = NUMBER OF UNSUBDIVIDED CHORDS ON TAIL           GEOMBX 00394
C          FEXLOC = LEADING EDGE LOCATIONS, BOTH PLANFORMS       GEOMBX 00395
C          TEXLOC = TRAILING EDGE LOCATIONS                      GEOMBX 00396
C
C          GET DIAPHRAGM VALUES FOR THE TAIL                    GEOMBX 00397
MYBBST = MYBT                                       GEOMBX 00398
IF (.NOT. CHECKPR) GO TO 515                       GEOMBX 00399
CALL PRNTBC(IBOXT(ISUBT,1),LBXCDT,IXBST, MXBST,MYBST,.T.)  GEOMBX 00400
II = MYBSW + 1                                     GEOMBX 00401
III = MYBSW + MYBT                                GEOMBX 00402
WRITE (NT6,7040) (FEXLOC(I), I = II,III)          GEOMBX 00403
WRITE (NT6,7045) (TEXLOC(I), I = II,III)          GEOMBX 00404
515 CONTINUE                                         GEOMBX 00405
IWK = 0                                            GEOMBX 00406
CALL BXCDI ( IWK, LBXCDT,LSCHDS, IBOXT(ISUBT,1) )  GEOMBX 00407
C          RETURNS - IBOXT, CODES 2 AND 3 ADDED FOR DIAPHRAGM AND WAKE  GEOMBX 00408
C          MYBBST = NUMBER OF SUBDIVIDED CHORDS, INCLUDING          GEOMBX 00409
C          DIAPHRAGM, FOR TAIL                                     GEOMBX 00410
C          MYBBT = NUMBER OF UNSUBDIVIDED CHORDS                  GEOMBX 00411
C
C          IF (.NOT. (PRBOX .OR. CHECKPR) ) GO TO 520             GEOMBX 00412
CALL PRNTBC(IBOXT(ISUBT,1),LBXCDT, IXBST, MXBST,MYBBST, .T.)  GEOMBX 00413
IF (NSUBDV .EQ. 1) GO TO 520                      GEOMBX 00414
IFR = (IXBT - IXBW)/NSUBDV + 1                   GEOMBX 00415
CALL PRNTBC(IBOXT(ISUBT,1),LBXCDT, IFR, MXBT, MYBBT, .F. )  GEOMBX 00416
520 CONTINUE                                         GEOMBX 00417
C
C          THE FOLLOWING LOOP DETERMINES THE LOCUS OF MAXIMUM AFTWARD  GEOMBX 00418
C          PROJECTIONS OF THE INTERSECTIONS OF THE TAIL MACH CONES     GEOMBX 00419
C          WITH THE WING PLANE (EXTENDED). MACH CONES FOR UNSUBDIVIDED  GEOMBX 00420
C          TAIL CHORDS ARE USED, BUT ALL ARITHMETIC IS IN THE SUBDIVIDED  GEOMBX 00421
C          COORDINATE SYSTEM.                                         GEOMBX 00422
C
C          LOOP ON TAIL CHORDS                                     GEOMBX 00423

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CAPLL = CAPL
      DO 600 JT = NSUBCN, MYBBST, NSUBDV )
C      JT = NSUBCN
 525 CONTINUE
      YCT = JT - .5
C      Y-OFFSET OF RECEIVING CHORD FROM CENTER-LINE, TAIL PLANE
C      GET ICT, THE I-LOCATION OF AFTMOST RECEIVING BOX ON TAIL CHRD
C      IS THE TAIL CHORD ON-PLANFORM OR DIAPHRAGM -
      IF (JT .GT. MYBST) GO TO 550
      JJ = JT + MYBSW
      ICT = TEXLOC(JJ) + EPS - AMOD(TEXLOC(JJ)-IXBW, XSUBDV)
      GO TO 555
 550 CONTINUE
      ICT = IXBT - NSUBDV
 555 CONTINUE
      ICTP1 = ICT + NSUBDV
      IF (ICTP1 .GT. MXBST) GO TO 570
C      CHECK FOR WAKE DIAPHRAGM AFT OF TAIL CHORD
      CALL DCDER( IBOXT(ISUBT,1), LBXCDT, ICTP1, JT, MXBST, JT,
      1 .T., ICODE)
      II = 1
      DO 580 I = ICTP1, MXBST, NSUBDV
      IF (ICODE(II).EQ. 0) GO TO 570
      ICT = I
      II = II + 1
 580 CONTINUE
 570 CONTINUE
C      ICT = X-LOCATION OF AFT-MOST TAIL BOX ON THE CHORD
      EL = COS(PSIW)*CAPL*XSUBDV + SIN(PSIDIF)*YCT
C      EL = PERPENDICULAR DISTANCE FROM RECEIVING CHORD TO RIGHT
C      WING PLANE, POSITIVE DOWNWARD.
C
C      ENTRY INTO THE LOOP FOR WASH SAMPLING CHORDS, FROM 705*
 590 CONTINUE
C      START OF LOOP ON WING CHORDS, ENDING AT 650
      JW = NSUBCN
 600 CONTINUE
      YJW = JW - .5
C      YJW = Y-OFFSET OF SENDING CHORD FROM CENTER LINE,
C      WING PLANE
      YMubar = -YJW + COS(PSIDIF)*YCT + SIN(PSIW)*CAPLL*XSUBDV
C      YMubar = Y-DISTANCE BETWEEN CHORD CENTERS, SENDING (WING)
C      PLANE
      IF (ABS(YMubar) .LE. HALFbx) GO TO 630
      IF (YMubar .LT. -HALFbx) YMubar = YMubar + HALFbx
      IF (YMubar .GT. HALFbx) YMubar = YMubar - HALFbx
C      YMubar = Y-DISTANCE TO NEAREST BOX EDGE, WING PLANE
      XNubar = SQRT(YMubar**2 + (EL*XSUBDV)**2 )
C      XNubar = DISTANCE FORWARD FROM RECEIVING CENTER TO NEAR-
C      EST PORTION OF SENDING CHORD
      GO TO 635
 630 CONTINUE
      XNubar = ABS(EL)*XSUBDV
 635 CONTINUE
      XNubar = XNubar + HALFbx
      INTRST = ICT - 1FIX(XNubar+EPS - AMOD(XNubar,XSUBDV) )
      IF (JW .GT. MYBSW) GO TO 640

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GEOMBX 00429
 FTNXI 00020
 FTNXI 00021
 FTNXI 00022
 GEOMBX 00431
 GEOMBX 00432
 GEOMBX 00433
 GEOMBX 00434
 GEOMBX 00435
 GEOMBX 00436
 GEOMBX 00437
 GEOMBX 00438
 GEOMBX 00439
 GEOMBX 00440
 GEOMBX 00441
 GEOMBX 00442
 GEOMBX 00443
 GEOMBX 00444
 GEOMBX 00445
 GEOMBX 00446
 GEOMBX 00447
 GEOMBX 00448
 GEOMBX 00449
 GEOMBX 00450
 GEOMBX 00451
 GEOMBX 00452
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 GEOMBX 00454
 GEOMBX 00455
 GEOMBX 00456
 GEOMBX 00457
 GEOMBX 00458
 GEOMBX 00459
 GEOMBX 00460
 GEOMBX 00461
 GEOMBX 00462
 GEOMBX 00463
 GEOMBX 00464
 GEOMBX 00465
 GEOMBX 00466
 GEOMBX 00467
 GEOMBX 00468
 GEOMBX 00469
 GEOMBX 00470
 GEOMBX 00471
 GEOMBX 00472
 GEOMBX 00473
 GEOMBX 00474
 GEOMBX 00475
 GEOMBX 00476
 GEOMBX 00477
 GEOMBX 00478
 GEOMBX 00479
 GEOMBX 00480
 GEOMBX 00481
 GEOMBX 00482
 GEOMBX 00483

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IWAKE(JW) = MAX0(IWAKE(JW),INTRST)           GEOMBX 00484
GO TO 650                                     GEOMBX 00485
640 CONTINUE                                     GEOMBX 00486
IF (INTRST .LE. JW-JDIAG) GO TO 660          GEOMBX 00487
MYBBSW = JW                                     GEOMBX 00488
IWAKE(JW) = INTRST                            GEOMBX 00489
650 CONTINUE                                     GEOMBX 00490
IF (NSUBD2 .EQ. 0) GO TO 657                  GEOMBX 00491
DO 655 I = 1,NSUBD2                           GEOMBX 00492
IWAKE(JW-I) = IWAKE(JW) - I                   GEOMBX 00493
IWAKE(JW+I) = IWAKE(JW) - I                   GEOMBX 00494
655 CONTINUE                                     GEOMBX 00495
657 CONTINUE                                     GEOMBX 00496
JW = JW + NSUBDV                             GEOMBX 00497
GO TO 600                                     GEIMBX 00498
C      END OF LOOP ON WING CHORDS             GEIMBX 00499
C
660 CONTINUE                                     GEOMBX 00500
IF (ISMPW .NE. 0) GO TO 706                  GEOMBX 00501
660 CONTINUE                                     GEOMBX 00502
JT = JT + NSUBDV                            GEOMBX 00503
IF (JT .LE. MYBBST) GO TO 525               FTNXI 00023
C      END OF LOOP ON TAIL CHORDS, FROM 548*   FTNXI 00024
C
665 CONTINUE                                     GEOMBX 00504
M0BBSW = MXBBSW                            GEOMBX 00505
DO 660 JW = NSUBCN,MYBBSW,NSUBDV            GEOMBX 00506
M0BBSW = MAX0(MXBBSW,IWAKE(JW))            GEOMBX 00507
660 CONTINUE                                     GEOMBX 00508
M0BBSW = MXBBSW                            GEOMBX 00509
IF (NSUBDV .GT. 1) MXBBSW = (MXBBSW-IXBW)/NSUBDV + 1
IF (CHECKPR) WRITE(NT6,7010) (IWAKE(I),I=1,MYBBSW)
GO TO 720                                     GEOMBX 00510
C
C      THE TWO SURFACES ARE COPLANAR. ENTER THE SECOND PLANFORM
C      INTO THE SAME BOX ARRAY                  GEOMBX 00511
C
700 CONTINUE                                     GEOMBX 00512
COPLAN = .T.                                 GEOMBX 00513
CALL BXCDPF(XTLE,YTLE,NTLE, XTTE,YTTE,NTTE, LBXCW,IB0KW)
M0BBSW = MXBST                               GEOMBX 00514
IF (.NOT. CHECKPR) GO TO 720                 GEOMBX 00515
CALL PRNTBC(IB0KW,LBXCW, IXBT, MXBT, MYBT,.T.)
II = MYBSW + 1                                GEOMBX 00516
III = MYBSW + MYBT                           GEOMBX 00517
WRITE (NT6,7040) (FEXLOC(I), I = II,III)     GEOMBX 00518
WRITE (NT6,7045) (TEXLOC(I), I = II,III)     GEOMBX 00519
GO TO 720                                     GEOMBX 00520
C      NO TAIL IS DEFINED. IS DOWNWASH SAMPLING DESIRED-
705 CONTINUE                                     GEOMBX 00521
COPLAN = .F.                                 GEOMBX 00522
IF (ISMPW .EQ. 0) GO TO 720                  GEOMBX 00523
C      BYPASS THE TAIL PLANFORM AND BOX CODE SETUP, AND LOOP ON SAMPL GEOMBX 00524
C      CHORDS TO DEFINE WING WAKE REGION          GEOMBX 00525
C      ( DO 706 JCHRD = 1,ISMPLW )
JCHRD = 1                                     FTNXI 00026
704 CONTINUE                                     FTNXI 00027
JT = (JCHRD(JCHRD)-1)*NSUBDV + NSUBCN       GEOMBX 00528

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IF (IBOXF(JCHRD) .EQ. INIT) IBOXF(JCHRD) = (TEXLOC(JT)-IXBW)/ NSUBDV + 1 GEOMBX 00537
IF (IBOXL(JCHRD) .EQ. INIT) IBOXL(JCHRD) = (TEXLOC(JT)-IXBW)/ NSUBDV + 1 GEOMBX 00538
YCT = JT - .5 GEOMBX 00539
ICT = (IBOXL(JCHRD)-1) * NSUBDV + IXBW GEOMBX 00540
EL = COS(PSIW) * ZLOC(JCHRD) - SIN(PSIW) * YCT GEOMBX 00541
CAPLL = ZLOC(JCHRD) GEOMBX 00542
GO TO 580 GEOMBX 00543
C THE LOGIC FOR A TAIL CHORD IS USED. AFTER THE WING WAKE GEOMBX 00544
C BOUNDS ARE DETERMINED FOR THIS TAIL CHORD. CONTROL IS RETURNED GEOMBX 00545
C TO THIS LOOP GEOMBX 00546
706 CONTINUE GEOMBX 00547
708 CONTINUE GEOMBX 00548
JCHRD = JCHRD + 1 GEOMBX 00549
IF (JCHRD .LE. ISMPLW) GO TO 704 FTNX1 00028
C END OF LOOP ON SAMPLE CHORDS FTNX1 00029
GO TO 685 GEOMBX 00550
C GET DIAPHRAGM BOXES CODES FOR THE WING GEOMBX 00551
720 CONTINUE GEOMBX 00552
CALL BXCDI (IWAKE,LBXCDW,LSCHDS, IBOXW) GEOMBX 00553
C RETURNS - IBOXW, CODES 2 AND 3 ADDED FOR DIAPHRAGM REGIONS GEOMBX 00554
C MYBBSW = NUMBER OF SUBDIVIDED CHORDS, INCLUDING DIAPHRAGM GEOMBX 00555
C MYBBW = NUMBER OF UNSUBDIVIDED CHORDS GEOMBX 00556
C PRINT BOX CODES GEOMBX 00557
IF (.NOT. (PRBOX .OR. CHECKPR) ) GO TO 725 GEOMBX 00558
CALL PRNTBC(iboxw,LBXCDW, 1, MXBBSW, MYBBSW, .T.)
IF (NSUBDV .NE. 1) CALL PRNTBC(iboxw,LBXCDW,
1, (MXBBSW-IXBW+NSUBDV)/NSUBDV, MYBBW, .F.)
725 CONTINUE GEOMBX 00559
C DETERMINE THE PLANAR AIC ARRAY SIZE GEOMBX 00560
NPLKRN = MAX0(MXBBSW, MXBT-IXBT/NSUBDV + 1) GEOMBX 00561
IF (COPLAN NPLKRN = MXBT GEOMBX 00562
C WRITE THE BOX CODE ARRAYS INTO THE GEOMETRY SCRATCH FILE GEOMBX 00563
REMD 1GEOSC
CALL RDINIT GEOMBX 00564
ITYPE = 5MHXED GEOMBX 00565
IVAL = 5HIBOXW GEOMBX 00566
PARM(1) = 0. GEOMBX 00567
PARM(2) = XMAXH GEOMBX 00568
M = MXBBSW GEOMBX 00569
N = (MYBBSW-1)/NBWD + 1 GEOMBX 00570
K = LBXCDW GEOMBX 00571
CALL WRTEMX(1GEOSC,MXWRIT,RANDOU,NF3,NM5,LS,NR,LW5,K, ID,
1 IBOXW, ITYPE, M,N, PARM, IRR )
IF (IRR .NE. 0) GO TO 8070 GEOMBX 00572
C IF (INSURF .EQ. 1) GO TO 730 GEOMBX 00573
IF (COPLAN) GO TO 730 GEOMBX 00574
IVAL = 5HIBOXT GEOMBX 00575
M = MXBST - IXBST + 1 GEOMBX 00576
N = (MYBBST - 1)/NBWD + 1 GEOMBX 00577

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K = LBXCDT                                GEOMBX 00592
CALL WRTEMX(IGEOSC,MXRIT,RANDU,NFS,NMS,LS,NMR,LWS,K, ID,    GEOMBX 00593
      IBOXT, ITYPE, M,N, PARM, IRR)          GEOMBX 00594
      IF (IRR .NE. 0) GO TO 8070            GEOMBX 00595
C                                         GEOMBX 00596
C     CHECK FOR DIAPHRAGMS CROSSING VERTICALLY   GEOMBX 00597
IF (YCROSS .LE. .5) GO TO 730              GEOMBX 00598
IF ( FLOAT(MYBBT)*COS(PSIT) .LT. YCROSS) GO TO 730          GEOMBX 00599
IF ( FLOAT(MYBBW)*COS(PSIW) .GE. YCROSS) GO TO 8080          GEOMBX 00600
C                                         GEOMBX 00601
C     WRITE THE LEADING AND TRAILING EDGE LOCATIONS ONTO SCRATCH  GEOMBX 00602
730 CONTINUE                               GEOMBX 00603
      M = 1                                 GEOMBX 00604
      M = MYBSW + MYBST                     GEOMBX 00605
      K = 1                                 GEOMBX 00606
      IVAL = GHTEXLOC                      GEOMBX 00607
      CALL WRTEMX(IGEOSC,MXRIT,RANDU,NFS,NMS,LS,NMR,LWS, K, ID,  GEOMBX 00608
      I          FELOC, ITYPE, M,N, PARM, IRR)        GEOMBX 00609
      IF (IRR .NE. 0) GO TO 8070            GEOMBX 00610
      IVAL = GHTEXLOC                      GEOMBX 00611
      CALL WRTEMX(IGEOSC,MXRIT,RANDU,NFS,NMS,LS,NMR,LWS, K, ID,  GEOMBX 00612
      I          TEXLOC, ITYPE, M,N, PARM, IRR)        GEOMBX 00613
      IF (IRR .NE. 0) GO TO 8070            GEOMBX 00614
C                                         GEOMBX 00615
C     DETERMINE THE ON-PLANFORM FRACTIONAL PART OF ALL UNSUBDIVIDED  GEOMBX 00616
C     BOXES CUT BY A PLANFORM EDGE          GEOMBX 00617
      CALL GMAREA(IBOXW,LBXCDW, .T., ALPHA,IJALPH, NALPHW)       GEOMBX 00618
      NALPH = NALPHW                         GEOMBX 00619
      IF (NSURF .EQ. 1 .OR. COPLAN) GO TO 740                  GEOMBX 00620
      CALL GMAREA(IBOXI(IISUB,1),LBXCDT, .F., ALPHA(NALPHW+1),  GEOMBX 00621
      I          IJALPH(NALPHW+1), NALPHI)           GEOMBX 00622
      NALPH = NALPH + NALPHI                 GEOMBX 00623
      740 CONTINUE                           GEOMBX 00624
      IF (CHECKPR) WRITE(NT6,7030) (IJALPH/I), ALPHA(I), I=1,NALPH )  GEOMBX 00625
C                                         GEOMBX 00626
C     WRITE THE AREA MULTIPLIERS             GEOMBX 00627
      M = 1                                 GEOMBX 00628
      N = NALPH                            GEOMBX 00629
      K = 1                                 GEOMBX 00630
      IPARM(3) = NALPHW                     GEOMBX 00631
      IVAL = SHALPHA                        GEOMBX 00632
      CALL WRTEMX(IGEOSC,MXRIT,RANDU,NFS,NMS,LS,NMR,LWS, K, ID,  GEOMBX 00633
      I          ALPHA, ITYPE, M,N, PARM, IRR)        GEOMBX 00634
      IF (IRR .NE. 0) GO TO 8070            GEOMBX 00635
      IVAL = SHIJALPH                      GEOMBX 00636
      CALL WRTEMX(IGEOSC,MXRIT,RANDU,NFS,NMS,LS,NMR,LWS, K, ID,  GEOMBX 00637
      I          IJALPH, ITYPE, M,N, PARM, IRR)        GEOMBX 00638
      IF (IRR .NE. 0) GO TO 8070            GEOMBX 00639
C                                         GEOMBX 00640
C     DETERMINE THE SPATIAL AIC PARAMETERS   GEOMBX 00641
C     THE MUATC ARRAYS ARE WRITTEN TEMPORARILY ON IVPSC FOR EDITTING  GEOMBX 00642
C     ONTO IWFSC. AFTER THE KPT-- ARRAYS ARE WRITTEN ON             GEOMBX 00643
C     IGEOSC. ALL NSPATK ARRAYS ARE TRANSFERRED TO IGEOSC            GEOMBX 00644
C     ISCR = NUMBER OF MUATCS TRANSFERRED TO IWFSC                  GEOMBX 00645
C     NSCR = TOTAL NUMBER OF MUATCS PRESENTLY ON IVPSC               GEOMBX 00646
C                                         GEOMBX 00647
      ISCR = 0                                GEOMBX 00648

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NBCR = 0	GEOMBX 00649
NRQHBR = 0	GEOMBX 00650
NRQHBL = 0	GEOMBX 00651
IPARM(3) = 0	GEOMBX 00652
IPARM(6) = 0	GEOMBX 00653
REWIND IWTFSC	GEOMBX 00654
REWIND IVPSC	GEOMBX 00655
M = 2	GEOMBX 00656
K = 2	GEOMBX 00657
NSPATK = 0	GEOMBX 00658
NNRK = 0	GEOMBX 00659
IF (SYM .EQ. 0) GO TO 904	GEOMBX 00660
IF (PSIW .EQ. 0 .OR. .NOT. DIHW) GO TO 800	GEOMBX 00661
C	GEOMBX 00662
C START OF LOOP FOR WING-WING PARAMETERS ON RECEIVING CHORDS,	GEOMBX 00663
C UNSUBDIVIDED (DETERMINES SPATIAL MUAIIC VALUES)	GEOMBX 00664
DO 790 JCOL = 1, MYBBW	GEOMBX 00665
C	GEOMBX 00666
C CALL MUAIIC (.T., IBOXW, LBXCW, IWAKE, JCOL)	GEOMBX 00667
C COMPUTES MUAIIC ARRAY FOR THE LEFT SURFACE CONTRIBUTION TO	GEOMBX 00668
C CHORD JCOL OF THE RIGHT SURFACE	GEOMBX 00669
C SURF = .T. INDICATES SOME RIGHT SURFACE CONTRIBUTION WAS	GEOMBX 00670
C FOUND	GEOMBX 00671
IF (.NOT. SURF) GO TO 800	GEOMBX 00672
NNRK = NNRK + 1	GEOMBX 00673
NSPATK = NNRK	GEOMBX 00674
KPTWW(NNRK) = NSPATK	GEOMBX 00675
IVAL = 10H WING-WING	GEOMBX 00676
IF (CHECKFR) WRITE(NT6,7020) IVAL, JCCL, YBAR, EL, NRWS, (MUAIIC(1,	GEOMBX 00677
1 NRWS-I+1), MUAIIC(2, NRWS-I+1), I=1, NRWS)	GEOMBX 00678
C	GEOMBX 00679
C WRITE MUAIIC ARRAY ON THE SCRATCH FILE	GEOMBX 00680
N = NRWS	GEOMBX 00681
PARM(4) = YBAR	GEOMBX 00682
PARM(5) = EL	GEOMBX 00683
CALL WRTEHX(IPVSC, MXWRIT, RANDU, NFS, NMS, LS, NMR, LWS, K, ID,	GEOMBX 00684
1 MUAIIC, ITYPE, M, N, PARM, IRR)	GEOMBX 00685
IF (IRR .NE. 0) GO TO 8075	GEOMBX 00686
NSCR = NSCR + 1	GEOMBX 00687
C	GEOMBX 00688
790 CONTINUE	GEOMBX 00689
C END OF LOOP ON RECEIVING CHORDS FOR WING-WING PARAMETERS,	GEOMBX 00690
C	GEOMBX 00691
800 CONTINUE	GEOMBX 00692
NTTK = 0	GEOMBX 00693
IF (NSURF .NE. 2) GO TO 900	GEOMBX 00694
IF (SYMT .EQ. 0) GO TO 900	GEOMBX 00695
IF (PSIT .EQ. 0 .OR. .NOT. DIHT) GO TO 900	GEOMBX 00696
IF (PSIDIF .EQ. 0.) REWIND IVPSC	GEOMBX 00697
C	GEOMBX 00698
C START OF LOOP FOR TAIL-TAIL PARAMETERS ON RECEIVING CHORDS,	GEOMBX 00699
C UNSUBDIVIDED	GEOMBX 00700
DO 895 JCOL = 1, MYBBT	GEOMBX 00701
C	GEOMBX 00702
C CALL MUAIIC (.F., IBOXT(1,1), LBXCDT, IWAKE, JCOL)	GEOMBX 00703
C COMPUTES MUAIIC ARRAY FOR THE CONTRIBUTION OF THE LEFT TAIL ON	GEOMBX 00704
C CHORD JCOL OF THE RIGHT TAIL SURFACE	GEOMBX 00705

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C      IF (.NOT. SURF) GO TO 900                                GEOMBX 00706
      NTTK = NTTK + 1                                         GEOMBX 00707
C      IVAL = 10H TAIL-TAIL                                     GEOMBX 00708
C          IF THE WING AND TAIL HAVE THE SAME DIHEDRAL, MODIFY THE WING   GEOMBX 00709
C          MUAIC ARRAY WHERE NEEDED BY THE TAIL, AND USE IT           GEOMBX 00710
C          IF (PSIDIF .EQ. 0.) GO TO 840                           GEOMBX 00711
C              OTHERWISE, WRITE THE MUAICS FOUND ONTO IVPSC            GEOMBX 00712
      NSPATK = NSPATK + 1                                         GEOMBX 00713
      KPTTT(NTTK) = NSPATK                                       GEOMBX 00714
      IVAL = 10H TAIL-TAIL                                     GEOMBX 00715
      IF (CHECKPR) WRITE(NT6,7020) IVAL,JCOL,YBAR,EL, NRWS,(MUAIC(1,
1      NRWS-I+1),MUAIC(2,NRWS-I+1), I=1,NRWS)                  GEOMBX 00716
C      WRITE MUAIC ARRAY ON SCRATCH FILE                         GEOMBX 00717
      N = NRWS                                                 GEOMBX 00718
      PARM(4) = YBAR                                           GEOMBX 00719
      PARM(5) = EL                                            GEOMBX 00720
      CALL WRTEMX(IPSC, MXWRIT,RANDU,NFS,NMS,LS,NMR,LWS, K, ID,
1      MUAIC, ITYPE, M,N, PARM, IRR)                            GEOMBX 00721
      IF (IRR .NE. 0) GO TO 8075                               GEOMBX 00722
      NSCR = NSCR + 1                                         GEOMBX 00723
C      GO TO 895                                              GEOMBX 00724
840 CONTINUE
      IF (ISCR .GE. NSCR) GO TO 850                           GEOMBX 00725
      CALL RDINIT                                           GEOMBX 00726
      CALL READMX(IPSC, MXWRIT,RANDU,NFS,NMS,LS,NMR, K, NID, ID,
1      ITYPE, LRS, MUAICL, M,N, PARM, IRR)                   GEOMBX 00727
      IF (IRR .NE. 0) GO TO 8090                               GEOMBX 00728
      ISCR = ISCR + 1                                         GEOMBX 00729
C      MERGE THE TWO MUAIC ARRAYS                           GEOMBX 00730
      DO 845 I = 1,N                                         GEOMBX 00731
      IF (I .GT. NRWS) GO TO 842                           GEOMBX 00732
      IF (MUAICL(1,I) .EQ. 0) GO TO 845                     GEOMBX 00733
      IF (MUAIC(1,I) .EQ. 0) GO TO 842                     GEOMBX 00734
      MUAIC(1,I) = MAXD(MUAIC(1,I),MUAICL(1,I))          GEOMBX 00735
      MUAIC(2,I) = MAXD(MUAIC(2,I),MUAICL(2,I))          GEOMBX 00736
      GO TO 845                                             GEOMBX 00737
842 MUAIC(1,I) = MUAICL(1,I)
      MUAIC(2,I) = MUAICL(2,I)
845 CONTINUE
      NRWS = MAXD(NRWS,N)
      KPTTT(NTTK) = ISCR
      IF (CHECKPR) WRITE(NT6,7020) IVAL,JCOL,YBAR,EL, NRWS,
1      (MUAIC(1,NRWS-I+1),MUAIC(2,NRWS-I+1), I = 1,NRWS)    GEOMBX 00738
      GO TO 895                                             GEOMBX 00739
C      THERE WERE NO MATRICES TO BE MERGED                    GEOMBX 00740
850 NSPATK = NSPATK + 1                                     GEOMBX 00741
      KPTTT(NTTK) = NSPATK                                    GEOMBX 00742
C      WRITE MERGED AICS ONTO 2ND SCRATCH FILE             GEOMBX 00743
855 CONTINUE
      N = NRWS
      CALL WRTEMX(IPSC, MXWRIT,RANDU,NFS,NMS,LS,NMR,LWS, K, ID,
1      MUAIC, ITYPE, M,N, PARM, IRR)                          GEOMBX 00744
      IF (IRR .NE. 0) GO TO 8110                           GEOMBX 00745

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C               GEOMBX 00763
C   893 CONTINUE GEOMBX 00764
C           END OF LOOP FOR TAIL-TAIL PARAMETERS, FROM 800*
C               GEOMBX 00765
C   900 CONTINUE GEOMBX 00766
C           COMPLETE ANY COPY FROM FIRST TO SECOND SCRATCH FILE GEOMBX 00767
C           IF (ISCR .EQ. 0) REWIND IVPSC GEOMBX 00768
C           IF (ISCR .GE. NSCR) GO TO 904 GEOMBX 00769
C           II = ISCR + 1 GEOMBX 00770
C           DO 902 I = II,NSCR GEOMBX 00771
C           CALL RDINIT GEOMBX 00772
C           CALL READMX(IPVSC, MXWRIT, RANDOU, NFS, NMS, LS, NMR, K, NID, ID,
C           I          ITYPE, LRS, MUAIIC, M, N, PARM, IRR) GEOMBX 00773
C           IF (IRR .NE. 0) GO TO 8090 GEOMBX 00774
C           CALL WRTEMX(IWTFSC, MXWRIT, RANDOU, NFS, NMS, LS, NMR, LWS, K, ID,
C           I          MUAIIC, ITYPE, M, N, PARM, IRR) GEOMBX 00775
C           IF (IRR .NE. 0) GO TO 8110 GEOMBX 00776
C   902 CONTINUE GEOMBX 00777
C   904 CONTINUE GEOMBX 00778
C           COMPUTE THE RIGHT AND LEFT WING INFLUENCE PARAMETERS ON THE GEOMBX 00779
C           TAIL OR SAMPLE CHORDS GEOMBX 00780
C           NRWTK = 0 GEOMBX 00781
C           NLWTK = 0 GEOMBX 00782
C           ISCR = 0 GEOMBX 00783
C           NSCR2 = 0 GEOMBX 00784
C           REWIND IVPSC GEOMBX 00785
C           IPARM(6) = 1 GEOMBX 00786
C           INITIALIZE THE MUAIIC ARRAYS GEOMBX 00787
C           DO 908 I = 1,50 GEOMBX 00788
C           MUAIIC(1,I) = I+I GEOMBX 00789
C           MUAIIC(2,I) = 0 GEOMBX 00790
C           MUAICL(1,I) = I+I GEOMBX 00791
C           MUAICL(2,I) = 0 GEOMBX 00792
C   908 CONTINUE GEOMBX 00793
C           IF (CCPLAN) GO TO 1015 GEOMBX 00794
C           IF (INSURF .NE. 2) GO TO 1120 GEOMBX 00795
C           CAPLL = CAPL GEOMBX 00796
C           YMUVSP = CAPLL*SIN(PSIW) GEOMBX 00797
C           JTCOL = MYBSW - NSUBD2 GEOMBX 00798
C
C           START OF LOOP ON TAIL CHORDS, TO COMPUTE GEOMBX 00799
C           WING - TAIL INFLUENCE PARAMETERS GEOMBX 00800
C           ( DO 1010 JCOL = 1,MYBBT ) FTNX1 00030
C           JCOL = 1 FTNX1 00031
C   909 CONTINUE FTNX1 00032
C           IF (JCOL .LE. MYBT) GO TO 910 GEOMBX 00801
C           IROW = (IXBT-IXBW)/NSUBDV + 1 GEOMBX 00802
C           GO TO 915 GEOMBX 00803
C   910 CONTINUE GEOMBX 00804
C           JTCOL = JTCOL + NSUBDV FTNX1 00033
C           IROW = (TEXLOC(JTCOL)-IXBW)/NSUBDV + 1 GEOMBX 00805
C           IF (IROW .EQ. MXBT) GO TO 930 GEOMBX 00806
C   915 CONTINUE GEOMBX 00807
C           CALL DCDCER(IBOXT(ISUBT,1),LBXCDT, IROW,JCOL, MXBT,JCOL,.F.,ICODE) GEOMBX 00808
C           II = 1 GEOMBX 00809
C           DO 917 I = IROW,MXBT GEOMBX 00810
C           IF ((CODE(II)) .NE. 0) GO TO 918 GEOMBX 00811

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II = II + 1                                GEOMBX 00818
J17 CONTINUE                                GEOMBX 00819
918 CONTINUE                                GEOMBX 00820
III = I                                     GEOMBX 00821
DO 920 I = III, MXBT                      GEOMBX 00822
IF (ICODE(II) .EQ. 0) GO TO 925            GEOMBX 00823
II = II + 1                                GEOMBX 00824
920 CONTINUE                                GEOMBX 00825
925 IRW = IRW + II - 2                     GEOMBX 00826
C      ENTRY INTO THE LOOP FROM SAMPLE WASH LOOP, FROM 1120*
930 CONTINUE                                GEOMBX 00827
CALL PPUTAIC( IBOXW,LBXCDW, IRW,JCOL, CAPLL, YMUVP )  GEOMBX 00828
C      GETS THE MUAIIC AND MUAICL ARRAYS FOR RIGHT AND LEFT CONTRI-  GEOMBX 00829
C      BUTIONS TO THE TAIL                                GEOMBX 00830
IF (SURF) GO TO 935                        GEOMBX 00831
IF (.NOT. SURFL) GO TO 1015                GEOMBX 00832
GO TO 985                                    GEOMBX 00833
935 NRWTK = NRWTK + 1                     GEOMBX 00834
C      DETERMINE WHETHER WING AND TAIL ARE PARALLEL    GEOMBX 00835
IF (PSIDIF .EQ. 0 .AND. NSURF .EQ. 2) GO TO 940  GEOMBX 00836
NSPATK = NSPATK + 1                         GEOMBX 00837
KPTRWT(NRWTK) = NSPATK                     GEOMBX 00838
C      WRITE NEW MUAIIC ARRAY ON SCRATCH          GEOMBX 00839
IVAL = 10HR WNG-TAIL                       GEOMBX 00840
N = NROWS                                     GEOMBX 00841
FARM(4) = YBAR                               GEOMBX 00842
FARM(5) = EL                                 GEOMBX 00843
CALL WRTEMX(IPSC, MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS, K, ID,  GEOMBX 00844
1           MUAIIC, ITYPE, M,N, PARM, IRR)        GEOMBX 00845
IF (IRR .NE. 0) GO TO 8075                  GEOMBX 00846
NSCR2= NSCR2 + 1                            GEOMBX 00847
IF (CHECKPR) WRITE(NT6,7020) IVAL,JCOL,YBAR,EL, NROWS,(MUAIIC(1,  GEOMBX 00848
1   NROWS-I+1),MUAIIC(2,NROWS-I+1), I=1,NROWS)       GEOMBX 00849
C      REINITIALIZE THE RIGHT MUAIIC ARRAY          GEOMBX 00850
DO 937 I = 1,NROWS                         GEOMBX 00851
MUAIIC(1,I) = I+I                           GEOMBX 00852
MUAIIC(2,I) = 0                             GEOMBX 00853
937 CONTINUE                                GEOMBX 00854
C      GO TO 982                                GEOMBX 00855
C      OLD MUAIIC ARRAY HAS BEEN FOUND WHICH MATCHES  GEOMBX 00856
940 CONTINUE                                GEOMBX 00857
C      ALL RIGHT HAND MUAIICS ARE THE SAME, PARALLEL SURFACES  GEOMBX 00858
IF (JCOL .GT. 1) GO TO 945                GEOMBX 00859
NSPATK = NSPATK + 1                         GEOMBX 00860
IPARAL = NSPATK                            GEOMBX 00861
945 CONTINUE                                GEOMBX 00862
KPTRWT(NRWTK) = IPARAL                     GEOMBX 00863
NROWSR = MAX0(NROWSR,NROWS)                 GEOMBX 00864
C
980 CONTINUE                                GEOMBX 00865
IVAL = 10HR WNG-TAIL                       GEOMBX 00866
IF (CHECKPR) WRITE(NT6,7020) IVAL,JCOL,YBAR,EL, NROWS,(MUAIIC(1,  GEOMBX 00867
1   NROWS-I+1),MUAIIC(2,NROWS-I+1), I=1,NROWS)       GEOMBX 00868
C
982 CONTINUE                                GEOMBX 00869
IF (.NOT. SURFL) GO TO 1000                GEOMBX 00870
GEOMBX 00871
GEOMBX 00872
GEOMBX 00873
GEOMBX 00874

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985 CONTINUE                                GEOMBX 00875
    NLWTK = NLWTK + 1                         GEOMBX 00876
C      DETERMINE WHETHER BOTH SURFACES HAVE NO DIHEDRAL   GEOMBX 00877
    IF (PSIW .EQ. 0. .AND. PSIDIF .EQ. 0. .AND. NSURF .EQ. 2) GO TO 990 GEOMBX 00878
    NSPATK = NSPATK + 1                         GEOMBX 00879
    KPTLWT(NLWTK) = NSPATK                     GEOMBX 00880
C      WRITE NEW MUASIC ARRAY ON SCRATCH        GEOMBX 00881
    IVAL = 10HL WNG-TAIL                      GEOMBX 00882
    N = NROWSL                                  GEOMBX 00883
    PARM(4) = YBARL                            GEOMBX 00884
    PARM(5) = ELL                               GEOMBX 00885
    CALL WRTEMX(IPVSC, MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS, K, ID,
    1           MUASICL, ITYPE, M,N, PARM, IRR)          GEOMBX 00886
    IF (IRR .NE. 0) GO TO 8075                  GEOMBX 00887
    NSCR2 = NSCR2 + 1                          GEOMBX 00888
    IF (CHECKPR) WRITE(NT6,7020) IVAL,JCOL,YBARL,ELL, NROWSL,
    1           (MUASICL(1,NROWSL-I+1),MUASICL(2,NROWSL-I+1), I=1,NROWSL)  GEOMBX 00889
C      REINITIALIZE THE LEFT MUASIC ARRAY       GEOMBX 00890
    DO 987 I = 1,NROWSL                      GEOMBX 00891
    MUASICL(1,I) = I+I                        GEOMBX 00892
    MUASICL(2,I) = 0                          GEOMBX 00893
987 CONTINUE                                 GEOMBX 00894
    GO TO 1000                                GEOMBX 00895
C      OLD MUASIC ARRAY HAS BEEN FOUND WHICH MATCHES  GEOMBX 00896
980 CONTINUE                                 GEOMBX 00897
C      USE THE SAME ARRAY FOR RIGHT AND LEFT CONTRIBUTIONS  GEOMBX 00898
    KPTLWT(NLWTK) = IPARAL                   GEOMBX 00899
    NROWSLL = MAX0(NROWSLL,NROWSL)            GEOMBX 00900
C
995 CONTINUE                                 GEOMBX 00901
    IVAL = 10HL WNG-TAIL                      GEOMBX 00902
    IF (CHECKPR) WRITE(NT6,7020) IVAL,JCOL,YBARL,ELL, NROWSL,
    1           (MUASICL(1,NROWSL-I+1),MUASICL(2,NROWSL-I+1), I=1,NROWSL)  GEOMBX 00903
1000 CONTINUE                                GEOMBX 00904
    IF (ISMLPW .NE. 0) GO TO 1150            GEOMBX 00905
1010 CONTINUE                                GEOMBX 00906
    JCOL = JCOL + 1                          FTN01 00033
    IF (JCOL .LE. MYBBT) GO TO 909          FTN01 00034
C      END OF LOOP ON CHORDS, FOR WING-TAIL PARAMETERS  GEOMBX 00907
C
C      PLACE ANY BUILT UP ARRAY DUE TO PARALLEL SURFACES ON SCRATCH  GEOMBX 00908
    IF (PSIDIF .NE. 0 .OR. PSIW .NE. 0) GO TO 1014  GEOMBX 00909
C      MERGE THE RIGHT AND LEFT ARRAYS          GEOMBX 00910
    NROWSX = MIN0(NROWSR,NROWSL)            GEOMBX 00911
    DO 1012 I = 1,NROWSX                    GEOMBX 00912
    MUASIC(1,I) = MIN0( MUASIC(1,I),MUASICL(1,I) )  GEOMBX 00913
    MUASIC(2,I) = MAX0( MUASIC(2,I),MUASICL(2,I) )  GEOMBX 00914
1012 CONTINUE                                GEOMBX 00915
1014 IF (PSIDIF .NE. 0) GO TO 1015          GEOMBX 00916
C      WRITE THE ARRAY ONTO IWTFSC            GEOMBX 00917
    IVAL = 10HPARAL TAIL                    GEOMBX 00918
    PARM(4) = YBAR                           GEOMBX 00919
    PARM(5) = EL                            GEOMBX 00920
    IPARM(6) = 2                           GEOMBX 00921
    N = MAX0( NROWSR,NROWSL)                GEOMBX 00922
    CALL WRTEMX(IWTFSC, MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS, K, ID,
    1           MUASIC, ITYPE, M,N, PARM, IRR)          GEOMBX 00923

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      IF (IRR .NE. 0) GO TO 8110                                GEOMBX 00930
      NSCR = NSCR + 1                                         GEOMBX 00931
C       PLACE THE KPT-- ARRAYS ON GEOMETRY SCRATCH AS ONE MATRIX.  GEOMBX 00932
1015 CONTINUE                                                 GEOMBX 00933
      IF (NSPATK .EQ. 0) ENDFILE IGEOSC                         GEOMBX 00934
      IF (NSPATK .EQ. 0) GO TO 2000                           GEOMBX 00935
      IPARM(3) = NMAX                                         GEOMBX 00936
      IPARM(4) = NTTK                                         GEOMBX 00937
      IPARM(5) = NRWTK                                         GEOMBX 00938
      IPARM(6) = NLWTK                                         GEOMBX 00939
      M = 0                                                       GEOMBX 00940
      N = 0                                                       GEOMBX 00941
      IF (NMAX .EQ. 0) GO TO 1030                            GEOMBX 00942
      M = 1                                                       GEOMBX 00943
      N = NMAX                                         GEOMBX 00944
      DO 1020 I = 1,NMAX                                     GEOMBX 00945
1020 KPT(I,I) = KPTW(I)                                    GEOMBX 00946
1030 CONTINUE                                                 GEOMBX 00947
      IF (NTTK .EQ. 0) GO TO 1050                            GEOMBX 00948
      M = M + 1                                              GEOMBX 00949
      N = MAX0(N,NTTK)                                       GEOMBX 00950
      DO 1040 I = 1,NTTK                                     GEOMBX 00951
1040 KPT(M,I) = KPTT(I)                                    GEOMBX 00952
1050 CONTINUE                                                 GEOMBX 00953
      IF (NRWTK .EQ. 0) GO TO 1070                            GEOMBX 00954
      M = M + 1                                              GEOMBX 00955
      N = MAX0(N,NRWTK)                                      GEOMBX 00956
      DO 1060 I = 1,NRWTK                                    GEOMBX 00957
1060 KPT(M,I) = KPTRWT(I)                                 GEOMBX 00958
1070 CONTINUE                                                 GEOMBX 00959
      IF (NLWTK .EQ. 0) GO TO 1090                            GEOMBX 00960
      M = M + 1                                              GEOMBX 00961
      N = MAX0(N,NLWTK)                                      GEOMBX 00962
      DO 1080 I = 1,NLWTK                                    GEOMBX 00963
1080 KPT(M,I) = KPTLWT(I)                                 GEOMBX 00964
1090 CONTINUE                                                 GEOMBX 00965
      K = 4                                                       GEOMBX 00966
      IVAL = 3HKPT                                         GEOMBX 00967
      CALL WRTENX(IGEOSC, MXWRIT, RANDOU, NFS, NMS, LS, NMR, LWS, K, ID,
1           KPT, ITYPE, M, N, PARM, IRR)                      GEOMBX 00968
      IF (IRR .NE. 0) GO TO 8070                           GEOMBX 00969
      END FILE IGEOSC                                     GEOMBX 00970
C
C       MOVE THE MUIC ARRAYS TO THE GEOMETRY SCRATCH TAPE
REWIND IWTFSC                                         GEOMBX 00971
REWIND IVPSC                                           GEOMBX 00972
      IVAL = 6HMUIC                                         GEOMBX 00973
      K = 2                                                       GEOMBX 00974
      IF (NSCR .LE. 0) GO TO 1096                            GEOMBX 00975
      DO 1094 I = 1,NSCR                                     GEOMBX 00976
      CALL RDINIT
      CALL READMX(IWTFSC, MXWRIT, RANDOU, NFS, NMS, LS, NMR, K, NID, ID,
1           ITYPE, LRS, MUIC, M, N, PARM, IRR)              GEOMBX 00977
      IF (IRR .NE. 0) GO TO 8100                           GEOMBX 00978
      CALL WRTENX(IGEOSC, MXWRIT, RANDOU, NFS, NMS, LS, NMR, LWS, K, ID,
1           MUIC, ITYPE, M, N, PARM, IRR)                  GEOMBX 00979
      IF (IRR .NE. 0) GO TO 8070                           GEOMBX 00980

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1094 CONTINUE                                         GEOMBX 00987
1096 IF (NSCR2 .LE. 0) GO TO 1110                  GEOMBX 00988
      DO 1100 I = 1,NSCR2                           GEOMBX 00989
      CALL READMX(IPVSC, MXWRIT,RANDOU,NFS,NMS,LS,NMR, K, NID,ID,
      1          ITYPE, LRS, MUAC, M,N, PARM, IRR)
      IF (IRR .NE. 0) GO TO 8090                   GEOMBX 00991
      CALL WRTEHX(IGEOSC, MXWRIT, RANDOU, NFS,NMS, LS,NMR,LWS, K, ID,
      1          MUAC, ITYPE, M,N, PARM, IRR)
      IF (IRR .NE. 0) GO TO 8070                   GEOMBX 00992
1100 CONTINUE                                         GEOMBX 00993
1110 CONTINUE                                         GEOMBX 00994
      END FILE IGEOSC                            GEOMBX 00995
      GO TO 2000                                     GEOMBX 00996
C
C          LOOP ON SAMPLE WASH CHORDS (USED IF ISMPLW .NE. 0) TO    GEOMBX 00997
C          DETERMINE MUAC ARRAYS FOR RIGHT AND LEFT WING INFLUENCE   GEOMBX 00998
C          ON EACH CHORD                                         GEOMBX 00999
1120 CONTINUE                                         GEOMBX 01000
      NRWTK = 0                                         GEOMBX 01001
      NLWTK = 0                                         GEOMBX 01002
      IF (ISMPLW .EQ. 0) GO TO 1015                  GEOMBX 01003
C      ( DO 1200 JCHRD = 1,ISMPLW )
      JCHRD = 1                                         GEOMBX 01004
1130 CONTINUE                                         GEOMBX 01005
      JCOL = ICHORD(JCHRD)                           GEOMBX 01006
      CAPLL = ZLOC(JCHRD)                           GEOMBX 01007
      YMUSP = CAPLL*STIN(PSIW)
      IRW = IBORL(JCHRD)
      GO TO 930                                         FTNXI 00035
C          THE LOGIC FOR A TAIL CHORD IS USED. AFTER THE MUAC ARRAYS
C          ARE DETERMINED AND STORED, CONTROL IS RETURNED TO THIS LOOP.  FTNXI 00036
1150 CONTINUE                                         FTNXI 00037
1200 CONTINU.                                         GEOMBX 01009
      JCHRD = JCHRD + 1                           GEOMBX 01010
      IF (JCHRD .LE. ISMPLW) GO TO 1130           GEOMBX 01011
C          END OF LOOP ON SAMPLE WASH CHORDS        GEOMBX 01012
C
      GO TO 1015                                         GEOMBX 01013
C
2000 CONTINUE                                         GEOMBX 01014
      ENDFILE IGEOSC                            GEOMBX 01015
      REMIND IGEOSC                            GEOMBX 01016
      REMIND IWTFSC                            GEOMBX 01017
      REMIND IPVSC                            FTNXI 00038
      RETURN                                         FTNXI 00039
C
C          OUTPUT FORMATS
C          CARD F
0010 FORMAT(1H0,15X,39H- - - GEOMETRIC PARAMETERS - - - /)
      1 1H0,29HCARDF -LOCAL AXES DEFINITION-,           GEOMBX 01021
      2 4X, 10HX-LOCATION, 4X,10HZ-LOCATION, 4X,           GEOMBX 01022
      2 7DIHEDRAL ANGLE (PSI) / 27X,5H WING , F10.3,4X, F10.3, 8X,
      3 .2, 8H DEGREES )                           GEOMBX 01023
      8012 FG. 4T(27X,5HTAIL , F10.3,4X, F10.3, 8X, F7.2, 8H DEGREES )
C          CARD G
0015 FORMAT(1H0,30HCARDG -BOX PATTERN DEFINITION-,5X,6HNCHRDS,10X,
      1 8HXCENTR,5X,2HCR,4X,5HXEDGE /37X, 13, 8X, F10.4, 6X,F10.4 )
                                         GEOMBX 01024
                                         GEOMBX 01025
                                         GEOMBX 01026
                                         GEOMBX 01027
                                         GEOMBX 01028
                                         GEOMBX 01029
                                         GEOMBX 01030
                                         GEOMBX 01031
                                         GEOMBX 01032
                                         GEOMBX 01033
                                         GEOMBX 01034
                                         GEOMBX 01035
                                         GEOMBX 01036
                                         GEOMBX 01037
                                         GEOMBX 01038
                                         GEOMBX 01039

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0017 FORMAT(7X,13H-SAMPLE WASH,I3, 8H CHORDS-,5X,CHICORD,6X,SHIBOKF, 1 5X,SHIBOKL, 6X, 4H2LOC / (37X, I3,8X, I3, 7X, I3, 6X, F7.2))	GEOBX 01040
C CARD H	GEOBX 01041
0021 FORMAT(1HD,40HCARDH -PLANFORM DEFINITION POINT COUNTS-,5X, 1 12HLEADING EDGE,4X,13HTRAILING EDGE /42X,4HWING, 17, 9X, 17)	GEOBX 01042
0022 FORMAT(42X,4HTAIL, 17,9X,17)	GEOBX 01043
C CARDS I TO L	GEOBX 01044
0029 FORMAT(1HD,37HCARDI TO CARDL -PLANFORM DEFINITIONS-,9X,1HX,9X,1HY, 1 4X,12H(LOCAL AXES))	GEOBX 01045
0030 FORMAT(32X,A10, F9.3, F10.3/ (41X,2F10.3))	GEOBX 01046
0040 FORMAT(1HD,7X,31H-BLK DIMENSIONS- B1 (LENGTH) = , E18.8,5X, 1 17HB1/BETA (WIDTH) = , E18.8)	GEOBX 01047
C	GEOBX 01048
C CHECK PRINT FORMATS, USED ONLY WHEN CHECPR = .T.	GEOBX 01049
7010 FORMAT(15HD)WAKE ARRAY - , 40I3 / (15X,40I3))	GEOBX 01050
7020 FORMAT(17HDMUAIIC ARRAY FOR ,A10, 8H, CHORD I2, 9H, YBAR = F8.3, 1 7H, EL = F7.2 / 10X, 3HROW,I3,2I4 / (16X,2I4))	GEOBX 01051
7030 FORMAT(51HD IJALPH (= J+1000 + I OCTAL) AND ALPHA ARRAYS, AS 1 TH STORED / 6(5X,13HIJALPH ALPHA) / (6(5X,06, F7.4)))	GEOBX 01052
7040 FORMAT(21HD CHECK PRINT, FEXLOC/(10F12.7))	GEOBX 01053
7045 FORMAT(1HD 14X, BHTEALOC / (10F12.7))	GEOBX 01054
C	GEOBX 01055
C DIAGNOSTIC FORMATS	GEOBX 01056
9010 FORMAT(52HD*** WARNING - XEDGE AND XCENTR WERE BOTH SPECIFIED. 1 29H XEDGE WILL BE IGNORED ***)	GEOBX 01057
9020 FORMAT(45HD*** WARNING - SAMPLE WASH SPECIFICATION SET I2,6H IS IN 1 55H ERROR. ONLY THE PRECEDING ONES WILL BE CALCULATED ***)	GEOBX 01058
9030 FORMAT(53HD*** WARNING - SAMPLING OF UPWASHES CANNOT BE DONE IF 1 35H A TAIL HAS BEEN DEFINED. ISMLPW =,I3,1SH WILL BE IGNORED 2 4H ***)	GEOBX 01059
9110 FORMAT(23HD*** ERROR - PARAMETER ,A6,23H WAS NOT SPECIFIED. IT 1 29H MUST ALWAYS BE GIVEN ***)	GEOBX 01060
9120 FORMAT(53HD*** ERROR - EITHER XEDGE OR XCENTR MUST BE SPECIFIED 1 4H ***)	GEOBX 01061
9130 FORMAT(13HD*** ERROR - ,A6,29H IS OUTSIDE ALLOWED RANGE ***)	GEOBX 01062
9150 FORMAT(13HD*** ERROR - ,A10,23HDEFINITION POINTS ERROR, I3, 1 29H, A COMBINATION OF- *** / 13X,24H1, NON-NONOTONIC Y-VALUE 2 1HS, 10X,29H2, NON-NONOTONIC X-VALUES / 13X,11H4, FIRST Y- 3 14HVALUE NON-ZERO, 10X,34H8, TIP T.E. Y-VALUE DISAGREES WITH 4 15H TIP L.E. VALUE)	GEOBX 01063
9160 FORMAT(51HD*** ERROR - XCENTR NOT WITHIN SD BOK LENGTHS (B1 = , 1 E15.8,20H) OF THE WING L.E. (,E15.8, 5H) ***)	GEOBX 01064
9170 FORMAT(52HD*** ERROR - WHILE WRITING ON GEOMETRY SCRATCH FILE A10, 1 15H, ERROR CODE = 14, 4H ***)	GEOBX 01065
9172 FORMAT(14X, BHARRAY ,A6,15H, DIMENSIONED (I4,1H,I4,11H) WAS BEING 1 3H WRITTEN)	GEOBX 01066
9173 FORMAT(14X,20HTHE MUAIIC ARRAY FOR A10,15H, DIMENSIONED (I4,1H,I4, 1 15H) WAS BEING WRITTEN)	GEOBX 01067
9180 FORMAT(52HD*** ERROR - THE TAIL AND WING, OR THEIR DIAPHRAGMS, 1 29H CROSS - ABOVE TO BELOW ***)	GEOBX 01068
9190 FORMAT(54HD*** ERROR - WHILE READING FROM GEOMETRY SCRATCH FILE 1 A10,15H, ERROR CODE = 14, 4H ***)	GEOBX 01069
9192 FORMAT(14X, BHARRAY ,A6,15H, DIMENSIONED (I4,1H,I4,11H) WAS BEING 1 3H READ)	GEOBX 01070
C	GEOBX 01071
C ERRORS - ALL ERRORS CALL FLUSH	GEOBX 01072
9010 CONTINUE	GEOBX 01073
	GEOBX 01074
	GEOBX 01075
	GEOBX 01076
	GEOBX 01077
	GEOBX 01078
	GEOBX 01079
	GEOBX 01080
	GEOBX 01081
	GEOBX 01082
	GEOBX 01083
	GEOBX 01084
	GEOBX 01085
	GEOBX 01086
	GEOBX 01087
	GEOBX 01088
	GEOBX 01089
	GEOBX 01090
	GEOBX 01091
	GEOBX 01092
	GEOBX 01093
	GEOBX 01094
	GEOBX 01095
	GEOBX 01096

WRITE (NT6,9110) IVAL	GEOMBX 01097
GO TO 8900	GEOMBX 01098
8015 CONTINUE	GEOMBX 01099
WRITE (NT6,9130) IVAL	GEOMBX 01100
GO TO 8900	GEOMBX 01101
8020 CONTINUE	GEOMBX 01102
WRITE (NT6,9120)	GEOMBX 01103
GO TO 8900	GEOMBX 01104
8030 CONTINUE	GEOMBX 01105
WRITE (NT6,9130) IVAL	GEOMBX 01106
GO TO 8900	GEOMBX 01107
8040 CONTINUE	GEOMBX 01108
GO TO 8900	GEOMBX 01109
8050 CONTINUE	GEOMBX 01110
WRITE (NT6,9150) IVAL,IRR	GEOMBX 01111
GO TO 8900	GEOMBX 01112
8060 CONTINUE	GEOMBX 01113
WRITE (NT6,9160) B1, XEDGEW	GEOMBX 01114
GO TO 8900	GEOMBX 01115
8070 CONTINUE	GEOMBX 01116
WRITE (NT6,9170) IGEOSC, IRR	GEOMBX 01117
WRITE (NT6,9172) IVAL, M,N	GEOMBX 01118
GO TO 8900	GEOMBX 01119
8075 CONTINUE	GEOMBX 01120
WRITE (NT6,9170) IVPSC,IRR	GEOMBX 01121
WRITE (NT6,9175) IVAL, M,N	GEOMBX 01122
GO TO 8900	GEOMBX 01123
8080 CONTINUE	GEOMBX 01124
WRITE (NT6,9180)	GEOMBX 01125
GO TO 8900	GEOMBX 01126
8090 CONTINUE	GEOMBX 01127
WRITE (NT6,9190) IVPSC,IRR	GEOMBX 01128
WRITE (NT6,9192) IVAL, M,N	GEOMBX 01129
GO TO 8900	GEOMBX 01130
8100 WRITE (NT6,9190) IWTFSC,IRR	GEOMBX 01131
WRITE (NT6,9192) IVAL, M,N	GEOMBX 01132
GO TO 8900	GEOMBX 01133
8110 WRITE (NT6,9170) IWTFSC,IRR	GEOMBX 01134
WRITE (NT6,9175) IVAL, M,N	GEOMBX 01135
8900 CALL FLUSH(1)	GEOMBX 01136
END	GEOMBX 01137

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SUBROUTINE EDGCHK (XEDGE,YEDGE,NEDGE,EDGE,IRR)          EDGCHK 00002
C                                                               EDGCHK 00003
C       CHECKS FOR MONOTONIC EDGE VALUES OF X FOR LEADING EDGES, AND    EDGCHK 00004
C       Y FOR EITHER LEADING OR TRAILING. CHECKS Y FIRST AND LAST      EDGCHK 00005
C       VALUES TO ENSURE DEFINITION FROM CENTERLINE TO TIP            EDGCHK 00006
C       XEDGE = X-VALUES FOR ONE PLATEFORM EDGE                      EDGCHK 00007
C       YEDGE = Y-VALUES                      EDGCHK 00008
C       NEDGE = NUMBER OF (XEDGE,YEDGE) SETS DEFINING THE PLATE-     EDGCHK 00009
C       FORM EDGE                      EDGCHK 00010
C       EDGE = 1 FOR LEADING EDGE, 2 FOR TRAILING EDGE                EDGCHK 00011
C       IRR = ERROR RETURN, 0 SUCCESSFUL
C           1, NON-MONOTONIC Y-VALUES                      EDGCHK 00012
C           2, NON-MONOTONIC X-VALUES, LEADING EDGE ONLY        EDGCHK 00013
C           4, FIRST Y-VALUE NON-ZERO                      EDGCHK 00014
C           8, TIP Y-VALUE OF A T.E. DISAGREES WITH PREVIOUS    EDGCHK 00015
C           L.E. TIP VALUE                      EDGCHK 00016
C           17, TIP Y-VALUE OF A T.E. DISAGREES WITH PREVIOUS    EDGCHK 00017
C           L.E. TIP VALUE                      EDGCHK 00018
C
C       DIMENSION XEDGE(1),YEDGE(1)          EDGCHK 00019
C
C       IRR = 0
C       IF (YEDGE(1) .NE. 0.) IRR = 4          EDGCHK 00020
C       DO 100 I = 2,NEDGE
C           IF (YEDGE(I) .LT. YEDGE(I-1) ) GO TO 150          EDGCHK 00021
C           GO TO (30,100), EDGE
C           50 IF (XEDGE(I) .LT. XEDGE(I-1) ) GO TO 200          EDGCHK 00022
C       100 CONTINUE
C           GO TO 250
C       150 IRR = IRR + 1
C           GO TO 250
C       200 IRR = IRR + 2
C       250 CONTINUE
C           GO TO (300,350), EDGE
C       300 YTIP = YEDGE(NEDGE)
C           GO TO 350
C       350 IF (YEDGE(NEDGE) .NE. YTIP) IRR = IRR + 8          EDGCHK 00030
C
C       500 RETURN
C
C       END          EDGCHK 00031
C
C

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SUBROUTINE EDGCHK(XEDGE,YEDGE,NEDGE,IEDGE,IRR)          EDGCHK 00002
C                                                               EDGCHK 00003
C       CHECKS FOR MONOTONIC EDGE VALUES OF X FOR LEADING EDGES, AND    EDGCHK 00004
C       Y FOR EITHER LEADING OR TRAILING. CHECKS Y FIRST AND LAST      EDGCHK 00005
C       VALUES TO ENSURE DEFINITION FROM CENTERLINE TO TIP            EDGCHK 00006
C       XEDGE = X-VALUES FOR ONE PLANFORM EDGE                      EDGCHK 00007
C       YEDGE = Y-VALUES                      EDGCHK 00008
C       NEDGE = NUMBER OF (XEDGE,YEDGE) SETS DEFINING THE PLAN-     EDGCHK 00009
C       FORM EDGE                      EDGCHK 00010
C       IEDGE = 1 FOR LEADING EDGE, 2 FOR TRAILING EDGE             EDGCHK 00011
C       IRR = ERROR RETURN, 0 SUCCESSFUL
C           1, NON-MONOTONIC Y-VALUES                         EDGCHK 00012
C           2, NON-MONOTONIC X-VALUES, LEADING EDGE ONLY        EDGCHK 00013
C           4, FIRST Y-VALUE NON-ZERO                          EDGCHK 00014
C           8, TIP Y-VALUE OF A T.E. DISAGREES WITH PREVIOUS    EDGCHK 00015
C           L.E. TIP VALUE                                     EDGCHK 00016
C
C       DIMENSION XEDGE(1),YEDGE(1)                                EDGCHK 00017
C
C       IRR = 0
C       IF (YEDGE(1) .NE. 0. ) IRR = 4
C       DO 100 I = 2,NEDGE
C       IF (YEDGE(I) .LT. YEDGE(I-1) ) GO TO 150
C       GO TO (50,100), IEDGE
C       50 IF (XEDGE(I) .LT. XEDGE(I-1) ) GO TO 200
C       100 CONTINUE
C       GO TO 250
C       150 IRR = IRR + 1
C       GO TO 250
C       200 IRR = IRR + 2
C       250 CONTINUE
C       GO TO (300,350),IEDGE
C       300 YTIP = YEDGE(NEDGE)
C       GO TO 500
C       350 IF (YEDGE(NEDGE) .NE. YTIP) IRR = IRR + 8
C
C       500 RETURN
C
C       DD

```

IB = IB + 1	DCODER 00055
1000 CONTINUE	DCODER 00056
GO TO 3000	DCODER 00057
C	DCODER 00058
PROGRAM WILL RETRIEVE NJ BOXES FROM CHORD J	DCODER 00059
1100 CONTINUE	DCODER 00060
JB8 = (J-1)/NBWRD + 1	DCODER 00061
JB = (NBWRD - MOD(J,NBWRD)) * 3	DCODER 00062
IF (JB .EQ. 80) JB = 0	DCODER 00063
IJMASK = SHIFT(MASK,JB)	DCODER 00064
NJB = -JB	DCODER 00065
DO 2000 II = I,IEND,ISKIP	DCODER 00066
IJWORD = IBOK(II,JB)	DCODER 00067
IJCODE = IJWORD.AND.IJMASK	DCODER 00068
ICODE(IB) = SHIFT(IJCODE,NJB)	DCODER 00069
IB = IB + 1	DCODER 00070
2000 CONTINUE	DCODER 00071
C	DCODER 00072
3000 CONTINUE	DCODER 00073
RETURN	DCODER 00074
END	DCODER 00075

```

SUBROUTINE NCODER(IBOX,LBOX, IA,JA, IL, ICODE )
DIMENSION IBOX(LBOX,1)                               NCODER 00002
C                                                 NCODER 00003
C ENCODES ONE INTEGER BOX CODE INTO THE PACKED CODE ARRAY ALONG A NCODER 00004
C PORTION OF A CHORD, REPLACING PREVIOUS VALUES NCODER 00005
C                                                 NCODER 00006
C                                                 NCODER 00007
C                                                 NCODER 00008
C IBOX - ARRAY OF BOX CODES, PACKED 20 PER WORD NCODER 00009
C LBOX - ROW DIMENSION OF BOX CODES ARRAY NCODER 00010
C IA - I-TH INDEX OF FIRST BOX CODE TO SET NCODER 00011
C JA - J-TH INDEX OF FIRST BOX CODE TO SET NCODER 00012
C IL - I-TH INDEX OF THE LAST BOX CODE TO SET NCODER 00013
C ICODE - CODE VALUE, 1,2, OR 3, FOR THE NCODER 00014
C
C INTEGER SHIFT
DATA MASK / ??????????????????DB /
DATA NBWRD/20/
JSB = (JA-1)/NBWRD + 1
JB = (NBWRD - MOD(JA,NBWRD) ) * 3
IF (JB .EQ. 60) JB = 0
C     JB = A LEFT SHIFT COUNT
ICOD = SHIFT(ICODE,JB)
IJMASK = SHIFT(MASK,JB)
DO 100 II = IA,IL
IJCODE = IJMASK .AND. IBOX(II,JSB)
100 IBOX(II,JSB) = IJCODE .OR. ICOD
RETURN
END

```

```

SUBROUTINE PRNTBC(IBOX,LBXCD, IFRST,ILAST, JLAST, SUBD)          PRNTBC 00002
DIMENSION IBOX(LBXCD,1),ICCCE(150)                                PRNTBC 00003
LOGICAL SUBD                                         PRNTBC 00004
C                                                 PRNTBC 00005
C         PRINTS BOX CODES, SUBDIVIDED OR UNSUBDIVIDED      PRNTBC 00006
C         IBOX - COMPRESSED BOX CODE ARRAY                 PRNTBC 00007
C         LBXCD - ROW DIMENSION OF BOX CODE ARRAY        PRNTBC 00008
C         IFRST - FIRST ROW DESIRED TO PRINT             PRNTBC 00009
C         ILAST - LAST ROW DESIRED                      PRNTBC 00010
C         JLAST - LAST CHORD DESIRED (FIRST IS ALWAYS ONE) PRNTBC 00011
C         SUBD - .T., SUBDIVIDED CODES DESIRED           PRNTBC 00012
C                     .F., UNSUBDIVIDED CODES DESIRED       PRNTBC 00013
C                                                 PRNTBC 00014
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,      GEOMTY 00002
1               B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,            GEOMTY 00003
2               MXBW,MXBBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW,        GEOMTY 00004
3               IXBW,XCENTR                                     GEOMTY 00005
LOGICAL COPLAN                                         GEOMTY 00006
COMMON /CONTRL/ PREVEX,XMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT, CONTRL 00002
1               DEFAULT                                      CONTRL 00003
LOGICAL PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT                CONTRL 00004
COMMON /PROBLM/ XMACH,NMOCES,NTSLOP,NKVALS,SMOOTH,NDEG,CRDFIT, PROBLM 00002
1               EXAIC,SUBDV,PLYWOOD                         PROBLM 00003
LOGICAL SMOOTH,CRDFIT,EXAIC,SUBDV,PLYWOOD                PROBLM 00004
COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP, FILES 00002
1               ICUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC      FILES 00003
DIMENSION BCD(4)                                         PRNTBC 00020
INTEGER BCD                                         PRNTBC 00021
DATA BCD/ 1H ,1H1,1H2,1H3 /                          PRNTBC 00022
DATA NOWRD/ 20 /                                     FTN01 00040
6001 FORMAT(1H1, 5X,8A10/1H0,20X,1SHBOX CODE PATTERN )      PRNTBC 00023
6002 FORMAT(19X,2DHFOR SUBDIVIDED BOXES,25X, 6HIXBW =,I2,11H (SUBDIVID PRNTBC 00024
1 30ED ROW OF UNSUBDIVIDED CENTER) )                  PRNTBC 00025
6005 FORMAT(22X,4HMACH,F11.7,56X,*CODE - 1 = PLANFORM BOX* /      PRNTBC 00026
1 19X,20(1H-),61X,*2 = DIAPHRAGM BOX* /100X,*3 = WAKE BOX * )    PRNTBC 00027
6010 FORMAT(1H0,4X,3I4 / ( 9X,3I4 ) )                  PRNTBC 00028
6012 FORMAT(1H )                                       PRNTBC 00029
6020 FORMAT(1X,I3,2X,63A2 / (12X,60A2) )              PRNTBC 00030
C                                                 PRNTBC 00031
        WRITE (NT6,6001) TITLE                           PRNTBC 00032
        IF (NSUBDV .EQ. 1) GO TO 100                   PRNTBC 00033
        IF (.NOT.SUBD) GO TO 100                      PRNTBC 00034
        WRITE (NT6,6002) IXBW                         PRNTBC 00035
100 CONTINUE                                         PRNTBC 00036
        WRITE (NT6,6005) XMACH                        PRNTBC 00037
C                                                 PRNTBC 00038
        WRITE (NT6,6010) (I, I=2,ILAST,2)             PRNTBC 00039
        WRITE (NT6,6012)                               PRNTBC 00040
        DO 250 IROW = IFRST,ILAST                   PRNTBC 00041
        CALL DCODE(IBOX,LBXCD, IROW,1, IROW,ILAST, SUBD, ICODE )     PRNTBC 00042
C                                                 PRNTBC 00043
C         CHANGE INTEGER CODES TO ALPHANUMERIC      PRNTBC 00044
C                                                 PRNTBC 00045
        DO 200 J= 1,ILAST                         PRNTBC 00046
        IF(ICODE(J).EQ.0) ICODE(J) = BCD(1)          PRNTBC 00047
        IF(ICODE(J).EQ.1) ICODE(J) = BCD(2)          PRNTBC 00048
        IF(ICODE(J).EQ.2) ICODE(J) = BCD(3)          PRNTBC 00049

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IF(ICODE(J).EQ.3) ICODE(J)=BCD(4)
200 CONTINUE
      WRITE (NTB,6020) IRW, (ICODE(J), J = 1,JLAST )
290 CONTINUE
      RETURN
END
```

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PRNTBC 00050
PRNTBC 00051
PRNTBC 00052
PRNTBC 00053
PRNTBC 00054
PRNTBC 00055
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SUBROUTINE BXCDPF(XLE,YLE,NLE,XTE,YTE,NTE, LSRQBS,IBOK ) BXCDPF 00002
C BXCDPF 00003
C GENERATES THE BOX CODES FOR THE ON-PLANFORM BOXES OF ONE BXCDPF 00004
C SURFACE. BXCDPF 00005
C INPUT PARAMETERS BXCDPF 00006
C XLE = X-VALUES, LEADING EDGE, NON-DIMENSIONAL BXCDPF 00007
C YLE = Y-VALUES, LEADING EDGE BXCDPF 00008
C NLE = NUMBER OF LEADING EDGE POINTS BXCDPF 00009
C XTE = X-VALUES, TRAILING EDGE BXCDPF 00010
C YTE = Y-VALUES, TRAILING EDGE BXCDPF 00011
C NTE = NUMBER OF TRAILING EDGE POINTS BXCDPF 00012
C LSRQBS = MAXIMUM NUMBER OF SUBDIVIDED ROWS ALLOWED BXCDPF 00013
C BXCDPF 00014
C OUTPUT PARAMETERS BXCDPF 00015
C BXCDPF 00016
C IBOK = COMPRESSED BOX CODES, SET 1 FOR PLANFORM BOXES, BXCDPF 00017
C UNCHANGED ELSEWHERE BXCDPF 00018
C BXCDPF 00019
C COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLATC,NSPAIC,NOUTP, FILES 00002
C IOLFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC FILES 00003
C COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF, GEOMTY 00002
C 1 B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW, GEOMTY 00003
C 2 MXBW,MXBBW,MYBW,MYBBW,MXBW,MYBWSW,MYBSW,MYBBSW, GEOMTY 00004
C 3 IXBW,XCENTR GEOMTY 00005
C LOGICAL COPLAN GEOMTY 00006
C COMMON /GEOM2 / TLAX,TLAZ,PSIT,MXB1,MYBT,MYBBT,MXB1ST,MYB1ST, GEOM2 00002
C 1 MYBB1ST,IXBT,IXB1ST,CAPL GEOM2 00003
C COMMON /EDGES / FEXLOC(250), TEXLOC(250),JDIAG EDGES 00002
C BXCDPF 00024
C LOGICAL WING BXCDPF 00025
C DIMENSION XLE(1),YLE(1),XTE(1),YTE(1) BXCDPF 00026
C DIMENSION IBOK(LSRQBS,1) BXCDPF 00027
C BXCDPF 00028
C INPUT COMMON PARAMETERS - BXCDPF 00029
C IXBT = SUBDIVIDED ROW OF FIRST UNSUBDIVIDED BOX ON TAIL BXCDPF 00030
C NSUBDV = (INTEGER) NUMBER OF SUBDIVISIONS BXCDPF 00031
C XSUBDV (REAL) BXCDPF 00032
C NSUBD2 = NSUBDV/2 BXCDPF 00033
C NSUBCN = NSUBDV/2 +1 , CENTER SUBDIVIDED BOX BXCDPF 00034
C BXCDPF 00035
C IN/OUT COMMON PARAMETERS - BXCDPF 00036
C IXBW = 0, WING BEING DONE, CHANGED TO SUBDIVIDED ROW BXCDPF 00037
C OF FIRST UNSUBDIVIDED BOX CENTER ON WING BXCDPF 00038
C .NE. 0, TAIL BEING DONE, NOT CHANGED BXCDPF 00039
C BXCDPF 00040
C BXCDPF 00041
C OUTPUT COMMON PARAMETERS - BXCDPF 00042
C MXBSW = NUMBER OF SUBDIVIDED ROWS TO AFT END OF (WING) BXCDPF 00043
C (TAIL) BXCDPF 00044
C MYBSW = NUMBER OF SUBDIVIDED CHORDS ON THE (WING) BXCDPF 00045
C (TAIL) BXCDPF 00046
C MXBT = NUMBER OF UNSUBDIVIDED ROWS ON (WING) BXCDPF 00047
C (TAIL) BXCDPF 00048
C MYBT = NUMBER OF UNSUBDIVIDED CHORDS ON THE (WING) BXCDPF 00049
C (TAIL) BXCDPF 00050
C FEALOC = ARRAY OF (LEADING) EDGE X-LOCATIONS, SUBDIVIDED BXCDPF 00051
C (TRAILING) BXCDPF 00052

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MYBS = 0                                BXCDPF 00053
DEL = 1.0/XSUBDV                         BXCDPF 00054
C      LOCATION OF FIRST CHORD           BXCDPF 00055
YMIN = .5*(1.0 + DEL)                     BXCDPF 00056
C      SLOPES OF FIRST LEADING AND TRAILING EDGE SEGMENTS
XREFLE = XLE(1)                           BXCDPF 00057
YREFLE = YLE(1)                           BXCDPF 00058
XREFTE = XTE(1)                           BXCDPF 00059
YREFTE =YTE(1)                            BXCDPF 00060
DLE   = (XLE(2)-XREFLE) / (YLE(2)-YREFLE)
DELTETE = (XTE(2)-XREFTE) / (YTE(2)-YREFTE)
ILE = 2                                   BXCDPF 00061
ITE = 2                                   BXCDPF 00062
IERR = 0                                   BXCDPF 00063
BXCDPF 00064
BXCDPF 00065
BXCDPF 00066
BXCDPF 00067
BXCDPF 00068
BXCDPF 00069
BXCDPF 00070
BXCDPF 00071
BXCDPF 00072
BXCDPF 00073
BXCDPF 00074
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BXCDPF 00096
BXCDPF 00097
BXCDPF 00098
BXCDPF 00099
BXCDPF 00100
BXCDPF 00101
BXCDPF 00102
BXCDPF 00103
BXCDPF 00104
BXCDPF 00105
BXCDPF 00106
BXCDPF 00107
BXCDPF 00108
BXCDPF 00109

C      WAS THIS CALL FOR WING OR TAIL -
IF (IXBW.EQ. 0) GO TO 120
C      SET UP COUNTERS FOR TAIL
SURF = 4HTAIL
WING = .F.
MYBT = IFIX(YLE(NLE))
NSCHRD = MYBT * NSUBDV
JELLOC = MYBW * NSUBDV + 1
IXB = IXBST
LSRR = LSRCWS
IF (.NOT. COPLAN) LSRR = LSRR + IXBST - 1
GO TO 130

C      SET UP COUNTERS FOR WING
120 CONTINUE
SURF = 4HWING
WING = .T.
NSCHRD = MYBW * NSUBDV
JELLOC = 1
XMIN = XREFLE + (YMIN-YREFLE)*DLE
IXBW = (1.-XMIN)*XSUBDV + 1
IXB = 1
LSRR = LSRCWS

C      START LOOP ON SUBDIVIDED CHORDS
130 CONTINUE
YCHORD = YMIN
DO 350 JCHRD = 1,NSCHRD
C      FIND LEADING EDGE OF THIS CHORD
140 CONTINUE
C      IS THE CURRENT L.E. SEGMENT STILL GOOD -
IF (YCHORD - YLE(ILE)) 180,170,150
C      NO, ANOTHER SEGMENT IS NEEDED
150 CONTINUE
YREFLE = YLE(ILE)
ILE = ILE + 1
C      CHECK FOR EXCEEDING LIMIT
IF (ILE.GT. NLE) GO TO 710
C      CHECK FOR EDGE SEGMENT PARALLEL TO (SKIP THE SEGMENT) OR
C      CUTTING BACK TOWARD CENTER-LINE (ERROR)
IF (YREFLE - YLE(ILE)) 160,150,730
C      SEGMENT HAS POSITIVE SLOPE

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160 XREFLE = XLE(ILE-1)
    DELTE = (XLE(ILE)-XREFLE) / (YLE(ILE) - YREFLE)
    GO TO 140
C
C      CHORD CENTER LIES ON SEGMENT ENDPOINT
170 CONTINUE
    FELOC(JEXLOC) = (XLE(ILE)-1.0)*XSUBDV + IXBW
    GO TO 190
C
C      CHORD CENTER LIES WITHIN THE SEGMENT
180 CONTINUE
    FELOC(JEXLOC) = ( XREFLE + DELTE*(YCHORD-YREFLE) - 1.0 )*XSUBDV
    1     + IXBW
C
190 CONTINUE
    FELOC(JEXLOC) = IFIX(FELOC(JEXLOC)) + .5
    ISTART = FELOC(JEXLOC) + 1
C
C      THE FOLLOWING CODE FINDS THE TRAILING EDGE OF THIS CHORD IN
C      THE SAME MANNER AS ABOVE.
240 CONTINUE
    IF (YCHORD - YTE(ITE)) 280,270,250
250 CONTINUE
    YREFTE = YTE(ITE)
    ITE = ITE + 1
    IF (ITE .GT. NTE) GO TO 720
    IF (YREFTE - YTE(ITE)) 260,250,740
260 XREFTE = XTE(ITE-1)
    DELTE = (XTE(ITE)-XREFTE) / (YTE(ITE)-YREFTE)
    GO TO 240
270 CONTINUE
    TELOC(JEXLOC) = (XTE(ITE)-1.0)*XSUBDV + IXBW
    GO TO 290
280 CONTINUE
    TELOC(JEXLOC) = ( XREFTE + DELTE*(YCHORD-YREFTE) - 1.0 )*XSUBDV
    1     + IXBW
290 CONTINUE
    TELOC(JEXLOC) = IFIX(TELOC(JEXLOC)) + .5
    IEND = TELOC(JEXLOC)
    IF (IEND .GT. LSRR) GO TO 770
C
C      SET BOX CODES TO 1 FOR PLANFORM BOXES OF THIS CHORD
300 CONTINUE
    CALL NCODER(IBOX,LSROWS, ISTART, JCHRD, IEND, 1 )
C
    MXBS = MAX0(MXBS,IEND)
    JEXLOC = JEXLOC+1
    YCHORD = YCHORD + DEL
350 CONTINUE
C      END OF LOOP ON CHORDS
C
    IF (WING) GO TO 380
    MXBST = MXBS
    MBST = (MXBS-IXBW)/XSUBDV + 1
    MYBST = NSCHRD
    GO TO 370
360 MX3SW = MXBS

```

BXCDPF 00110
 BXCDPF 00111
 BXCDPF 00112
 BXCDPF 00113
 BXCDPF 00114
 BXCDPF 00115
 BXCDPF 00116
 BXCDPF 00117
 BXCDPF 00118
 BXCDPF 00119
 BXCDPF 00120
 BXCDPF 00121
 BXCDPF 00122
 BXCDPF 00123
 BXCDPF 00124
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 BXCDPF 00162
 BXCDPF 00163
 BXCDPF 00164
 BXCDPF 00165
 BXCDPF 00166

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MXBW = (MXBS-IXBW)/NSUBDV + 1          BXCDPF 00167
MYBSW = NSCHRD                         BXCDPF 00168
370 CONTINUE                            BXCDPF 00169
    IF (IERR .NE. 0) GO TO 750           BXCDPF 00170
C
C      RETURN
C
C      ERROR DIAGNOSTICS
C
C      LIMIT EXCEEDED
710 IERR = 1                            BXCDPF 00171
    EDGE = 8H LEADING                   BXCDPF 00172
    ISEC = ILE - 1                     BXCDPF 00173
    GO TO 750                           BXCDPF 00174
720 IERR = 1                            BXCDPF 00175
    EDGE = 8HTRAILING                  BXCDPF 00176
    ISEC = ITE - 1                     BXCDPF 00177
    GO TO 750                           BXCDPF 00178
730 IERR = 2                            BXCDPF 00179
    EDGE = 8H LEADING                   BXCDPF 00180
    ISEC = ILE - 1                     BXCDPF 00181
    GO TO 750                           BXCDPF 00182
740 IERR = 2                            BXCDPF 00183
    EDGE = 8HTRAILING                  BXCDPF 00184
    ISEC = ITE - 1                     BXCDPF 00185
C      BAD EDGE DEFINITION
750 IERR = 2                            BXCDPF 00186
    EDGE = 8H LEADING                   BXCDPF 00187
    ISEC = ILE - 1                     BXCDPF 00188
    GO TO 750                           BXCDPF 00189
755 WRITE(NT6,750) ISEC                BXCDPF 00190
7550 FORMAT(5X, 8HSECTION,I2,24H IS BEYOND THOSE DEFINED )
    GO TO 800                           BXCDPF 00191
760 WRITE(NT6,760) ISEC                BXCDPF 00192
7600 FORMAT(5X, 8HSECTION,I2,36H OF THE EDGE DOUBLES BACK TOWARD THE
    1 12H CENTER LINE )                 BXCDPF 00193
    GO TO 800                           BXCDPF 00194
C      PLANFORM EXCEEDS BOX PATTERN LIMIT
770 CONTINUE                            BXCDPF 00195
    IERR = 3                            BXCDPF 00196
    EDGE = 8HTRAILING                  BXCDPF 00197
    ISEC = ITE - 1                     BXCDPF 00198
    WRITE (NT6,770) ISEC,SURF,JOHRD,IZEND
7700 FORMAT(20H *** ERROR - SECTION,13,29H OF THE TRAILING EDGE OF THE
    1  A4,14H CAUSES CHORD 13,14H TO GO TO ROW 13,15H, WHICH EXCEEDS
    2  14H THE LIMIT *** )
    IEND = LSROWS                      BXCDPF 00199
C      GO BACK TO FINISH THE SURFACE, THEN PRINT PLANFORM AND FLUSH
    GO TO 300                           BXCDPF 00200
C
    800 CA     PRNTBC(1BOX,LSROWS, IXB,MXB, NSCHRD, .T.)
C
    8000 CALL "LUSH(1)
C
    END

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SUBROUTINE BXCDI(IWAKE,LSROWS,LSCHDS,IBOX)
DIMENSION IWAKE(1),IBOX(LSROWS,1)

C      DETERMINES BOX CODES FOR DIAPHRAGM REGIONS
C
C      IWAKE - ARRAY OF WAKE LIMITS, AS DICTATED BY A TAIL SURFACE
C      LSROWS - ROW DIMENSION OF THE BOX CODE ARRAY
C      LSCHDS - MAXIMUM NUMBER OF BOX CODES ALLOWED PER ROW
C      IBOX - BOX CODE ARRAY, COMPRESSED TO 20 CODES PER WORD

C
COMMON /FILES/ NT5,NT6,INTAPE,INFSP,NPLAIC,NPLAIC,NOUTP,
1           IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,
1           B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,
2           MXBW,MXBWPW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW,
3           IXBW,XCENTR
LOGICAL COPLAN
COMMON /GEOM2/ TLAX,TLAZ,PS1,MXBST,MYBT,MYBBT,MXBEST,MYBEST,
1           MYBBST,IXB1,IXBST,CAPL
COMMON /SAMPLW/ ISMPLW,ICHORD(10),IBOXF(10),IBOXL(10),ZLCC(10)
C      COMMON INPUT VALUES-
C           MXBBSW, MXBBST, IXBST, MYBSW, MYBST, MYBBSW, MYBBST, NSUBDV
C           ISMFLW
C      COMMON OUTPUT VALUES-
C           MYBBSW, MYBBST, MYBBW, MYBBT
DIMENSION ICODE(160)
LOGICAL WING

C
C      DETERMINE WHETHER THIS IS A WING OR TAIL
IF (IWAKE(1) .EQ. 0) GO TO 80
WING = .T.
IXBS = 1
IXBS1 = 2
MXBBS = MXBBW+NSUBDV + IXBW - NSUBCN
IF (COPLAN) MXBBS = MXBST
JEDLOC = 1
MYBBS = MYBBW
MYBS = MYBSW
GO TO 100

C      THIS IS A TAIL SURFACE
80 WING = .F.
IXBS = IXBST
IXBS1 = IXBS + 1
MXBBS = MXBST
JEDLOC = MYBSW + 1
MYBBS = MYBBST
MYBS = MYBST
100 CONTINUE
MXBBS1 = MXBBS-1

C
C      DETERMINE LEADING EDGE DIAPHRAGM
DO 130 J = 2,MYBS
CALL DCODE(IBOX,LSROWS, IXBS,J-1, IXBS1,J-1, .T., ICODE(2) )
DO 120 I = IXBS1,MXBBS1
CALL DCODE(IBOX,LSROWS, I,J, I,J, .T., ICOD1 )
ICODE(1) = ICODE(2)
ICODE(2) = ICODE(3)

BXCDI 00002
BXCDI 00003
BXCDI 00004
BXCDI 00005
BXCDI 00006
BXCDI 00007
BXCDI 00008
BXCDI 00009
BXCDI 00010
BXCDI 00011
FILES 00002
FILES 00003
GEOMTY 00002
GEOMTY 00003
GEOMTY 00004
GEOMTY 00005
GEOMTY 00006
GEOM2 00002
GEOM2 00003
SAMPLW 00002
BXCDI 00015
BXCDI 00016
BXCDI 00017
BXCDI 00018
BXCDI 00019
BXCDI 00020
BXCDI 00021
BXCDI 00022
BXCDI 00023
BXCDI 00024
BXCDI 00025
BXCDI 00026
BXCDI 00027
BXCDI 00028
BXCDI 00029
BXCDI 00030
BXCDI 00031
BXCDI 00032
BXCDI 00033
BXCDI 00034
BXCDI 00035
BXCDI 00036
BXCDI 00037
BXCDI 00038
BXCDI 00039
BXCDI 00040
BXCDI 00041
BXCDI 00042
BXCDI 00043
BXCDI 00044
BXCDI 00045
BXCDI 00046
BCSGEB 00001
BXCDI 00048
BXCDI 00049
BXCDI 00050
BXCDI 00051

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CALL DCODER(1BOX,LSROWS, I+1,J-1, I+1,J-1, .T., ICODE(3))	BXCDI 00052
IF (ICOD .NE. 0) GO TO 120	BXCDI 00053
IF (ICODE(1) .EQ. 0 .OR. ICODE(3) .EQ. 0) GO TO 120	BXCDI 00054
CALL NCODER(1BOX,LSROWS, I,J, I, 2)	BXCDI 00055
120 CONTINUE	BXCDI 00056
130 CONTINUE	BXCDI 00057
C END OF DOUBLE LOOP TO DETERMINE LEADING EDGE DIAPHRAGM AREAS	BXCDI 00058
C	BXCDI 00059
C DETERMINE TRAILING EDGE (WAKE) DIAPHRAGM	BXCDI 00060
IWK = 0	BXCDI 00061
DO 180 J = 1,MYBS	BXCDI 00062
IF (WING) IWK = IWAKE(J)	BXCDI 00063
CALL DCODER(1BOX,LSROWS, IXBS,J, IXBS,J, .T., ICOD)	BXCDI 00064
DO 170 I = IXBS1,MXBBS	BXCDI 00065
ICODMI = ICOD	BXCDI 00066
CALL DCODER(1BOX,LSROWS, I,J, I,J, .T., ICOD)	BXCDI 00067
IF (ICOD .EQ. 1) GO TO 170	BXCDI 00068
IF (ICODMI .NE. 1 .AND. ICODMI .NE. 3) GO TO 170	BXCDI 00069
IF (I .LE. IWK) GO TO 180	BXCDI 00070
IF (I .GE. MXBBS) GO TO 180	BXCDI 00071
C THE BOX IS A CANDIDATE. SEARCH DIAGONALLY FOR POSSIBLE	BXCDI 00072
C RECEIVING BOXES DOWNSTREAM.	BXCDI 00073
JP = J	BXCDI 00074
JM = J	BXCDI 00075
IS = I+1	BXCDI 00076
DO 190 II = IS,MXBBS	BXCDI 00077
IF (JM .GT. 1) JM = JM - 1	BXCDI 00078
CALL DCODER(1BOX,LSROWS, II,JM, II,JM, .T., ICOD)	BXCDI 00079
IF (ICOD .NE. 0) GO TO 190	BXCDI 00080
IF (JP .GE. MYBBS) GO TO 190	BXCDI 00081
JP = JP + 1	BXCDI 00082
CALL DCODER(1BOX,LSROWS, II,JP, II,JP, .T., ICOD)	BXCDI 00083
IF (ICOD .NE. 0) GO TO 190	BXCDI 00084
IF (.NOT. WING) GO TO 190	BXCDI 00085
IF (II .LE. IWAKE(JM)) GO TO 190	BXCDI 00086
IF (II .LE. IWAKE(JP)) GO TO 190	BXCDI 00087
190 CONTINUE	BXCDI 00088
C END OF LOOP ON DIAGONAL SEARCH	BXCDI 00089
GO TO 170	BXCDI 00090
C	BXCDI 00091
C CONDITIONS HAVE BEEN FOUND FOR A VALID WAKE BOX	BXCDI 00092
180 CONTINUE	BXCDI 00093
CALL NCODER(1BOX,LSROWS, I,J, I, 3)	BXCDI 00094
ICOD = 3	BXCDI 00095
170 CONTINUE	BXCDI 00096
C END OF LOOP ON ROWS, AND	BXCDI 00097
180 CONTINUE	BXCDI 00098
C END OF LOOP ON CHORDS, FOR WAKE DIAPHRAGM, FROM 130*	BXCDI 00099
C	BXCDI 00100
C DETERMINE THE TIP DIAPHRAGM REGION	BXCDI 00101
LBB = 2	BXCDI 00102
DO 300 I = IXBS1,MXBBS1	BXCDI 00103
C SEARCH FOR LAST NON-ZERO BOX CODE ON THE ROW, FROM LBB OUTWARD	BXCDI 00104
CALL DCODER(1BOX,LSROWS, 1,LBB, 1,MYBBS, .T., ICODE(LBB))	BXCDI 00105
200 LBB = LBB+1	BXCDI 00106
DO 2 J J = LBBP1,MYBBS	BXCDI 00107
IF (ICODE(J) .EQ. 0) GO TO 220	BXCDI 00108

210	CONTINUE	BXCDI	00109
	LBB = MYBBS	BXCDI	00110
	GO TO 225	BXCDI	00111
220	LBB = J - 1	BXCDI	00112
C	LBB = THE SUBSCRIPT OF THE LAST NON-ZERO BOX ON THE ROW	BXCDI	00113
C	TEST BOX DIRECTLY AHEAD FOR NON-ZERO VALUE	BXCDI	00114
225	CONTINUE	BXCDI	00115
	CALL DCODER(1BOX,LSROWS, I-1,LBB, I-1,LBB, .T., ICOD)	BXCDI	00116
	IF (ICOD .EQ. 0) GO TO 280	BXCDI	00117
C	SEARCH FOR A NON-ZERO CODE ON THE INBOARD DIAGONAL AFT	BXCDI	00118
	J = LBB	BXCDI	00119
	IP1 = I + 1	BXCDI	00120
	IF (IP1 .GT. MXBBS) GO TO 310	BXCDI	00121
	DO 230 II = IP1,MXBBS	BXCDI	00122
	CALL DCODER(1BOX,LSROWS, II,J, II,J, .T., ICOD)	BXCDI	00123
	IF (ICOD .NE. 0) GO TO 255	BXCDI	00124
	J = J - 1	BXCDI	00125
230	CONTINUE	BXCDI	00126
C	NO DIAGONAL BOX WAS FOUND	BXCDI	00127
C	IF (INSURF .EQ. 1 .AND. ISMPLW .EQ. 0) GO TO 310	BXCDI	00128
C	IF (.NOT. WING) GO TO 310	BXCDI	00129
C	SEARCH BACK ALONG THE DIAGONAL FOR A DIAPHRAGM REGION	BXCDI	00130
C	CAUSED BY A TAIL SURFACE	BXCDI	00131
	II = MXBBS + 1	BXCDI	00132
	DO 240 III = IP1,MXBBS	BXCDI	00133
	II = II - 1	BXCDI	00134
	J = J + 1	BXCDI	00135
	IF (IWAKE(J) .GE. II) GO TO 250	BXCDI	00136
240	CONTINUE	BXCDI	00137
C	NO DIAPHRAGM FOUND	BXCDI	00138
	GO TO 310	BXCDI	00139
C	CONDITION FOUND REQUIRING DIAPHRAGM BOXES ON THE DIAGONAL.	BXCDI	00140
250	JJ = J	BXCDI	00141
	GO TO 260	BXCDI	00142
255	JJ = J + 1	BXCDI	00143
	II = II - 1	BXCDI	00144
260	CONTINUE	BXCDI	00145
C	TEST FOR EXCEEDING BOX CODE ARRAY	BXCDI	00146
	LBB = LBB + 1	BXCDI	00147
	IF (LBB .GT. LSCHDS) GO TO 8500	BXCDI	00148
C	SET DIAGONAL ELEMENTS	BXCDI	00149
	DO 270 J = JJ,LBB	BXCDI	00150
	CALL MCODER(1BOX,LSROWS, II,J, II, 2)	BXCDI	00151
	II = II - 1	BXCDI	00152
270	CONTINUE	BXCDI	00153
	ICODE(LBB) = 2	BXCDI	00154
	MYBBS = MAX0(MYBBS,LBB)	BXCDI	00155
	GO TO 260	BXCDI	00156
C	DETERMINE LAST NON-ZERO BOX ON NEXT ROW	BXCDI	00157
280	CONTINUE	BXCDI	00158
	MYBBS = MAX0(MYBBS,LBB)	BXCDI	00159
	II = I + 1	BXCDI	00160
	DO 290 K = 1,LBB	BXCDI	00161
	J = LBB - K + 1	BXCDI	00162
		BXCDI	00163
		BXCDI	00164
		BXCDI	00165

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CALL DCODER(IBOX,LSRCHS, II,J, II,J, .T., ICOD) BXCDI 00166
IF (ICOD .NE. 0) GO TO 295 BXCDI 00167
290 CONTINUE BXCDI 00168
    GO TO 300 BXCDI 00169
295 LBB = J BXCDI 00170
300 CONTINUE BXCDI 00171
C      END OF LOOP ON ROWS DETERMINING TIP DIAPHRAGM CODES, FROM 180* BXCDI 00172
C BXCDI 00173
310 CONTINUE BXCDI 00174
    IF (WING) GO TO 350 BXCDI 00175
    MYBBST = MYBBS
    MYBBT = (MYBBS+NSUBD2)/NSUBDV BXCDI 00176
    GO TO 500 BXCDI 00177
350 MYBBSW = MYBBS BXCDI 00178
    MYBBW = (MYBBS+NSUBD2)/NSUBDV BXCDI 00179
    IF (.NOT. COPLAN) GO TO 500 BXCDI 00180
    MYBBST = MYBBSW BXCDI 00181
    MYBBT = MYBBW BXCDI 00182
C BXCDI 00183
    500 RETURN BXCDI 00184
C BXCDI 00185
C BXCDI 00186
8300 WRITE (NT6,9300) BXCDI 00187
9300 FORMAT(51H*** ERROR - TOO MANY CHORDS FOR BOX CODE ARRAY *** ) BXCDI 00188
    CALL PRNTBC(IBOX,LSRCHS,IXBS,MXBBS,MYBBS, .T. ) BXCDI 00189
    CALL FLUSH(1) BXCDI 00190
C BXCDI 00191
C BXCDI 00192
END

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SUBROUTINE GMAREA(IBOX,LBXCD, WING, ALPHA,IJALPH, NALPH)          GMAREA 00002
C
C      IBOX = ARRAY OF BOX TYPE CODES                                GMAREA 00003
C      LBXCD = ROW DIMENSION OF BOX CODE ARRAY                      GMAREA 00004
C      WING = PLANFORM INDICATOR                                     GMAREA 00005
C      A' PHA = ARRAY OF ALPHAS (NORMALIZED AREAS)                  GMAREA 00006
C      IJALPH = SUBSCRIPTS FOR IBOX ARRAY OF CELLS THAT HAVE ALPHAS GMAREA 00007
C                      NOT EQUAL TO 0.0 OR 1.0                         GMAREA 00008
C      NALPH = NUMBER OF ALPHAS STORED                               GMAREA 00009
C
C      COMMON /GEOMTY/  COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,    GMAREA 00010
C                         B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,           GMAREA 00011
C                         MXBW,MXBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW,           GEOMTY 00002
C                         IXBW,XCENTR                                     GEOMTY 00003
C
LOGICAL COPLAN
COMMON /GEOM2/  TLAX,TLAZ,PSIT,MXB1,MYB1,MXB1T,MYB1T,MXB1ST,MYB1ST,   GEOM2 00002
C                         MYB1ST,IXBT,!XBST,CAPL                           GEOM2 00003
COMMON /PLANXY/  NMLE,NMTE,NTLE,NTTE,  XMLE(10),YMLE(10),               PLANXY 00002
C                         XMLE("0"),YMLE(10), XMLE(10),YTLE(10),           PLANXY 00003
C                         XTTE(10),YTTE(10)                           PLANXY 00004
COMMON /EDGES /  FELOC(250), TELOC(250), JDIAG                      EDGES 00002
C
C      COMMON PARAMETERS USED
C      MXB = LENGTH OF BOX PATTERN (X-DIRECTION)                   GMAREA 00016
C      MYB = MAXIMUM ON-PLANFORM SPAN (Y-DIRECTION)                 GMAREA 00017
C      COPLAN = .T., SECOND SURFACE EXISTS FOR PLANFORM          GMAREA 00018
C                      = .F., SINGLE SURFACE                         GMAREA 00019
C      NMLE = NUMBER OF POINTS DEFINING LEADING EDGE OF THE WING   GMAREA 00020
C      NMTE = NUMBER OF POINTS DEFINING TRAILING EDGE OF THE WING  GMAREA 00021
C      NTLE = NUMBER OF POINTS DEFINING LEADING EDGE OF THE TAIL   GMAREA 00022
C      NTTE = NUMBER OF POINTS DEFINING TRAILING EDGE OF THE TAIL  GMAREA 00023
C      XMLE = X COORDINATE OF THE LEADING EDGE DEFINITION POINT   GMAREA 00024
C                      FOR THE FIRST PLANFORM                         GMAREA 00025
C      YMLE = Y COORDINATE OF THE LEADING EDGE DEFINITION POINT   GMAREA 00026
C                      FOR THE FIRST PLANFORM                         GMAREA 00027
C      XMTE = X COORDINATE OF THE TRAILING EDGE DEFINITION POINT  GMAREA 00028
C                      FOR THE FIRST PLANFORM                         GMAREA 00029
C      YMTE = Y COORDINATE OF THE TRAILING EDGE DEFINITION POINT  GMAREA 00030
C                      FOR THE FIRST PLANFORM                         GMAREA 00031
C      XTLE = X COORDINATE OF THE LEADING EDGE DEFINITION POINT   GMAREA 00032
C                      FOR THE SECOND PLANFORM                        GMAREA 00033
C      YTLE = Y COORDINATE OF THE LEADING EDGE DEFINITION POINT   GMAREA 00034
C                      FOR THE SECOND PLANFORM                        GMAREA 00035
C      XTTE = X COORDINATE OF THE TRAILING EDGE DEFINITION POINT  GMAREA 00036
C                      FOR THE SECOND PLANFORM                        GMAREA 00037
C      YTTE = Y COORDINATE OF THE TRAILING EDGE DEFINITION POINT  GMAREA 00038
C                      FOR THE SECOND PLANFORM                        GMAREA 00039
C
COMMON /LAREA / LEFT,RIGHT,ICODE                                     LAREA 00040
DIMENSION IBX(50)                                                 GMAREA 00041
DIMENSION ALPHA(1), IJALPH(1)                                     GMAREA 00042
LOGICAL WING
REAL LIN1,LIN2,LIN3,LIN4,LEFT
NALPH = 1
IF (WING) GO TO 5
MYB = MYBT
IXB = (IXBT-IXBW)/NSUBDV + 1

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MXB = MXBT
CEN2 = 0.
GO TO 8
5 MYB = MYBW
IF (COPLAN) MYB = MAX0(MYB,MYBT)
IXB = 1
MXB = MXBW
IF (COPLAN) MXB = MXBT
8 CONTINUE
C
C      LOOP ON CHORDS
DO 90 J=1,MYB
YJ = J
RIGHT = YJ + 0.5
LEFT = YJ - 0.5
IF (.NOT. WING) GO TO 110
C
C      CALL NTRCEP TO DETERMINE LEFT, RIGHT AND CENTER LINE
C      INTERCEPTS, AND THE BREAK POINTS OF EDGES OVER THIS
C      CHORD.
C
IF (YJ .GT. YMLE(NMLE)) GO TO 10
CALL NTRCEP(J, YMEL, XMLE, LIN1,CEN1,RIN1,NBK1,KINK1, 1)
CALL NTRCEP(J, YTLE, XTLE, LIN2,CEN2,RIN2,NBK2,KINK2, 2)
10 IF (INSURF .EQ. 1 .OR. .NOT. COPPLAN) GO TO 20
C      COMPUTE SLOPE AND INTERCEPTS FOR SECOND PLANFORM.
110 IF (YJ .GT. YTLE(NTLE)) GO TO 20
CALL NTRCEP(J, YTLE, XTLE, LIN3,CEN3,RIN3,NBK3,KINK3, 1)
CALL NTRCEP(J, YTTE, XTTE, LIN4,CEN4,RIN4,NBK4,KINK4, 2)
20 CONTINUE
C      SLOPE AND INTERVALS COMPLETED.
C
C      LOOP DOWN THE CHORD
CALL DCODE(IIBOX,LBXCD, IXB,J, MXB,J, .F., IBX)
II = 1
DO 85 I=IXB,MXB
XI = I
IF (IBX(II) .NE. 1) GO TO 80
BORL = XI - 0.5
BOTR = XI + 0.5
IF (.NOT. WING) GO TO 40
IF (YJ .GT. YMLE(NMLE)) GO TO 40
IF (XI .GT. CEN2) GO TO 40
C      BOX IS ON PLANFORM 1
ICODE = 1
C          ICODE = 1, 1ST L.E. BOX ON CHORD
C          = 2, LAST T.E. BOX ON CHORD
C          = 3, INTERNAL CUT BOX
IF (II .EQ. 1) GO TO 24
IF (IBX(II-1) .NE. 1) GO TO 24
ICODE = 2
IF (I .EQ. MXB) GO TO 24
IF ((II+1) .NE. 1) GO TO 24
IF (I .GT. CEN2) GO TO 24
ICOD = 3
IUE=0
IB=0

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GMAREA	00052
GMAREA	00053
GMAREA	00054
GMAREA	00055
GMAREA	00056
GMAREA	00057
GMAREA	00058
GMAREA	00059
GMAREA	00060
GMAREA	00061
GMAREA	00062
GMAREA	00063
GMAREA	00064
GMAREA	00065
GMAREA	00066
GMAREA	00067
GMAREA	00068
GMAREA	00069
GMAREA	00070
GMAREA	00071
GMAREA	00072
GMAREA	00073
GMAREA	00074
GMAREA	00075
GMAREA	00076
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GMAREA	00094
GMAREA	00095
GMAREA	00096
GMAREA	00097
GMAREA	00098
GMAREA	00099
GMAREA	00100
GMAREA	00101
GMAREA	00102
GMAREA	00103
GMAREA	00104
GMAREA	00105
GMAREA	00106
GMAREA	00107
GMAREA	00108

IF(RIN1.GT.BOXLE) IU=1	GMAREA 00109
IF(RIN2.LT.BOXTE) IB=1	GMAREA 00110
IF(LIN1.LT.BOXTE) IB=1	GMAREA 00111
IF (IU+IB.EQ.0) GO TO 80	GMAREA 00112
C BOX IS NOT ENTIRELY ON PLANFORM. COMPUTE AREA.	GMAREA 00113
24 CONTINUE	GMAREA 00114
CALL ALPHAC(XI, XYLE,YYLE,XYTE,YYTE,	GMAREA 00115
1LIN1,CEN1,RIN1,NBK1,KINK1,LIN2,CEN2,RIN2,NBK2,KINK2,ALPHA(NALPH))	GMAREA 00116
GO TO 75	GMAREA 00117
C	GMAREA 00118
40 CONTINUE	GMAREA 00119
IF (NSURF .EQ. 1) GO TO 80	GMAREA 00120
C BOX IS ON PLANFORM 2.	GMAREA 00121
ICODE =2	GMAREA 00122
IF(I.EQ.MXB) GO TO 44	GMAREA 00123
ICODE =1	GMAREA 00124
IF(XI-1. .LT.CENB) GO TO 44	GMAREA 00125
IF(IBX(II-1) .NE. 1) GO TO 44	GMAREA 00126
ICODE =2	GMAREA 00127
IF(IBX(II+1) .NE. 1) GO TO 44	GMAREA 00128
ICODE =3	GMAREA 00129
IU=0	GMAREA 00130
IB=0	GMAREA 00131
IF(RINB.GT.BOXLE) IU=1	GMAREA 00132
IF(RIN4.LT.BOXTE) IB=1	GMAREA 00133
IF(LIN4.LT.BOXTE) IB=1	GMAREA 00134
IF (IU+IB.EQ.0) GO TO 80	GMAREA 00135
C BOX IS NOT ENTIRELY ON PLANFORM. COMPUTE AREA.	GMAREA 00136
44 CONTINUE	GMAREA 00137
CALL ALPHAC(XI, XYLE,YYLE,XYTE,YYTE,	GMAREA 00138
1LINB,CENB,RINB,NBK3,KINK3,LIN4,CEN4,RIN4,NBK4,KINK4,ALPHA(NALPH))	GMAREA 00139
75 IJALPH(NALPH) = J*512 + I	GMAREA 00140
NALPH = NALPH + 1	GMAREA 00141
80 CONTINUE	GMAREA 00142
II = II + 1	GMAREA 00143
85 CONTINUE	GMAREA 00144
90 CONTINUE	GMAREA 00145
NALPH = NALPH -1	GMAREA 00146
RETURN	GMAREA 00147
END	GMAREA 00148

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SUBROUTINE ALPHAC(X,XLED,YLED,XTED,YTED,
1      L1,C1,R1,NBK1,K1,L2,C2,R2,NBK2,K2,AREA)          ALPHAC 00002
C
C      X      = X COORDINATE OF CELL CENTER                  ALPHAC 00003
C      L1     = X COORDINATE OF LEADING EDGE LEFT SIDE INTERSECTION ALPHAC 00004
C      C1     = X COORDINATE OF LEADING EDGE CENTER LINE INTERSECTION ALPHAC 00005
C      R1     = X COORDINATE OF LEADING EDGE RIGHT SIDE INTERSECTION ALPHAC 00006
C      K1     = FLAG TO INDICATE LEADING EDGE KINK            ALPHAC 00007
C      L2     = X COORDINATE OF TRAILING EDGE LEFT SIDE INTERSECTION ALPHAC 00008
C      C2     = X COORDINATE OF TRAILING EDGE CENTER LINE INTERSECTION ALPHAC 00009
C      R2     = X COORDINATE OF TRAILING EDGE RIGHT SIDE INTERSECTION ALPHAC 00010
C      K2     = FLAG TO INDICATE TRAILING EDGE KINK           ALPHAC 00011
C      AREA   = AREA COMPUTED FOR THE CELL                   ALPHAC 00012
C
C      COMMON /LAREA / LEFT,RIGHT,ICODE
C      LEFT   = Y COORDINATE OF LEFT SIDE OF CHORD           ALPHAC 00013
C      RIGHT  = Y COORDINATE OF RIGHT SIDE OF CHORD          ALPHAC 00014
C      ICODE  = 1, 1ST L.E. BOX ON CHORD                      ALPHAC 00015
C              = 2, LAST T.E. BOX ON CHORD                     ALPHAC 00016
C              = 3, INTERNAL CUT BOX                         ALPHAC 00017
C
C      DIMENSION XC(6),  YC(6)                                LAREA 00002
C      DIMENSION XLED(1) ,YLED(1),  XTED(1),  YTED(1)          ALPHAC 00017
C      REAL LEFT, L1, L2
C      EPS = 1.0E-04
C      BXLE = X-0.5
C      BXTE = X + 0.5
C      XU = X - 1.0
C      XL = X + 1.0
C      AREA = 0.0
C      ISLICE = 0
C      IF (ICODE.EQ.3) GO TO 5000
C      IF(C1.GT.XU.AND.C2.LT.XL) GO TO 3000
1110 IF(ICODE.EQ.1)  GO TO 1000
1120 IF(ICODE.EQ.2)  GO TO 2000
      GO TO 4000
C
C      LEADING EDGE BOX
1000 CONTINUE
      NTRAPS = NBK1 + 1
      NTM1 = NTRAPS - 1
      NXC = NTRAPS + 1
      XC(1) = L1
      YC(1) = LEFT
      XC(NXC) = R1
      YC(NXC) = RIGHT
      IF (NTRAPS.EQ.1) GO TO 110
      DO 100 NA=2,NTRAPS
      KIDX = K1+NA-2
      XC(NA) = XLED(KIDX)
      YC(NA) = YLED(KIDX)
100 CONTINUE
110 CONTINUE
      DO 300 NX = 1,NTRAPS
      IF(XC(NX).GE.BXTE) GO TO 300
      A = BXTE - XC(NX)
      DY = YC(NX+1) - YC(NX)
ALPHAC 00002
ALPHAC 00003
ALPHAC 00004
ALPHAC 00005
ALPHAC 00006
ALPHAC 00007
ALPHAC 00008
ALPHAC 00009
ALPHAC 00010
ALPHAC 00011
ALPHAC 00012
ALPHAC 00013
ALPHAC 00014
ALPHAC 00015
LAREA 00002
ALPHAC 00017
ALPHAC 00018
ALPHAC 00019
ALPHAC 00020
ALPHAC 00021
ALPHAC 00022
ALPHAC 00023
ALPHAC 00024
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ALPHAC 00049
ALPHAC 00050
ALPHAC 00051
ALPHAC 00052
ALPHAC 00053
ALPHAC 00054
ALPHAC 00055
ALPHAC 00056
ALPHAC 00057
ALPHAC 00058

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IF(DY.LT.EPS) GO TO 300          ALPHAC 00059
IF(XC(NX+1).GT.BOXTE) GO TO 250  ALPHAC 00060
B = BOXTE - XC(NX+1)             ALPHAC 00061
200 AREA = AREA + 0.5*(A+B)*DY   ALPHAC 00062
GO TO 300                         ALPHAC 00063
C
C      EDGE CROSSES BOXTE. COMPUTE INTERSECTION FOR DY
250 CONTINUE                      ALPHAC 00064
B = 0.0                            ALPHAC 00065
DX = XC(NX+1) - XC(NX)            ALPHAC 00066
S = DY/DX                          ALPHAC 00067
DY = S*A                           ALPHAC 00068
ALPHAC 00069
ALPHAC 00070
ALPHAC 00071
ALPHAC 00072
ALPHAC 00073
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ALPHAC 00103
ALPHAC 00104
ALPHAC 00105
ALPHAC 00106
ALPHAC 00107
ALPHAC 00108
ALPHAC 00109
ALPHAC 00110
ALPHAC 00111
ALPHAC 00112
ALPHAC 00113
ALPHAC 00114
ALPHAC 00115
C
C      TRAILING EDGE BOX
2000 CONTINUE
NTRAPS = NBK2 + 1
NTM1 = NTRAPS - 1
NMC = NTRAPS + 1
XC(1) = L2
YC(1) = LEFT
XC(NMC) = R2
YC(NMC) = RIGHT
IF(NTRAPS.EQ.1) GO TO 2110
DO 2100 NA = 2,NTRAPS
KIDX = K2 + NA - 2
XC(NA) = XTED(KIDX)
YC(NA) = YTED(KIDX)
2100 CONTINUE
2110 CONTINUE
DO 2300 NDC=1,NTRAPS
IF(XC(NX).LT.BOXLE.AND.XC(NX+1).LT.BOXLE) GO TO 2300
DY = YC(NX+1) - YC(NX)
IF(DY.LT.EPS) GO TO 2300
IF(XC(NX).LT.BOXLE.OR.XC(NX+1).LT.BOXLE) GO TO 2250
C
C      DOES NOT INTERSECT BOXLE
A = XC(NX) - BOXLE
B = XC(NX+1) - BOXLE
2225 AREA = AREA + 0.5*(A+B)*DY
GO TO 2300
C
C      INTERSECTS BOXLE
2250 CONTINUE
DX = XC(NX+1) - XC(NX)
S = DY/DX
A = BOXLE - XC(NX)
YINT = YC(NX) + S*A
IF(S.LT.0.0) GO TO 2275
C
C      SLOPE POSITIVE

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A = 0                                ALPHAC 00116
B = XC(NX+1) - BOXLE                 ALPHAC 00117
DY = YC(NX+1) - YINT                 ALPHAC 00118
GO TO 2225                            ALPHAC 00119
C
C      SLOPE NEGATIVE
2275 CONTINUE
A = -A                                ALPHAC 00120
B = 0                                  ALPHAC 00121
DY = YINT - YC(NX)                     ALPHAC 00122
GO TO 2225                            ALPHAC 00123
2300 CONTINUE
IF(R1.LT.BOXLE) ISLICE =3             ALPHAC 00124
IF(ISLICE.NE.0) GO TO 5000            ALPHAC 00125
GO TO 4000                            ALPHAC 00126
C
C      CHORD HAS ONLY 1 BOX
3000 CONTINUE
BOXLE = R1                             ALPHAC 00127
BOXTE = R1                             ALPHAC 00128
GO TO 1000                            ALPHAC 00129
C :
C
C      THE FOLLOWING IS THE CALCULATIONS FOR A BOX WITH FORE AND
C      AFT BOXES ON PLANFORM. ONE OF 3 CORNER (L.L., L.R., OR U.B.)
C      IS CUT OFF.
C
5000 CONTINUE
IF(ISLICE.NE.0) GO TO 5005
AREA = (RIGHT-LEFT)
5005 CONTINUE
TA = 0.0
IF(ISLICE.EQ.3) GO TO 5020
IF(L2.LT.BOXTE) GO TO 5100
5010 IF(R2.LT.BOXTE) GO TO 5122
5020 CONTINUE
IF(ISLICE.EQ.1.OR.ISLICE.EQ.2) GO TO 5400
IF(R1.GT.BOXLE) GO TO 5300
GO TO 5400
5100 ITAG = 1
I = 1
XC(1) = L2
YC(1) = LEFT
GO TO 5110
5102 ITAG = 2
I = 1
XC(1) = R2
YC(1) = RIGHT
5110 I = I+1
IF(NBK2.EQ.0) GO TO 5150
KIDX = K2 + I - 2
IF(I .G.EQ.2) KIDX = K2 + NBK2-1 - I + 2
IF(XTED(KIDX).GT.BOXTE) GO TO 5125
XC(1) = XTED(KIDX)
YC(1) = YTED(KIDX)
GO TO 5110
5125 CONTINUE

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ALPHAC 00130
 ALPHAC 00131
 ALPHAC 00132
 ALPHAC 00133
 ALPHAC 00134
 ALPHAC 00135
 ALPHAC 00136
 ALPHAC 00137
 ALPHAC 00138
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 ALPHAC 00168
 ALPHAC 00169
 ALPHAC 00170
 ALPHAC 00171
 ALPHAC 00172

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IF(ITAG.EQ.2) KIDX=KIDX+1          ALPHAC 00173
XC(I) = BOXTE                      ALPHAC 00174
DX = XTED(KIDX) - XTED(KIDX-1)      ALPHAC 00175
DY = YTED(KIDX) - YTED(KIDX-1)      ALPHAC 00176
IF(ITAG.EQ.2) GO TO 5130            ALPHAC 00177
C                                     ALPHAC 00178
C           LOWER LEFT CORNER        ALPHAC 00179
DIST = BOXTE - XC(I-1)              ALPHAC 00180
YC(I) = YTED(KIDX-1) + DIST*(DY/DX)  ALPHAC 00181
GO TO 5160                          ALPHAC 00182
C                                     ALPHAC 00183
C           LOWER RIGHT CORNER       ALPHAC 00184
5130 DIST = BOXTE - XC(I-1)          ALPHAC 00185
YC(I) = YC(I-1)+ DIST*(DY/DX)       ALPHAC 00186
GO TO 5160                          ALPHAC 00187
5150 CONTINUE
XC(I) = BOXTE                      ALPHAC 00188
DX = R2 - L2                        ALPHAC 00189
DY = RIGHT - LEFT                   ALPHAC 00190
IF(ITAG.EQ.2) GO TO 5155            ALPHAC 00191
DIST = BOXTE - L2                   ALPHAC 00192
YC(I) = LEFT + DIST*(DY/DX)         ALPHAC 00193
GO TO 5160                          ALPHAC 00194
5155 DIST = BOXTE - R2              ALPHAC 00195
YC(I) = RIGHT+ DIST*(DY/DX)         ALPHAC 00196
5160 CONTINUE
NTRAPS = I-1                        ALPHAC 00197
DO 5175 NX=1,NTRAPS                ALPHAC 00198
A= BOXTE - XC(NX)                  ALPHAC 00199
B= BOXTE - XC(NX+1)                ALPHAC 00200
IF(A.GT.1.) A = 1.0                 ALPHAC 00201
IF(B.GT.1.) B = 1.0                 ALPHAC 00202
H= YC(NX+1) - YC(NX)               ALPHAC 00203
IF(ITAG.EQ.2) H = -H               ALPHAC 00204
TA = TA + 0.5*(A+B)*H              ALPHAC 00205
5175 CONTINUE
IF(ITAG.EQ.2) GO TO 5020            ALPHAC 00206
GO TO 5010                          ALPHAC 00207
C                                     ALPHAC 00208
C           COMPUTE FOR UPPER RIGHT HAND CORNER
5300 CONTINUE
I = 1                               ALPHAC 00209
XC(I) = R1                          ALPHAC 00210
YC(I) = RIGHT                         ALPHAC 00211
5310 I = I+1
IF(NBK1.EQ.0) GO TO 5350            ALPHAC 00212
KIDX = K1 + NBK1-1 -I +2             ALPHAC 00213
IF(XLED(KIDX).LT.BOXLE) GO TO 5325  ALPHAC 00214
XC(I) = XLED(KIDX)                  ALPHAC 00215
YC(I) = YLED(KIDX)                  ALPHAC 00216
GO TO 5310                          ALPHAC 00217
5325 CONTINUE
XC(I) = BOXLE                      ALPHAC 00218
DX = XLED(KIDX+1)-XLED(KIDX)        ALPHAC 00219
DY = YLED(KIDX+1)-YLED(KIDX)        ALPHAC 00220
DIST = BOXLE - XLED(KIDX)            ALPHAC 00221
YC(I)= YLED(KIDX) + DIST*(DY/DX)     ALPHAC 00222
ALPHAC 00223
ALPHAC 00224
ALPHAC 00225
ALPHAC 00226
ALPHAC 00227
ALPHAC 00228
ALPHAC 00229

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GO TO 5360                               ALPHAC 00230
5350 CONTINUE                           ALPHAC 00231
  XC(I) = BOXLE                          ALPHAC 00232
  DX = R1 - L1                           ALPHAC 00233
  DY = RIGHT - LEFT                      ALPHAC 00234
  DIST = BOXLE - L1                      ALPHAC 00235
  YC(I)= LEFT + DIST*(DY/DX)            ALPHAC 00236
5360 CONTINUE                           ALPHAC 00237
  NTRAPS = I-1                           ALPHAC 00238
  DO 5375 NX = 1,NTRAPS                ALPHAC 00239
  A = XC(NX) - BOXLE                     ALPHAC 00240
  B = XC(NX+1)-BOXLE                    ALPHAC 00241
  IF(A.LT.1..AND.B.LT.1.) GO TO 5370   ALPHAC 00242
  IF(A.GT.1..AND.B.GT.1.) GO TO 5365   ALPHAC 00243
C
C      A.GT.B1 AND B.LT.B1               ALPHAC 00244
  DX = XC(NX) - XC(NX+1)                ALPHAC 00245
  DY = YC(NX) - YC(NX+1)                ALPHAC 00246
  IF(DY.LT.EPS) GO TO 5375             ALPHAC 00247
  DIST = BOXLE - XC(NX+1)               ALPHAC 00248
  YINT = YC(NX+1) + DIST*(DY/DX)        ALPHAC 00249
  TA = (YC(NX)-YINT)      +TA          ALPHAC 00250
  YC(NX) =YINT                         ALPHAC 00251
  A = 1.0                                ALPHAC 00252
  GO TO 5370                            ALPHAC 00253
5365 CONTINUE                           ALPHAC 00254
  TA = TA +      (YC(NX)-YC(NX+1))    ALPHAC 00255
  GO TO 5375                            ALPHAC 00256
5370 CONTINUE                           ALPHAC 00257
  H = YC(NX) - YC(NX+1)                ALPHAC 00258
  TA = TA + 0.5*(A+B)*H               ALPHAC 00259
5375 CONTINUE                           ALPHAC 00260
5400 AREA = AREA -TA                  ALPHAC 00261
4000 CONTINUE                           ALPHAC 00262
  RETURN                                ALPHAC 00263
  END                                   ALPHAC 00264
                                         ALPHAC 00265

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SUBROUTINE NTRCEP(J, YEDG,XEDG,           L1,C1,R1,NBK1,K1,INDEX) NTRCEP 00002
C
C      J      = INDEX OF CHORD NUMBER                                NTRCEP 00003
C      NBK1   = NUMBER OF BREAK POINTS ON EDGE FOR THIS CHORD.    NTRCEP 00004
C      L1     = X COORDINATE OF LEADING EDGE LEFT SIDE INTERSECTION NTRCEP 00005
C      C1     = X COORDINATE OF LEADING EDGE CENTER LINE INTERSECTION NTRCEP 00006
C      R1     = X COORDINATE OF LEADING EDGE RIGHT SIDE INTERSECTION NTRCEP 00007
C      K1     = INDEX OF XLEA ANDYLEA ARRAYS THAT DEFINE A KINK IF    NTRCEP 00008
C                  ONE EXISTS                                         NTRCEP 00009
C
COMMON /LAREA / LEFT,RIGHT,ICODE          LAREA  00002
REAL LEFT,L1,L2                          NTRCEP 00012
C      LEFT   = Y COORDINATE OF LEFT SIDE OF CHORD                 NTRCEP 00013
C      RIGHT  = Y COORDINATE OF RIGHT SIDE OF CHORD                NTRCEP 00014
DIMENSION XEDG(1), YEDG(1)                NTRCEP 00015
YJ = J                                     NTRCEP 00016
EPS = 1.0E-04                               NTRCEP 00017
K=2                                         NTRCEP 00018
1 IF(LEFT.LT.YEDG(K)-EPS) GO TO 2        NTRCEP 00019
K= K+1                                     NTRCEP 00020
GO TO 1                                     NTRCEP 00021
2 DX = XEDG(K) - XEDG(K-1)                 NTRCEP 00022
DY = YEDG(K) - YEDG(K-1)                   NTRCEP 00023
DIST = LEFT - YEDG(K-1)                    NTRCEP 00024
L1 = XEDG(K-1) + (DX/DY) * DIST           NTRCEP 00025
C
C      FIND CEN1 AND REBIN COUNTING BREAKS
NBK1 = 0                                     NTRCEP 00026
K1 = 0                                       NTRCEP 00027
NTRCEP 00028
3 IF( YJ .LT.YEDG(K) +EPS) GO TO 4        NTRCEP 00029
NTRCEP 00030
C
C      KINK(S) BETWEEN LEFT AND CENTER LINE
IF(K1.EQ.0) K1 = K                         NTRCEP 00031
NTRCEP 00032
NBK1 = NBK1 + 1                            NTRCEP 00033
NTRCEP 00034
K = K + 1                                    NTRCEP 00035
NTRCEP 00036
GO TO 3                                     NTRCEP 00037
4 IF(INDEX.EQ.1) GO TO 40                  NTRCEP 00038
IF(ABS(YEDG(K)-YJ) .GT.EPS) GO TO 40      NTRCEP 00039
IF(K1.EQ.0) K1 = K                         NTRCEP 00040
NTRCEP 00041
NBK1 = NBK1 +1                            NTRCEP 00042
NTRCEP 00043
IF (YEDG(K+1)-YEDG(K).GT.EPS) GO TO 104   NTRCEP 00044
NTRCEP 00045
NBK1 = NBK1 + 1                           NTRCEP 00046
NTRCEP 00047
K = K + 1                                    NTRCEP 00048
NTRCEP 00049
104 CONTINUE                                 NTRCEP 00049
C1 = XEDG(K)                                NTRCEP 00050
NTRCEP 00051
GO TO 5                                     NTRCEP 00052
40 DX = XEDG(K) - XEDG(K-1)                 NTRCEP 00053
DY = YEDG(K) - YEDG(K-1)                   NTRCEP 00054
DIST = YJ - YEDG(K-1)                      NTRCEP 00055
C1 = XEDG(K-1) + (DX/DY) * DIST            NTRCEP 00056
NTRCEP 00057
C
C      FIND R1 IN SAME MANNER AS CEN1
5 IF(RIGHT.LT.YEDG(K)+EPS) GO TO 6         NTRCEP 00058
C
C      KINKS BETWEEN CENTER LINE AND RIGHT SIDE OF CHORD
IF(K1.EQ.0) K1 = K                         NTRCEP 00059
NTRCEP 00060
NBK1 = NBK1 + 1                            NTRCEP 00061
NTRCEP 00062
K = K + 1                                    NTRCEP 00063

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```
GO TO 5
6 DX = XEDG(K) - XEDG(K-1)
  DY = YEDG(K) - YEDG(K-1)
  DIST = RIGHT - YEDG(K-1)
  R1 = XEDG(K-1) + (DX/DY) * DIST
RETURN
END
```

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NTRCEP 00059
NTRCEP 00060
NTRCEP 00061
NTRCEP 00062
NTRCEP 00063
NTRCEP 00064
NTRCEP 00065
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SUBROUTINE PWAIC(WING,IBOX,LBOX,IWAKE, JCCL)          PWAIC 00002
C
C      COMPUTES THE POINTER ARRAY (MUAIC) FOR THE SPATIAL AIC ARRAY    PWAIC 00003
C      OF THE LEFT WING (TAIL) ON ONE CHORD OF THE RIGHT WING (TAIL)    PWAIC 00004
C
C      WING = WING/TAIL INDICATOR                                     PWAIC 00005
C      IBOX = BOX CODE ARRAY TO USE                                    PWAIC 00006
C      IWAKE = ARRAY OF WAKE EDGE LOCATIONS FOR WING                  PWAIC 00007
C      JCCL = THE (UNSUBDIVIDED) CHORD NUMBER OF INTEREST             PWAIC 00008
C      SURF = INDICATOR OF WHETHER ANY LEFT SURFACE IS INTER-        PWAIC 00009
C             CEPTED BY THE MACH CONE FOR THIS CORD                      PWAIC 00010
C
C      COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,     PWAIC 00011
C      1           B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,                 PWAIC 00012
C      2           MXBW,MXBBW,MYBW,MYBBW,MXBW,MYBWSW,MYBBSW,            PWAIC 00013
C      3           IXBW,XCENTR
C      LOGICAL COPLAN
C      COMMON /GEOM2 / TLAX,TLAZ,PSIT,MGBT,MYBT,MYBBT,MXBST,MYBST,    GEOM2 00002
C      1           MYBBST,IXBT,IXBST,CAPL
C      COMMON /MUAICS/ YBAR,EL,MUAIC(2,50),NROWS,SURF,                MUAICS 00002
C      1           YBARL,ELL, MUAICL(2,50),NROWSL,SURFL,PSIDL
C      LOGICAL SURF,SURFL
C      COMMON /EDGES / FEXLOC(250), TEXLOC(250), JDIAG
C
C      LOGICAL WING
C      DIMENSION IBOX(LBOX,8), ICODE(50), IWAKE(1)
C      DATA EPS / 1.0E-4 /
C
C      IF (WING) GO TO 100
C
C      THE CALL IS FOR A TAIL CHORD
C      PBIZ = PSIT + PSIT
C      IXB = (IXBT-IXBW)/NSUBDV + 1
C      MXB = MXBT
C      IF (JCCL .LE. MYBT) GO TO 120
C      THE CHORD IS ON THE TIP DIAPHRAGM
C      IFIRST = IXB
C      NU = MXBT-IFIRST+1
C      GO TO 130
C
C      THE CALL IS FOR A WING CHORD
100 CONTINUE
C      PBIZ = PSIW + PSIW
C      IXB = 1
C      IF (COPLAN) GO TO 110
C      MXB = MXBBW
C      IF (JCCL .GT. MYBW) GO TO 115
C      ISUB = JCCL+NSUBDV - NSUBD2
C      GO TO 125
C
C      THE CALL IS FOR A COPLANAR WING-TAIL
110 CONTINUE
C      MXB = MXBT
C      IF (JCCL .LE. MYBT) GO TO 120
C      THE CHORD IS ON THE TIP DIAPHRAGM
115 CONTINUE
C      IFIRST = 1

```

```

NU = MXB
GO TO 130

C
C      THE CHORD IS ON PLANFORM
120 CONTINUE
ISUB = NYBSW + JCCL*NSUBDV - NSUBD2
125 CONTINUE
IFRST = (TEXLOC(ISUB)-IXBW) / NSUBDV + 1
NU = MXB - IFRST + 1
C
130 CONTINUE
CALL DCODER(IBOK,LBOK,IFRST,JCCL,MXB,JCCL,.F.,ICODE)
IROW = IFRST
DO 135 I = 1,NU
IA = I
IF (ICODE(I) .NE. 0) GO TO 140
IROW = IRROW + 1
135 CONTINUE
140 CONTINUE
IF (IA .GE. NU) GO TO 155
DO 145 I = IA,NU
IF (ICODE(I) .EQ. 0) GO TO 150
IROW = IRROW + 1
145 CONTINUE
150 CONTINUE
IROW = IRROW - 1
155 CONTINUE
NROWS = IRROW - IXB + 1
C
C      COMPUTE HORIZONTAL AND VERTICAL OFFSETS
YMUSND = (JCCL-.5)*COS(PSI2)
      = Y-OFFSET ON THE SENDING SURFACE OF THE PROJECTION OF THE
C      RECEIVING CHORD
JBAR = IFIX(YMUSND) + 1
C      = CHORD CONTAINING YMUSND
YBAR = YMUSND - JBAR + .5
C      = DISTANCE FROM NEAREST SENDING CHORD CENTER TO PROJE-
C      CTION OF THE RECEIVING CHORD, POSITIVE RIGHT.
EL = (JCCL - .5) * SIN(PSI2)
C      = VERTICAL SEPARATION BETWEEN THE SENDING PLANE AND THE
C      RECEIVING CHORD
IF (YBAR) 160,165,170
160 JMIN = JBAR - 1
NBOXES = 2
GO TO 180
165 JMIN = JBAR
NBOXES = 1
GO TO 180
170 JMIN = JBAR
NBOXES = 2
C
180 CONTINUE
JL = 1
SURF = .F.
C
C      START OF LOOP ON ROWS, FORWARD FROM RECEIVING BOX CENTER, TO
C      DEFINE THE MUASIC ARRAY

```

PWASIC	00052
PWASIC	00053
PWASIC	00054
PWASIC	00055
PWASIC	00056
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PWASIC	00058
PWASIC	00059
PWASIC	00060
PWASIC	00061
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PWASIC	00102
PWASIC	00103
PWASIC	00104
PWASIC	00105
PWASIC	00106
PWASIC	00107
PWASIC	00108

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DO 280 I = 1,NROWS
XI = I - .5 - ABS(EL)
IF (XI .LE. EPS) GO TO 280
IF (JMIN .GT. 0) GO TO 280
C      CENTER LINE HAS BEEN CROSSED, THEREFORE THERE MAY BE CONTRIBU-
C      TION FROM THE LEFT WING FOR THIS ROW
JM = -JMIN + 1
JMM = JM-JL+1
CALL DCODE(IBOX,LBOX, IROW,1, IROW,JMM, .F., ICODE)
DO 240 J = JL,JM
IF (ICODE(JMM) .NE. 0) GO TO 250
JMM = JMM+1
240 CONTINUE
NROWS = I - 1
GO TO 290
C      CONTRIBUTING BOXES HAVE BEEN FOUND FOR THIS ROW
290 CONTINUE
SURF = .T.
JL = J
IF (YBAR .GE. 0) GO TO 255
MUAC(1,I) = NBOXES - JM + 1
MUAC(2,I) = NBOXES - JL + 1
GO TO 270
255 CONTINUE
MUAC(1,I) = JL
MUAC(2,I) = JM
GO TO 270
C      CENTER LINE HAS NOT BEEN CROSSED
280 MUAC(1,I) = 0
MUAC(2,I) = 0
C
270 CONTINUE
NBOXES = NBOXES + 2
JMIN = JMIN - 1
IROW = IROW - 1
280 CONTINUE
C      END OF LOOP FORWARD ON ROWS, FROM 180*
C
290 CONTINUE
RETURN
END

```

PWAIC 00109
 PWAIC 00110
 PWAIC 00111
 PWAIC 00112
 PWAIC 00113
 PWAIC 00114
 PWAIC 00115
 PWAIC 00116
 PWAIC 00117
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 PWAIC 00119
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 PWAIC 00150

OVERLAY (AFMBOX,1,3)	MODES	00002
PROGRAM MODES	MODES	00003
COMMON /PROBLM/ YMACH,NMODES,NTSLOP,NKVALS,SMOOTH,NDEC,CRDFIT,	PROBLM	P0002
1 EXAIC,SUBDV,PLYWOOD	PROBLM	00003
LOGICAL SMOOTH,CRDFIT,EXAIC,SUBDV,PLYWOOD	PROBLM	00004
COMMON /CTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT,	CTRL	00012
1 DEFAULT	CTRL	00003
LOGICAL PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT	CTRL	00004
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,	GEOMTY	00002
1 B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,	GEOMTY	00003
2 MXBW,MXBBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW,	GEOMTY	00004
3 IXBW,XCENTR	GEOMTY	00005
LOGICAL COPLAN	GEOMTY	00006
COMMON /GEOM2/ TLAX,TLAZ,PSIT,MXB7,MYBT,MYBBT,MXB7T,MYB7T,	GEOM2	00002
1 MYBBT,IXBT,IXB7T,CAPL	GEOM2	00003
COMMON /FILES/ NT5,NT6,INTAPE,INFSP,NPLAIC,NPAIC,NOUTP,	FILES	00002
1 IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC	FILES	00003
EQUIVALENCE (IWTFSC,ITSLSC)	MODES	00009
COMMON /IOCONT/ OPLAIC,OSPAIC,WTGEOM,WTGNF,WTSL,WTBL,PRBOX,	IOCONT	00002
1 PRPAIC,PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,	IOCONT	00003
2 PRBL,PRDCP,PRGNAC,PRSL,PRLW,PRNW,PRCM	BCSFRB	00001
EQUIVALENCE (PRUW,PRDW)	IOCONT	00005
LOGICAL OPLAIC,OSPAIC,WTGEOM,WTGNF,WTSL,WTBL,PRBOX,PRATC,	IOCONT	00006
1 PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNF,	IOCONT	00007
2 PRDCP,PRGNAC,PRUW,PRLW,PRNW,PRCM	BCSFRB	00002
COMMON /TAPEIO/ NFS,NMS,LS,NHR,IB(20),NID,ITYPE,LRS,LWS,M,N,	TAPEIO	00002
1 PARM(10),IRR	TAPEIO	00003
DIMENSION IPARM(10)	TAPEIO	00004
EQUIVALENCE (PARM,IPARM)	TAPEIO	00005
COMMON /ARRAYS/ KBXCDW,LBXCDW,LBXCC,KBXCDT,LBXCDT,KJALPH,LJALPH,	ARRAYS	00002
1 KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,DEFLSL,KELPHI,	ARRAYS	00003
2 LMODES,KPNTRD,LPNTRD,KSDW,LSDW,KPTNDW,LPTNDW,	ARRAYS	00004
3 KDW,LDW,KTVP,LTVF	ARRAYS	00005
COMMON / MODES/ SYH,SYMT,MTYPEW,MTYPEP	MODECOM	00002
COMMON /CHECKPR/ DPPCPR,GEOCPR,MODCPR,AICCP,RNSCPR,SMCPR,GAFCPR	CHECKPR	00002
LOGICAL DPPCPR, GEOCPR, MODCPR, AICCP, RNSCPR, SMCPR, GAFCPR	CHECKPR	00003
EQUIVALENCE (CHECKPR,MODCPR)	MODES	00015
LOGICAL CHECKPR	MODES	00016
C DEFSL(2,NBOXES), XX(NPTS), YY,ZZ SAME, A (NO OF COEF)	MODES	00017
DIMENSION DEFSL(2,1000), XX(100),YY(100),ZZ(100), A(21)	MODES	00018
COMMON /INDEX/ IS(100),NOC(100),JS(100),JOC(100)	MODES	00019
DIMENSION IPNTRM(2,100)	MODES	00020
DIMENSION XP(6),YP(6),X1(100),Y1(100)	MODES	00021
DIMENSION DOS(50)	MODES	00022
DIMENSION FEXLOC(250), TEXLOC(250)	MODES	00023
C FEXLOC(MYBSW+MYB7T), TEXLOC SAME)	MODES	00024
C	MODES	00026
LOGICAL MXREAD,RANDIN,MXRWIT,RANDOU	MODES	00027
NAMELIST /CARDM/ NMODES,NTSLOP	FTNX1	00043
MXREAD = .FALSE.	MODES	00028
MXRWIT = .FALSE.	MODES	00029
RANDIN = .FALSE.	MODES	00030
RW::NWU = .FALSE.	MODFS	00031
EPS = 1.0E-04	MODES	00032
GAMMA = 1.4	MODES	00033
GAMC = XMACH*(GAMMA+1.0)/2.	MODES	00034
IF(.NOT.PRVMODE) GO TO 100	MODES	00035

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IF(.NOT.PRGEOEM) GO TO 50      MODES 00036
WRITE (NT5,7005)                MODES 00037
RETURN                          MODES 00038
C                                MODES 00039
50 CONTINUE                      MODES 00040
IF(MTYPEEW.EQ.3) GO TO 75      MODES 00041
IF(NSURF.EQ.2.AND.MTYPET.EQ.3) GO TO 75
WRITE (NT5,7010)
GO TO 125
C                                MODES 00042
75 CONTINUE                      MODES 00043
WRITE (NT5,7015)
CALL FLUSH(1)
C                                MODES 00044
C                                MODES 00045
100 CONTINUE                     MODES 00046
C                                MODES 00047
C                                MODES 00048
NMODES = 0                      MODES 00049
NTSLOP = 0                       MODES 00050
READ(NT5,CARDM)
125 CONTINUE                     MODES 00051
REWIND ICEOSC
C                                MODES 00052
C                                MODES 00053
C                                MODES 00054
READ FEXLOC AND TEXLOC ARRAY FROM GEOMETRY SCRATCH FILE MODES 00055
MNAME =GFEXLOC                 MODES 00056
CALL RDINIT                      MODES 00057
ITYPE = SHMIXED                  MODES 00058
NMS = 1                           MODES 00059
IF(.NOT.COPLAN.AND.NSURF.EQ.2) NMS = 2
CALL READMX(ICEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
1           LRS,FEXLOC, M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6010
C                                MODES 00060
MNAME =GTEXLOC                  MODES 00061
CALL RDINIT                      MODES 00062
ITYPE = SHMIXED                  MODES 00063
CALL READMX(ICEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
1           LRS,TEXLOC, M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6010
C                                MODES 00064
C                                MODES 00065
C                                MODES 00066
C                                MODES 00067
C                                MODES 00068
C                                MODES 00069
C                                MODES 00070
C                                MODES 00071
C                                MODES 00072
C                                MODES 00073
C                                MODES 00074
C                                MODES 00075
C                                MODES 00076
C                                MODES 00077
C                                MODES 00078
C                                MODES 00079
130 CONTINUE                     MODES 00080
C                                MODES 00081
C                                MODES 00082
C DETERMINE STARTING BOXES AND NUMBER OF BOXES PER CHORD. MODES 00083
IYB1 = (NBUDDV+1)/2            MODES 00084
DO 300 NS=1,NSURF               MODES 00085
IF(NS.EQ.2) GO TO 200
NC =1
NCH = NYBW
NCF = 0
ICN = IXBW - IYB1
IXB = IXBW
GO TO 225

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200 CONTINUE                                MODES 00093
  NC = MYBW + 1                            MODES 00094
  NCH = MYBW + MYBT                         MODES 00095
  NCF = MYBW * NSUBDV                      MODES 00096
  IFBT = (IXBT-IXBW)/NSUBDV + 1             MODES 00097
225 CONTINUE                                MODES 00098
  IYB = IYB1 + NCF                          MODES 00099
  DO 250 J=NC,NCH                           MODES 00100
    IS(J) = FELOC(IYB) + 1.0                 MODES 00101
    ITEI = TEXLOC(IYB)
    IF(NSUBDV.EQ.1) GO TO 240                MODES 00102
    IS(J) = (IS(J)-ICN)/NSUBDV + 1          MODES 00103
    ITEI = (ITEI-IXB)/NSUBDV + 1             MODES 00104
240 CONTINUE                                MODES 00105
  NOC(J) = ITEI-IS(J) + 1                  MODES 00106
  IYB = IYB + NSUBDV                      MODES 00107
250 CONTINUE                                MODES 00108
300 CONTINUE                                MODES 00109
  CALL ROPER                                 MODES 00110
  MODES 00111
C
C
C      FIND OVERLAP OF 2 PLANFORMS IF THEY ARE NON-COPLANAR
  IOVLAP = 0                                  MODES 00112
  NPNTRS = MXBW+1                            MODES 00113
  IF(NSURF.EQ.1) GO TO 325                  MODES 00114
  IF(COPLAN) GO TO 324                      MODES 00115
  IF(IFBT.GT.MXBW) GO TO 324                MODES 00116
  IOVLAP = MXBW - IFBT + 1                  MODES 00117
  NPNTRS = MXBT + IOVLAP + 1                MODES 00118
  GO TO 325                                 MODES 00119
324 CONTINUE                                MODES 00120
  NPNTRS = MXBT + 1                         MODES 00121
325 CONTINUE                                MODES 00122
C      COMPUTE POINTER ARRAY AND STORE ON MODESC
  REWIND MODESC                             MODES 00123
  MODES 00124
C
  IPNTRM(1,1) = 1                           MODES 00125
  IPNTRM(2,1) = JS(1)                        MODES 00126
  DO 320 I=2,NPNTRS                         MODES 00127
    IPNTRM(1,I) = IPNTRM(1,I-1) + JOC(I-1)
    IPNTRM(2,I) = JS(I)
320 CONTINUE                                MODES 00128
  CALL RDINIT                               MODES 00129
  IPARM(3) = IOVLAP                         MODES 00130
  ITYPE = SHMIXED                           MODES 00131
  CALL WTEMX(MODESC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
  1           IPNTRM,ITYPE,2,NPNTRS,PARM,IRR)   MODES 00132
  IF(IRR.NE.0) GO TO 6030                  MODES 00133
C      FIRST LOOP DETERMINES MODE SHAPES.
C      SECOND LOOP DETERMINES THICKNESS SLOPES.
C
  DO 3 0 IPASS=1,2                           MODES 00134
  IF(I .SS.EQ.2) GO TO 2100                MODES 00135
C      DOOP ON NUMBER OF SURFACES
  DO 2000 MS=1,NSURF                         MODES 00136
  IF(NS.EQ.NSURF) GO TO 330                MODES 00137
C
  MODES 00138
  MODES 00139
  MODES 00140
  MODES 00141
  MODES 00142
  MODES 00143
  MODES 00144
  MODES 00145
  MODES 00146
  MODES 00147
  MODES 00148
  MODES 00149

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      IFILE = IVPSC          MODES 00150
      REWIND IVPSC          MODES 00151
      GO TO 340             MODES 00152
 330 CONTINUE           MODES 00153
      IFILE = MODESC        MODES 00154
 340 CONTINUE           MODES 00155
C
C      LOOP ON NUMBER OF MODES    MODES 00156
      DO 1500 NM =1,NMODES      MODES 00157
C
C      ZERO OUT THE DEFLSL ARRAY MODES 00158
      DO 350 I = 1,LMODES      MODES 00159
          DEFLSL(1,I) = 0.0     MODES 00160
          DEFLSL(2,I) = 0.0     MODES 00161
 350 CONTINUE           MODES 00162
BCSMDA 00002
C
C      ZERO OUT THE COEFFICIENT ARRAY MODES 00163
      DO 355 I = 1,21         MODES 00164
          A(I) = 0.0           MODES 00165
 355 CONTINUE           MODES 00166
BCSMDA 00003
C
C      INPUT FIRST PLANEFORM IF THERE IS A TAIL SECTION MODES 00167
      IF(NB.EQ.1) GO TO 400     MODES 00168
      READ (IVPSC) DEFLSL      MODES 00169
C
C
 400 CONTINUE           MODES 00170
      IF(.NOT.PREMODE) GO TO 450     MODES 00171
      CALL RDINIT              MODES 00172
      IF(NB.EQ.1.AND.NM.EQ.1) NFS = 2     MODES 00173
      MNAME = SH COEF.          MODES 00174
      CALL READMX(IGEOBC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
      1           LRS,A,M,N,PARM,IRR)     MODES 00175
      IF(IRR.NE.0) GO TO 6010     MODES 00176
      NFS = 0                   MODES 00177
      MODES 00178
C
      GO TO 551               MODES 00179
 450 CONTINUE           MODES 00180
      ITYPE = MTYPEW            MODES 00181
      IF(NB.EQ.2) ITYPE = MTYPEP     MODES 00182
      GO TO (501,502,503),ITYPE     MODES 00183
C
      READ IN POLYNOMIAL COEFFICIENTS     MODES 00184
 501 CONTINUE           MODES 00185
      READ(NTS,8010) IDEG        MODES 00186
      IF (IDEF .LT. 0 .OR. IDEG .GT. 5) GO TO 6000     MODES 00187
 8010 FORMAT(2I5)        MODES 00188
      MDEG = IDEG + 1           MODES 00189
      DEG = MDEG                MODES 00190
      DEG2= DFG/2.              MODES 00191
      NC = DEG+DEG2 + DEG2 + EPS     MODES 00192
      READ(NTS,8015)(A(I),I=1,NC)     MODES 00193
 8015 FORMAT(7E10.0)        MODES 00194
      IFLAG = 1                 MODES 00195
      GO TO 550               MODES 00196
C
      MODES 00197
      MODES 00198
      MODES 00199
      MODES 00200
      MODES 00201

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C      READ IN DEFLECTIONS AT SELECTED LOCATIONS AND FIT A POLYNOMIAL MODES 00202
C      OF DEGREE IDEG TO THE POINTS USING METHOD OF LEAST SQUARES. MODES 00203
502 CONTINUE MODES 00204
      READ (NPTS,8010) IDEG,NPTS MODES 00205
      IF (IDEGLT.0.OR.IDEG.GT.5) GO TO 6005 MODES 00206
      IF (NPTS.GT.100.OR.NPTS.LT.1) GO TO 6005 MODES 00207
      READ (NPTS,8020) (XX(I),YY(I),ZZ(I),I=1,NPTS) MODES 00208
8020 FORMAT(6E10.0) MODES 00209
      IDIM = 1 MODES 00210
      CN = 1.0 MODES 00211
C
C      CN IS A SCALE FACTOR TO REDUCE THE MAGNITUDE OF THE NUMBERS MODES 00212
C      IDIM IS A DIMENSION VARIABLE SET TO 1 TO INDICATE FIT IS MODES 00213
C      BEING MADE ON REAL VALUES . IDIM = 2 FOR COMPLEX Z VALUES. MODES 00214
      CALL FITTER(IDEG,NPTS,XX,YY,ZZ,A,CN,IDIM) MODES 00215
C
      IFLAG = 2 MODES 00216
      IDEG = IDEG + 1 MODES 00217
      DEG = IDEG + 1 MODES 00218
      DEG2 = DEG/2. MODES 00219
      NC = DEG*DEG2 + DEG2 + EPS MODES 00220
C
C      550 CONTINUE MODES 00221
C
C      STORE THE COEFFICIENTS ON THE THIRD FILE OF THE IGEOSC FILE. MODES 00222
C      IF THE COEFFICIENTS ARE TO BE PRINTED THE ONES FOR THE FIRST MODES 00223
C      SURFACE MUST BE STORED ON A SCRATCH FILE TEMPORARILY. MODES 00224
C
      CALL RDINIT MODES 00225
      IF(NS.EQ.1.AND.NM.EQ.1) NFS = 2 MODES 00226
      IPARM(3) = IDEG MODES 00227
      IPARM(4) = IFLAG MODES 00228
      ITYPE = SHMIXED MODES 00229
      CALL WRTMX(IGEOSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,1,1D,
1           A,ITYPE,1,NC,FARM,IRR) MODES 00230
      IF(IRR.NE.0) GO TO 6050 MODES 00231
      NFS = 0 MODES 00232
C
      IF(.NOT.FRCOEF) GO TO 3550 MODES 00233
      IF(NS.EQ.2.OR.NSURF.EQ.1) GO TO 3550 MODES 00234
      IF(NM.EQ.1) REWIND IAICSC MODES 00235
C
      CALL WRTMX(IAICSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,1,1D,
1           A,ITYPE,1,NC,FARM,IRR) MODES 00236
      IF(IRR.NE.0) GO TO 6060 MODES 00237
C
      3550 CONTINUE MODES 00238
      IF(NS.EQ.2.AND.NM.EQ.1) REWIND IAICSC MODES 00239
C
C      EVALUATE THE POLYNOMIAL EQUATION FOR DEFLECTIONS. MODES 00240
C      (E PARTIAL DERIVATIVE WITH RESPECT TO X TO GET SLOPES. MODES 00241
C
C      551 CONTINUE MODES 00242
      IF(NM.NE.1) GO TO 560 MODES 00243
      IF(NS.EQ.2) GO TO 556 MODES 00244

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C
C      CALCULATE X,Y COORDINATES FOR EVALUATION OF POLYNOMIAL      MODES 00259
      X1(I) = XCENTR      MODES 00260
      Y1(I) = 0.5*B1BETA      MODES 00261
      IF(NSURF.EQ.2) GO TO 552      MODES 00262
      MMAX = MAXD(MXBW,MYBW)      MODES 00263
      GO TO 554      MODES 00264
552 CONTINUE      MODES 00265
      MMAX = MAXD(MXBT,MYBT,MYBW)      MODES 00266
554 CONTINUE      MODES 00267
      DO 555 I= 2,MMAX      MODES 00268
      X1(I) = X1(I-1)+B1      MODES 00269
      Y1(I) = Y1(I-1)+B1BETA      MODES 00270
555 CONTINUE      MODES 00271
      GO TO 580      MODES 00272
C      MODES 00273
556 CONTINUE      MODES 00274
      XADJ = TLAX - WLAX      MODES 00275
      DO 557 I=1,MMAX      MODES 00276
      X1(I) = X1(I) - XADJ      MODES 00277
557 CONTINUE      MODES 00278
      GO TO 580      MODES 00279
C      MODES 00280
C      MODES 00281
580 CONTINUE      MODES 00282
      IF(NS.EQ.2) GO TO 580      MODES 00283
      IC = 0      MODES 00284
      ILIM = MXBW      MODES 00285
      IBEG = 1      MODES 00286
      NCH = 0      MODES 00287
      GO TO 564      MODES 00288
580 CONTINUE      MODES 00289
      IBEG = IFBT      MODES 00290
      ILIM = MXBT      MODES 00291
      NCH = MYBW      MODES 00292
      IC = 0      MODES 00293
      IUP = MXBW      MODES 00294
      IF(COPLAN) IUP = IFBT-1      MODES 00295
      DO 563 I=1,IUP      MODES 00296
      IC = IC + JOC(I)      MODES 00297
563 CONTINUE      MODES 00298
564 CONTINUE      MODES 00299
      DO 575 IX=IBEG,ILIM      MODES 00300
      I = IX      MODES 00301
      IF(NS.EQ.2) I = IX + ICULAP      MODES 00302
      XP(I) = 1.      MODES 00303
      DO 561 IP=2,MDEG      MODES 00304
561 XP(IP) = XP(IP-1) * XI(IX)      MODES 00305
      JI= JS(I)      MODES 00306
      JT= JOC(I)+ JI -1      MODES 00307
      DO 570 J=JI,JT      MODES 00308
      IC = IC +1      MODES 00309
      IB = IB(J+NCH)      MODES 00310
      IT = IB + NOC(J+NCH) -1      MODES 00311
      IF(IX.LT.IB) GO TO 570      MODES 00312
      IF(IX.GT.IT) GO TO 570      MODES 00313
      YP(I) = 1.      MODES 00314
      MODES 00315

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DO 562 JP =2,MDEG      MODES 00316
562 YP(JP) = YP(JP-1)*Y1(J) MODES 00317
D = A(1)                MODES 00318
S = 0.0                 MODES 00319
IF (MDEG .LT. 2) GO TO 567 BCSMDA 00008
IA = 1                  MODES 00320
DO 565 L2=2,MDEG       MODES 00321
DO 565 L3=1,L2          MODES 00322
L4 = L2-L3+1            MODES 00323
IA = IA + 1             MODES 00324
D = D + XP(L4)*YP(L3)*A(IA) MODES 00325
IF(L4.EQ.1) GO TO 565   MODES 00326
L5 = L4 - 1             MODES 00327
S = S + L5*XP(L5)*YP(L3)*A(IA) MODES 00328
565 CONTINUE             MODES 00329
567 CONTINUE             BCSMDA 00009
DEFSL(1,IC) = D          MODES 00330
DEFSL(2,IC) = S          MODES 00331
570 CONTINUE             MODES 00332
575 CONTINUE             MODES 00333
GO TO 900               MODES 00334
C                         MODES 00335
C     READ IN DEFLECTIONS AND SLOPES AT BOX CENTERS MODES 00336
503 CONTINUE             MODES 00337
IF(INTAPE.EQ.0.OR.INTAPE.EQ.5) GO TO 700 MODES 00338
C                         MODES 00339
C     MODES ON TAPE. CALL SPECIAL ROUTINE TO HANDLE. MODES 00340
CALL TAPMOD(NS,NM,DEFSL) MODES 00341
GO TO 900               MODES 00342
700 CONTINUE             MODES 00343
IF(NS.EQ.2) GO TO 720   MODES 00344
C                         MODES 00345
C     FIRST PLANFORM MODES 00346
NCH = MYBW               MODES 00347
NC = 1                   MODES 00348
GO TO 725               MODES 00349
720 CONTINUE             MODES 00350
NC = MYBW + 1            MODES 00351
NCH = MYBW + MYBT        MODES 00352
C                         MODES 00353
C     READ AND STORE DEFLECTIONS MODES 00354
725 CONTINUE             MODES 00355
DO 750 J=NC,NCH          MODES 00356
IST = IS(J)               MODES 00357
NK = NOC(J) + IST - 1    MODES 00358
JSUM = 0                  MODES 00359
ITROW = IST               MODES 00360
IF(.NOT.COPLAN.AND.NS.EQ.2) ITROW = IST + IOVLAP MODES 00361
DO 730 I=1,ITROW          MODES 00362
730 JSUM = JSUM + JOC(I)  MODES 00363
JSU = JSUM - JOC(ITROW) + 1 MODES 00364
REA (INT5,9015) (DOS(I),I=IST),NK) MODES 00365
DO 750 I=IST,NK           MODES 00366
IX = ?                   MODES 00367
IF(.NOT.COPLAN.AND.NS.EQ.2) IX = I + IOVLAP MODES 00368
ISUB = JSUM + J - JS(IX) - NC + 1 MODES 00369
DEFSL(1,ISUB) = DOS(I)    MODES 0037

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        JSUM = JSUM + JOC(IX)          MODES 00371
750 CONTINUE
C
C      READ AND STORE SLOPES
DO 775 J=NC,NCH                MODES 00372
    IST = IS(J)
    NK = NOC(J) +IST -1          MODES 00373
    JSUM = 0                      MODES 00374
    ITROW = IST                  MODES 00375
    IF(.NOT.COPLAN.AND.NS.EQ.2) ITROW = IST + IOVLAP MODES 00376
    DO 770 I=1,ITROW              MODES 00377
    DO 770 JSUM = JSUM + JOC(I)   MODES 00378
        JSUM = JSUM - JOC(ITROW) + 1 MODES 00379
        READ(NT5,9015) (DOS(I),I=IST,NK) MODES 00380
        DO 775 I=IST,NK             MODES 00381
        IX = I                     MODES 00382
        IF(.NOT.COPLAN.AND.NS.EQ.2) IX = I + IOVLAP MODES 00383
        ISUB = JSUM + J - JS(IX) - NC + 1 MODES 00384
        DEFSL(2,ISUB) = DOS(I)       MODES 00385
        JSUM = JSUM + JOC(IX)         MODES 00386
775 CONTINUE
900 CONTINUE
C
C      WRITE THE DEFSL ARRAY ONTO MODESC FILE
IF(NB.EQ.NSURF) GO TO 925        MODES 00394
WRITE (IFILE) DEFSL              MODES 00395
GO TO 950                         MODES 00396
925 CONTINUE
CALL RDINIT                        MODES 00397
ITYPE = SHMIXED                   MODES 00398
N = IPNTRM(1,NPNTRS)-1           MODES 00399
CALL WRTMX(IFILE,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1           DEFSL,ITYPE,2,N,PARM,IRR) MODES 00400
IF(IRR.NE.0) GO TO 6020           MODES 00401
930 CONTINUE
IF(NB.NE.NSURF) GO TO 1500        MODES 00402
IF(.NOT.PRMOVS.AND..NOT. PRCOEF) GO TO 1500 MODES 00403
C
C      PRINT MODES, COEFFICIENTS OR BOTH
C
WRITE (NT6,9500) TITLE,NM,XMACH MODES 00404
IF(.NOT.PRCOEF) GO TO 975        MODES 00405
IF(MTYPEW.EQ.3) GO TO 960        MODES 00406
C
C      PRINT COEFFICIENTS
C
IF(NSURF.EQ.1) GO TO 980        MODES 00407
CALL RDINIT                        MODES 00408
ITYPE = SHMIXED                   MODES 00409
CALL READMX(IAICSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,1D,ITYPE,
1           LRS,XX,M,N,PARM,IRR) MODES 00410
IF(IRR.NE.0) GO TO 6070           MODES 00411
C
IFLG = IPARM(4)                  MODES 00412
IDG1 = IPARM(3)                  MODES 00413
CALL PRECOF(IDG1,XX,IFLG)        MODES 00414
C

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980 CONTINUE                                MODES 00428
  IF(NB.EQ.1.AND.MTYPEW.EQ.3) GO TO 975      MODES 00429
  IF(NB.EQ.2.AND.MTYPET.EQ.3) GO TO 975      MODES 00430
  CALL PRECOF(Ideg,A,IFLAG)                  MODES 00431
C
  975 CONTINUE                                MODES 00432
    NROWS = MXBW
    IF(NSURF.EQ.2) NROWS = MXBT + IOVLAP      MODES 00433
C
C       CALL ROUTINE TO PRINT THE MODE SHAPES   MODES 00434
  CALL MODOUT(DEFSL,JS,JCC,NROWS,NM,IOVLAP)   MODES 00435
C
  1500 CONTINUE                                MODES 00436
    END FILE IFILE                           MODES 00437
    M = 1                                     MODES 00438
    N = 400                                    MODES 00439
    CALL WRTEMX(IFILE,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,1,ID,
1           IS,ITYPE,M,N,PARM,IRR)            MODES 00440
    IF(IRR.NE.0) GO TO 6040                   MODES 00441
C
    END FILE IFILE                           MODES 00442
    REWIND IFILE                            MODES 00443
  2000 CONTINUE                                MODES 00444
    GO TO 3000                               MODES 00445
  2100 CONTINUE                                MODES 00446
C
C       DETERMINE THICKNESS SLOPES          MODES 00447
C
    NBV = IPNTRM(1,NPNTRS) - 1                MODES 00448
    REWIND ITSLSC                            MODES 00449
    IF(NTSLOP.NE.0) GO TO 2225               MODES 00450
C
C       WRITE ARRAY OF ONES                 MODES 00451
C
    DO 2200 I=1,NBV                         MODES 00452
      DEFSL(I,I) = 1.0                      MODES 00453
  2200 CONTINUE                                MODES 00454
C
    CALL RDINIT                                MODES 00455
    ITYPE = SHMIXED                           MODES 00456
    M = 1                                     MODES 00457
    N = NBV                                   MODES 00458
    CALL WRTEMX(ITSLSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1           DEFSL, ITYPE, M, N, PARM, IRR)     MODES 00459
    IF(IRR.NE.0) GO TO 6080                   MODES 00460
C
    END FILE ITSLSC                          MODES 00461
    REWIND ITSLSC                            MODES 00462
    GO TO 3000                               MODES 00463
C
  2225 CONTINUE                                MODES 00464
    DO 2600 NS=1,NSURF                      MODES 00465
    IF(NS.EQ.NSURF) GO TO 2230              MODES 00466
    IFILE = IVPSC                            MODES 00467
    REWIND IVPSC                            MODES 00468
    GO TO 2240                               MODES 00469
  2230 CONTINUE                                MODES 00470
C
    DO 2800 NS=1,NSURF                      MODES 00471
    IF(NS.EQ.NSURF) GO TO 2230              MODES 00472
    IFILE = IVPSC                            MODES 00473
    REWIND IVPSC                            MODES 00474
    GO TO 2240                               MODES 00475
C
    DO 2600 NS=1,NSURF                      MODES 00476
    IF(NS.EQ.NSURF) GO TO 2230              MODES 00477
    IFILE = IVPSC                            MODES 00478
    REWIND IVPSC                            MODES 00479
    GO TO 2240                               MODES 00480
C
    DO 2800 NS=1,NSURF                      MODES 00481
    IF(NS.EQ.NSURF) GO TO 2230              MODES 00482
    IFILE = IVPSC                            MODES 00483
    REWIND IVPSC                            MODES 00484
    GO TO 2240                               MODES 00485
  2230 CONTINUE                                MODES 00486

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      IFILE = ITSLSC          MODES 00485
2240 CONTINUE          MODES 00486
C          MODES 00487
      DO 2700 NSL=1,NTSLOP  MODES 00488
C          MODES 00489
C          ZERO OUT THE ARRAY  MODES 00490
      DO 2250 I=1,500  MODES 00491
      DEFSL(1,I) = 0.0  MODES 00492
2250 CONTINUE          MODES 00493
C          MODES 00494
      IF(NS.EQ.2) GO TO 2325  MODES 00495
      NCH = NYBW  MODES 00496
      NC = 1  MODES 00497
      GO TO 2350  MODES 00498
2325 CONTINUE          MODES 00499
      NC = NYBW + 1  MODES 00500
      NCH = NYBW + NYBT  MODES 00501
      READ (IVPSC) DEFSL  MODES 00502
2350 CONTINUE          MODES 00503
C          MODES 00504
      DO 2500 J=NC,NCH  MODES 00505
      IST = IS(J)  MODES 00506
      NK = NOC(J) + IST - 1  MODES 00507
      JSUM = 0  MODES 00508
      ITROW = IST  MODES 00509
      IF(.NOT.COPLAN.AND.NS.EQ.2) ITROW = IST + IOVLAP  MODES 00510
      DO 2400 I=1,ITROW  MODES 00511
2400 JSUM = JSUM + JCC(I)  MODES 00512
      JSUM = JSUM - JCC(ITROW) + 1  MODES 00513
      READ (NT5,9015) (DOS(I),I=IST,NK)  MODES 00514
      DO 2500 I=IST,NK  MODES 00515
      IX = I  MODES 00516
      IF(.NOT.COPPLAN.AND.NS.EQ.2) IX = I + IOVLAP  MODES 00517
      ISUB = JSUM + J - JS(IX) - NC + 1  MODES 00518
      DEFSL(1,ISUB) = 1. + GAMC * DOS(I)  MODES 00519
      JSUM = JSUM + JCC(IX)  MODES 00520
2500 CONTINUE          MODES 00521
C          MODES 00522
      IF(NB.EQ.NSURF) GO TO 2550  MODES 00523
      WRITE (IVPSC) DEFSL  MODES 00524
      GO TO 2600  MODES 00525
2550 CONTINUE          MODES 00526
      CALL RDINIT  MODES 00527
      ITYPE = SHMIXED  MODES 00528
      M = 1  MODES 00529
      N = NBV  MODES 00530
      CALL WRTEHX(IFILE,MXWRIT,RANDOU,NFS,NHS,L9,NMR,LWS,2,ID,
1           DEFSL, ITYPE,M,N,PARM,IRR)  MODES 00531
      IF(IRR.NE.0) GO TO 0000  MODES 00532
C          MODES 00533
      2600 CONTINUE          MODES 00534
C          MODES 00535
      2700 CONTINUE          MODES 00536
      END FILE IFILE  MODES 00537
      REWIND IFILE  MODES 00538
2800 CONTINUE          MODES 00539
3000 CONTINUE          MODES 00540

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C                               MODES 00542
9015 FORMAT(7E10.0)           MODES 00544
      RETURN                   MODES 00545
C                               MODES 00546
C           INPUT DATA ERRORS  MODES 00547
C                               MODES 00548
C           6000 WRITE (NT6,9000) IDEG   MODES 00549
      GO TO 6199                 MODES 00550
  6005 WRITE (NT6,9000) IDEG, NPTS MODES 00551
      GO TO 6199                 MODES 00552
C                               MODES 00553
C           AN ERROR FROM READING OR WRITING A MATRIX FROM TAPE OR MODES 00554
C           DISK FILE OCCURRED. PRINT MESSAGES AND FLUSH MODES 00555
C                               MODES 00556
C           6010 CONTINUE            MODES 00557
      WRITE (NT6,9010) IGEOSC,IRR  MODES 00558
      WRITE (NT6,9011) MNAME       MODES 00559
      GO TO 6100                 MODES 00560
  6020 CONTINUE                MODES 00561
      WRITE (NT6,9020) MODESC,IRR  MODES 00562
      WRITE (NT6,9021) NM          MODES 00563
      GO TO 6100                 MODES 00564
C                               MODES 00565
  6030 CONTINUE                MODES 00566
      WRITE (NT6,9020) MODESC,IRR  MODES 00567
      WRITE (NT6,9022)             MODES 00568
C                               MODES 00569
      GO TO 6100                 MODES 00570
  6040 CONTINUE                MODES 00571
      WRITE (NT6,9020) MODESC,IRR  MODES 00572
      WRITE (NT6,9023)             MODES 00573
      GO TO 6100                 MODES 00574
C                               MODES 00575
  6050 CONTINUE                MODES 00576
      WRITE (NT6,9050) IGEOSC,IRR  MODES 00577
      WRITE (NT6,9051) NM          MODES 00578
      GO TO 6100                 MODES 00579
C                               MODES 00580
  6060 WRITE (NT6,9050) IAICSC,IRR MODES 00581
      WRITE (NT6,9051) NM          MODES 00582
      GO TO 6100                 MODES 00583
C                               MODES 00584
  6070 CONTINUE                MODES 00585
      WRITE (NT6,9070) IAICSC,IRR  MODES 00586
      WRITE (NT6,9071) NM          MODES 00587
      GO TO 6100                 MODES 00588
C                               MODES 00589
  6080 CONTINUE                MODES 00590
      WRITE (NT6,9080) ITSLSC,IRR  MODES 00591
      WRITE (NT6,9081) NSL         MODES 00592
C                               MODES 00593
C                               MODES 00594
  6100 CONTINUE                MODES 00595
      WRITE (NT6,9101) ID(1),ID(2) MODES 00596
      WRITE (NT6,9102) PARM,IPARM  MODES 00597
      WRITE (NT6,9103) NFS,NMS     MODES 00598
      WRITE (NT6,9104) ITYPE,M,N    MODES 005

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6199 CONTINUE                               MODES 00600
    WRITE(NT5,9900)                           MODES 00601
C                                           MODES 00602
    CALL FLUSH(1)                            MODES 00603
C                                           MODES 00604
7005 FORMAT(40 PREVIOUS MODES AND GEOMETRY HAVE BEEN SPECIFIED. *)   MODES 00605
7010 FORMAT(8GH0*** WARNING - PREVIOUS MODE SHAPES HAVE BEEN SPECIFIED, MODES 00606
    1 BUT GEOMETRY HAS CHANGED. *** )        MODES 00607
7015 FORMAT(8DH0*** ERROR - PREVIOUS MODE SHAPES HAVE BEEN SPECIFIED, B MODES 00608
    1UT THE GEOM HAS CHANGED. / 13X,41HPREVIOUS MODE SHAPES WERE AT BOX MODES 00609
    2 CENTERS., 26X,4H *** )                 MODES 00610
9000 FORMAT(43H0*** ERROR - SPECIFIED POLYNOMIAL DEGREE OF 15,           MODES 00611
    1 22H IS OUTSIDE LIMITS, OR, 16,24H IS TOO MANY POINTS *** )        MODES 00612
9010 FORMAT(53H0*** ERROR - WHILE READING THE GEOMETRY SCRATCH FILE A10 MODES 00613
    1, 15H, ERROR CODE = I4,4H *** )        MODES 00614
9011 FORMAT(5X,32HAN ATTEMPT WAS MADE TO READ THE A6, 8H MATRIX.//)      MODES 00615
9020 FORMAT(52H0*** ERROR - WHILE WRITING ON THE MODE SCRATCH FILE A10, MODES 00616
    1 15H, ERROR CODE = I4,4H *** )        MODES 00617
9021 FORMAT(5X,40HAN ATTEMPT WAS MADE TO WRITE MODE SHAPE I3,//)        MODES 00618
9022 FORMAT(5X,47HAN ATTEMPT WAS MADE TO WRITE THE POINTER ARRAY. // )    MODES 00619
9023 FORMAT(5X,41HAN ATTEMPT WAS MADE TO WRITE INDEX ARRAY. // )         MODES 00620
9030 FORMAT(57H0*** ERROR - WHILE WRITING THE COEFFICIENT ARRAY ON FILE MODES 00621
    1 A10,15H, ERROR CODE = I4,4H *** )    MODES 00622
9051 FORMAT(5X,44HAN ATTEMPT WAS MADE TO WRITE FOR MODE SHAPE I4 )       MODES 00623
9070 FORMAT(59H0*** ERROR - WHILE READING THE COEFFICIENT ARRAY FROM FI MODES 00624
    1LE A10,15H, ERROR CODE = I4,4H *** )    MODES 00625
9071 FORMAT(5X,43HAN ATTEMPT WAS MADE TO READ FOR MODE SHAPE I4 )        MODES 00626
9080 FORMAT(56H0*** ERROR - WHILE WRITING ON THE THICKNESS SLOPE FILE MODES 00627
    1 2X,A10,15H, ERROR CODE = ,I4,4H *** )    MODES 00628
9081 FORMAT(5X,45HAN ATTEMPT WAS MADE TO WRITE THICKNESS SLOPE I4,//)    MODES 00629
9500 FORMAT(1H1,8A10,// 46X,* MODE SHAPE NUMBER *,I3,                      MODES 00630
    1                                / 46X,*MACH NUMBER =*,F11.6,/46X,24(1H-),/ ) MODES 00631
9101 FORMAT(5X,*MATRIX ID = *, A10, I10)                                MODES 00632
9102 FORMAT(5X,*PARAMETERS *,10E11.3, /10X,* (INTEGER)*, I7, 9111 )    MODES 00633
9103 FORMAT(5X,*FILE SPACING = *,I3,* MATRIX SPACING = *,I3)            MODES 00634
9104 FORMAT(5X,*MATRIX TYPE -*,A10,* DIMENSIONED (*I4,24 X,I4,1H) )    MODES 00635
9900 FORMAT(40     ERROR OCCURRED IN MODES SECTION (MAIN PROGRAM).*)    MODES 00636
END                                     MODES 00637

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SUBROUTINE TAPMOD(NS,NM,DEFLS)
DIMENSION DEFLS(2,500), D(500), S(500)

C
C      THIS PROGRAM WILL READ MODE SHAPES FROM TAPE IN THE SAME
C      FORMAT AS PROGRAM TEV059. THE PROGRAM WILL BE REPLACED BY
C      THE AIR FORCE FOR ITS USE AND BOEING MAY MODIFY OR REPLACE
C      AS INPUT TAPES ARE MODIFIED OR REPLACED.
C
C      NS    - =1, FIRST PLANFORM
C              -=2, SECOND PLANFORM
C      NM    - MODE SHAPE NUMBER
C      DEFLS - ARRAY WHERE MODE SHAPES FOR ENTIRE PLANFORM(S) IS
C              STORED.
C
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,
1          B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,
2          MXBW,MXBBW,MYBW,MYBBW,MXBW,MYBWSW,MYBDSW,
3          IXBW,XCENTR
LOGICAL COPLAN
COMMON /GEOM2/ TLAX,TLAZ,PSIT,MXBT,MYBT,MYBBT,MXBST,MYBST,
1          MYBBST,IXBT,IXBST,CAPL
COMMON /FILES/ NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP,
1          IOWFSP,MODESC,IVPSC,IGEOBC,IWTFSC,IAICSC
COMMON /TAPEIO/ NFS,NMS,LS,NMR,TD(20),NID,ITYPE,LRS,LWS,M,N,
1          PARM(10),IRR
DIMENSION IFARM(10)
EQUIVALENCE (PARM,IFARM)
COMMON /INDEX/ IS(100),NOC(100),JS(100),JOC(100)

C
LOGICAL MXREAD,RANDIN
MXREAD = .FALSE.
RANDIN = .FALSE.

C
CALL RDINIT
IF(NM.NE.1) GO TO 710
IF(NS.EQ.1) REWIND INTAPE
NMS = 2
NFS = INFSP
710 CONTINUE
ITYPE = SHMIXED
CALL READMX(INTAPE,MXREAD,RANDIN,NFS,NMS,LS,NMR,1, NID, ID, ITYPE,
1          LRS, D, M, N, PARM, IRR)
IF(IRR.NE.0) GO TO 6020
CALL RDINIT
ITYPE = SHMIXED
CALL READMX(INTAPE,MXREAD,RANDIN,NFS,NMS,LS,NMR,1, NID, ID, ITYPE,
1          LRS, S, M, N, PARM, IRR)
IF(IRR.NE.0) GO TO 6020

C
IF(NS.EQ.2) GO TO 720
C
C      1 RST PLANFORM
NC = MYBW
NC = 1
GO TO 725
C
C      SECOND PLANFORM

```

TAPMOD 00002
TAPMOD 00003
TAPMOD 00004
TAPMOD 00005
TAPMOD 00006
TAPMOD 00007
TAPMOD 00008
TAPMOD 00009
TAPMOD 00010
TAPMOD 00011
TAPMOD 00012
TAPMOD 00013
TAPMOD 00014
TAPMOD 00015
GEOMTY 00002
GEOMTY 00003
GEOMTY 00004
GEOMTY 00005
GEOMTY 00006
GEOM2 00002
GEOM2 00003
FILES 00002
FILES 00003
TAPEIO 00002
TAPEIO 00003
TAPEIO 00004
TAPEIO 00005
TAPMOD 00020
TAPMOD 00021
TAPMOD 00022
TAPMOD 00023
TAPMOD 00024
TAPMOD 00025
TAPMOD 00026
TAPMOD 00027
TAPMOD 00028
TAPMOD 00029
TAPMOD 00030
TAPMOD 00031
TAPMOD 00032
TAPMOD 00033
TAPMOD 00034
TAPMOD 00035
TAPM 00036
TAPMOD 00037
TAPMOD 00038
TAPMOD 00039
TAPMOD 00040
TAPMOD 00041
TAPMOD 00042
TAPMOD 00043
TAPMOD 00044
TAPMOD 00045
TAPMOD 00046
TAPMOD 00047
TAPMOD 00048
TAPMOD 00049

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720 CONTINUE          TAPMOD 00050
    NC = NYBW + 1      TAPMOD 00051
    NCH = NYBW + NYBT   TAPMOD 00052
C
C      STORE DEFLECTIONS AND SLOPES
725 CONTINUE          TAPMOD 00053
    ITS = 0             TAPMOD 00054
    DO 730 J=NC,NCH     TAPMOD 00055
    IST = IS(J)          TAPMOD 00056
    NK = NOC(J) + IST - 1 TAPMOD 00057
    JSUM = 0              TAPMOD 00058
    DO 730 I=1,IST       TAPMOD 00059
    ITS = ITS + 1        TAPMOD 00060
    ISUB = JSUM + J - NC + 1 - JS(I) TAPMOD 00061
    DEFLS(1,ISUB) = D(ITS) TAPMOD 00062
    DEFLS(2,ISUB) = S(ITS) TAPMOD 00063
    JSUM= JSUM + JOC(I) TAPMOD 00064
730 CONTINUE          TAPMOD 00065
C
C      RETURN           TAPMOD 00066
C
C      AN ERROR DURING READING A MATRIX FROM TAPE OR
C      DISK FILE OCCURRED. PRINT MESSAGES AND FLUSH
C
6010 CONTINUE          TAPMOD 00067
6020 CONTINUE          TAPMOD 00068
    WRITE (NT6,9020) MODESC,IRR TAPMOD 00069
    WRITE (NT6,9021) NM TAPMOD 00070
6100 CONTINUE          TAPMOD 00071
    WRITE(NT6,9101) ID(1),ID(2) TAPMOD 00072
    WRITE(NT6,9102) PARM,IPARM TAPMOD 00073
    WRITE(NT6,9103) NFS,NMS TAPMOD 00074
    WRITE(NT6,9104) ITYPE,M,N TAPMOD 00075
    WRITE(NT6,9900) TAPMOD 00076
C
C      CALL FLUSH(1)     TAPMOD 00077
C
C
9020 FORMAT(54H*** ERROR - WHILE READING FROM THE MODE SCRATCH FILE
1 A10,15H, ERROR CODE = I4,4H *** )
9021 FORMAT(5X,39HAN ATTEMPT WAS MADE TO READ MODE SHAPE I3,//)
9101 FORMAT(5X,*MATRIX ID = *, A10, I10) TAPMOD 00078
9102 FORMAT(5X,*PARAMETERS *,10E11.3, /10X,*(INTEGER)*, I7, 9I11 ) TAPMOD 00079
9103 FORMAT(5X,*FILE SPACING = *,I3,* MATRIX SPACING = *,I3) TAPMOD 00080
9104 FORMAT(5X,*MATRIX TYPE -*,A10,* DIMENSIONED (*I4,2H X,I4,1H) ) TAPMOD 00081
9900 FORMAT(40      ERROR OCCURRED IN MODES SECTION (SUBROUTINE TAPMOD).* FTNX1 00044
1      )               FTNX1 00045
END                  TAPMOD 00098

```

```

SUBROUTINE ROPER          ROPER 00002
C
C      SUBROUTINE TO DETERMINE THE BOXES ON EACH ROW THAT SHOULD   ROPER 00003
C      HAVE MODE SHAPES.                                         ROPER 00004
C
C      IS(J) - ROW INDEX OF FIRST PLANFORM BOX FOR CHORD J.   ROPER 00005
C      NOC(J) - NUMBER OF PLANFORM BOXES ON CHORD J             ROPER 00006
C      JS(I) - COL INDEX OF FIRST PLANFORM BOX FOR SPAN I.    ROPER 00007
C      JOC(I) - NUMBER OF BOXES BETWEEN FIRST AND LAST PLANFORM ROPER 00008
C                  BOX ON SPAN I.                                ROPER 00009
C
C
C      COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF, ROPER 00010
C                         B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW, ROPER 00011
C                         MXBW,MXBBW,MYBW,MYBBW,MXBW,MYBSW,MYBBSW, ROPER 00012
C                         IXBW,XCENTR.                           ROPER 00013
C
C      LOGICAL COPLAN
C      COMMON /GEOM2 / TLAX,TLAZ,PSIT,MXBT,MYBT,MYBBT,MXBST,MYBST, ROPER 00014
C                         MYBBST,IXBT,IXBST,CAPL.                 ROPER 00015
C      COMMON /INDEX/ IS(100),NOC(100),JS(100),JOC(100)
C      IF(COPLAN) GO TO 100
C      NOH = MYBW
C      NOB = MXBW
C      GO TO 200
100 CONTINUE
      NOH = MYBW + MYBT
      NOB = MXBT
200 CONTINUE
      DO 500 I = 1,MXB
      JS(I) =0
      JOC(I)=0
      JCUT = 0
      DO 400 J=1,NOH
      IF(I.LT.IS(J)) GO TO 400
      ILAST = IS(J) + NOC(J) -1
      IF(I.GT.ILAST) GO TO 400
      IF(JS(I).NE.0) GO TO 300
      JS(I) = J
      IF(J.GT.MYBW) JS(I) = J - MYBW
300 CONTINUE
      IF(JCUT.EQ.1) GO TO 400
      JV = J
      IF(J.GT.MYBW) JV = J - MYBW
      IF(JV.LT.JS(I)) GO TO 350
      JOC(I) = JV-JS(I) +1
      GO TO 400
350 CONTINUE
      JOC(I) = JS(I)-JV + JOC(I)
      JS(I) = JV
      JCUT = 1
400 CONTINUE
425 CONTINUE
500 CONTINUE
C
C      CALCULATE FOR SECOND PLANFORM. THIS IS ONLY USED FOR   ROPER 00049
C      NON COPLANAR PLANFORMS.                                 ROPER 00050
      IF(NSURF.EQ.1) GO TO 1500
      ROPER 00051
      ROPER 00052
      ROPER 00053

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```

IF(COPLAN) GO TO 1500
NCH = MYBW + MYBT
JCH = MYBW + 1
IFBT = (IXBT-IXBW)/NSUBDV +1
IOVLAP = 0
IF(IFBT.LE.MXBW) IOVLAP = MXBW - IFBT + 1
DO 1000 I=IFBT,MXBТ
IX = I + IOVLAP
JS(IX) = 0
JOC(IX)= 0
DO 800 J=JCH,NCH
IF(I.LT.IS(J)) GO TO 800
ILAST = IS(J) + NOC(J) -1
IF(I.GT.ILAST) GO TO 800
IF(JS(IX).EQ.0) JS(IX) = J-MYBW
JV = J- MYBW
JOC(IX) = JV - JS(IX) + 1
800 CONTINUE
825 CONTINUE
1000 CONTINUE
1500 CONTINUE
RETURN
END

```

ROPER	00054
ROPER	00055
ROPER	00056
ROPER	00057
ROPER	00058
ROPER	00059
ROPER	00060
ROPER	00061
ROPER	00062
ROPER	00063
ROPER	00064
ROPER	00065
ROPER	00066
ROPER	00067
ROPER	00068
ROPER	00069
ROPER	00070
ROPER	00071
ROPER	00072
ROPER	00073
ROPER	00074
ROPER	00075
ROPER	00076

```

SUBROUTINE FITTER(M,N,X,Y,Z,C,CN,NDIM)
DIMENSION X(100), Y(100), Z(IDIM,100), C(IDIM,66)
DIMENSION AI(66), A(66,66), XP(11), YP(11)
DIMENSION VS(10)
LOGICAL COMPLX

C
C      M - DEGREE OF POLYNOMIAL EQUATION
C      N - NUMBER OF DATA POINTS TO FIT CURVE THROUGH
C      X - X COORDINATE OF DATA POINT
C      Y - Y COORDINATE OF DATA POINT
C      Z - Z COORDINATE OF DATA POINT
C      C - OUTPUT COEFFICIENT ARRAY
C      CN - SCALE FACTOR
C      CN - SCALE FACTOR
C      IDIM - INDICATOR OF REAL OR COMPLEX FUNCTION
C            = 1, FUNCTION IS REAL
C            = 2, FUNCTION IS COMPLEX
C            IF COMPLEX SET DIMENSIONS OF FUNCTION AND COEFFICIENTS
C            TO (IDIM * --- )
C
C      DETERMINE NUMBER OF COEFFICIENTS
C
EPS = 1.0E-04
COMPLX = .FALSE.
IF(IDIM.EQ.2) COMPLX = .TRUE.

C
C      SCALE DATA TO REDUCE MAGNITUDE OF MATRIX TERMS.
C      SHOULD AVOID BOMB OUTS DUE TO OVERFLOW CONDITIONS.
IF(CN.EQ.0) CN=1.0
IF(CN.EQ.1.1) GO TO 15
DO 5 I=1,N
X(I) = X(I)/CN
Y(I) = Y(I)/CN
5 CONTINUE
15 CONTINUE
XM = M + 1
XMR = XM/2.
NC = XM*XMR + XMR + EPS
2F(NC.LE.ND) GO TO 25
M = M-1
GO TO 15
25 CONTINUE
C
NAC = NC

C
DETERMINE THE MAXIMUM DEGREE THAT CAN BE COMPUTED IN
EACH DIRECTION AND SET UP ORDER OF SOLUTION.
C
NDV = 1
MDX = M
VS(1) = X(1)
DO F 1 I=1,N
DO 3 J=1,NDV
IF(X(I).EQ.VS(J)) GO TO 55
50 CONTINUE
NDV = NDV + 1
VS(NDV) = X(I)

```

FITTER	00002
FITTER	00003
FITTER	00004
FITTER	00005
FITTER	00006
FITTER	00007
FITTER	00008
FITTER	00009
FITTER	00010
FITTER	00011
FITTER	00012
FITTER	00013
FITTER	00014
FITTER	00015
FITTER	00016
FITTER	00017
FITTER	00018
FITTER	00019
FITTER	00020
FITTER	00021
FITTER	00022
FITTER	00023
FITTER	00024
FITTER	00025
FITTER	00026
FITTER	00027
FITTER	00028
FITTER	00029
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FITTER	00046
FITTER	00047
FITTER	00048
FITTER	00049
FITTER	00050
FITTER	00051
FITTER	00052
FITTER	00053
FITTER	00054
FITTER	00055
FITTER	00056
FITTER	00057
FITTER	00058

```

      IF(NCV-1.EQ.M) GO TO 65          FITTER 00059
55 CONTINUE                         FITTER 00060
60 CONTINUE                         FITTER 00061
      NDX = NCV -1                     FITTER 00062
65 CONTINUE                         FITTER 00063
C
      NDY = 1                          FITTER 00064
      NDY = M                          FITTER 00065
      VS(1) = Y(1)                     FITTER 00066
      DO 80 I=1,N                      FITTER 00067
      DO 70 J=1,NCV                   FITTER 00068
      IF(Y(I).EQ.VS(J)) GO TO 75       FITTER 00069
70 CONTINUE                         FITTER 00070
      NDY = NCV + 1                   FITTER 00071
      VS(NDY) = Y(I)                  FITTER 00072
      IF(NCV-1.EQ.M) GO TO 85         FITTER 00073
75 CONTINUE                         FITTER 00074
80 CONTINUE                         FITTER 00075
      NDY = NCV - 1                   FITTER 00076
85 CONTINUE                         FITTER 00077
C
      ITOT = NC +1                   FITTER 00078
      ITOT1 = ITOT                    FITTER 00079
      IF(COMPLX) ITOT = ITOT + 1      FITTER 00080
C
C      ZERO OUT THE A ARRAY          FITTER 00081
C
      DO 95 I=1,NC                   FITTER 00082
      C(1,I) = 0.0                   FITTER 00083
      IF(.NOT.COMPLX) GO TO 90        FITTER 00084
      C(2,I) = 0.0                   FITTER 00085
90 CONTINUE                         FITTER 00086
      DO 95 J=1,ITOT                 FITTER 00087
95 A(I,J) = 0.0                     FITTER 00088
C
C      DETERMINE DEVIATION EQUATION AND SQUARE THE EQUATION
C
      AI(1) =1.0                     FITTER 00089
      XP(1) =1.0                     FITTER 00090
      YP(1) =1.0                     FITTER 00091
      MM = M + 1                     FITTER 00092
      DO 200 K=1,N                   FITTER 00093
      DO 10 L=2,MM                   FITTER 00094
      XP(L) = XP(L-1)*X(K)           FITTER 00095
      YP(L) = YP(L-1)*Y(K)
10 CONTINUE                         FITTER 00096
C
      I = 1                          FITTER 00097
      DO 40 L=2,MM                   FITTER 00098
      DO 20 LL=1,L                   FITTER 00099
      IL= L - LL +1                 FITTER 00100
      IF(LL-1.GT.NDY) GO TO 30       FITTER 00101
      IF(IL-1.GT.NDX) GO TO 20       FITTER 00102
      I = I + 1                     FITTER 00103
      AI(I) = XP(IL)*YP(LL)
20 CONTINUE                         FITTER 00104
30 CONTINUE                         FITTER 00105

```

```

40 CONTINUE
    AI(I+1) = Z(1,K)
    IF(COMPLX) AI(I+2) = Z(2,K)
    IF(K.GT.1) GO TO 45
    NC = I
    ITOT = NC + 1
    ITOT1 = ITOT
    IF(COMPLX) ITOT = ITOT + 1
45 CONTINUE
C
    DO 1100 I=1,NC
    DO 1100 J=I,ITOT
        ASAV = AI(I)*AI(J)
        A(I,J)=A(I,J)+ASAV
1100 CONTINUE
200 CONTINUE
C
C      SQUARE ROOT METHOD
C      INTERMEDIATE MATRIX
    DO 1200 I=1,NC
    IM1 = I-1
    TMP=0.0
    IF(I.EQ.1) GO TO 1150
    DO 1120 L=1,IM1
1120 TMP= TMP+A(L,I)**2
1150 CONTINUE
    T = A(I,I) - TMP
    IF(T.GT.EPS) GO TO 4
    A(I,I) = 0.0
    GO TO 1200
4 CONTINUE
    A(I,I) = SQRT(T)
    IF(A(I,I).GT.EPS) GO TO 1155
    A(I,ITOT) = 0.0
    GO TO 1200
1155 CONTINUE
C
    JS = I+1
    DO 1180 . = JS,ITOT
    TMP= 0.0
    IF(I.EQ.1) GO TO 1175
    DO 1160 L=1,IM1
1160 TMP = TMP + A(L,I)*A(L,J)
1175 A(I,J) =(A(I,J)-TMP)/A(I,I)
1180 CONTINUE
1200 CONTINUE
C
C      BACK SUBSTITUTE FOR COEFFICIENTS
    DO 1400 K=1,NC
    I = NC - K + 1
    TMP1=I 1
    TMP2 = 0.0
    TMP3 = 0.0
    IF(A(I,I).GT.EPS) GO TO 1325
    C(I,I) = 0.0
    IF(COMPLX) C(2,I) = 0.0

```

FITTER 00116
 FITTER 00117
 FITTER 00118
 FITTER 00119
 FITTER 00120
 FITTER 00121
 FITTER 00122
 FITTER 00123
 FITTER 00124
 FITTER 00125
 FITTER 00126
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 FITTER 00164
 FITTER 00165
 FITTER 00166
 FITTER 00167
 FITTER 00168
 FITTER 00169
 FITTER 00170
 FITTER 00171
 FITTER 00172

```

GO TO 1400
1325 CONTINUE
IF(I.EQ.NC) GO TO 1375
DO 1350 L=IP1,NC
TMP1 = TMP1 + A(I,L)*C(1,L)
IF(.NOT.COMPLX) GO TO 1350
TMP2 = TMP2 + A(I,L)*C(2,L)
1350 CONTINUE
1375 CONTINUE
C(1,I) = (A(I,ITOT1)-TMP1)/A(I,I)
IF(.NOT.COMPLX) GO TO 1400
C(2,I) = (A(I,ITOT) - TMP2)/A(I,I)
1400 CONTINUE
C
C      REORDER THE COEFFICIENTS IN CORRECT POWERS
C      OF X AND Y.
C
IF(NAC.EQ.NC) GO TO 1475
C
IZ = 1
I = 1
DO 1440 L=2,NN
DO 1420 LL=1,L
IL = L - LL + 1
I = I + 1
IF(LL-1.LE.MDY.AND.IL-1.LE.MDX) GO TO 1410
X(I) = 0.0
Y(I) = 0.0
GO TO 1420
1410 CONTINUE
IZ = IZ + 1
X(I) = C(1,IZ)
IF(COMPLX) Y(I) = C(2,IZ)
1420 CONTINUE
1440 CONTINUE
C
DO 1450 I=2,NAC
C(1,I) = X(I)
IF(COMPLX) C(2,I) = Y(I)
1450 CONTINUE
1475 CONTINUE
C
C      ELIMINATE THE SCALE FACTOR FROM THE COEFFICIENTS.
C
IF(CN.EQ.1.0) GO TO 1700
I=1
CP= 1.0/CN
DO 1600 L1=2,NN
DO 1500 L2=1,L1
I = I+1
C(1,I) = C(1,I)*CP
C(2,I) = C(2,I)*CP
1500 CONTINUE
CP= CP/CN
1600 CONTINUE
1700 CONTINUE

```

FITTER 00173
 FITTER 00174
 FITTER 00175
 FITTER 00176
 FITTER 00177
 FITTER 00178
 FITTER 00179
 FITTER 00180
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 FITTER 00182
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 FITTER 00209
 FITTER 00210
 FITTER 00211
 FITTER 00212
 FITTER 00213
 FITTER 00214
 FITTER 00215
 FITTER 00216
 FITTER 00217
 FITTER 00218
 FITTER 00219
 FITTER 00220
 FITTER 00221
 FITTER 00222
 FITTER 00223
 FITTER 00224
 FITTER 00225
 FITTER 00226
 FITTER 00227
 FITTER 00228
 FITTER 00229

C
C THE C ARRAY NOW CONTAINS THE COEFFICIENTS.
C
RETURN
END

FITTER 00230
FITTER 00231
FITTER 00232
FITTER 00233
FITTER 00234

```

SUBROUTINE MODOUT(DEFSL,JS,JOC,NROWS,NM,IOMLAP)
DIMENSION DEFSL(2,500),JS(100),JOC(100)
DIMENSION JPH(15),DS(50)
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,
1           B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,
2           MXBW,MXBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW,
3           IXBW,XCENTR
LOGICAL COPLAN
COMMON /FILES/ NTS,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP,
1           IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC
COMMON /PROBLM/ XMACH,NNODES,NTSLOP,NKVALS,SMOOTH,NDEG,CDFIT,
1           EXAIC,SUBDV,PLYWOOD
LOGICAL SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD
DIMENSION SLOMAT(3)
EQUIVALENCE (SLOMAT(2),IMAT)
DATA SLOMAT/10H(I3,3X),10H      0,10H,15F7.3) /
DATA IMAT1,IMAT2 /10H          0, 10.      -0 /

C
C      MODOUT WILL PRINT THE MODE SHAPES OUT IN ROW/COLUMN FORM
C
C      DEFSL - MOD SHAPES IN INTERNAL STORAGE
C      JS(I) - FIRST COLUMN FOR WHICH THERE IS A MODE SHAPE ON ROW I
C      JOC(I) - NUMBER OF BOXES BETWEEN FIRST AND LAST PLANFORM
C                  BOX ON ROW I
C      NROWS - NUMBER OF ROWS
C      NM    - MODE SHAPE NUMBER
C      IOMLAP - NUMBER OF BOXES OVERLAP BETWEEN PLANFORMS FOR
C                  NON-COPLANAR SURFACES

C
C      IMAT = 10H          0
C      NSETS = 0
DO 50 I=1,NROWS
JL = JS(I)+JOC(I)-1
IF(NSETS.LT.JL) NSETS=JL
50 CONTINUE
NSETS = (NSETS-1)/15 + 1
DO 1000 NP=1,2

C
C      FIND LARGEST VALUE
C
C      VALUE = 0.0
DO 100 L=1,500
AVAL = ABS(DEFSL(NP,L))
IF(AVAL.GT.VALUE) VALUE = AVAL
100 CONTINUE
POW = 1
TSCALE = 10.
IF(VALUE.GE.10.) TSCALE = 0.1
DO 110 N=1,8
POW = POW *TSCALE
TTEN = VALUE * POW
IF(TTEN.GE.10.) GO TO 110
IF(TTEN.LT.1. ) GO TO 110
NPA = N
GO TO 115
110 CONTINUE
NPA= 0

```

MODOUT	00002
MODOUT	00003
MODOUT	00004
GEOMTY	00002
GEOMTY	00003
GEOMTY	00004
GEOMTY	00005
GEOMTY	00006
FILES	00002
FILES	00003
PROBLM	00002
PROBLM	00003
PROBLM	00004
MODOUT	00008
FTND	00046
MODOUT	00009
MODOUT	00011
MODOUT	00012
MODOUT	00013
MODOUT	00014
MODOUT	00015
MODOUT	00016
MODOUT	00017
MODOUT	00018
MODOUT	00019
MODOUT	00020
MODOUT	00021
MODOUT	00022
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MODOUT	00044
MODOUT	00045
MODOUT	00046
MODOUT	00047
MODOUT	00048
MODOUT	00049
MODOUT	00050
MODOUT	00051

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115 CONTINUE                               MODOUT 00052
    IF(VALUE.LE.1.00) GO TO 120               MODOUT 00053
C                                         MODOUT 00054
C     THE ARRAY MUST BE SCALED DOWN.        MODOUT 00055
C                                         MODOUT 00056
C     NPX = -NPA                           MODOUT 00057
C     IMAT = IMAT2                         MODOUT 00058
C     GO TO 122                           MODOUT 00059
C                                         MODOUT 00060
C     THE ARRAY MUST BE SCALED UP.         MODOUT 00061
C                                         MODOUT 00062
120 CONTINUE                               MODOUT 00063
    NPX = NPA                            MODOUT 00064
    IMAT = IMAT1                         MODOUT 00065
122 CONTINUE                               MODOUT 00066
    IMAT = IMAT + NPA                     MODOUT 00067
    IF(NP.EQ.2) GO TO 124                 MODOUT 00068
    WRITE (NT6,9005) NPX                  MODOUT 00069
    GO TO 125                           MODOUT 00070
124 CONTINUE                               MODOUT 00071
    WRITE (NT6,9010) NPX                  MODOUT 00072
125 CONTINUE                               MODOUT 00073
    DO 900 JPS=1,JSETS                  MODOUT 00074
    JBASE = (JPS-1)*15                   MODOUT 00075
    DO 150 JC=1,15                      MODOUT 00076
150 JFH(JC) = JC + JBASE                MODOUT 00077
    JL=15                                MODOUT 00078
    IF(JPS.EQ.JSETS) JL= NSETS - 15*(JSETS-1)
    WRITE(NT6,9015) (JFH(J),J=1,JL)      MODOUT 00079
    ION = 0                                MODOUT 00080
    ITOT = 0                                MODOUT 00081
    DO 800 I=1,NROWS                     MODOUT 00082
C                                         MODOUT 00083
C     ZERO OUT PRINTY ARRAY              MODOUT 00084
    DO 200 J=1,50                        MODOUT 00085
200 DS(J) = 0.0                          MODOUT 00086
C                                         MODOUT 00087
C     PUT THE VALUES INTO PRINT ARRAY   MODOUT 00088
    JI = JS(I)                           MODOUT 00089
    IF(JI.EQ.0) GO TO 800                 MODOUT 00090
    JL = JS(I) +JOC(I) -1                MODOUT 00091
    DO 300 J=JI,JL                      MODOUT 00092
    ITOT = ITOT+1                       MODOUT 00093
300 DS(J) = DEFLS(NP,ITOT)              MODOUT 00094
C                                         MODOUT 00095
C     PRINT THE ONES IN THIS SET       MODOUT 00096
    JIP=(JPS-1)*15 +1                   MODOUT 00097
    JIL= JIP +14                        MODOUT 00098
    IF(JL.LT.JIP) GO TO 800             MODOUT 00099
    IF(JI.GT.JIL) GO TO 800             MODOUT 00100
    IF(JIL.GT.JL) JIL=JL                MODOUT 00101
    M = I                                MODOUT 00102
    IF(I .NE.0) GO TO 350               MODOUT 00103
    IF(I .LE.MXBW) GO TO 350           MODOUT 00104
    M = I-IONLAP                         MODOUT 00105
    IF(ION.NE.0) GO TO 350               MODOUT 00106
    ION = 1                                MODOUT 00107
                                         MODOUT 00108

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```
      WRITE(NT6,9015)                               MODOUT 00109
350 CCONTINUE                                     MODOUT 00110
      WRITE(NT6,SLOMAT) M,(DS(J),J=1,IP,JIL)       MODOUT 00111
600 CCONTINUE                                     MODOUT 00112
900 CONTINUE                                      MODOUT 00113
1000 CONTINUE                                     MODOUT 00114
      RETURN                                         MODOUT 00115
9005 FORMAT(1HC,46X,*DEFLECTIONS X 1.0E *,I2,/ 47X,21(1H-)) MODOUT 00116
9010 FORMAT(1HO,//47X,* SLOPES X 1.0E *,I2,/ 47X,21(1H-)) MODOUT 00117
9015 FORMAT(1HO,5X,15(14,3X))                   MODOUT 00118
9020 FORMAT(I3,3X,15F7.3)                        MODOUT 00119
      END                                            MODOUT 00120
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SUBROUTINE PRECOF(IDEGL,A,IFR)
COMMON /FILES/ NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP,
1           IOLFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC
DIMENSION A(21),BLNK(7),BNK2(6)
EQUIVALENCE (BLNK(2),BNK2(1))
DIMENSION IXP(7),IYP(7)
DATA      BLNK / 7*1H /
C
C      THIS SUBROUTINE PRINTS THE COEFFICIENTS USED IN THE
C      POLYNOMIAL EQUATION USED IN CALCULATION OF MODE SHAPES
C
C      IDEG - DEGREE OF POLYNOMIAL EQUATION
C      A     - ARRAY OF COEFFICIENTS
C      IFR   - FLAG INDICATING HOW COEFFICIENTS ARE OBTAINED
C              =1, READ FROM CARDS
C              =2, FROM LEAST SQUARES SURFACE FIT
C
C      IF(IFR.EQ.1) WRITE(NT6,9055) A(1)
C      IF(IFR.EQ.2) WRITE(NT6,9065) A(1)
C
C      IF(IDEGL.EQ.0) GO TO 550
INDEX = 2
DO 520 I=1,IDEGL
NCL = I+1
LDEX = INDEX + I
DO 541 NXP=1,NCL
IXP(NXP) = NCL-NXP
IYP(NXP) = NXP-1
541 CONTINUE
WRITE(NT6,9060) (BLNK(NXP),IXP(NXP),IYP(NXP),NXP=1,NCL)
WRITE(NT6,9061) (BNK2(NXP),NXP=1,NCL)
WRITE(NT6,9062) (A(J),J=INDEX,LDEX)
INDEX = LDEX + 1
520 CONTINUE
550 CONTINUE
RETURN
9055 FORMAT(13X,*MODAL POLYNOMIAL COEFFICIENTS* 20X,*FROM CARD INPUT*
1 /13X,29(1H-) //0 CONSTANT* /2X,10.1H-/) E12.4 )
9060 FORMAT(1H0,6(A1,4HX ** I1,4HY ** I1,3X) )
9061 FORMAT(1X,6(A1,10H-----,3X) )
9062 FORMAT(E12.4,5E14.4)
9063 FORMAT(1H0,12X,*MODAL POLYNOMIAL COEFFICIENTS*/13X,*BY LEAST SQUAR PRECOF 00042
1ES SURFACE FIT*/13X,29(1H-)//0 CONSTANT* /2X,10(1H-)/ E12.4 ) PRECOF 00043
END          PRECOF 00044

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ITYPE = SHMIXED                                VICMAIN 00095
MXARRY = GHATAB                                VICMAIN 00096
CALL REACHX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,500,NID,ID,ITYPE,
1           LRS, ATAB,M,N,PARM,IRR)               VICMAIN 00097
IF(IRR.NE.0) GO TO 6060                         VICMAIN 00098
REWIND NSPAIC                                     VICMAIN 00100
C
C           IF(NV.NE.1) GO TO 555                   VICMAIN 00101
C
C           PRINT THE TABLE OF CONTENTS            VICMAIN 00102
WRITE (NT6,9215)
9215 FORMAT(1H1,14X,*SPATIAL AIC TAPE TABLE OF CONTENTS* /15X,34(1H-),
1   // 5X,**NO.* 4X,**MACH* 6X,*K1-VALUE* 6X,*ERROR* 4X,*SIZE*,5X,
2   *YBAR* 5X,*ZBAR* / )
NDELT = 0                                         VICMAIN 00107
DO 550 I=1,NKST                                 VICMAIN 00108
IF(AMACH(I).GT.0) GO TO 525                     VICMAIN 00109
NDELT = NDELT +1                               VICMAIN 00110
GO TO 550                                         VICMAIN 00111
525 WRITE (NT6,9020) I, AMACH(I),AKVAL(I),AERR(I),ISIZE(I),YBARS(I),
1           VERTS(I)                           VICMAIN 00112
550 CONTINUE                                     VICMAIN 00113
WRITE (NT6,9025) NDELT                         VICMAIN 00114
C
C           SEARCH FOR MATRICES WITH CORRECT K-VALUE, MACH, ERROR, SIZE,
C           AND YBAR.                            VICMAIN 00115
C
C           555 CONTINUE                         VICMAIN 00116
IVAL = 0                                         VICMAIN 00117
NBIZ = 0                                         VICMAIN 00118
DO 600 I=1,NKST                                 VICMAIN 00119
IF(ABS(AMACH(I)-XMACH).GT.1.0E-05) GO TO 600
IF(ABS(AKVAL(I)-K1) .GT.1.0E-07) GO TO 600
IF(AERR(I).GT.ERR) GO TO 600
IF(ABS(VERTS(I)-EL).GT.1.0E-04) GO TO 600
IF(ABS(YBARS(I)-YBAR) .GT.1.0E-04) GO TO 600
C
C           THERE IS A GOOD MATRIX ON TAPE. DETERMINE IF SIZE IS ADEQUATE
C
C           IF(IVAL.NE.0) GO TO 575                 VICMAIN 00122
IF(NROWS.GT.ISIZE(I)+10.AND.ERR.GT.AERR(I)) GO TO 600
IVAL = I                                         VICMAIN 00123
NBIZ = ISIZE(I)                                VICMAIN 00124
575 CONTINUE                                     VICMAIN 00125
IF(ISIZE(I).GE.NROWS) GO TO 600                VICMAIN 00126
C
C           THE SIZE IS NOT LARGE ENOUGH. SEE IF THIS IS LARGER THAN ANY
C           PREVIOUS MATRIX.                      VICMAIN 00127
C
C           IF(ISIZE(I).LE.NBIZ) GO TO 600          VICMAIN 00128
IF(NROWS.GT.ISIZE(I)+10.AND.ERR.GT.AERR(I)) GO TO 600
IVAL = I                                         VICMAIN 00129
NBIZ = ISIZE(I)                                VICMAIN 00130
600 CONTINUE                                     VICMAIN 00131
C
C           DETERMINE IF THERE WAS A MATRIX ON TAPE THAT COULD BE ENLARGED
C           IF(IVAL.EQ.0) GO TO 25                  VICMAIN 00132
VICMAIN 00133
VICMAIN 00134
VICMAIN 00135
VICMAIN 00136
VICMAIN 00137
VICMAIN 00138
VICMAIN 00139
VICMAIN 00140
VICMAIN 00141
VICMAIN 00142
VICMAIN 00143
VICMAIN 00144
VICMAIN 00145
VICMAIN 00146
VICMAIN 00147
VICMAIN 00148
VICMAIN 00149
VICMAIN 00150
VICMAIN 00151

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C          CALCULATE 2 PLANAR AICS IF SUBDIVISION IS APPLIED.          VICKMAIN 00038
C          NPK = 1          VICKMAIN 00039
C          IF(NSUBDV.GT.1) NPK = 2          VICKMAIN 00040
C          NVCS = NSPAK + NPK          VICKMAIN 00041
C          DO 1000 NV = 1,NVCS          VICKMAIN 00042
C          VICKMAIN 00043
C          SET K1 VALUE IF SUBDIVIDED AIC          VICKMAIN 00044
C          VICKMAIN 00045
C          IF(NPK.EQ.2.AND.NV.EQ.NVCS) K1= K1/FLOAT(NSUBDV)          VICKMAIN 00046
C          VICKMAIN 00047
C          LENZ = LKERNEL          VICKMAIN 00048
C          IF(NV.EQ.NVCS.AND.NPK.EQ.2) LENZ = LSKERN          VICKMAIN 00049
C          DO 100 I=1,LENZ          VICKMAIN 00050
C          C(I) = (0.,0.)          VICKMAIN 00051
C          W(I) = (0.,0.)          VICKMAIN 00052
C          V(I) = (0.,0.)          VICKMAIN 00053
C          100 CONTINUE          VICKMAIN 00054
C          VICKMAIN 00055
C          110 CONTINUE          VICKMAIN 00056
C          IF(NV.GT.NVCS-NPK) GO TO 10          VICKMAIN 00057
C          VICKMAIN 00058
C          READ MUASIC ARRAY FROM IGEOSC. FIRST FILE MUST BE SKIPPED          VICKMAIN 00059
C          PRIOR TO FIRST READ.          VICKMAIN 00060
C          VICKMAIN 00061
C          CALL RDINIT          VICKMAIN 00062
C          IF(NV.NE.1) GO TO 200          VICKMAIN 00063
C          REWIND IGEOSC          VICKMAIN 00064
C          NFS = 1          VICKMAIN 00065
C          200 CONTINUE          VICKMAIN 00066
C          ITYPE = SHMIXED          VICKMAIN 00067
C          MXARRY = GMUASIC          VICKMAIN 00068
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,          VICKMAIN 00069
C          1           LRS,MUASIC,M,NROWS,FARM,IRR)          VICKMAIN 00070
C          IF(IRR.NE.0) GO TO 6010          VICKMAIN 00071
C          VICKMAIN 00072
C          YBAR = PARM(4)          VICKMAIN 00073
C          EL = PARM(5)          VICKMAIN 00074
C          NN = 1          VICKMAIN 00075
C          VICKMAIN 00076
C          DETERMINE IF SPATIAL AICS ARE ON TAPE AND GET THEM          VICKMAIN 00077
C          IF POSSIBLE.          VICKMAIN 00078
C          VICKMAIN 00079
C          NKST = 0          VICKMAIN 00080
C          IF(.NOT.OSPAIC) GO TO 25          VICKMAIN 00081
C          VICKMAIN 00082
C          AICS ARE ON TAPE. GET TABLE OF CONTENTS.          VICKMAIN 00083
C          REWIND NSPAIC          VICKMAIN 00084
C          CALL RDINIT          VICKMAIN 00085
C          NFS = 1          VICKMAIN 00086
C          ITYPE = SHMIXED          VICKMAIN 00087
C          MXARRY = GHTAB          VICKMAIN 00088
C          CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,500,NID,ID,ITYPE,          VICKMAIN 00089
C          1           LRS, TAB,NKST,N,FARM,IRR)          VICKMAIN 00090
C          IF(IRR.NE.0) GO TO 6060          VICKMAIN 00091
C          VICKMAIN 00092
C          CALL RDINIT          VICKMAIN 00093
C          VICKMAIN 00094

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C          CALCULATE 2 PLANAR AICS IF SUBDIVISION IS APPLIED.      VICMAIN 00038
C          NPK = 1                                              VICMAIN 00039
C          IF(NSUBDV.GT.1) NPK = 2                                VICMAIN 00040
C          NVCS = NSPATK + NPK                                 VICMAIN 00041
C          DO 1000 NV = 1,NVCS                                VICMAIN 00042
C
C          SET K1 VALUE IF SUBDIVIDED AIC                      VICMAIN 00043
C
C          IF(NPK.EQ.2.AND.NV.EQ.NVCS) K1= K1/FLOAT(NSUBDV)    VICMAIN 00044
C
C          LENZ = LKERNEL                                     VICMAIN 00045
C          IF(NV.EQ.NVCS.AND.NPK.EQ.2) LENZ = LSKERN        VICMAIN 00046
C          DO 100 I=1,LENZ                                  "ICMAIN 00047
C          C(I) = (0.,0.)                                VICMAIN 00048
C          W(I) = (0.,0.)                                VICMAIN 00049
C          V(I) = (0.,0.)                                VICMAIN 00050
C          100 CONTINUE                               VICMAIN 00051
C
C          100 CONTINUE                               VICMAIN 00052
C          IF(NV.GT.NVCS-NPK) GO TO 10                  VICMAIN 00053
C
C          READ MUASIC ARRAY FROM IGEOSC. FIRST FILE MUST BE SKIPPED
C          PRIOR TO FIRST READ.                         VICMAIN 00054
C
C          CALL RDINIT                                VICMAIN 00055
C          IF(NV.NE.1) GO TO 200                         VICMAIN 00056
C          REWIND IGEOSC                            VICMAIN 00057
C          NFS = 1                                    VICMAIN 00058
C          200 CONTINUE                               VICMAIN 00059
C          ITYPE = SHMIXED                           VICMAIN 00060
C          MXARRY = GHMUASIC                         VICMAIN 00061
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
C          1           LRS,MUASIC,M,NRMS,FARM,IRR)      VICMAIN 00062
C          IF(IRR.NE.0) GO TO 6010                     VICMAIN 00063
C
C          YBAR = PARM(4)                            VICMAIN 00064
C          EL = PARM(5)                             VICMAIN 00065
C          NN = 1                                    VICMAIN 00066
C
C          DETERMINE IF SPATIAL AICS ARE ON TAPE AND GET THEM
C          IF POSSIBLE.                            VICMAIN 00067
C
C          NKST = 0                                 VICMAIN 00068
C          IF(.NOT.QSPAIC) GO TO 25                 VICMAIN 00069
C
C          AICS ARE ON TAPE. GET TABLE OF CONTENTS.      VICMAIN 00070
C          REWIND NSPATC                            VICMAIN 00071
C          CALL RDINIT                            VICMAIN 00072
C          NFS = 1                                VICMAIN 00073
C          ITYPE = SHMIXED                           VICMAIN 00074
C          MXARRY = GHTAB                            VICMAIN 00075
C          CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,500,NID,ID,ITYPE,
C          1           TAB,NKST,N,FARM,IRR)            VICMAIN 00076
C          IF(IRR.NE.0) GO TO 6060                   VICMAIN 00077
C
C          CALL RDINIT

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C          THERE IS A MATRIX THAT CAN BE ENLARGED.          VICMAIN 00152
C          AMACH(IVAL) = -AMACH(IVAL)                      VICMAIN 00153
C          WRITE (NT6,9030) IVAL, AERR(IVAL),ISIZE(IVAL),NROWS   VICMAIN 00154
C          VICMAIN 00155
C          VICMAIN 00156
C          VICMAIN 00157
C          VICMAIN 00158
C          VICMAIN 00159
C          VICMAIN 00160
C          VICMAIN 00161
C          VICMAIN 00162
C          VICMAIN 00163
C          VICMAIN 00164
C          VICMAIN 00165
C          VICMAIN 00166
C          VICMAIN 00167
C          VICMAIN 00168
C          VICMAIN 00169
C          VICMAIN 00170
C          VICMAIN 00171
C          VICMAIN 00172
C          VICMAIN 00173
C          VICMAIN 00174
C          VICMAIN 00175
C          VICMAIN 00176
C          VICMAIN 00177
C          VICMAIN 00178
C          VICMAIN 00179
C          VICMAIN 00180
C          VICMAIN 00181
C          VICMAIN 00182
C          VICMAIN 00183
C          VICMAIN 00184
C          VICMAIN 00185
C          VICMAIN 00186
C          VICMAIN 00187
C          VICMAIN 00188
C          VICMAIN 00189
C          VICMAIN 00190
C          VICMAIN 00191
C          VICMAIN 00192
C          VICMAIN 00193
C          VICMAIN 00194
C          VICMAIN 00195
C          VICMAIN 00196
C          VICMAIN 00197
C          VICMAIN 00198
C          VICMAIN 00199
C          VICMAIN 00200
C          VICMAIN 00201
C          VICMAIN 00202
C          VICMAIN 00203
C          VICMAIN 00204
C          VICMAIN 00205
C          VICMAIN 00206
C          VICMAIN 00207
C          VICMAIN 00208

C          SPACE TO CORRECT ARRAY ON TAPE
CALL RDINIT
NMS = (IVAL-1)*4
ITYPE = SHMIXED
MXARRY = GHMUTWO
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,MUTWO,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6060

C          CALL RDINIT
ITYPE = SHMIXED
MXARRY = GH C
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS, C, M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6060

C          CALL RDINIT
ITYPE = SHMIXED
MXARRY = GH W
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS, W, M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6060

C          CALL RDINIT
ITYPE = SHMIXED
MXARRY = GH V
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS, V, M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6060

C          GO TO 25
600 CONTINUE

C          THERE IS A GOOD MATRIX ON TAPE. READ THE TAPE, PRINT MESSAGE,
C          MAIL RESUME ON THIS ONE ANYWAY.
C          VICMAIN 00190
C          VICMAIN 00191
C          VICMAIN 00192
C          VICMAIN 00193
C          VICMAIN 00194
C          VICMAIN 00195
C          VICMAIN 00196
C          VICMAIN 00197
C          VICMAIN 00198
C          VICMAIN 00199
C          VICMAIN 00200
C          VICMAIN 00201
C          VICMAIN 00202
C          VICMAIN 00203
C          VICMAIN 00204
C          VICMAIN 00205
C          VICMAIN 00206
C          VICMAIN 00207
C          VICMAIN 00208

C          SET NSIZ EQUAL TO NROWS SO THAT MATRIX WILL NOT BE WRITTEN
C          ON TAPE
NSIZ = NROWS
CALL RDINIT
NMS = (I-1)*4
ITYPE = SHMIXED
MXARRY = GHMUTWO
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,MUTWO,M,N,PARM,IRR)
IF(' ? .NE.0) GO TO 6060

C          CALL RDINIT
ITYPE = SHMIXED
MXARRY = GH C
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,

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1           LRS,C,M,N, PARM,IRR)          VICHMAIN 00209
IF(IRR.NE.0) GO TO 6060                      VICHMAIN 00210
C
CALL RDINIT                                     VICHMAIN 00211
ITYPE = SHMIXED                                VICHMAIN 00212
MXARRY = SH W                                    VICHMAIN 00213
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,W,M,N, PARM,IRR)          VICHMAIN 00214
IF(IRR.NE.0) GO TO 6060                      VICHMAIN 00215
C
CALL RDINIT                                     VICHMAIN 00216
ITYPE = SHMIXED                                VICHMAIN 00217
MXARRY = SH V                                    VICHMAIN 00218
CALL READMX(NSPAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,V,M,N, PARM, IRR)          VICHMAIN 00219
IF(IRR.NE.0) GO TO 6060                      VICHMAIN 00220
WRITE (NTS,6005) I, AERR(I)                  VICHMAIN 00221
GO TO 25                                         VICHMAIN 00222
10 CONTINUE                                       VICHMAIN 00223
YBAR = 0.0                                       VICHMAIN 00224
EL = 0.0                                         VICHMAIN 00225
C
DETERMINE THE SIZE AND LOCATIONS OF THE PLANAR AIC ARRAYS.
THE UNSUBDIVIDED WILL BE CALCULATED FIRST AND STORED IN      VICHMAIN 00226
THE PROPER PLACE IN BLANK COMMON.                           VICHMAIN 00227
THE SUBDIVIDED WILL BE CALCULATED SECOND, OVERLAYING SOME OF      VICHMAIN 00228
THE UNSUBDIVIDED NUMBERS.                                 VICHMAIN 00229
VICHMAIN 00230
C
LSKERN = SIZE OF UNSUBDIVIDED OR SUBDIVIDED ARRAY ALONE.      VICHMAIN 00231
ISUB  = NUMBER OF ROWS ON UNSUBDIVIDED AIC THAT SUBDIVIDED      VICHMAIN 00232
AIC ARRAY WILL OVERLAY                                VICHMAIN 00233
LTOBKN = NUMBER OF BOXES OF UNSUBDIVIDED AIC THAT WILL BE      VICHMAIN 00234
OVERLAID                                              VICHMAIN 00235
IPKERN = SUBSCRIPT OF WHERE FIRST BOX OF UNSUBDIVIDED BOX      VICHMAIN 00236
WOULD BE IF IT WERE NOT OVERLAID. THIS ALLOWS      VICHMAIN 00237
PROGRAM TO REFERENCE UNSUBDIVIDED ARRAY WITH PROPER      VICHMAIN 00238
SUBSCRIPT.                                            VICHMAIN 00239
MAXL = LENGTH OF COMPUTED AIC ARRAY                  VICHMAIN 00240
VICHMAIN 00241
C
XNA = NPLKRN                                      VICHMAIN 00242
LUKERN = (XNA/2.) * (XNA + 1.) + .001            VICHMAIN 00243
NSKRN = NPLKRN                                     VICHMAIN 00244
IST = 0                                           VICHMAIN 00245
IF(NSUBDV.EQ.1) GO TO 340                         VICHMAIN 00246
VICHMAIN 00247
C
IF THE EFFECTIVE AREA WAS INPUT ON CARD C USE THAT      VICHMAIN 00248
FIND IF THE PLANFORM LIMITS THE SIZE OF THE EFFECTIVE AREA.  VICHMAIN 00249
VICHMAIN 00250
C
IF(NRWEA.GT.20) NRWEA = 20                         VICHMAIN 00251
IF(NRWEA.NE.0) MXSKRN = NSUBDV * NRWEA            VICHMAIN 00252
VICHMAIN 00253
C
NBOXES = MYBBSW                                     VICHMAIN 00254
NBROW = 1                                         VICHMAIN 00255
NBORRW = (MYBBSW-1)/2                            VICHMAIN 00256
300 CONTINUE                                       VICHMAIN 00257
IF(NBOXES .GT. LSDW) GO TO 323                  VICHMAIN 00258
VICHMAIN 00259
C
VICHMAIN 00260
VICHMAIN 00261
VICHMAIN 00262
VICHMAIN 00263
VICHMAIN 00264
VICHMAIN 00265

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NBOXES = NBOXES + NBOXRW          VIMAIN 00266
NBOXRW = NBOXRW - 2                VIMAIN 00267
IF(NBOXRW.LE.0) GO TO 330          VIMAIN 00268
NSROW = NSROW + 2                 VIMAIN 00269
GO TO 300                          VIMAIN 00270
325 CONTINUE
NSROW = NSROW - 1                 VIMAIN 00271
IF (NBOXES-NBOXRW/2 .GT. LSDW) NSROW = NSROW - 1
GO TO 335                          VIMAIN 00272
330 CONTINUE
NSROW = MXSKRN                     VIMAIN 00273
335 CONTINUE
XNA = MXSKRN                      VIMAIN 00274
LSKERN = XNA*(XNA/2.) + (XNA/2.) + 0.001
ISUB = MXSKRN/NSUBDV              VIMAIN 00275
IF (NSROW.LT. MXSKRN) ISUB = NSROW/NSUBDV
SUB = ISUB                         VIMAIN 00276
LTOKBN = SUB*(SUB/2.) + (SUB/2.) + 0.001
IST = LSKERN - LTOKBN             VIMAIN 00277
340 CONTINUE
IPKERN = IST + 1                  VIMAIN 00278
MAXL = IST + LKERN                VIMAIN 00279
IF(MAXL.LE.LKERN) GO TO 21        VIMAIN 00280
WRITE (NT6,9305) MAXL,LKERN      VIMAIN 00281
9305 FORMAT(59HD*** ERROR - THE SIZE OF THE AIC ARRAY FOR THIS PLANFORM
1 IS,IS,29H, THE MAXIMUM SIZE ALLOWED IS,IS, 5H. *** )
CALL FLUSH(1)                      VIMAIN 00282
C
21 CONTINUE                         VIMAIN 00283
C
IF(NPK.EQ.2.AND.NV.EQ.NVCS-1) GO TO 22
NN = 1                               VIMAIN 00284
NRWS = MXSKRN                      VIMAIN 00285
GO TO 23                             VIMAIN 00286
22 CONTINUE                         VIMAIN 00287
NN = IPKERN                         VIMAIN 00288
NRWS = NFLKRN                       VIMAIN 00289
23 CONTINUE                         VIMAIN 00290
DO 24 I=1,NRWS
MUAIC(1,I) = I                      VIMAIN 00291
MUAIC(2,I) = I + I - 1              VIMAIN 00292
24 CONTINUE                         VIMAIN 00293
C
DETERMINE IF KERNELS EXIST ON TAPE AND GET THEM FROM TAPE
IF POSSIBLE.                         VIMAIN 00294
C
NKOT = 0                             VIMAIN 00295
IF(.NOT.OPLAIC) GO TO 25            VIMAIN 00296
C
KERNELS ARE ON TAPE, GET TABLE OF CONTENTS
REWIND OPLAIC                        VIMAIN 00297
CALL RINIT                           VIMAIN 00298
NFS = 1                              VIMAIN 00299
ITYPE = SHMTXED                      VIMAIN 00300
MXARRY = GHTAR                        VIMAIN 00301
CALL READMX(NPLAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,500,NID,TD,ITYPE,    VIMAIN 00302

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2 31X,1HC,34X,1HW,34X,1HV,/3X,2HNU,3X,2HMU,7X,*VELOCITY POTENTIAL VICKMAIN 00437
3COEFFICIENT* 10X,*UPWASH COEFFICIENT*,15X,*SIDEWASH COEFFICIENT* / VICKMAIN 00438
4 2(3X,2H--),7X, 32(1H-),2X,32(1H-),2X,32(1H-) // VICKMAIN 00439
VICKMAIN 00440
C IF(.NOT.PRNT) GO TO 52 VICKMAIN 00441
K=0 VICKMAIN 00442
KN = 0 VICKMAIN 00443
IF(NPK.EQ.2.AND.NV.EQ.NVCS-1) KN = IPKERN -1 VICKMAIN 00444
VICKMAIN 00445
C DO 50 I=1,NROWS VICKMAIN 00446
M= I-1 VICKMAIN 00447
MH = M/2 VICKMAIN 00448
JS = 2*(I-1) +1 VICKMAIN 00449
IF(YBAR.NE.0.0) JS = JS+1 VICKMAIN 00450
IF(EL.EQ.0.0) JS = I VICKMAIN 00451
DO 50 J=1,JS VICKMAIN 00452
K=K+1 VICKMAIN 00453
KN = KN + 1 VICKMAIN 00454
N = I - J VICKMAIN 00455
IF(EL.EQ.0.0) N = 1- J VICKMAIN 00456
IF(YBAR.LT.0.0) N = -N VICKMAIN 00457
WRITE (NT6,9210) M,N,C(KN),WK(K),V(K) VICKMAIN 00458
9210 FORMAT(2I5,5X,6E17.8) VICKMAIN 00459
50 CONTINUE VICKMAIN 00460
GO TO 53 VICKMAIN 00461
C 52 CONTINUE VICKMAIN 00462
RN = NROWS VICKMAIN 00463
K = RN*(RN/2.) + (RN/2.) + 1.0E-05 VICKMAIN 00464
IF(EL.EQ.0.) GO TO 53 VICKMAIN 00465
K = K+K VICKMAIN 00466
IF(YBAR.EQ.0.) K = K - NROWS VICKMAIN 00467
53 CONTINUE VICKMAIN 00468
VICKMAIN 00469
C C IF (NSPATK .EQ. 0 .OR. NV .GT. NVCS-NPK) GO TO 55 VICKMAIN 00470
IF(NV.EQ.1) REWIND IAICSC VICKMAIN 00471
C C WRITE THE SPATIAL AICS ON A SCRATCH FILE VICKMAIN 00472
C CALL RDINIT VICKMAIN 00473
PARM(1) = K1 VICKMAIN 00474
PARM(2) = XMACH VICKMAIN 00475
PARM(4) = YBAR VICKMAIN 00476
PARM(5) = EL VICKMAIN 00477
ITYPE = SHMIXED VICKMAIN 00478
MXARRY = GHMUAC VICKMAIN 00479
CALL WRTDX(Iaicsc,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1 MUAC,ITYPE,2,NROWS,PARM,IRR) VICKMAIN 00480
IF(IRR.NE.0) GO TO 6170 VICKMAIN 00481
C MXARRY = GH C VICKMAIN 00482
CALL WRTDX(Iaicsc,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1 C, ITYPE, 2,K, PARM, IRR) VICKMAIN 00483
IF(IRR.NE.0) GO TO 6170 VICKMAIN 00484
C MXARRY = SH W VICKMAIN 00485

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AMACH(IVAL) = -AMACH(IVAL)                                VIMAIN 00380
WRITE (NT6,9030) IVAL, AERR(IVAL),ISIZE(IVAL),NPLKRN      VIMAIN 00381
9030 FORMAT(4SH0THE PROGRAM IS GOING TO ENLARGE AIC ARRAY NO. I3,
1 29H, GENERATED AT AN ACCURACY OF F8.5 /19H IT IS NECESSARY TO VIMAIN 00382
2 29H ENLARGE THE SAVED ARRAY FROM I3, 3H TO, I3 )          VIMAIN 00383
VIMAIN 00384
VIMAIN 00385
C
C     SPACE TO CORRECT ARRAY ON TAPE
REWIND NPLAIC                                         VIMAIN 00386
CALL RDINIT                                           VIMAIN 00387
NMS = IVAL-1                                         VIMAIN 00388
ITYPE = SHMIXED                                       VIMAIN 00389
MXARRY = GH C                                         VIMAIN 00390
CALL READMX(NPLAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,C(NN), M, N, PARM, IRR)                  VIMAIN 00391
IF(IRR.NE.0) GO TO 6050                               VIMAIN 00392
VIMAIN 00393
VIMAIN 00394
C
C     SET MUASIC ARRAY FOR EXPANSION AIC CALCULATION
DO 1700 I=1,NSIZE                                     VIMAIN 00395
MUASIC(1,I) = 0                                       VIMAIN 00396
MUASIC(2,I) = 0                                       VIMAIN 00397
1700 CONTINUE                                         VIMAIN 00398
GO TO 25                                              VIMAIN 00399
1800 CONTINUE                                         VIMAIN 00400
VIMAIN 00401
VIMAIN 00402
C
C     THERE IS A GOOD MATRIX ON TAPE. READ THE TAPE, PRINT MESSAGE,
C     DO NOT MAIL RESUME.
C     SET NSIZE EQUAL TO NPLKRN SO THAT MATRIX WILL NOT BE WRITTEN VIMAIN 00403
C     ON TAPE.                                               VIMAIN 00404
VIMAIN 00405
VIMAIN 00406
VIMAIN 00407
VIMAIN 00408
NSIZE = NPLKRN                                         VIMAIN 00409
CALL RDINIT                                           VIMAIN 00410
REWIND NPLAIC                                         VIMAIN 00411
NMS = I-1                                             VIMAIN 00412
ITYPE = SHMIXED                                       VIMAIN 00413
MXARRY = GH C                                         VIMAIN 00414
CALL READMX(NPLAIC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2, NID,ID,ITYPE,
1           LRS,C(NN), M, N, PARM, IRR)                  VIMAIN 00415
IF(IRR.NE.0) GO TO 6050                               VIMAIN 00416
VIMAIN 00417
VIMAIN 00418
C
WRITE (NT6,6005) I, AERR(I)                           VIMAIN 00419
GO TO 35                                              VIMAIN 00420
25 CONTINUE                                         VIMAIN 00421
VIMAIN 00422
VIMAIN 00423
VIMAIN 00424
C
CALL KERNEL(XMACH,K1,ERR,C(NN),W,V)
C
35 CONTINUE                                         VIMAIN 00425
PRNT = .FALSE.
IF (NV.GT.NVCS-NPK) GO TO 40
IF (PRSAIC) PRNT = .TRUE.
GO TO 45
40 CONTINUE                                         VIMAIN 00426
IF (PRPAIC) PRNT = .TRUE.
45 CONTINUE                                         VIMAIN 00427
IF (.RNT) WRITE (NT6,9005) TITLE,XMACH,K1,ERR,EL,YBAR
9005 FORMAT(1H1, 6A10, // 40X, *AIC CALCULATIONS*, //,
X       17X, *MACH =*, F10.5, 5X, *K1 =*, F10.7, 5X,*ERR =*, F12.5, 5X,*EL =*, F6.2, 5X,*YBAR =*, F6.2, //
1   F10.7, 5X,*ERR =*, F12.5, 5X,*EL =*, F6.2, 5X,*YBAR =*, F6.2, //,
VIMAIN 00428
VIMAIN 00429
VIMAIN 00430
VIMAIN 00431
VIMAIN 00432
VIMAIN 00433
VIMAIN 00434
VIMAIN 00435
VIMAIN 00436

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2 31X,1HC,34X,1HW,34X,1HV,/3X,2HNU,3X,2HNU,7X,*VELOCITY POTENTIAL VICMAIN 00437
3COEFFICIENT* 10X,*UPWASH COEFFICIENT*,15X,*SIDEWASH COEFFICIENT* / VICMAIN 00438
4 2(3X,2H--),7X, 32(1H-),2X,32(1H-),2X,32(1H-) //) VICMAIN 00439
C
IF(.NOT.PRNT) GO TO 52 VICMAIN 00440
K=0 VICMAIN 00441
KN = 0 VICMAIN 00442
IF(NPK.EQ.2.AND.NV.EQ.NVCS-1) KN = IPKERN -1 VICMAIN 00443
C
DO 50 I=1,NROWS VICMAIN 00444
ME= I-1 VICMAIN 00445
MH = MH/2 VICMAIN 00446
JS = 2*(I-1) +1 VICMAIN 00447
IF(YBAR.NE.0.0) JS = JS+1 VICMAIN 00448
IF(EL.EQ.0.0) JS = I VICMAIN 00449
DO 50 J=1,JS VICMAIN 00450
K=K+1 VICMAIN 00451
KN = KN + 1 VICMAIN 00452
N = I - J VICMAIN 00453
IF(EL.EQ.0.0) N = 1- J VICMAIN 00454
IF(YBAR.LT.0.0) N = -N VICMAIN 00455
WRITE (NT6,9210) M,N,C(KN),W(K),V(K) VICMAIN 00456
9210 FORMAT(215,5X,6E17.8)
50 CONTINUE VICMAIN 00457
GO TO 53 VICMAIN 00458
C
52 CONTINUE VICMAIN 00459
RN = NROWS VICMAIN 00460
K = RN* (RN/2.) + (RN/2.) + 1.0E-05 VICMAIN 00461
IF(EL.EQ.0.) GO TO 53 VICMAIN 00462
K = K+K VICMAIN 00463
IF(YBAR.EQ.0.) K = K - NROWS VICMAIN 00464
53 CONTINUE VICMAIN 00465
C
IF (NSPATK .EQ. 0 .OR. NV .GT. NVCS-NPK) GO TO 55 VICMAIN 00466
IF(NV.EQ.1) REWIND IAICSC VICMAIN 00467
C
      WRITE THE SPATIAL AICS ON A SCRATCH FILE VICMAIN 00468
C
CALL RDINIT VICMAIN 00469
PARM(1) = K1 VICMAIN 00470
PARM(2) = X0ACH VICMAIN 00471
PARM(4) = YBAR VICMAIN 00472
PARM(5) = EL VICMAIN 00473
ITYPE = SHMIXED VICMAIN 00474
MXARRY = GHMUAIIC VICMAIN 00475
CALL WTEMX(IAICSC,MXWRIT,RANDU,NFS,NMS,LS,NMR,LWS,2,1D,
1           MUAIIC,ITYPE,2,NROWS,PARM,IRR) VICMAIN 00476
IF(IRR.NE.0) GO TO 6170 VICMAIN 00477
C
MXARRY = GH C VICMAIN 00478
CALL WTEMX(IAICSC,MXWRIT,RANDU,NFS,NMS,LS,NMR,LWS,2,1D,
1           C, ITYPE, 2,K, PARM, IRR) VICMAIN 00479
IF(IRR.NE.0) GO TO 6170 VICMAIN 00480
C
MXARRY = SH W VICMAIN 00481

```

```

CALL WRTENX(IAICSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1           W,      ITYPE, 2,K,      PARM, IRR)          VICKMAIN 00494
1 IF(IRR.NE.0) GO TO 617J          VICKMAIN 00495
C
MXARRY = 3H V          VICKMAIN 00496
CALL WRTENX(IAICSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1           V,      ITYPE, 2,K,      PARM, IRR)          VICKMAIN 00497
1 IF(IRR.NE.0) GO TO 617O          VICKMAIN 00498
C
IF(MV.NE.NVCS-NPK) GO TO 55          VICKMAIN 00499
END FILE IAICSC          VICKMAIN 00500
REWIND IAICSC          VICKMAIN 00501
55 CONTINUE          VICKMAIN 00502
IF(NPLAIC.EQ.0.AND.NSFAIC.EQ.0) GO TO 1000          VICKMAIN 00503
C
C       WRITE THE KERNEL ON TAPE          VICKMAIN 00504
IF(MV.LE.NVCS-NPK) GO TO 900          VICKMAIN 00505
C
C       WRITE ON THE PLANAR KERNEL TAPE          VICKMAIN 00506
IF(NPLAIC.EQ.0) GO TO 1000          VICKMAIN 00507
C
C       DETERMINE IF A PREVIOUS MATRIX WAS ON TAPE.          VICKMAIN 00508
C       IF NKOT = 0 IT IS A NEW TAPE AND THERE ARE NO OLD ONES          VICKMAIN 00509
C       IF NSIZE IS LESS THAN NROWS A MATRIX WAS EXPANDED OR THERE          VICKMAIN 00510
C       WAS NONE WITH CORRESPONDING PARAMETERS          VICKMAIN 00511
C
IF(NKOT.EQ.0) GO TO 60          VICKMAIN 00512
IF(NSIZE.GE.NROWS) GO TO 1000          VICKMAIN 00513
C
60 CONTINUE          VICKMAIN 00514
NKOT = NKOT + 1          VICKMAIN 00515
AMACH(NKOT) = XMACH          VICKMAIN 00516
AKVAL(NKOT) = K1          VICKMAIN 00517
AERR(NKOT) = ERR          VICKMAIN 00518
ISIZE(NKOT) = NROWS          VICKMAIN 00519
REWIND NPLAIC          VICKMAIN 00520
CALL RDINIT          VICKMAIN 00521
NMS = NKOT - 1          VICKMAIN 00522
ITYPE = SHMIXED          VICKMAIN 00523
MXARRY = 6H C          VICKMAIN 00524
CALL WRTENX(NPLAIC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1           C(INN), ITYPE, 2, K, PARM, IRR)          VICKMAIN 00525
1 IF(IRR.NE.0) GO TO 6150          VICKMAIN 00526
C
END FILE NPLAIC          VICKMAIN 00527
CALL RDINIT          VICKMAIN 00528
ITYPE = SHMIXED          VICKMAIN 00529
MXARRY = 6HTAB          VICKMAIN 00530
CALL WRTENX(NPLAIC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,500,ID,
1           TAB,ITYPE,NKOT,3,PARM,IRR)          VICKMAIN 00531
1 IF(IRR.NE.0) GO TO 6150          VICKMAIN 00532
C
MXARRY = 6HSIZE          VICKMAIN 00533
CALL WRTENX(NPLAIC,MXWRIT,RANDOU,NFS,NMS,LS,NMR, LWS,1, ID,
1           ISIZE,ITYPE,1,NKOT,PARM,IRR)          VICKMAIN 00534
1 IF(IRR.NE.0) GO TO 6150          VICKMAIN 00535
END FILE NPLAIC          VICKMAIN 00536

```

```

ENC FILE NSPAIC
NSPAIC = .TRUE.
GO TO 1000
100 CONTINUE
C
C      THIS AREA WILL WRITE SPATIAL KERNELS ON TAPE
C      THIS TAPE WILL BE USED IN DOWNWASH AND VELOCITY POTENTIAL
C      CALCULATIONS.
C      IN THE DEVELOPMENT STAGE IT WILL BE ASCERTAINED IF A TAPE
C      SHOULD BE SAVED FOR SUBSEQUENT RUNS.
C
C      IF(NSPAIC.EQ.0) GO TO 1000
C      IF(NKST.NE.0) GO TO 910
C
C      THERE WAS NO OLD KERNEL TAPE THEREFORE SKIP TO NV-NPK AND
C      ADD TO TABLE OF CONTENTS.
C
C      NKST = NV
C      GO TO 920
910 CONTINUE
C
C      IF NSIZ IS LESS THAN NROWS A MATRIX WAS EXPANDED OR THERE
C      WAS NONE WITH CORRESPONDING PARAMETERS.
C
C      IF(NSIZ.GE.NROWS) GO TO 1000
C
C      THERE WAS AN OLD KERNEL TAPE THEREFORE SKIP TO NKST+1 AND
C      ADD TO TABLE OF CONTENTS
C
C      NKST = NKST + 1
920 CONTINUE
AMACH(NKST) = XMACH
AKVAL(NKST) = K1
AERR(NKST) = ERR
YBARS(NKST) = YBAR
MERTS(NKST) = EL
ISIZE(NKST) = NROWS
REWIND NSPAIC
CALL RDINIT
ITYPE = SHMIXED
NMS =(NKST - 1)*4
MXARRY = GHMUAC
CALL WRTDX(NSPAIC,MXWRIT,RANDU,NFS,NMS,LS,NMR,LWS,2,ID,
1           MUAC,ITYPE,2,NROWS,PARM,IRR)
IF(IRR.NE.0) GO TO 6160
C
NMS = 0
ITYPE = SHMIXED
MXARRY = GH C
CALL WRTDX(NSPAIC,MXWRIT,RANDU,NFS,NMS,LS,NMR,LWS,2,ID,
1           C, ITYPE,2,K,PARM,IRR)
IF(IRR.NE.0) GO TO 6160
C
MXARRY = GH W
CALL WRTDX(NSPAIC,MXWRIT,RANDU,NFS,NMS,LS,NMR,LWS,2,ID,
1           W, ITYPE,2,K,PARM,IRR)
IF(IRR.NE.0) GO TO 6160

```

VICMAIN 00551
VICMAIN 00552
VICMAIN 00553
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VICMAIN 00597
VICMAIN 00598
VICMAIN 00599
VICMAIN 00600
VICMAIN 00601
VICMAIN 00602
VICMAIN 00603
VICMAIN 00604
VICMAIN 00605
VICMAIN 00606
VICMAIN 00607

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C
      MXARRY = 0H V          VICMAIN 00608
      CALL WRTDMX(NSPAIC,MXRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,  VICMAIN 00609
      1           V, ITYPE,2,K,PARM,IRR)  VICMAIN 00610
      IF(IRR.NE.0) GO TO 6160  VICMAIN 00611
C
      END FILE NSPAIC  VICMAIN 00612
C
      WRITE TABLE OF CONTENTS ARRAYS  VICMAIN 00613
      ITYPE = SHMIXED  VICMAIN 00614
      MXARRY = 0HTAB  VICMAIN 00615
      CALL WRTDMX(NSPAIC,MXRIT,RANDOU,NFS,NMS,LS,NMR,LWS,500,ID,  VICMAIN 00616
      1           TAB,ITYPE,NKST,3,PARM,IRR)  VICMAIN 00617
      IF(IRR.NE.0) GO TO 6160  VICMAIN 00618
C
      MXARRY = 0HTAB  VICMAIN 00619
      CALL WRTDMX(NSPAIC,MXRIT,RANDOU,NFS,NMS,LS,NMR,LWS,500,ID,  VICMAIN 00620
      1           ATAB,ITYPE,NKST,3,PARM,IRR)  VICMAIN 00621
      IF(IRR.NE.0) GO TO 6160  VICMAIN 00622
C
      MXARRY = 0HTAB  VICMAIN 00623
      CALL WRTDMX(NSPAIC,MXRIT,RANDOU,NFS,NMS,LS,NMR,LWS,500,ID,  VICMAIN 00624
      1           ATAB,ITYPE,NKST,3,PARM,IRR)  VICMAIN 00625
      IF(IRR.NE.0) GO TO 6160  VICMAIN 00626
C
      END FILE NSPAIC  VICMAIN 00627
      END FILE NSPAIC  VICMAIN 00628
      IF(NV.EQ.NVCS-NPK) OSFAIC = .TRUE.  VICMAIN 00629
1000 CONTINUE  VICMAIN 00630
      RETURN  VICMAIN 00631
C
      6010 CONTINUE  VICMAIN 00632
      WRITE (NT6,6010) ICEOSC,IRR  VICMAIN 00633
      WRITE (NT6,6011) MXARRY  VICMAIN 00634
      GO TO 6100  VICMAIN 00635
C
      6050 CONTINUE  VICMAIN 00636
      WRITE (NT6,6050) NPLAIC,IRR  VICMAIN 00637
      WRITE (NT6,6051) MXARRY  VICMAIN 00638
      GO TO 6100  VICMAIN 00639
C
      6060 CONTINUE  VICMAIN 00640
      WRITE (NT6,6060) NSPAIC,IRR  VICMAIN 00641
      WRITE (NT6,6061) MXARRY  VICMAIN 00642
      GO TO 6100  VICMAIN 00643
C
      6150 CONTINUE  VICMAIN 00644
      WRITE (NT6,6150) NPLAIC,IRR  VICMAIN 00645
      WRITE (NT6,6151) MXARRY  VICMAIN 00646
      GO TO 6100  VICMAIN 00647
C
      6160 CONTINUE  VICMAIN 00648
      WRITE (NT6,6160) NSPAIC,IRR  VICMAIN 00649
      WRITE (NT6,6161) MXARRY  VICMAIN 00650
      GO TO 6100  VICMAIN 00651
C
      6170 CONTINUE JE  VICMAIN 00652
      WRITE (NT6,6160) IAICSC,IRR  VICMAIN 00653
      WRITE (NT6,6161) MXARRY  VICMAIN 00654
C
      6100 CONTINUE  VICMAIN 00655
      WRITE (NT6,6101) ID(1),ID(2)  VICMAIN 00656
      VICMAIN 00657
      VICMAIN 00658
      VICMAIN 00659
      VICMAIN 00660
      VICMAIN 00661
      VICMAIN 00662
      VICMAIN 00663
      VICMAIN 00654

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WRITE (NT6,9102) PARM,IPARM	VICMAIN 00665
WRITE (NT6,9103) NFS,NMS	VICMAIN 00666
WRITE (NT6,9104) ITYPE,M,N	VICMAIN 00667
WRITE (NT6,9900)	VICMAIN 00668
C	VICMAIN 00669
CALL FLUSH(1)	VICMAIN 00670
C	VICMAIN 00671
8005 FORMAT(*OAIC ARRAY NO.* ,I3,* , GENERATED AT AN ACCURACY OF *,F6.4, 1 * IS BEING USED. *)	VICMAIN 00672
C	VICMAIN 00673
9010 FORMAT(53H0*** ERROR - WHILE READING THE GEOMETRY SCRATCH FILE A10 1 , 15H, ERROR CODE = I4,4H ***)	VICMAIN 00674
9011 FORMAT(5X,31HAN ATTEMPT WAS MADE TO READ THE A6,8H MATRIX.//)	VICMAIN 00675
9050 FORMAT(48H0*** ERROR - WHILE READING THE PLANAR AIC FILE A10, 1 15H, ERROR CODE = I4,4H ***)	VICMAIN 00676
9060 FORMAT(47H0*** ERROR - WHILE READING THE SPATIAL AIC FILE A10, 1 15H, ERROR CODE = I4,4H ***)	VICMAIN 00677
9150 FORMAT(48H0*** ERROR - WHILE WRITING THE PLANAR AIC FILE A10, 1 15H, ERROR CODE = I4,4H ***)	VICMAIN 00678
9151 FORMAT(5X,32HAN ATTEMPT WAS MADE TO WRITE THE A6,8H MATRIX.//)	VICMAIN 00679
9160 FORMAT(47H0*** ERROR - WHILE WRITING THE SPATIAL AIC FILE A10, 1 15H, ERROR CODE = I4,4H ***)	VICMAIN 00680
C	VICMAIN 00681
9101 FORMAT(5X,*MATRIX ID = *, A10, I10)	VICMAIN 00682
9102 FORMAT(5X,*PARAMETERS *,10E11.3, /10X,*(INTEGER)*, I7, 9I11)	VICMAIN 00683
9103 FORMAT(5X,*FILE SFACING = *,I3,* MATRIX SFACING = *,I3)	VICMAIN 00684
9104 FORMAT(5X,*MATRIX TYPE -*,A10,* , DIMENSIONED (*I4,2H X,I4,1H))	VICMAIN 00685
C	VICMAIN 00686
9900 FORMAT(*0 ERROR OCCURRED IN AIC SECTION (VICMAIN).*)	VICMAIN 00687
END	VICMAIN 00688
	VICMAIN 00689
	VICMAIN 00690
	VICMAIN 00691
	VICMAIN 00692
	VICMAIN 00693
	VICMAIN 00694

```

SUBROUTINE KERNEL(XMACH,K1,ERR, C, W, V)
COMMON /FILES / NT5,NT6,INTAPE,INFS~,NPLAIC,NSPAIC,NOUTP,
1           ICLFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC
COMMON /VICPAR/ YBAR,EL,NROWS,MUAI(2,150)
DIMENSION XIL(4),IUC(3)
DIMENSION C(:), W(:), V(:)
COMPLEX C,W,V, CSV,WSV,VSV
DIMENSION CTM(2), WTM(2), VTM(2)
EQUIVALENCE (CSV,CTM), (WSV,WTM), (VSV,VTM)
COMMON /BESFUN/ XIB(5), A(50,5)
REAL K1,K1BAR
COMPLEX ZERO
DATA EPS / 1.0E-4 /

C
C      XMACH - MACH NUMBER
C      K1   - REDUCED FREQUENCY
C      ERR  - CONVERGENCE CRITERIA (RELATIVE, NOT ABSOLUTE)
C      C   - VELOCITY POTENTIAL AERODYNAMIC INFLUENCE COEFFICIENTS
C      W   - UPWASH AERODYNAMIC INFLUENCE COEFFICIENTS
C      V   - SIDEWASH AERODYNAMIC INFLUENCE COEFFICIENTS
C
C      ZERO = (0.,0.)
TMACH = XMACH*XMACH
K1BAR = (K1*TMACH)/(TMACH - 1.0)
EL2 = EL*EL
ITOT = 0
NTP = 0
IF(YBAR.EQ.0.0) NTP = -1
DO 1000 I=1,NROWS
C
C      NTP = NTP + 2
C
C      IS = ITOT+1
IF(EL.EQ.0.0) GO TO 50
ITOT = ITOT + NTP
GO TO 75
50 CONTINUE
ITOT = ITOT + 1
C
75 CONTINUE
IF(MUAI(2,1).EQ.0) GO TO 1000
C
C      DETERMINE IF THERE ARE ANY BOXES ON THE I-TH ROW CUT BY THE
C      MACH HYPERBOLA. (I=1 IS THE FIRST ROW)
VBARU = FLOAT(I) - 0.5
VBARL = VBARU - 1.0
IF (ABS(EL)+EPS .GT. VBARU) GO TO 950
XLOW = VBARL
IF (ABS(EL) .GT. VBARL) XLOW = ABS(EL)
XIB(1) = XLOW
XINC = 0.25 * (VBARU-XLOW)
DO 115 J=2,5
XIB(J) = XIB(J-1) + XINC
105 CONTINUE
DO 106 J=1,250
A(J) = 0.0
106 CONTINUE

```

KERNEL 00002
FILE 00002
FILE 00003
KERNEL 00003
KERNEL 00004
KERNEL 00005
KERNEL 00006
KERNEL 00007
KERNEL 00008
BCSAICB 00001
KERNEL 00010
KERNEL 00011
KERNEL 00012
KERNEL 00013
KERNEL 00014
KERNEL 00015
KERNEL 00016
KERNEL 00017
KERNEL 00018
KERNEL 00019
KERNEL 00020
KERNEL 00021
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KERNEL 00127
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KERNEL 00038
KERNEL 00039
KERNEL 00041
KERNEL 00042
KERNEL 00043
KERNEL 00044
KERNEL 00045
KERNEL 00046
KERNEL 00047
KERNEL 00048
KERNEL 00049
BCSAICB 00002
BCSAICB 00003
KERNEL 00052
KERNEL 00013
BCSAICB 00004
KERNEL 00051
KERNEL 00056

```

IWARN = 0                                KERNEL 00057
DO 108 J=1,5                            BCSAICB 00005
TAU = SQRT(XIB(J)*XIB(J) - EL2)          KERNEL 00059
TAUKH = (K1BAR/XMACH) * TAU              KERNEL 00060
XIB(J) = TAUHK                          KERNEL 00061
CALL RANGE(TAUHK,N)                      KERNEL 00062
IF(N.LE.100) GO TO 107                  KERNEL 00063
WRITE (NT6,9005) N                      KERNEL 00064
9005 FORMAT( 99HD*** THE ARGUMENT FOR A BESSSEL FUNCTION YIELDS AN ORDER
1 GREATER THAN 100. ORDER REDUCED TO 100. *** )
N = 100                                  KERNEL 00065
IF(IWARN.EQ.1) GO TO 107                KERNEL 00066
IWARN = 1                                KERNEL 00067
107 CONTINUE                            KERNEL 00068
    CALL BESSEL(TAUHK,A(1,J),N)
108 CONTINUE                            KERNEL 00069
C                                     KERNEL 00070
C             THERE ARE BOXES ON THIS ROW. FIND LEFT MOST BOX AND PROCEED
C             FROM LEFT TO RIGHT.                               KERNEL 00071
C             ULEFT = SQRT(VBARU*VBARU - EL2)                 KERNEL 00072
C             URIGHT = -ULEFT                           KERNEL 00073
C             IHALF = (NTP+1)/2                         KERNEL 00074
C             REM = ABS(YBAR) - 0.5                     KERNEL 00075
C             IL = ULEFT - REM                         KERNEL 00076
C             IL = IHALF - IL                         KERNEL 00077
C             IR = REM - URIGHT + 1.0                  KERNEL 00078
C             IR = IHALF + IR                         KERNEL 00079
C             IF(IL.LT.MUIC(1,I)) GO TO 110            KERNEL 00080
C                                     KERNEL 00081
C             HYPERBOLA IS LESS THAN ALLOWED, REDUCE LIMITS.
C             MUIC(1,I) = IL                         KERNEL 00082
C             GO TO 120                                KERNEL 00083
C                                     KERNEL 00084
C             HYPERBOLA CROSSED A BOUNDARY, REDUCE CALCULATIONS.
C             110 IL = MUIC(1,I)                      KERNEL 00085
C             120 CONTINUE                            KERNEL 00086
C                                     KERNEL 00087
C             TEST RIGHT SIDE                        KERNEL 00088
C             IF(IR.GT.MUIC(2,I)) GO TO 130            KERNEL 00089
C                                     KERNEL 00090
C             HYPERBOLA IS LESS THAN ALLOWED, REDUCE LIMITS.
C             MUIC(2,I) = IR                         KERNEL 00091
C             GO TO 140                                KERNEL 00092
C                                     KERNEL 00093
C             HYPERBOLA CROSSES A BOUNDARY, REDUCE CALCULATIONS.
C             130 IR = MUIC(2,I)                      KERNEL 00094
C             140 CONTINUE                            KERNEL 00095
C                                     KERNEL 00096
C             DETERMINE INTEGRALS FOR BOXES IL TO IR.
C             IF (IL .GT. IR) GO TO 950            KERNEL 00097
C             DO 900 ID=IL,IR                      KERNEL 00098
C             U = IHALF - ID                      KERNEL 00099
C             IU = U                                KERNEL 00100
C             ULEFT = U + 0.5 + ABS(YBAR)           KERNEL 00101
C             URIGHT = ULEFT - 1.0                 KERNEL 00102
C             YMUBAR = ULEFT - 0.5                 KERNEL 00103
C                                     KERNEL 00104
C             IF (IL .GT. IR) GO TO 950            KERNEL 00105
C             DO 900 ID=IL,IR                      KERNEL 00106
C             U = IHALF - ID                      KERNEL 00107
C             IU = U                                KERNEL 00108
C             ULEFT = U + 0.5 + ABS(YBAR)           KERNEL 00109
C             URIGHT = ULEFT - 1.0                 KERNEL 00110
C             YMUBAR = ULEFT - 0.5                 KERNEL 00111
C                                     KERNEL 00112
C                                     KERNEL 00113

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```

CBARL = SORT(ULEFT+ULEFT + EL2)
CBARR = SORT(URIGHT+URIGHT+EL2)
C
IF(CBARL.LE.VBARL.AND.CBARR.LE.VBARL) GO TO 500
C
IF(IU) 300,400,200
C
C      BOX IS TO THE LEFT OF THE CENTER LINE OR APEX
200 CONTINUE
IF(CBARL.LT.VBARU) GO TO 220
C
C      EDGE BOX, ONLY HAS ONE SEGMENT TO INTEGRATE
NINT = 1
XIL(1) = CBARR
XIL(2) = VBARU
IUC(1) = 1
GO TO 700
C
C      DOUBLY CUT BOX, HAS TWO SEGMENTS TO INTEGRATE
220 NINT = 2
XIL(1) = VBARL
IF(CBARR.GT.VBARL) XIL(1) = CBARR
XIL(2) = CBARL
XIL(3) = VBARU
IUC(1) = 1
IUC(2) = 0
GO TO 700
C
C      BOX IS ON THE RIGHT OF THE CENTER LINE OR APEX
300 CONTINUE
IF(CBARR.LT.VBARU) GO TO 320
C
C      EDGE BOX, HAS ONLY ONE SEGMENT
NINT = 1
XIL(1) = CBARL
XIL(2) = VBARU
IUC(1) = 2
GO TO 700
C
C      DOUBLY CUT BOX, HAS TWO SEGMENTS
320 NINT = 2
XIL(1) = VBARL
IF(CBARL.GT.VBARL) XIL(1) = CBARL
XIL(2) = CBARR
XIL(3) = VBARU
IUC(1) = 2
IUC(2) = 0
GO TO 700
C
C      CENTER LINE OR APEX BOX
400 CONTINUE
IF(ABS(EL).LT.VBARL) GO TO 475
IF(CBARL.LT.VBARU.OR.CBARR.LT.VBARU) GO TO 420
C
C      ONLY BOX ON ROW, ONLY ONE SEGMENT TO INTEGRATE
NINT = 1
XIL(1) = ABS(EL)

```

KERNEL 00114
 KERNEL 00115
 KERNEL 00116
 KERNEL 00117
 KERNEL 00118
 KERNEL 00119
 KERNEL 00120
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 KERNEL 00169
 KERNEL 00170

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XIL(2) = VBARU          KERNEL 00171
IUC(1) = 3               KERNEL 00172
GO TO 700                KERNEL 00173
KERNEL 00174
KERNEL 00175
KERNEL 00176
KERNEL 00177
KERNEL 00178
KERNEL 00179
KERNEL 00180
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KERNEL 00223
KERNEL 00224
KERNEL 00225
KERNEL 00226
KERNEL 00227

C
420 CONTINUE
IF(CBRL.LT.VBARU.AND.CBARR.LT.VBARU) GO TO 440

C
ONLY HAS 2 SEGMENTS
NINT = 2
XIL(1) = ABS(EL)
XIL(2) = CBARR
XIL(3) = VBARU
IUC(1)=3
IUC(2)=1
GO TO 700

C
440 CONTINUE
C
C      UNLESS THE HYPERBOLA CENTER IS ON A BOX SIDE LINE,
C      I.E. YBAR = 0.5, THEN THERE WILL BE 2 SEGMENTS.
C
IF(ABS(YBAR).NE.0.5) GO TO 445
XIL(1) = ABS(EL)
XIL(2) = CBRL
XIL(3) = VBARU
IUC(1) = 1
IUC(2) = 0
NINT = 2
GO TO 700

C
445 CONTINUE
C      WILL HAVE THREE SEGMENTS IF YBAR .NE. ZERO
XIL(1) = ABS(EL)
XIL(2) = CBARR
IUC(3)=3
IF(ABS(YBAR).NE.0.0) GO TO 450

C
C      TWO SEGMENTS
NINT = 2
XIL(3) = VBARU
IUC(2)=0
GO TO 700

C
C      THREE SEGMENTS
450 NINT = 3
XIL(3) = CBRL
IUC(2)=1
XIL(4) = VBARU
IUC(3)=0
GO TO 700

C
C      CENTER LINE BOX, BUT NOT APEX, HAS THREE SEGMENTS
479 CONTINUE
IF(CBARR.LE.VBRL) GO TO 220
NINT = 3
XIL(1) = VBRL
XIL(2) = CBARR

```

```

XIL(3) = CBARL          KERNEL 00228
XIL(4) = VBARU          KERNEL 00229
IUC(1) = 3               KERNEL 00230
IUC(2) = 1               KERNEL 00231
IUC(3) = 0               KERNEL 00232
GO TO 700               KERNEL 00233
C
C      FULL BOX, ONLY ONE SEGMENT TO INTEGRATE
500 CONTINUE
NINT = 1                KERNEL 00234
XIL(1) = VRARL          KERNEL 00235
XIL(2) = VBARU          KERNEL 00236
IUC(1) = 0               KERNEL 00237
C
C      LIMITS AND TYPES FOR ALL SEGMENTS ARE COMPLETED. INTEGRATE.
700 CONTINUE
IDX = ID                KERNEL 00238
IF(EL.EQ.0.0) IDX = ID - I + 1
IX = IS + IDX - 1       KERNEL 00239
IF(C(IX).NE.0) GO TO 900
DO 800 INT=1,NINT
CSV = (0.,0.)
WSV = (0.,0.)
VSV = (0.,0.)
IFLAG=0
C
C      CALL ROMBERG INTEGRATION FOR REAL PART
CALL ROMBER(XIL(INT),XIL(INT+1),IUC(INT),ERR,IFLAG,K1BAR,YMUBAR,
1           EL,XMACH,CTM(1),WTM(1),VTM(1))
IF(K1.EQ.0.0) GO TO 750
IFLAG = 1
C
C      CALL ROMBERG INTEGRATION FOR IMAGINARY PART
CALL ROMBER(XIL(INT),XIL(INT+1),IUC(INT),ERR,IFLAG,K1BAR,YMUBAR,
1           EL,XMACH,CTM(2),WTM(2),VTM(2))
750 CONTINUE
C(IX) = C(IX) + CSV
W(IX) = W(IX) + WSV
V(IX) = V(IX) + VSV
800 CONTINUE
C
900 CONTINUE
GO TO 1000
C
950 CONTINUE
MUMIC(1,I) = 0
MUMIC(2,I) = 0
C
1000 CONTINUE
RETURN
END

```

KERNEL 00240
 KERNEL 00241
 KERNEL 00242
 KERNEL 00243
 KERNEL 00244
 KERNEL 00245
 KERNEL 00246
 KERNEL 00247
 KERNEL 00248
 KERNEL 00249
 KERNEL 00250
 KERNEL 00251
 KERNEL 00252
 KERNEL 00253
 KERNEL 00254
 KERNEL 00255
 KERNEL 00256
 KERNEL 00257
 KERNEL 00258
 KERNEL 00259
 KERNEL 00260
 KERNEL 00261
 KERNEL 00262
 KERNEL 00263
 KERNEL 00264
 KERNEL 00265
 KERNEL 00266
 KERNEL 00267
 KERNEL 00268
 KERNEL 00269
 KERNEL 00270
 KERNEL 00271
 KERNEL 00272
 KERNEL 00273
 KERNEL 00274
 KERNEL 00275
 KERNEL 00276
 KERNEL 00277
 KERNEL 00278

```

SUBROUTINE ROMBER(XILL,XILU,IUC,ERR,IFLAG,K1BAR,YMUBAR,EL ,XMACH,    ROMBER 00002
1          C, W, V )                                         ROMBER 00003
DIMENSION XI(512),FXIC(512),FXIW(512),FXIV(512)                  ROMBER 00004
DIMENSION A(11,11), AW(11,11), AV(11,11), VT(2)                   ROMBER 00005
REAL      K1BAR                                         ROMBER 00006
PIE = 3.141592654                                         ROMBER 00007
C
C      XILL - XI LOWER LIMIT OF INTEGRATION                 ROMBER 00008
C      XILU - XI UPPER LIMIT OF INTEGRATION                 ROMBER 00009
C      IUC - FLAG INDICATING TYPE OF BOX OR EDGE CONDITION OF ROMBER 00010
C           INTERVAL TO BE INTEGRATED.
C           IUC = 0, FULL BOX                                ROMBER 00011
C           = 1, LEFT SIDE OF INTERVAL IS EDGE OF MACH HYP.  ROMBER 00012
C           = 2, RIGHT SIDE OF INTERVAL IS EDGE OF MACH HYP.  ROMBER 00013
C           = 3, BOTH SIDES OF INTERVAL IS EDGE OF MACH HYP.  ROMBER 00014
C           = 4, MACH HYP. IS ON THE EDGE OF INTERVAL        ROMBER 00015
C           = 5, MACH HYP. IS ON THE EDGE OF INTERVAL        ROMBER 00016
C           = 6, MACH HYP. IS ON THE EDGE OF INTERVAL        ROMBER 00017
C           = 7, MACH HYP. IS ON THE EDGE OF INTERVAL        ROMBER 00018
C           = 8, MACH HYP. IS ON THE EDGE OF INTERVAL        ROMBER 00019
C           = 9, MACH HYP. IS ON THE EDGE OF INTERVAL        ROMBER 00020
C           = 10, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00021
C           = 11, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00022
C           = 12, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00023
C           = 13, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00024
C           = 14, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00025
C           = 15, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00026
C           = 16, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00027
C           = 17, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00028
C           = 18, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00029
C           = 19, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00030
C           = 20, MACH HYP. IS ON THE EDGE OF INTERVAL       ROMBER 00031
C
C      CALCULATE INITIAL VALUES AT END POINTS
C
C      EL2 = EL*EL
ROMBER 00032
C      VK = -XMACH/(PIE*K1BAR)
ROMBER 00033
C      XI(1) = XILL
ROMBER 00034
C      XI(2) = XILU
ROMBER 00035
CALL FUNCT(2,XI,FXIC,FXIW,IFLAG,K1BAR,EL ,YMUBAR,IUC,XMACH,BESSW) ROMBER 00036
IF(EL.EQ.0.0) GO TO 101
ROMBER 00037
IF(K1BAR.EQ.0.0) GO TO 101
ROMBER 00038
IF(IUC.EQ.3) GO TO 101
ROMBER 00039
CALL VFUNC (2, XI,FXIV,IFLAG,K1BAR,EL,YMUBAR,IUC,XMACH,1,VT)
ROMBER 00040
101 CONTINUE
ROMBER 00041
TERM1 = FXIC(1)/XI(1)
ROMBER 00042
TERM2 = FXIC(2)/XI(2)
ROMBER 00043
HINT = 0.5*(XILU-XILL)
ROMBER 00044
TC = HINT *(FXIC(1)+FXIC(2))
ROMBER 00045
TW = HINT *(FXIW(1)+FXIW(2))
ROMBER 00046
A(1,1) = TC
ROMBER 00047
AW(1,1) = TW
ROMBER 00048
IF(EL.EQ.0.0) GO TO 102
ROMBER 00049
IF(K1BAR.EQ.0.0) GO TO 102
ROMBER 00050
IF(IUC.EQ.3) GO TO 102
ROMBER 00051
TV = HINT *(FXIV(1)+FXIV(2))
ROMBER 00052
AV(1,1) = TV
ROMBER 00053
102 CONTINUE
ROMBER 00054
C
DO 30 M=2,11
ROMBER 00055
I= 2*(M-1)
ROMBER 00056
H = (XILU - XILL)/I
ROMBER 00057
ROMBER 00058

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C
C      DETERMINE XI LOCATIONS TO EVALUATE FUNCTION AT
DO 3 J=2,I,2
JJ = J/2
XI(JJ) = XILL + (J-1)*H
3 CONTINUE
CALL FUNCT(JJ,XI,FXIC,FXIW,IFLAG,K1BAR,EL ,YMUBAR,IUC,XMACH,DUMMY) ROMBER 00059
IF(EL.EQ.0.0) GO TO 103 ROMBER 00060
IF(K1BAR.EQ.0.0) GO TO 103 ROMBER 00061
IF(IUC.EQ.3) GO TO 103 ROMBER 00062
CALL VFUNC (JJ,XI,FXIV,IFLAG,K1BAR,EL,YMUBAR,IUC,XMACH,0,DUMMY) ROMBER 00063
ROMBER 00064
103 CONTINUE ROMBER 00065
C
C      DETERMINE TRAPEZOIDAL AREA WITH THE NEW FUNCTION EVALUATIONS
TMNC = 0.0 ROMBER 00066
TMNW = 0.0 ROMBER 00067
TMNV = 0.0 ROMBER 00068
DO 5 J=1,JJ ROMBER 00069
TMNC = TMNC + FXIC(J) ROMBER 00070
TMNW = TMNW + FXIW(J) ROMBER 00071
IF(EL.EQ.0.0) GO TO 5 ROMBER 00072
IF(K1BAR.EQ.0.0) GO TO 5 ROMBER 00073
IF(IUC.NE.3) TMNV = TMNV + FXIV(J) ROMBER 00074
5 CONTINUE ROMBER 00075
TC = 0.5*TC + H*TMNC ROMBER 00076
TW = 0.5*TW + H*TMNW ROMBER 00077
ROMBER 00078
C
C      PUT THE NEW AREAS INTO THE ARRAY AND PERFORM EXTRAPOLATION
A(M,1) = TC ROMBER 00079
AW(M,1)= TW ROMBER 00080
IF(EL.EQ.0.0) GO TO 104 ROMBER 00081
IF(K1BAR.EQ.0.0) GO TO 104 ROMBER 00082
IF(IUC.EQ.3) GO TO 104 ROMBER 00083
TV = 0.5*TV + H*TMNV ROMBER 00084
AV(M,1) = TV ROMBER 00085
ROMBER 00086
104 CONTINUE ROMBER 00087
DO 10 N=2,M ROMBER 00088
A(M,N) = ((4**((N-1)))*A(M,N-1)-A(M-1,N-1))/(4**((N-1)-1)) ROMBER 00089
AW(M,N)= ((4**((N-1)))*AW(M,N-1)-AW(M-1,N-1))/(4**((N-1)-1)) ROMBER 00090
IF(EL.EQ.0.0) GO TO 10 ROMBER 00091
IF(K1BAR.EQ.0.0) GO TO 10 ROMBER 00092
IF(IUC.EQ.3) GO TO 10 ROMBER 00093
AV(M,N)= ((4**((N-1)))*AV(M,N-1)-AV(M-1,N-1))/(4**((N-1)-1)) ROMBER 00094
10 CONTINUE ROMBER 00095
C
C      DETERMINE IF THE TECHNIQUE HAS REACHED SUFFICIENT CONVERGENCE
C = A(M,M) ROMBER 00096
W = AW(M,M) ROMBER 00097
IF(EL.EQ.0.0) GO TO 105 ROMBER 00098
IF(K1BAR.EQ.0.0) GO TO 105 ROMBER 00099
IF(IUC.EQ.3) GO TO 105 ROMBER 00100
V = AV(M,M) ROMBER 00101
105 CONTINUE ROMBER 00102
ERR = ABS(ERR*C) ROMBER 00103
MM1 = M-1 ROMBER 00104
UDIF = ABS(A(M,MM1)-A(MM1,MM1)) ROMBER 00105
RDIF = ABS(A(M,M) -A(M ,MM1)) ROMBER 00106
ROMBER 00107
ROMBER 00108
ROMBER 00109
ROMBER 00110
ROMBER 00111
ROMBER 00112
ROMBER 00113
ROMBER 00114
RCMBR 00115

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DELS = 0.5*(UDIF+RDIF)           ROMBER 00116
IF(DELS.GT.RERR) GO TO 30        ROMBER 00117
C
C      C HAS CONVERGED, TEST FOR W CONVERGENCE
IF(EL.EQ.0.0) GO TO 50           ROMBER 00118
RERR = ABS(ERR*W)                ROMBER 00119
UDIF = ABS(AW(M,MM1) - AW(MM1,MM1)) ROMBER 00120
RDIF = ABS(AW(M,M) - AW(M ,MM1))  ROMBER 00121
DELS = 0.5 *(UDIF + RDIF)        ROMBER 00122
IF(DELS.GT.RERR) GO TO 30        ROMBER 00123
C
C      C AND W HAVE CONVERGED, TEST FOR V CONVERGENCE
C
IF(K1BAR.EQ.0.0 .OR. IUC.EQ.3) GO TO 50           ROMBER 00124
RERR = ABS(ERR*V)                ROMBER 00125
UDIF = ABS(AV(M,MM1) - AV(MM1,MM1)) ROMBER 00126
RDIF = ABS(AV(M,M) - AV(M ,MM1))  ROMBER 00127
DELS = 0.5 *(UDIF + RDIF)        ROMBER 00128
IF(DELS.LE.RERR) GO TO 50        ROMBER 00129
C
C      HAS NOT CONVERGED MAKE ANOTHER LOOP.
30 CONTINUE                         ROMBER 00130
50 CONTINUE                         ROMBER 00131
C = -C/PIE                          ROMBER 00132
IF(EL.EQ.0.0) GO TO 70             ROMBER 00133
W = (EL/PIE)*(WTERME-TERM1 + BESSW)  ROMBER 00134
IF(K1BAR.EQ.0.0) GO TO 65          ROMBER 00135
IF(IUC.EQ.3) GO TO 70             ROMBER 00136
V = VK*(V + VT(2)-VT(1))         ROMBER 00137
GO TO 70                            ROMBER 00138
65 CONTINUE                         ROMBER 00139
ETAL = YMUBAR - 0.5               ROMBER 00140
IF(IUC.EQ.2) ETAL = - SQRT(XILU*XILU - EL2)  ROMBER 00141
ETAU = YMUBAR + 0.5               ROMBER 00142
IF(IUC.EQ.1) ETAU =   SQRT(XILU*XILU - EL2)  ROMBER 00143
XILU2 = XILU * XILU              ROMBER 00144
XILL2 = XILL * XILL              ROMBER 00145
ETAU2 = ETAU * ETAU              ROMBER 00146
ETAL2 = ETAL * ETAL              ROMBER 00147
SU = ETAU2 + EL2                 ROMBER 00148
SL = ETAL2 + EL2                 ROMBER 00149
S1 = XILU2 - SU                  ROMBER 00150
S2 = XILU2 - SL                  ROMBER 00151
S3 = XILL2 - SU                  ROMBER 00152
S4 = XILL2 - SL                  ROMBER 00153
V1 = 0.0                           ROMBER 00154
V2 = 0.0                           ROMBER 00155
V3 = 0.0                           ROMBER 00156
V4 = 0.0                           ROMBER 00157
IF(S1.GT.0.0) V1 = ALOG((XILU+SQRT(S1))/SQRT(SU)) ROMBER 00158
IF(S2.GT.0.0) V2 = ALOG((XILU+SQRT(S2))/SQRT(SL)) ROMBER 00159
IF(S3.GT.0.0) V3 = ALOG((XILL+SQRT(S3))/SQRT(SU)) ROMBER 00160
IF(S4.GT.0.0) V4 = ALOG((XILL+SQRT(S4))/SQRT(SL)) ROMBER 00161
V = (-1.0/PIE)*(V1-V2-V3+V4)     ROMBER 00162
70 CONTINUE                         ROMBER 00163
RETURN                            ROMBER 00164
END                               ROMBER 00165
IF(S1.GT.0.0) V1 = ALOG((XILU+SQRT(S1))/SQRT(SU)) ROMBER 00166
IF(S2.GT.0.0) V2 = ALOG((XILU+SQRT(S2))/SQRT(SL)) ROMBER 00167
IF(S3.GT.0.0) V3 = ALOG((XILL+SQRT(S3))/SQRT(SU)) ROMBER 00168
IF(S4.GT.0.0) V4 = ALOG((XILL+SQRT(S4))/SQRT(SL)) ROMBER 00169
V = (-1.0/PIE)*(V1-V2-V3+V4)     ROMBER 00170
70 CONTINUE                         ROMBER 00171
RETURN                            ROMBER 00172

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SUBROUTINE FUNCT(K,XI,FXIC,FXIW,IFLAG,K1BAR,EL ,YMUBAR,IUC,
1                 XMACH,BESSY)
DIMENSION XI(512),FXIC(512),FXIW(512),A(50)

C
C      K      - NUMBER OF FUNCTIONS TO EVALUATE
C      XI     - VARIABLE OF INTEGRATION
C      FXIC   - FUNCTIONAL VALUE FOR C EQUATION
C      FXIW   - FUNCTIONAL VALUE FOR W EQUATION
C      IFLAG  - INDICATOR OF REAL OR IMAGINARY PARTS
C                  IFLAG = 0, REAL PART
C                  IFLAG = 1, IMAGINARY PART
C      K1BAR  - FUNCTION OF REDUCED FREQUENCY AND MACH NUMBER
C      EL     - DISTANCE OF RECEIVING BOX ABOVE SENDING PLANE
C      YMUBAR - COORDINATE HORIZONTALLY OF PULSE SENDING BOX
C      IUC    - FLAG INDICATING TYPE OF BOX OR EDGE CONDITION OF
C                  INTERVAL TO BE INTEGRATED.
C      XMACH  - MACH NUMBER
C      BESSY  - EVALUATION OF END POINTS FOR W COEFFICIENTS.
C

REAL      K1BAR
PIE = 3.141592654
PIE2 = 1.570796327
EL2 = EL*EL
BESSY = 0.0

C      DO 1000  I=1,K
C
C      SET UP CONSTANTS
TAU = SQRT(XI(I)*XI(I) - EL2)
TAUKM = (K1BAR/XMACH)*TAU
EPOW = K1BAR*X1(I)
IF (ABS(TAU).LT.1.0E-06) GO TO 25
THETAU = (YMUBAR + 0.5)/TAU
THETAL = (YMUBAR-0.5)/TAU
GO TO 50
25 CONTINUE
THETAL = 0.0
THETAU = 0.0
50 CONTINUE
C      IF(IFLAG.EQ.0) GO TO 100
C
C      IMAGINARY PART
EXPIN = -SIN(EPOW)
EXPINW = (EPOW*COS(EPOW) - SIN(EPOW))/(XI(I)*XI(I))
GO TO 200
C
C      REAL PART
100 EXPIN = COS(EPOW)
EXPINW = (COS(EPOW) + EPOW*SIN(EPOW))/(XI(I)*XI(I))
C
200 CONTINUE
CALL BFUNC(TAUKM,A,N)
IF(EL .EQ.0.0) GO TO 250
IF(I.GT.1) GO TO 250
IF(IUC.NE.3) GO TO 250
EXL = ABS(EL) - XI(I)

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FUNCT	00002
FUNCT	00003
FUNCT	00004
FUNCT	00005
FUNCT	00006
FUNCT	00007
FUNCT	00008
FUNCT	00009
FUNCT	00010
FUNCT	00011
FUNCT	00012
FUNCT	00013
FUNCT	00014
FUNCT	00015
FUNCT	00016
FUNCT	00017
FUNCT	00018
FUNCT	00019
FUNCT	00020
FUNCT	00021
FUNCT	00022
FUNCT	00023
FUNCT	00024
FUNCT	00025
FUNCT	00026
FUNCT	00027
FUNCT	00028
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FUNCT	00054
FUNCT	00055
FUNCT	00056
FUNCT	00057
FUNCT	00058

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      IF(ABS(EXL).GT.1.0E-05) GO TO 250          FUNCT 00059
      BESSY = (EXPW*PIE)/EL                      FUNCT 00060
  250 CONTINUE                                     FUNCT 00061
      BESSO = A(1)                                FUNCT 00062
      PTERM = 0.0                                  FUNCT 00063
      IF(IUC.EQ.0.OR.IUC.EQ.2) GO TO 300          FUNCT 00064
C
C           LEFT SIDE IS BOUNDARY CONDITION
      C1 = PIE2                                    FUNCT 00065
      GO TO 400                                     FUNCT 00066
C
C           300 CONTINUE
      IF(ABS(THETAU).GE.1.0) GO TO 350          FUNCT 00067
      C1 = ASIN(THETAU)                           FUNCT 00068
      GO TO 400                                     FUNCT 00069
  350 CONTINUE                                     FUNCT 00070
      C1 = SIGN(PIE2,THETAU)                      FUNCT 00071
C
C           400 CONTINUE
      IF(IUC.LE.1) GO TO 500                      FUNCT 00072
C
C           RIGHT SIDE IS BOUNDARY CONDITION
      C2 = -PIE2                                    FUNCT 00073
      GO TO 600                                     FUNCT 00074
C
C           500 CONTINUE
      IF(ABS(THETAL).GE.1.0) GO TO 550          FUNCT 00075
      C2 = ASIN(THETAL)                           FUNCT 00076
      GO TO 600                                     FUNCT 00077
  550 CONTINUE                                     FUNCT 00078
      C2 = SIGN(PIE2,THETAL)                      FUNCT 00079
  600 CONTINUE                                     FUNCT 00080
C
C           IF(IUC.EQ.3) GO TO 900
      IF (N.EQ. 1) GO TO 900                      FUNCT 00081
C
      SIGNX = -1.0                                 FUNCT 00082
      R= 0                                         FUNCT 00083
      PSIGN = 1.0                                  FUNCT 00084
      N = (N+1)/2                                 FUNCT 00085
      DO 800 IR=2,N                               FUNCT 00086
      R = R +1                                    FUNCT 00087
      PSIGN = PSIGN * SIGNX                      FUNCT 00088
      PTERM = PSIGN/R                            FUNCT 00089
      BTERM = BTERM + PTERM*A(IR)*(SIN(2.0*R+C1) - SIN(2.0*R+C2))
      800 CONTINUE                                 FUNCT 00090
C
C           900 CONTINUE
      FXIC(I) = EXPW*(BESSO*(C1-C2) + BTERM )   FUNCT 00091
      FXIW(I) = 0.0                                 FUNCT 00092
      IF(EL2.EQ.0.0) GO TO 1000                  FUNCT 00093
      FXIW(I) = EXPW*(BESSO*(C1-C2) + BTERM )
C
C           1000 CONTINUE
      RETURN                                       FUNCT 00094
      END                                         FUNCT 00095
      FTNDI 00049
      FUNCT 00096
      FUNCT 00097
      FUNCT 00098
      FUNCT 00099
      FUNCT 00100
      FTNDI 00050
      FUNCT 00102
      FUNCT 00103
      FUNCT 00104
      FUNCT 00105
      FUNCT 00106
      FUNCT 00107
      FUNCT 00108
      FUNCT 00109
      FUNCT 00110
      FUNCT 00111
      FUNCT 00112
      FUNCT 00113
      FUNCT 00114

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SUBROUTINE BESEL(K12,A,NA)                               BESEL 00002
DIMENSION A(1), AV(150)                                BESEL 00003
REAL K12                                              BESEL 00004
C
C      K12 - FUNCTION OF XI VALUE, MACH NUMBER AND REDUCED FREQUENCY BESEL 00005
C      A      = EVALUATION OF THE BESEL FUNCTION                         BESEL 00006
C      NA     - ORDER OF THE BESEL FUNCTION TO BE EVALUATED             BESEL 00007
C
C      ALPHA = 1.E-25                                              BESEL 00008
C      NT = NA + 1                                              BESEL 00009
C      BETA=.0008                                              BESEL 00010
C      IF(K12-BETA) 76,76,78                                     BESEL 00011
78 CONTINUE
C      IF(K12-ALPHA) 76,76,20                                     BESEL 00012
76 CONTINUE
C      (2) = 0.                                              BESEL 00013
C      A(1) = 1.                                              BESEL 00014
C      GO TO 99                                              BESEL 00015
20 SUM = 0.
C      K = 1.5*K12 + 1.                                         BESEL 00016
C      NP = MAX0(K,NT)                                         BESEL 00017
C      I = NP+11                                              BESEL 00018
C      AV(I+2) = 0.0                                           BESEL 00019
C      AV(I+1) = ALPHA                                         BESEL 00020
30 AV(I) = AV(I+1)*I*2./K12-AV(I+2)                   BESEL 00021
C      IF (I-1) 40, 40, 50                                     BESEL 00022
40 IF (MOD(I,2)) 60, 70, 60                           BESEL 00023
60 SUM = SUM + AV(I)                                     BESEL 00024
70 I = I-1                                              BESEL 00025
GO TO 30
40 C = 1./(2.*SUM+AV(1))                                BESEL 00026
I = 1
DO 90 II=1,NA,2
A(I) = AV(II) * C                                       BESEL 00027
I = I + 1
IF(I.EQ.50) GO TO 99
90 CONTINUE
99 CONTINUE
RETURN
END

```

SUBROUTINE RANGE(K12,NA)	RANGE	00002
REAL K12	RANGE	00003
C	RANGE	00004
C CALCULATES THE RANGE ON THE VARIABLE N FOR SUBROUTINE BESSL	RANGE	00005
C	RANGE	00006
C	RANGE	00007
C K12 = FUNCTION OF X VALUE, MACH NUMBER AND REDUCED	RANGE	00008
C FREQUENCY	RANGE	00009
C NA - ORDER OF THE BESSSEL FUNCTION TO BE EVALUATED	RANGE	00010
C	RANGE	00011
400 CONTINUE	RANGE	00012
C	RANGE	00013
IF(K12-.0.01) 101,98,98	RANGE	00014
98 IF(K12-.3.00) 102,102,99	RANGE	00015
99 IF(K12-19.00) 103,103,100	RANGE	00016
100 GO TO 104	RANGE	00017
C	RANGE	00018
101 CONTINUE	RANGE	00019
NA=4	RANGE	00020
RETURN	RANGE	00021
102 CONTINUE	RANGE	00022
NA= 3.0*K12 +7.0	RANGE	00023
RETURN	RANGE	00024
103 CONTINUE	RANGE	00025
NA= 2.0*K12 +7.0	RANGE	00026
RETURN	RANGE	00027
104 CONTINUE	RANGE	00028
NA= (10.0/9.)*K12 + 29.	RANGE	00029
RETURN	RANGE	00030
END	RANGE	00031

```

SUBROUTINE VFUNC(K,XI,FXIV,IFLAG,K1BAR,EL,YMUBAR,IUC,XMACH,IND,VT) VFUNCT 00002
DIMENSION XI(256),FXIV(256),VT(2) VFUNCT 00003
REAL K1BAR VFUNCT 00004
C VFUNCT 00005
C THIS PROGRAM CALCULATES THE FUNCTION VALUES OF INTEGRATION VFUNCT 00006
C FOR THE V COEFFICIENTS. VFUNCT 00007
C VFUNCT 00008
C K - NUMBER OF VALUES TO CALCULATE VFUNCT 00009
C XI - VARIABLE ARRAY AT WHICH VALUES ARE CALCULATED. VFUNCT 00010
C FXIV - FUNCTIONAL VALUES VFUNCT 00011
C IFLAG = FLAG INDICATING REAL OR COMPLEX PART VFUNCT 00012
C IFLAG = 0, REAL PART VFUNCT 00013
C = 1, IMAGINARY PART VFUNCT 00014
C K1BAR - FUNCTION OR REDUCED FREQUENCY AND MACH NUMBER. VFUNCT 00015
C EL - DISTANCE OF RECEIVING BOX ABOVE SENDING PLANE. VFUNCT 00016
C YMUBAR - COORDINATE HORIZONTALLY OF PULSE SENDING PLANE. VFUNCT 00017
C IUC - FLAG INDICATING TYPE OF BOX OR EDGE CONDITION OF VFUNCT 00018
C INTERVAL TO BE INTEGRATED. VFUNCT 00019
C XMACH - MACH NUMBER VFUNCT 00020
C IND - INDICATOR TO CALCULATE VT TERMS VFUNCT 00021
C = 0, DO NOT CALCULATE VFUNCT 00022
C = 1, CALCULATE VFUNCT 00023
C VT - EXTRA TERMS CALCULATE AT THE LIMITS OF INTEGRATION VFUNCT 00024
C VFUNCT 00025
C EPS = 1.0E-04 VFUNCT 00026
C EL2 = EL*EL VFUNCT 00027
C DO 500 I=1,K VFUNCT 00028
C VFUNCT 00029
C CALCULATE CONSTANTS VFUNCT 00030
C EP0W = K1BAR*X1(I) VFUNCT 00031
C XI2 = XI(I)*XI(I) VFUNCT 00032
C FREQM = K1BAR/XMACH VFUNCT 00033
C YMUP2 = (YMUBAR + 0.5)*(YMUBAR + 0.5) VFUNCT 00034
C YMUM2 = (YMUBAR - 0.5)*(YMUBAR - 0.5) VFUNCT 00035
C VFUNCT 00036
C IF(IFLAG.EQ.0) GO TO 100 VFUNCT 00037
C VFUNCT 00038
C IMAGINARY PART VFUNCT 00039
C EXPN = (EP0W*COS(EP0W)-SIN(EP0W))/XI2 BCSAICA 00002
C IF(IND.EQ.1) EXPNV = -SIN(EP0W) BCSAICA 00003
C GO TO 200 VFUNCT 00042
C VFUNCT 00043
C REAL PART VFUNCT 00044
C 100 EXPN = (COS(EP0W)+EP0W*SIN(EP0W))/XI2 BCSAICA 00004
C IF(IND.EQ.1) EXPNV = COS(EP0W) BCSAICA 00005
C VFUNCT 00047
C 200 CONTINUE VFUNCT 00048
C C1 = 0.0 VFUNCT 00049
C C2 = 0.0 VFUNCT 00050
C IF(IUC.EQ.1) GO TO 300 VFUNCT 00051
C VFUNCT 00052
C C1R = XI2 - YMUP2 - EL2 VFUNCT 00053
C IF(ABS(C1R).LT.EPS) GO TO 300 VFUNCT 00054
C C1 = SIN(FREQM*SQRT(C1R)) VFUNCT 00055
C VFUNCT 00056
C 300 CONTINUE VFUNCT 00057
C IF(IUC.EQ.2) GO TO 400 VFUNCT 00058

```

C
C2R = XI2 - YI2 - EL2
IF(ABS(C2R).LT.EPS) GO TO 400
C2 = SIN(FREQ*MSQRT(C2R))
C
400 CONTINUE
FXIV(I) = EXPN * (C1-C2)
IF(IND.EQ.1) VT(I) = (EXPNV/XI(I)) * (C1-C2)
500 CONTINUE
RETURN
END

VFUNCT 00059
VFUNCT 00060
VFUNCT 00061
VFUNCT 00062
VFUNCT 00063
VFUNCT 00054
BCSAICA 00006
BCSAICA 00007
VFUNCT 00067
VFUNCT 00068
VFUNCT 00069

```

SUBROUTINE BFUNC(X,AV,N)
C
C      X - ARGUMENT FOR THE BESSSEL FUNCTION
C      AV - EVALUATION OF BESSSEL FUNCTION
C      N - MAX ORDER OF BESSSEL FUNCTION
C
C      *MON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSPATC,NOUTP,
C           IQUFSP,MQDESC,IVPSC,IGEOSC,IWTFSC,IAICSC
COMMON /BESFUNV/ XIB(5), A(50,5)
DIMENSION AV(1)

C      FIND INTERVAL X IS IN
IF(X.GE.XIB(1).AND.X.LE.XIB(5)) GO TO 50
EPS = 1.0E-04
I = 1
IF (ABS(X-XIB(I)) .LE. EPS*XIB(I)) GO TO 150
I= 5
IF (ABS(X-XIB(I)) .LE. EPS*XIB(I)) GO TO 150
WRITE (NT6,9005) X,XIB(1),XIB(5)
9005 FORMAT( 68H0*** ERROR - THE ARGUMENT FOR A BESSSEL FUNCTION IS OUT
1OF RANGE. *** 14H0ARGUMENT = F11.6/14H LOWER LIMIT = F11.6 /
2    14H UPPER LIMIT = F11.6 )
CALL FLUSH(1)
50 CONTINUE
I = 2
100 IF(X-XIB(I)) 200,150,125
125 CONTINUE
I = I + 1
GO TO 100
150 CONTINUE
C
C      X EQUALS XIB(I) DO NOT INTERPOLATE
C
N = 1
160 CONTINUE
AV(N) = A(N,I)
IF(A(N+1,I).EQ.0) GO TO 400
N = N + 1
GO TO 160
200 CONTINUE
DXX = (X-XIB(I-1))/(XIB(I) - XIB(I-1))
N = 1
300 CONTINUE
AV(N) = A(N,I-1) + DXX * (A(N,I) - A(N,I-1))
IF(A(N+1,I-1).EQ.0.) GO TO 400
N = N + 1
GO TO 300
400 CONTINUE
RETURN
END

```

BFUNC	00002
BFUNC	00003
BFUNC	00004
BFUNC	00005
BFUNC	00006
BFUNC	00007
FILES	00002
FILES	00003
BCSAICB	00006
BFUNC	00010
BFUNC	00011
BFUNC	00012
BCSAICB	00007
BFUNC	00014
BFUNC	00015
BFUNC	00016
BCSAICB	00008
BFUNC	00018
BCSAICB	00009
BFUNC	00020
BFUNC	00021
BFUNC	00022
BFUNC	00023
BFUNC	00024
BFUNC	00025
BFUNC	00026
BFUNC	00027
BFUNC	00028
BFUNC	00029
BFUNC	00030
BFUNC	00031
BFUNC	00032
BFUNC	00033
BFUNC	00034
BFUNC	00035
BFUNC	00036
BFUNC	00037
BFUNC	00038
BFUNC	00039
BFUNC	00040
BFUNC	00041
BFUNC	00042
BFUNC	00043
BFUNC	00044
BFUNC	00045
BFUNC	00046
BFUNC	00047
BFUNC	00048
BFUNC	00049
BFUNC	00050

OVERLAY (AFMBOX,1,5)

PROGRAM NAVPMBX

C
C THIS SECTION CONTROLS THE COMPUTATION OF BOX NORMAL WASH
C VALUES AND VELOCITY POTENTIAL DIFFERENCES. THE NORMAL WASH
C VALUES MAY BE PRINTED, BUT OTHERWISE ARE NOT SAVED. THE
C VELOCITY POTENTIAL DIFFERENCES ARE PLACED ON SCRATCH FILE
C IVPS, TWO MATRICES PER MODE SHAPE
C
COMMON PKERNL(1640)
COMPLEX PKERNL
COMMON /CONTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM, PRVMODE, DIHW, DIHT, CONTRL 00002
1 DEFAULT
LOGICAL PRVGEOM, PRVMODE, DIHW, DIHT, DEFAULT
COMMON /PROBLM/ XMAHC,NMOCES,NTSLOP,NKVALS,SMOOTH,NDEC,CDFIT,
1 EXAIC,SUBDV,PLYWOOD
LOGICAL SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NEURF,
1 B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,
2 MXBW,MXBW,MYBW,MYBBW,MXBW,MYSBW,MYBSW,MYBBSW,
3 IXBW,XCENTR
LOGICAL COPLAN
COMMON /GEOM/ TLAX,TLAZ,PSIT,MXB7,MYBT,MYBBT,MXBST,MYBST,
1 MYBBST,IXBT,IXBST,CARL
COMMON / KERN / ERR,MXSKRN,IPKERN,NPLKRN,NSPATK,NRCWEA
COMMON /KVAL / IKVAL, XKVAL(20), XKS(20)
COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSFAIC,NOUTP,
1 IOUFS,NCDESC,IVPS,IGEOSC,IWTFSC,IAICSC
COMMON /IOCONT/ OPLAIC,OSPAIC,WTGEOM,WTGNAF,WTSI,WTLB,PRBOX,
1 PRPAIC,PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,
2 PRBL,PRDCP,PRGNAC,PRGNAC,PRSL,PRLW,PRNW,PRCM
EQUIVALENCE (PRW, PRDW)
LOGICAL OPLAIC,OSPAIC,WTGEOM,WTGNAF,WTSI,WTLB,PRBOX,PRPAIC,
1 PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAC,
2 PRDCP,PRGNAC,PRW,PRLW,PRNW,PRCM
COMMON /TAPEIO/ NFS,NMS,LS,NMR,ID(20),NID,ITYPE,LRS,LWS,M,N,
1 PARM(10),IRR
DIMENSION IPARM(10)
EQUIVALENCE (PARM,IPARM)
COMMON / MODES/ SYM,SYMT,MTYPEW,MTYPEP
COMMON /ARRAYS/ KBXCW,LBXCDW,LBXOC,KBXCDT,LBXCDT,KJALPH,LJALPH,
1 KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFSL,KELPHI,
2 LMODES,KPNTSD,LPNTRM,KSDW,LSDW,KPNTDW,LPNTRW,
3 KDW,LDW,KTVP,LTVP
COMMON /SAMPLE/ ISAMPLE,ICHORD(10),IBOXF(10),IBOXL(10),ZLOC(10)
COMMON /PAICS / NAK, NTK, NRWTK, XWTK, PAIC(4,50)
INTEGER PAIC
DIMENSION NK(4)
EQUIVALENCE (NAK,NK(1))
COMMON /MUACIS/ YBAR,EL,MUAC(2,50),NRWBS,SURF,
1 YBARL,ELL,MUACL(2,50),NRWBSL,SURFL,PSIDIF
LOGICAL SURF,BURFL
COMMON /AICS / XKVL, C(1640),W(1640),V(1640)
COMPLEX C, W, V
DELPHI(LMODES),TVP(LTVP),TEXLOC(LTVP)
COMMON /DELTAP/ DELPHI(1000), TVP(250), TEXLOC(250), FEXLOC(250),
1 IPNTRM(2,100),NPNTR, IOWLP
NWPMBX 00003
NWPMBX 00004
NWPMBX 00005
NWPMBX 00006
NWPMBX 00007
NWPMBX 00008
NWPMBX 00009
NWPMBX 00010
NWPMBX 00011
NWPMBX 00012
CONTRL 00002
CONTRL 00003
CONTRL 00004
PROBLM 00002
PROBLM 00003
PROBLM 00004
GEOMTY 00002
GEOMTY 00003
GEOMTY 00004
GEOMTY 00005
GEOMTY 00006
GEOMTY 00007
GEOMTY 00008
KERN 00002
KVAL 00002
FILES 00002
FILES 00003
IOCONT 00002
IOCONT 00003
BCSFNB 00001
IOCONT 00005
IOCONT 00006
IOCONT 00007
BCSFNB 00002
TAPEIO 00002
TAPEIO 00003
TAPEIO 00004
TAPEIO 00005
MODCOM 00002
ARRAYS 00002
ARRAYS 00003
ARRAYS 00004
ARRAYS 00005
SAMPLE 00002
PAICS 00002
PAICS 00003
PAICS 00004
PAICS 00005
MUACIS 00002
MUACIS 00003
MUACIS 00004
AICS 00002
AICS 00003
DELTAP 00002
DELTAP 00003
DELTAP 00004

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COMPLEX      DELPHI,      TVP          DELTAP 00005
C           DEFSL(2,LMODES)    DELTAP 00006
DIMENSION    DEFSL(2,1000)    DELTAP 00007
EQUIVALENCE  (DELPHI(81), DEFSL)   DELTAP 00008
C           ARRAYS DELPHI AND DEFSL ARE DELTAP 00009
C           EQUIVALENCED TO GIVE A 2 ROW UN-OVERLAPPED SECTION DELTAP 00010
DIMENSION TVPX(500), XINITX(2)      FTNXI  00051
COMPLEX XINIT      FTNXI  00052
EQUIVALENCE (TVP,TVFX), (XINIT,XINITX)  FTNXI  00053
COMMON /NWASHES/ IPNTDW(2,100), ENRUS(1275), ENRLS(1275), IOVLAPN NWASHES 00002
COMPLEX          ENRUS,        ENRLS       NWASHES 00003
COMMON /SNWASH/  IPNTSD(2,50), ENSUBD(2,600), IPNTIN,IPNTOT,IPNTLS SNWASH 00002
C           IPNTSD(IPNTSD), ENSUBD(2*LSDW) SNWASH 00003
COMPLEX ENSUBD      SNWASH 00004
C           IBOKW(LBXCDW,LBOXC), WHERE LBOXC = LSDW/20 BXCODES 00002
COMMON /BXCDES/ IBOKW(150,8)        BXCODES 00003
C           IBOKW IS USED FOR BOTH WING AND TAIL BOX CODES BXCODES 00004
C           IPNTRM(2,NROWS), IPNTDW(2,NROWS), IPNTSD(2,NSROWS) NWVPMBX 00033
C           COMMON /LROT / LROT NWVPMBX 00034
COMMON /CHECKPR/ DPPCPR, GEOCPR, MDCPR, AICCP, NMSCPR, SMCPR, GAFCP, NWVPMBX 00035
LOGICAL DPICPR, GEOCPR, MDCPR, AICCP, NMSCPR, SMCPR, GAFCP
LOGICAL CHECKPR
EQUIVALENCE (CHECKPR, NMSCPR)
DIMENSION TITL(3)          NWVPMBX 00036
DIMENSION PARMW(10),IPARMW(10)    CHECKPR 00002
EQUIVALENCE (PARMW,IPARMW)      CHECKPR 00003
LOGICAL MXWRIT, RANDU, MXRD, MXWRT NWVPMBX 00038
DATA TVFX / 500* 60000000000200377777B /
DATA MXWRIT,RANDU, MXRD,MXWRT / .F.,.F.,.F.,.F. /
DATA XINITX / 2* 377040000000000000000B /
C           PSIDIF = PSIT - PSTW NWVPMBX 00040
C           IPNTLS = LPNTSD NWVPMBX 00041
C           LMOWS = LSDW NWVPMBX 00042
C           LROT = NSUPDV + NSUBCN NWVPMBX 00043
C           NTVP = (MYBSW + MYBST) FTNXI  00054
C           PARMW(2) = B1 NWVPMBX 00044
C           PARMW(3) = XMACH FTNXI  00055
C           IPARMW(4) = NMCCES FTNXI  00055
C           NWVPMBX 00047
C           NWVPMBX 00048
C           NWVPMBX 00049
C           NWVPMBX 00050
C           NWVPMBX 00051
C           NWVPMBX 00052
C           NWVPMBX 00053
C           NWVPMBX 00054
C           NWVPMBX 00055
C           NWVPMBX 00056
C           NWVPMBX 00057
C           NWVPMBX 00058
C           NWVPMBX 00059
C           NWVPMBX 00060
C           NWVPMBX 00061
C           NWVPMBX 00062
C           NWVPMBX 00063
C           NWVPMBX 00064
C           NWVPMBX 00065
C           NWVPMBX 00066
C           NWVPMBX 00067
C           NWVPMBX 00068
C           NWVPMBX 00069
C           NWVPMBX 00070
CALL READMX (IGEO6C, MXRD, .F., NFS,NMS,LS, NMR, K, NID, ID, ITYPE, NWVPMBX 00071

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1      LRS, IBOXW, M,N, PARM, IRR)          NWVPMBX 00072
IF (IRR .NE. 0) GO TO 910                 NWVPMBX 00073
MXBBS = N                                 NWVPMBX 00074
IF (INSURF .EQ. 1 .OR. COPLAN) GO TO 20   NWVPMBX 00075
C
C      TAIL BOX CODES                     NWVPMBX 00076
MXARRY = 10HTAIL CODES                   NWVPMBX 00077
CALL RDINIT                                NWVPMBX 00078
ITYPE = SHMIXED                            NWVPMBX 00079
CALL READMX (IGEOSC, MXRD, .F., NFS,NMS,LS, NMR, K, NID, ID, ITYPE, NWVPMBX 00080
1      LRS, IBOXW(MXBBS+1,1), M,N, PARM, IRR) NWVPMBX 00081
IF (IRR .NE. 0) GO TO 910                 NWVPMBX 00082
ISUBT = MXBBS + 2 - IXBST                NWVPMBX 00083
C      = SUBSCRIPT FOR IBOXW TO GET TAIL CODES NWVPMBX 00084
NWVPMBX 00085
C      20 CONTINUE                         NWVPMBX 00086
MXARRY 10H      FEXLOC                  NWVPMBX 00087
CALL RDINIT                                NWVPMBX 00088
ITYPE = SHMIXED                            NWVPMBX 00089
CALL READMX(IGEOSC, MXRD,.F., NFS,NMS,LS, NMR, 1, NID, ID, ITYPE, NWVPMBX 00090
1      LRS, FEXLOC, M,N, PARM, IRR)        NWVPMBX 00091
IF (IRR .NE. 0) GO TO 910                 NWVPMBX 00092
C      MXARRY =10H      TEXLOC            NWVPMBX 00093
CALL RDINIT                                NWVPMBX 00094
ITYPE = SHMIXED                            NWVPMBX 00095
CALL READMX(IGEOSC, MXRD,.F., NFS,NMS,LS, NMR, 1, NID, ID, ITYPE, NWVPMBX 00096
1      LRS, TEXLOC, M,N, PARM, IRR)        NWVPMBX 00097
IF (IRR .NE. 0) GO TO 910                 NWVPMBX 00098
IF(M .NE. 1) GO TO 930                   NWVPMBX 00100
C      IF (PSIW .NE. 0 .AND. DIHW) GO TO 30 NWVPMBX 00101
IF (ISMPLW .NE. 0) GO TO 30               NWVPMBX 00102
IF (INSURF .EQ. 1) GO TO 40               NWVPMBX 00103
IF (CAPL .NE. 0) GO TO 30               NWVPMBX 00104
IF (PSIDIF .NE. 0) GO TO 30              NWVPMBX 00105
IF (PSIT .NE. 0 .AND. DIHT) GO TO 30    NWVPMBX 00106
GO TO 40                                  NWVPMBX 00107
C      READ THE ARRAY OF AIC TABLE & CONTENTS NWVPMBX 00108
30 CONTINUE                               NWVPMBX 00109
MXARRY = 9HSPAT. TOC                      NWVPMBX 00110
CALL RDINIT                                NWVPMBX 00111
NMS = 2                                    NWVPMBX 00112
K = 4                                    NWVPMBX 00113
CALL READMX (IGEOSC, MXRD, .F., NFS,NMS,LS, NMR, K, NID, ID, ITYPE, NWVPMBX 00114
1      LRS, PAIC, M,N, PARM, IRR)        NWVPMBX 00115
IF (IRR .NE. 0) GO TO 910                 NWVPMBX 00116
DO 35 I = 1,4                           NWVPMBX 00117
NK(I) = IPARM(I+2)                        NWVPMBX 00118
35 CONTINUE                               NWVPMBX 00119
40 CONTINUE                               NWVPMBX 00120
REWIND IGEOSC                            NWVPMBX 00121
C      GET POINTER ARRAY FOR MODES       NWVPMBX 00122
CALL RDINIT                                NWVPMBX 00123
ITYPE = SHMIXED                            NWVPMBX 00124
NWVPMBX 00125
NWVPMBX 00126
NWVPMBX 00127
NWVPMBX 00128

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CALL READMX(MODESC, MXRD,.F., NFS,NMS,LS, NMR, 2, NID, ID, ITYPE, NWVPMBX 00129
1 LRS, IPNTRM, M,N, PARM, IRR) NWVPMBX 00130
IF (IRR .NE. 0) GO TO 912 NWVPMBX 00131
IF (M .NE. 2) GO TO 931 NWVPMBX 00132
NPNTRS = N NWVPMBX 00133
IOMLAP = IPARM(3) NWVPMBX 00134
C IOMLAP = NUMBER OF ROWS TO ALLOW FOR TAIL OVERLAP (TAIL ONLY) NWVPMBX 00135
C NPNTRS = TOTAL NUMBER OF ROWS ON BOTH SURFACES, + 1 . NWVPMBX 00136
C (INCLUDES OVERLAP IF SPATIAL) NWVPMBX 00137
C NWVPMBX 00138
C SET UP POINTER ARRAY FOR UNSUBDIVIDED DOWNWASHES NWVPMBX 00139
IP = 1 NWVPMBX 00140
MYBB = MYBBW NWVPMBX 00141
IF (COPLAN) MYBB = MAX0(MYBB,MYBBT) NWVPMBX 00142
MXBBS = MYBB*NSUBDV NWVPMBX 00143
MXBB = MXBBW NWVPMBX 00144
IF (COPLAN) MXBB = MXBT NWVPMBX 00145
CALL POINTR(1,MXBB,MYBB, .F., .T., IBOXW,LBXCDW, LPNTDW,1, NWVPMBX 00146
1 , IP, IPNTDW) NWVPMBX 00147
MXB = MXBW NWVPMBX 00148
IF (COPLAN) MXB = MXBT NWVPMBX 00149
MYB = MYBW NWVPMBX 00150
IF(COPLAN) MYB = MYBT NWVPMBX 00151
IOMLPN = 0 NWVPMBX 00152
IF (INSURF .EQ. 1 .OR. COPLAN) GO TO 50 NWVPMBX 00153
MXB = MXBT NWVPMBX 00154
IXBUT = (IXBT-IXBW)/NSUBDV + 1 NWVPMBX 00155
IP = MXBBW + 1 NWVPMBX 00156
IPNT = IPNTDW(1,IP) NWVPMBX 00157
CALL POINTR(IXBUT, MXBT-IXBUT+1, MYBBT, .F., .T., IROWW(ISUBT,1), NWVPMBX 00158
1 LBXCDW,LPNTDW, IPNT, IP, IPNTDW) NWVPMBX 00159
IF (MXBBW .GE. IXBUT) IOMLPN = MXBBW - IXBUT + 1 NWVPMBX 00160
50 CONTINUE NWVPMBX 00161
C NWVPMBX 00162
C LOOP ON MODE SHAPES NWVPMBX 00163
DO 500 IMODE = 1,NMOCES NWVPMBX 00164
C NWVPMBX 00165
C ZERO OUT THE DOWNWASH AND VELOCITY POTENTIAL ARRAYS NWVPMBX 00166
LIM = IPNTDW(1,IP) - 1 NWVPMBX 00167
DO 80 I = 1,LIM NWVPMBX 00168
ENRUS(I) = XINIT NWVPMBX 00169
ENRLS(I) = XINIT NWVPMBX 00170
80 CONTINUE NWVPMBX 00171
LIM = IPNTRM(1,NPNTRS) - 1 NWVPMBX 00172
DO 85 I = 1,LIM NWVPMBX 00173
DELPHI(I) = (0.,0.) NWVPMBX 00174
85 CONTINUE NWVPMBX 00175
C NWVPMBX 00176
C NWVPMBX 00177
C READ IN MODE SHAPE NWVPMBX 00178
CALL RDINIT NWVPMBX 00179
MXARRY = 10HMODE SHAPE NWVPMBX 00180
ITYPE = 4HREAL NWVPMBX 00181
CALL READMX(MODESC, MXRD,.F., NFS,NMS,LS, NMR, 2, NID, ID, ITYPE, NWVPMBX 00182
1 LRS, DEFSL, M,N, PARM, IRR) NWVPMBX 00183
IF (IRR .NE. 0) GO TO 912 NWVPMBX 00184
IF (M .NE. 2) GO TO 931 NWVPMBX 00185

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C          COMPUTE DOWNWASHES AND VELOCITY POTENTIALS FOR ONE MODE      NAVPMBX 00186
C          CALL VELPOT ( IBOXW,LBXCDW, PKERNL(IPKERN), PKERNL, .T., DIHW)  NAVPMBX 00187
C          OPTIONAL PRINT OF NORMAL WASHES                               NAVPMBX 00188
C          IF ( .NOT. (CHECKPR .OR. PRNW) ) GO TO 90                      NAVPMBX 00189
C          IF (NSUBDV .EQ. 1) GO TO 87                                     NAVPMBX 00190
C          IF (.NOT. CHECKPR) GO TO 87                                     NAVPMBX 00191
C          TITL(1) = 10HEN SUBDIVI                                         NAVPMBX 00192
C          TITL(2) = 10HDED, UPPER                                         NAVPMBX 00193
C          TITL(3) = 10H, PARTIAL                                         NAVPMBX 00194
C          IF (IPNTIN .LT. IPNTOT) GO TO 86                           NAVPMBX 00195
C          CALL PRINTR(TITL,IMODE,ENSUBD,2, 1, IPNTIN-1, MYBBS,IPNTSD)  NAVPMBX 00196
C          GO TO 87                                                 NAVPMBX 00197
86 CONTINUE
C          CALL PRINTR(TITL,IMODE,ENSUBD,2,IPNTOT,IPNTLS-1,MYBBS,IPNTSD)  NAVPMBX 00198
C          CALL PRINTR(TITL,IMODE,ENSUBD,2, 1, IPNTIN-1, MYBBS,IPNTSD)  NAVPMBX 00199
87 CONTINUE
C          TITL(1) = 10H WING UPPER                                         NAVPMBX 00200
C          TITL(2) = 10H SURFACE N                                         NAVPMBX 00201
C          TITL(3) = 10H NORMAL WASH                                       NAVPMBX 00202
C          IF (COPLAN) TITL(1) = 10H WING/TA                            NAVPMBX 00203
C          IF (COPLAN) TITL(2) = 10HIL UPPER N                           NAVPMBX 00204
C          CALL PRINTR( TITL,IMODE, ENRUS,1, 1,MXBB,MYBB, IPNTDW)  NAVPMBX 00205
C          IF (.N. COPPLAN) TITL(1) = 10H WING LOWER                   NAVPMBX 00206
C          IF ( COPPLAN) TITL(2) = 10HIL LOWER N                         NAVPMBX 00207
C          CALL PRINTR( TITL,IMODE, ENRLS,1, 1,MXBB,MYBB, IPNTDW)  NAVPMBX 00208
C
C          90 CONTINUE
C          IF (NSURF .EQ. 1 .OR. COPPLAN) GO TO 140
C
C          DETERMINE WHICH WING SURFACE CONTRIBUTES TO THE TAIL.
C          IF (CAPL .GT. 0) GO TO 130                                     NAVPMBX 00209
C          IF (CAPL .EQ. 0 .AND. PSIDIF .GT. 0 ) GO TO 130               NAVPMBX 00210
C          THE LOWER WING SURFACE CONTRIBUTES TO THE TAIL
C          LIM = IPNTDW(1,MYBBW1) - 1                                     NAVPMBX 00211
C          DO 120 I = 1,LIM                                              NAVPMBX 00212
C          ENRUS(I) = ENRLS(I)
C
120 CONTINUE
C
C          COMPUTE THE TAIL NORMAL WASHES AND VELOCITY POTENTIALS
130 CONTINUE
C          CALL VELPOT ( IBOXW(ISURT,1),LBXCDW, PKERNL(IPKERN), PKERNL,
C                         .F., DIHT)                                     NAVPMBX 00213
C          OPTIONAL PRINT OF NORMAL WASHES
C          IF ( .NOT. (CHECKPR .OR. PRNW) ) GO TO 135                  NAVPMBX 00214
C          IF (NSUBDV .EQ. 1 .OR. .NOT. CHECKPR) GO TO 133            NAVPMBX 00215
C          TITL(1) = 10HEN SUBDIVI                                         NAVPMBX 00216
C          TITL(2) = 10HDED, UPPER                                         NAVPMBX 00217
C          TITL(3) = 10H, PARTIAL                                         NAVPMBX 00218
C          IF (IPNTIN .LT. IPNTOT) GO TO 131                           NAVPMBX 00219
C          CALL PRINTR(TITL,IMODE,ENSUBD,2, 1, IPNTIN-1, MYBBS,IPNTSD)  NAVPMBX 00220
C          GO TO 133                                                 NAVPMBX 00221
131 CONTINUE
C          CALL PRINTR(TITL,IMODE,ENSUBD,2,IPNTOT,IPNTLS-1,MYBBS,IPNTSD)  NAVPMBX 00222
C          CALL PRINTR(TITL,IMODE,ENSUBD,2, 1, IPNTIN-1, MYBBS,IPNTSD)  NAVPMBX 00223
133 CONTINUE
C          TITL(1) = 10HTAIL UPPER                                         NAVPMBX 00224

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CALL PRINTR( TITL,IMODE, ENRLS,1, IXBUT,MXBT,MYBT,          NAVPMBX 00243
  1           IPNTRW(1,IOVLAPN + 1) )                         NAVPMBX 00244
  TITL(1) = 10HTAIL LOWER                                     NAVPMBX 00245
  CALL PRINTR( TITL,IMODE, ENRLS,1, IXBUT,MXBT,MYBT,          NAVPMBX 00246
  1           IPNTRW(1,IOVLAPN + 1) )                         NAVPMBX 00247
135 CONTINUE                                                 NAVPMBX 00248
C
C      WRITE THE RESULTS ON SCRATCH FILES
140 CONTINUE                                                 NAVPMBX 00249
  PARMW(I) = YKVL                                         NAVPMBX 00250
  DO 210 I = 1,4                                         NAVPMBX 00251
210 PARM(I) = PARMW(I)                                     NAVPMBX 00252
C
C      XX      WRITE DELTA PHIS, TEMPORARILY AS A REAL MATRIX, WRTETP    XX
  P = 2                                                       NAVPMBX 00253
  !TYPE = 4REAL                                           NAVPMBX 00254
  N = IPNTRM(1,MXB+IOVLAP+1) - 1                         NAVPMBX 00255
  CALL WRTEMX(IPVSC, .F., .F., NFS,NMS,LS, NMR,LWS, 2, ID, DELPHI,
  1     ITYPE, M,N,FARM, IRR)                            NAVPMBX 00256
  IF (IRR .NE. 0) GO TO 920                               NAVPMBX 00257
C
  M = 2                                                       NAVPMBX 00258
  N = NTVP                                                 NAVPMBX 00259
  CALL WRTEMX(IPVSC, .F., .F., NFS,NMS,LS, NMR, LWS, 2, ID, TVP,
  1     ITYPE, M,N,FARM, IRR)                            NAVPMBX 00260
  IF (IRR .NE. 0) GO TO 920                               NAVPMBX 00261
C
  IF (.NOT. PRVP) GO TO 230                             NAVPMBX 00262
  TITL(1) = 8H   WING                                     NAVPMBX 00263
  TITL(2) = 10HVELOCITY P                                NAVPMBX 00264
  TITL(3) = 10HOTENTIALS                                 NAVPMBX 00265
  M = MXBW                                               NAVPMBX 00266
  IF (.N. COPLAN) GO TO 220                             NAVPMBX 00267
  TITL(1) = 10HMING/TAIL                                NAVPMBX 00268
  M = NPNTRS - 1                                         NAVPMBX 00269
220 CALL PRINTR( TITL, IMODE, DELPHI,1, 1,M, MYB, IPNTRM)
  IF (NSURF .EQ. 1 .OR. COPLAN) GO TO 230
  TITL(1) = 8H   TAIL                                    NAVPMBX 00270
  CALL PRINTR( TITL, IMODE, DELPHI, 1, IXBUT,MXBT, MYBT,
  1           IPNTRW(1,IOVLAPN+1) )                         NAVPMBX 00271
230 CONTINUE                                                 NAVPMBX 00272
C      ARE SAMPLE WASHES DESIRED -
  IF (NSURF .EQ. 2 .OR. ISMPLW .EQ. 0) GO TO 500
C      YES.  IS SAMPLE WASH PRINTOUT DESIRED
  IF (.NOT. (PRDW .OR. PRSW)) GO TO 500
C      LOOP ON CHORDS FOR WHICH SAMPLE-WASH IS DESIRED
  DO 300 JCHRD = 1,ISMPLW                               NAVPMBX 00273
  JT = 1CHRD(JCHRD)                                     NAVPMBX 00274
  IFRST = 1BOKF(JCHRD)                                  NAVPMBX 00275
  ILAST = 1BOKL(JCHRD)                                  NAVPMBX 00276
  CALL SMPLW( 1BOXW,LBXCDW,JCHRD,JT,IFRST,ILAST)
300 CONTINUE                                                 NAVPMBX 00277
C
  500 CONTINUE                                                 NAVPMBX 00278
C      END OF LOOP ON MODE SHAPES, FROM STATEMENT 50*
C
  RETURN                                                 NAVPMBX 00279

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C          DIAGNOSTICS - ALL CALL FLUSH                               NAVPMBX 00300
C          READ FROM SCRATCH FILE                                NAVPMBX 00301
C          WRITING TO SCRATCH FILE                                NAVPMBX 00302
C          INCORRECT DIMENSIONS READ                            NAVPMBX 00303
C          ERROR DETECTED READING A MATRIX                      NAVPMBX 00304
C          ERROR DETECTED WRITING A MATRIX                      NAVPMBX 00305
C          MATRIX DESCRIPTION                                 NAVPMBX 00306
C          CALL FLUSH(1)                                     NAVPMBX 00307
C          DIAGNOSTIC FORMATS                                NAVPMBX 00308
C          FORMAT(47H0*** ERROR WHILE READING GEOMETRY SCRATCH FILE ,A10,    NAVPMBX 00309
C          1 4H *** )                                         NAVPMBX 00310
C          FORMAT(44H0*** ERROR WHILE READING MODES SCRATCH FILE ,A10,      NAVPMBX 00311
C          1 4H *** )                                         NAVPMBX 00312
C          FORMAT(51H0*** ERROR WHILE WRITING VELOCITY POTENTIAL SCRATCH   NAVPMBX 00313
C          1 4H *** )                                         NAVPMBX 00314
C          I = 1                                           NAVPMBX 00315
C          GO TO 932                                         NAVPMBX 00316
C          I = 2                                           NAVPMBX 00317
C          WRITE (NT6,9300) I                                NAVPMBX 00318
C          IF (MXRD) GO TO 960                           NAVPMBX 00319
C          GO TO 962                                         NAVPMBX 00320
C          WRITE (NT6,9500) IRR                           NAVPMBX 00321
C          IF (MXRD) GO TO 960                           NAVPMBX 00322
C          WRITE (NT6,9520) IRR                           NAVPMBX 00323
C          IF (MXWT) GO TO 960                           NAVPMBX 00324
C          GO TO 962                                         NAVPMBX 00325
C          WRITE (NT6,9630) MXARRY                         NAVPMBX 00326
C          GO TO 962                                         NAVPMBX 00327
C          WRITE (NT6,9600) (ID(I),I=1,10),(ID(I),I=1,10)  NAVPMBX 00328
C          WRITE (NT6,9622) PARM,PARM                      NAVPMBX 00329
C          WRITE (NT6,9614) NMR,NMR,LRS,LWS                NAVPMBX 00330
C          GO TO 964                                         NAVPMBX 00331
C          WRITE (NT6,9620) ID(1),ID(2)                  NAVPMBX 00332
C          WRITE (NT6,9622) PARM,PARM                      NAVPMBX 00333
C          WRITE (NT6,9624) NFS,NMS                        NAVPMBX 00334
C          WRITE (NT6,9640) ITYPE,M,N                     NAVPMBX 00335
C          WRITE (NT6,9630) MXARRY                         NAVPMBX 00336
C          GO TO 990                                         NAVPMBX 00337
C          WRITE (NT6,9900)                                NAVPMBX 00338
C          CALL FLUSH(1)                                    NAVPMBX 00339
C          CALL FLUSH(1)                                    NAVPMBX 00340
C          CALL FLUSH(1)                                    NAVPMBX 00341
C          CALL FLUSH(1)                                    NAVPMBX 00342
C          CALL FLUSH(1)                                    NAVPMBX 00343
C          CALL FLUSH(1)                                    NAVPMBX 00344
C          CALL FLUSH(1)                                    NAVPMBX 00345
C          CALL FLUSH(1)                                    NAVPMBX 00346
C          CALL FLUSH(1)                                    NAVPMBX 00347
C          CALL FLUSH(1)                                    NAVPMBX 00348
C          CALL FLUSH(1)                                    NAVPMBX 00349
C          CALL FLUSH(1)                                    NAVPMBX 00350
C          CALL FLUSH(1)                                    NAVPMBX 00351
C          FORMAT(47H0*** ERROR WHILE READING GEOMETRY SCRATCH FILE ,A10,    NAVPMBX 00352
C          1 4H *** )                                         NAVPMBX 00353
C          FORMAT(44H0*** ERROR WHILE READING MODES SCRATCH FILE ,A10,      NAVPMBX 00354
C          1 4H *** )                                         NAVPMBX 00355
C          FORMAT(51H0*** ERROR WHILE WRITING VELOCITY POTENTIAL SCRATCH   NAVPMBX 00356

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1 GH FILE ,A10, 4H ***)	NNVPMBX 00357
9300 FORMAT(1H0, 48H*** MATRIX READ ERROR. THE M DIMENSION SHOULD	NNVPMBX 00358
1 4H BE ,I2, 4H ***)	NNVPMBX 00359
9500 FORMAT(1SH0 *** ERROR CODE ,I5, 28H WHILE READING THE FOLLOWING	NNVPMBX 00360
1 S1H MATRIX ***)	NNVPMBX 00361
9520 FORMAT(1SH0 *** ERROR CODE ,I5, 28H WHILE WRITING THE FOLLOWING	NNVPMBX 00362
1 S1H MATRIX ***)	NNVPMBX 00363
9600 FORMAT(5X, #MATRIX ID = *,10A10 / (20X,10A10))	NNVPMBX 00364
9614 FORMAT(5X,22HMATRIX INDEX (NAME) = ,I5,2H (A10,1H) /	NNVPMBX 00365
1 5X,33HLEVEL NUMBER READ (OR WRITTEN) = 02,2H (,02,1H))	FTNX1 00056
9620 FORMAT(5X, #MATRIX ID = *, A10, I10)	NNVPMBX 00367
9622 FORMAT(5X,11HPARAMETERS, 1UE11.3 /10X, 9H(INTEGER), I7,9I11)	NNVPMBX 00368
9624 FORMAT(5X,15HFILE SPACING = ,I3, 19H, MATRIX SPACING = ,I3)	NNVPMBX 00369
9630 FORMAT(5X,A10,21H ARRAY WAS BEING USED)	NNVPMBX 00370
9640 FORMAT(5X, #MATRIX TYPE - *,A10, *, DIMENSIONED (#I4,* X#,I4,*))	NNVPMBX 00371
9900 FORMAT(54H*** ERROR OCCURRED DURING VELOCITY POTENTIAL CALCULAT	NNVPMBX 00372
1 SHIONS ***)	NNVPMBX 00373
C	NNVPMBX 00374
ENC	NNVPMBX 00375

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SUBROUTINE VELPOT(IBOX,LBXCD, PKERNL,SKERNL, WING, DIHS)          VELPOT 00002
C
C   VELPOT IS CALLED ONCE FOR EACH MODE, TO COMPUTE NORMAL WASHES    VELPOT 00003
C   AND VELOCITY POTENTIALS FOR THAT MODE.                            VELPOT 00004
C
C   IBOX   = BOX CODES FOR THE SURFACE                                VELPOT 00005
C   LBXCD = BOX CODE ARRAY ROW DIMENSION                            VELPOT 00006
C   PKERNL = ARRAY CONTAINING C(NU,MU,0)                            VELPOT 00007
C   SKERNL = ARRAY CONTAINING SUBDIVIDED C(NU,MU,0)                  VELPOT 00008
C   WING   = LOGICAL, .T. FOR WING OR COPLANAR, .F. FOR TAIL        VELPOT 00009
C   DIHS   = LOGICAL, .T. TO INCLUDE DIHEDRAL EFFECTS, .F. TO      VELPOT 00010
C           IGNORE, FOR WING/WING (TAIL/TAIL)                          VELPOT 00011
C
C   DIMENSION IBOX(LBXCD,1), ICODE(5)                                 VELPOT 00012
C   COMPLEX PKERNL(1), SKERNL(1)                                     VELPOT 00013
C   LOGICAL WING,DIHS                                              VELPOT 00014
C
C   OUTPUTS -
C   DELPHI = DELTA PHI (VELOCITY POTENTIAL) ARRAY                   VELPOT 00015
C
C   COMMON PARAMETERS USED
C   NSUBDV = NUMBER OF SUBDIVISIONS                                VELPOT 00016
C   B1     = BOX LENGTH                                              VELPOT 00017
C   B1S    = SUBDIVIDED BOX LENGTH                                 VELPOT 00018
C
C   COMMON /FILES / NT5,NT6,INTAPE,INFS/,NPLAIC,NSPAIC,NOUTH,       VELPOT 00019
C   1      IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC                FILES 00002
C   COMMON /ARRAYS/ KBXCDW,LBXCDW,IBOXC,KBXCDT,LBXCDT,KJALPH,LJALPH,  FILES 00003
C   1      KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFSL,KELPHI,        ARRAYS 00002
C   2      LMODES,KPNTSD,LPNTRD,KSDW,LSDW,KPNTDW,LPNTDW,            ARRAYS 00003
C   3      KDW,LDW,KTVP,LTVP                                         ARRAYS 00004
C
C   COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,      GEOMTY 00005
C   1      B1,B1BETA,B1S,B1STAS,WLAX,WLAZ,PSIW,                      GEOMTY 00006
C   2      MYBW,MXBHW,MYBW,MYBBW,MXBHW,MYBSW,MYBBSW,                 GEOMTY 00007
C   3      IXBW,YCENTR                                         GEOMTY 00008
C
C   LOGICAL COPLAN
C   COMMON /GEOM2 / TLAX,TLAZ,PSIT,MXBT,MYBT,MXBST,MYBGT,           GEOM2  00002
C   1      MYBBS2,IXBT,IXBT,CAFL                                     GEOM2  00003
C   COMMON /MOCES/ SYM,SYNT,MTYPEW,MTYPEF                           MOCOM  00002
C
C   COMMON /MUAICS/ YBAR,EL,MUAIC(2,50),NRCHS,SURF,                 VEL.POT 00032
C   1      YBARL,ELL,MUAICL(2,50),NROWSL,SURFL,PSIDIF               MUAICS 00002
C   LOGICAL SURF,SURFL                                         MUAICS 00003
C   COMMON /AICS / XKVL, C(1640),W(1640),V(1640)                   AICS   00004
C   COMPLEX          C, W, V                                         AICS   00005
C
C   DELPHI(LMOCES),TVP(LTVP),TEXLOC(LTV)                           DELTAP 00002
C   COMMON /DELTAP/ DELPHI(1000), TVP(250), TEXLOC(250), FEXLOC(250),  DELTAP 00003
C   1      IPNTRM(2,100),NPNTRS, IONLAP                            DELTAP 00004
C   COMPLEX          DELPHI, TVP                                     DELTAP 00005
C
C   DEFSL(2,LMODES)
C   DIMENSION DEFSL(2,1000)                                         DELTAP 00006
C   EQUIVALENCE (DELPHI(81), DEFSL)                                  DELTAP 00007
C
C   ARRAYS DELPHI AND DEFSL ARE
C   EQUIVALENCED TO GIVE A 2 ROW UN-OVERLAPPED SECTION             DELTAP 00008
C
C   COMMON /NWASHES/ IPNTDW(2,100),ENRUS(1275),ENRLS(1275),ICMLAPN  NWASHES 00002
C   COMPLEX          ENRUS, ENRLS                                    NWASHES 00003
C   COMMON /SMASH/ IPNTSD(2,50),ENSUBD(2,600),IPNTIN,IPNTOT,IPNTLS SMASH 00002

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C	PNTSD(LPNTSD), ENSUBD(2+LSCW)	SNASH 00003
C	COMPLEX ENSUBD	SNASH 00004
C	IBOXW(LBXCDW,LBOXC), WHERE LBOXC = LSCWD/20	BXCDES 00002
C	COMMON /BXCDES/ IBOXW(150,8)	BXCDES 00003
C	IBOXW IS USED FOR BOTH WING AND TAIL BOX CODES	BXCDES 00004
C	COMMON /CHECKPR/ DPPCPR,GEOPCR,MODCPR,AICCPR,NMSCPR,SMCPR,GFCPR	CHECKPR 00002
C	LOGICAL DPPCPR, GEOPCR, MODCPR, AICCPR, NMSCPR, SMCPR, GFCPR	CHECKPR 00003
C	EQUIVALENCE (CHECKPR,NMSCPR)	VELPOT 00040
C	LOGICAL CHECKPR	VELPOT 00041
C	COMPLEX AZERO, DELPH, DELPHB, DELPHC, B, SDELPH, COEF	VELPOT 00042
C	COMPLEX ENRULU(50),ENRILL(50),ENRURW(50),ENRULW(50), DPHIL(50)	VELPOT 00043
C	COMPLEX ENDIF, ENSUM, ENSRUS, ENSRLS	VELPOT 00044
C	COMPLEX ENSBD(2)	VELPOT 00045
C	EQUIVALENCE (ENSRUS,ENSBD(1)), (ENSRLS,ENSBD(2))	VELPOT 00046
C	LOGICAL CROW, FROW, LROW, CBOX, FBOX, LBOX, SUBOFF	VELPOT 00047
C	LOGICAL FULLBX(50)	VELPOT 00048
C	INTEGER WW,TT,RWT,LWT	VELPOT 00049
C	COMPLEX XINIT	VELPOT 00050
C	DIMENSION XINITX(2)	VELPOT 00051
C	EQUIVALENCE (XINIT,XINITX)	FTNXI 00057
C	DATA WW,TT,RWT,LWT /1,2,3,4/	FTNXI 00058
C	THESE VALUES MAY BE MODIFIED BY ACTUAL PAIC-- ARRAYS READ IN	FTNXI 00059
C	DATA XINITX / 2* 37704000000000000000B /	VELPOT 00052
C	SET CONSTANTS	VELPOT 00053
C	IXB = SUBDIVIDED SUBSCRIPT OF FIRST PLANFORM CONTROL POINT	VELPOT 00054
C	IXBS = SUBSCRIPT OF FIRST PLANFORM SUBDIVIDED BOX	VELPOT 00055
C	IXBU = UNSUBDIVIDED SUBSCRIPT OF FIRST PLANFORM CONTROL PT.	VELPOT 00056
C	MYBB = NUMBER OF UNSUBDIVIDED CHORDS TO CONSIDER, INCLUDING	VELPOT 00057
C	DIAPHRAGM	VELPOT 00058
C	MYBBS = NUMBER OF SUBDIVIDED CHORDS TO CONSIDER, INCL. DIAPH.	VELPOT 00059
C	MXB = LAST UNSUBDIVIDED ROW TO CONSIDER	VELPOT 00060
C	MXBS = LAST SUBDIVIDED ROW TO CONSIDER (TO LAST CONTROL PNT)	VELPOT 00061
C	IF (WING) GO TO 80	VELPOT 00062
C	IOVLP = IOVLAP	VELPOT 00063
C	IOVLPN = IOVLAPN	VELPOT 00064
C	PSIS = PSIT	VELPOT 00065
C	IXB = IXBT	VELPOT 00066
C	IXBS = IXBST	VELPOT 00067
C	IXBU = (IXBT - IXBW)/NSUBDV + 1	VELPOT 00068
C	MYBB = MYBBT	VELPOT 00069
C	MYBBS = MYBBST	VELPOT 00070
C	SYHTY = SYMT	VELPOT 00071
C	GO TO 90	VELPOT 00072
C	90 CONTINUE	VELPOT 00073
C	IOVLP = 0	VELPOT 00074
C	IOVLPN = 0	VELPOT 00075
C	PSIS = PSIW	VELPOT 00076
C	IXB = IXBW	VELPOT 00077
C	IXBS = 1	VELPOT 00078
C	IXBU = 1	VELPOT 00079
C	SYHTY = SYM	VELPOT 00080
C	IF (COPLAN .AND. NSURF .EQ. 2) GO TO 85	VELPOT 00081
C	MYBB = MYBBW	VELPOT 00082
C	MYBBS = MYBBSW	VELPOT 00083
C		VELPOT 00084
C		VELPOT 00085
C		VELPOT 00086
C		VELPOT 00087

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MXB = MXBBW          VEL POT 00088
MXBS = (MXB-1)*NSUBDV + IXBW      VEL POT 00089
MXBS = MAX0(MYBS, MXBSW)          VEL POT 00090
GO TO 100              VEL POT 00091
05 CONTINUE            VEL POT 00092
MYBB = MAX0(MYBBW, MYBBT)        VEL POT 00093
MYBBS = MAX0(MYBBSW, MYBBST)      VEL POT 00094
90 CONTINUE            VEL POT 00095
MXB = MXBT            VEL POT 00096
MXBS = MXBST           VEL POT 00097
C
100 CONTINUE            VEL POT 00098
PSIS2 = 2*PSIS          VEL POT 00099
PSIUM = PSIW + PSIT        VEL POT 00100
CPSIS2 = COS(PSIS2)        VEL POT 00101
SPSIS2 = SIN(PSIS2)        VEL POT 00102
CPSISM = COS(PSIUM)        VEL POT 00103
SPSISM = SIN(PSIUM)        VEL POT 00104
IRFB = IXB - NSUBD2       VEL POT 00105
C
= ROW OF FIRST SUBDIVIDED BOX IN THE FIRST ROW OF THE
UNSUBDIVIDED BOXES          VEL POT 00106
C
MYBBSX = ((MYBBS+NSUBD2)/NSUBDV)*NSUBDV          VEL POT 00107
IF (NSUBDV .EQ. 1) GO TO 110          VEL POT 00108
SUBOFF = .F.          VEL POT 00109
VEL POT 00110
MYBBPI = MYBB+1          VEL POT 00111
DO 105 I = 1,MYBBPI      VEL POT 00112
ENRULU(I) = (0.,0.)      VEL POT 00113
ENRULLL(I) = (0.,0.)      VEL POT 00114
ENRURW(I) = (0.,0.)      VEL POT 00115
ENRULW(I) = (0.,0.)      VEL POT 00116
FULLBX(I) = .T.          VEL POT 00117
VEL POT 00118
105 CONTINUE            VEL POT 00119
GO TO 120              VEL POT 00120
110 SUBOFF = .T.          VEL POT 00121
CROW = .T.          VEL POT 00122
LROW = .T.          VEL POT 00123
FBOX = .T.          VEL POT 00124
FBOK = .T.          VEL POT 00125
LBOK = .T.          VEL POT 00126
VEL POT 00127
C
C
LOOP ON ALL (SUBDIVIDED) ROWS OF THE SURFACE
C
120 CONTINUE            VEL POT 00128
FLIROW = FLOAT(IXBS) - 1.0      VEL POT 00129
DO 1300 IROW = IXBS,MXBS      VEL POT 00130
FLIROW = FLIROW + 1.0          VEL POT 00131
C
C
SET FLAGS FOR FIRST, CENTER AND LAST SUBDIVIDED ROW IN UN-
SUBDIVIDED ROW          VEL POT 00132
C
IF (SUBOFF) GO TO 270          VEL POT 00133
FROW = .F.          VEL POT 00134
VEL POT 00135
IF (IROW - IXB) 230,220,210      VEL POT 00136
210 IF (MOD((IROW-IXB), NSUBDV) .NE. 0) GO TO 240
220 CROW = .T.          VEL POT 00137
GO TO 230              VEL POT 00138
230 CROW = .F.          VEL POT 00139
IF (IROW .EQ. IXBS .OR. IROW .EQ. IRFB) FROW = .T.
VEL POT 00140
VEL POT 00141
VEL POT 00142
VEL POT 00143
VEL POT 00144

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GO TO 250
 240 IROW = .F.
 I = MOD(IROW+NSUBCN - IXB, NSUBDV)
 IF (I - 1) 280,245,250
 245 IROW = .T.
 250 LROW = .F.
 IF (IROW .NE. MXBS) GO TO 270
 260 LROW = .T.
 C
 C GET THE SUBDIVIDED ROW NUMBER FOR THE CENTER OF THE ASSOCIATED VEL.POT 00145
 C UNSUBDIVIDED BOX, ICENT VEL.POT 00146
 270 CONTINUE VEL.POT 00147
 IF (IROW) GO TO 340 VEL.POT 00148
 IF (IROW - IRFB) 310,320,330 VEL.POT 00149
 C NO FULL ROW VEL.POT 00150
 310 ICENT = 0 VEL.POT 00151
 IUCENT = 0 VEL.POT 00152
 NDCEN = 0 VEL.POT 00153
 CBOX = .F. VEL.POT 00154
 LBOX = .F. VEL.POT 00155
 GO TO 355 VEL.POT 00156
 320 ICENT = IXB VEL.POT 00157
 GO TO 350 VEL.POT 00158
 330 ICENT = ((IROW-IRFB)/NSUBDV)*NSUDV + IXB VEL.POT 00159
 IF (ICENT .LT. MXRS) ICENT = ICENT - NSUBDV VEL.POT 00160
 GO TO 350 VEL.POT 00161
 340 ICENT = IROW VEL.POT 00162
 C
 350 CONTINUE VEL.POT 00163
 IUCENT = (ICENT-IXBW)/NSUBDV + 1 VEL.POT 00164
 355 CONTINUE VEL.POT 00165
 JEWLOC = 1 VEL.POT 00166
 IF (.NOT. WING) JEWLOC = MYBSW + 1 VEL.POT 00167
 C
 C LOOP ON ALL (SUBDIVIDED) CHORDS FOR THE SURFACE AND DIAPHRAGM VEL.POT 00168
 C
 DO 1200 JCCL = 1,MYBSX VEL.POT 00169
 C
 C GET THE CURRENT (SUBDIVIDED) BOX CODE VEL.POT 00170
 CALL DCODER(IBOX,LBXCD, IROW,JCCL, IROW,JCL, .T., NDCBOX) VEL.POT 00171
 IF (SUBOFF) GO TO 480 VEL.POT 00172
 C
 C - - - - - VEL.POT 00173
 C
 C GET INFORMATION ABOUT POSITION WITHIN UNSUBDIVIDED BOX VEL.POT 00174
 C
 C ICENT = I-LOCATION (SUBDIVIDED) OF THE CONTROL POINT VEL.POT 00175
 C JCENT = J-LOCATION (SUBDIVIDED) OF THE CONTROL POINT VEL.POT 00176
 C IUCENT = UNSUBDIVIDED I-LOCATION OF ASSOCIATED CONTROL POINT VEL.POT 00177
 C JUCENT = UNSUBDIVIDED J-LOCATION OF ASSOCIATED CONTROL POINT VEL.POT 00178
 C IPCENT = UNSUBDIVIDED I-LOCATION OF THE NEAREST PLANFORM VEL.POT 00179
 CONTROL POINT, IF THE SUBDIVIDED BOX IS ON-PLANFORM VEL.POT 00180
 C CBOX = .T., CURRENT BOX IS A CENTER BOX VEL.POT 00181
 C LBOX = .T., THIS IS THE LAST BOX ASSOCIATED WITH THE CONTROL VEL.POT 00182
 POINT VEL.POT 00183
 C NDCEN = CODE FOR CONTROL POINT VEL.POT 00184
 C NDCBOX = CODE FOR THE CURRENT SUBDIVIDED BOX VEL.POT 00185
 C

C	IICENT = I-LOCATION (SUBDIVIDED) FOR THE NEAREST PLANFORM	VELPOT 00202
C	CONTROL POINT	VELPOT 00203
C	LBOX = .F.	VELPOT 00204
	JCENT = (JCOL * NSUBCN) / XSUBDV + 1.5	VELPOT 00205
	JCENT = NSUBDV * JCENT - NSUBD2	VELPOT 00206
	IF (.NOT. COPLAN) GO TO 410	VELPOT 00207
	IF (JCENTER .NE. JCOL) GO TO 410	VELPOT 00208
	CBOX = .T.	VELPOT 00209
	NDCEN = NCDBOX	VELPOT 00210
	IICENT = ICROW	VELPOT 00211
	IF (NCDBOX) 450,1100,450	VELPOT 00212
C	GET CENTER BOX CODE, NDCEN	VELPOT 00213
410	CBOX = .F.	VELPOT 00214
	NDCEN = 0	VELPOT 00215
	IF (IICENT .LE. 0) GO TO 414	VELPOT 00216
	CALL DCODER (IBOX, LBXCD, ICEN1, JCEN1, IICENT, JCENT, .T., NDCEN)	VELPOT 00217
	IICENT = ICENT	VELPOT 00218
	GO TO 418	VELPOT 00219
	414 IICENT = IXB - NSUBDV	VELPOT 00220
C	418 CONTINUE	VELPOT 00221
	IF (NCDBOX - 1) 420,424,450	VELPOT 00222
C	SUBDIVIDED BOX IS NOT CONSIDERED (CODE = 0). IF CENTER CODE	VELPOT 00223
C	IS ALSO ZERO, LOOP TO NEXT BOX. OTHERWISE, CHECK FOR LAST BOX	VELPOT 00224
420	IF (NDCEN) 450,1100,450	VELPOT 00225
C	SUBDIVIDED BOX CODE = 1. CHECK WHETHER ITS CONTROL PT = 1	VELPOT 00226
424	CONTINUE	VELPOT 00227
	IF (.NOT. COPLAN) GO TO 431	VELPOT 00228
	IF (TEXLOC(JEXLOC) .LT. FLIROW) JEXLOC = JEXLOC + MYBSW	VELPOT 00229
	IF (FEXLOC(JEXLOC) .GT. FLIROW) JEXLOC = JEXLOC - MYBSW	VELPOT 00230
C	DETERMINE WHETHER SUBDIVIDED BOX IS ON SAME PLANFORM AS	VELPOT 00231
C	IICENT (LOCATION OF NEAREST CONTROL POINT)	VELPOT 00232
	IF (JEXLOC .EQ. JCOL) GO TO 428	VELPOT 00233
C	SUBDIVIDED BOX IS ON THE TAIL. IS IICENT ON THE WING -	VELPOT 00234
	IF (FLOAT(IICENT) .LE. TEXLOC(JCENT)) GO TO 432	VELPOT 00235
C	NO. CHECK FOR IICENT OFF-PLANFORM.	VELPOT 00236
	GO TO 431	VELPOT 00237
C	SUBDIVIDED BOX IS ON THE WING. IS IICENT AFT OF THE WING T.E.	VELPOT 00238
428	IF (FLOAT(IICENT) .GT. TEXLOC(JCENT)) GO TO 432	VELPOT 00239
C	NO. CHECK FOR IICENT OFF-PLANFORM (L.E. DIAPHRAGM)	VELPOT 00240
431	CONTINUE	VELPOT 00241
C	CHECK CODE AT IICENT (NEAREST CONTROL POINT)	VELPOT 00242
	IF (NDCEN .EQ. 1) GO TO 450	VELPOT 00243
C	SUBDIVIDED ON-PLANFORM BOX DOES NOT LIE WITHIN AN UNSUBDIVIDED	VELPOT 00244
C	BOX WHOSE CONTROL POINT IS ON PLANFORM. SEARCH FORE AND AFT	VELPOT 00245
C	FOR THE NEAREST CONTROL POINT ON THE SURFACE.	VELPOT 00246
432	CONTINUE	VELPOT 00247
	IFCOMR = INT(FEXLOC(JEXLOC)) + 1	VELPOT 00248
	ILCOMR = TEXLOC(JEXLOC)	VELPOT 00249
	IMAX = 2 * NSUBDV	VELPOT 00250
DO 432	I = NSUBDV, IMAX * NSUBDV	VELPOT 00251
	IICENT = ICENT>I	VELPOT 00252
	IF (IICENT .GT. ILCOMR) GO TO 434	VELPOT 00253
	CALL DCODER (IBOX, LBXCD, IICENT, JCEN1, IICENT, JCEN1, .T., NCD)	VELPOT 00254
	IF (NCD .EQ. 1) GO TO 440	VELPOT 00255
		VELPOT 00256
		VELPOT 00257
		VELPOT 00258

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434 CONTINUE
  ICENT = ICENT-1
  IF (ICENT .LT. IFCOM) GO TO 438
  CALL DCODER(IBOX,LBXCD, ICENT,JCENT, ICENT,JCENT, .T., NCD)
  IF (NCD .EQ. 1) GO TO 440
438 CONTINUE
C      NO CENTER BOX FOUND. A WARNING DIAGNOSTIC WILL BE PRINTED.
C      THEN COMPUTATION WILL CONTINUE AT 420
C      GO TO 3010.
C      A BOX WAS FOUND
440 CONTINUE
C
C      THE ASSOCIATED CONTROL POINT HAS BEEN FOUND. GET THE UNSUB-
C      DIVIDED SUBSCRIPT.
450 CONTINUE
  ICENT = (ICENT-1)XBW/NSUBDV + 1
  IF (LROW) GO TO 470
  IF (.NOT. FROW) GO TO 460
  IF (JCOL .NE. JCEN-NSUBD2) GO TO 460
  FBOX = .T.
  GO TO 500
460 CONTINUE
  FBOX = .F.
  GO TO 500
470 CONTINUE
  IF (JCOL .EQ. JCEN+NSUBD2) LBOX = .T.
  FBOX = .F.
  GO TO 500
C
C      - - - - -
C
C      SET UP VALUES FOR AN UNSUBDIVIDED CASE
C      TEST FOR NON-ZERO BOX CODE -
480 CONTINUE
  IF (NCDBOX .EQ. 0) GO TO 1100
  ICEN = IRCA
  ICENT = ICEN
  ICENT = ICEN
  JCEN = JCEN
  JCEN = JCEN
  CBOX = .T.
  LBOX = .T.
  NDCEN = NCDBOX
  FBOX = .T.
  IPCENT = ICEN
C
500 CONTINUE
  IF (NCDBOX .GT. 0) GO TO 510
  FULLBX(JCEN) = .F.
  IF (LBOX) GO TO 1040
  IF (FBOX) GO TO 515
  GO TO 1100
C
C      THE BOX IS TO BE CONSIDERED. ARE N-HAT TERMS NECESSARY -
510 CONTINUE
C      ARE N-HAT TERMS ALREADY AVAILABLE -
  IF (FBOX) GO TO 515

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VEL POT 00259
 VEL POT 00260
 VEL POT 00261
 VEL POT 00262
 VEL POT 00263
 VEL POT 00264
 VEL POT 00265
 VEL POT 00266
 VEL POT 00267
 VEL POT 00268
 VEL POT 00269
 VEL POT 00270
 VEL POT 00271
 VEL POT 00272
 VEL POT 00273
 VEL POT 00274
 VEL POT 00275
 VEL POT 00276
 VEL POT 00277
 VEL POT 00278
 VEL POT 00279
 VEL POT 00280
 VEL POT 00281
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 VEL POT 00298
 VEL POT 00299
 VEL POT 00300
 VEL POT 00301
 VEL POT 00302
 VEL POT 00303
 VEL POT 00304
 VEL POT 00305
 VEL POT 00306
 VEL POT 00307
 VEL POT 00308
 VEL POT 00309
 VEL POT 00310
 VEL POT 00311
 VEL POT 00312
 VEL POT 00313
 VEL POT 00314
 VEL POT 00315

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IF (NCCBOX .NE. 1) GO TO 860 VEL.POT 00316
IF (.NOT. WING) GO TO 850 VEL.POT 00317
IF (IPCENT .NE. JUCENT) GO TO 830 VEL.POT 00318
GO TO 850 VEL.POT 00319
C VEL.POT 00320
C - - - - - VEL.POT 00321
C COMPUTE N-HAT TERMS FOR THIS (UNSUBDIVIDED) BOX VEL.POT 00322
C VEL.POT 00323
C VEL.POT 00324
515 CONTINUE VEL.POT 00325
ENRULU(JUCENT) = (0.,0.) VEL.POT 00326
ENRLLL(JUCENT) = (0.,0.) VEL.POT 00327
DPAHIL(JUCENT) = (0.,0.) VEL.POT 00328
C ARE LEFT SURFACE CONTRIBUTIONS POSSIBLE - VEL.POT 00329
IF (IUCENT-IXBU .LT. JUCENT) GO TO 800 VEL.POT 00330
IF (PSIS .EQ. 0 .OR. .NOT. DIHS) GO TO 800 VEL.POT 00331
C GET AIC ARRAYS W AND V FOR LEFT SURFACE INFLUENCE ON RT SURFACE VEL.POT 00332
CALL GETAIC(JUCENT,NW,0, IR) VEL.POT 00333
IF (IR .NE. 0) GO TO 800 VEL.POT 00334
NUBMIN = JUCENT VEL.POT 00335
NUBMAX = IUCENT - IXBU VEL.POT 00336
I = IUCENT - JUCENT VEL.POT 00337
YMUBAR = COS(2*PSIS)* (JUCENT-.5) VEL.POT 00338
JBAR = YMUBAR + .5 VEL.POT 00339
C GET REFERENCE LOCATION IN AIC ARRAYS VEL.POT 00340
IF (YBAR) 520,525,530 VEL.POT 00341
520 JINCR = 1 VEL.POT 00342
GO TO 535 VEL.POT 00343
525 IAIC = NUBMIN+2 VEL.POT 00344
INCAIC = 2*NUBMIN + 1 VEL.POT 00345
JINCR = -1 VEL.POT 00346
GO TO 540 VEL.POT 00347
530 JINCR = -1 VEL.POT 00348
535 IAIC = NUBMIN+2 + NUBMIN VEL.POT 00349
INCAIC = 2*NUBMIN + 2 VEL.POT 00350
C VEL.POT 00351
C LOOP FORWARD OF BOX FOR WING/WING (TAIL/TAIL) N-HAT TERMS VEL.POT 00352
540 CONTINUE VEL.POT 00353
DO 590 NUBAR = NUBMIN,NUBMAX VEL.POT 00354
MUAIC1 = MUAIC(1,NUBAR+1) VEL.POT 00355
MUAIC2 = MUAIC(2,NUBAR+1) VEL.POT 00356
IF (MUAIC2 .EQ. 0) GO TO 585 VEL.POT 00357
IF (YBAR .GE. 0) GO TO 550 VEL.POT 00358
JCOLL = -JBAR - NUBAR + MUAIC1 VEL.POT 00359
GO TO 560 VEL.POT 00360
550 JCOLL = -JBAR + NUBAR - MUAIC1 + 2 VEL.POT 00361
560 CONTINUE VEL.POT 00362
C VEL.POT 00363
C LOOP LEFT OF RECEIVING CHORD TO GET LEFT SURFACE CONTRIBUTIONS VEL.POT 00364
DO 580 MUA1 = MUAIC1,MUAIC2 VEL.POT 00365
IF (JCOLL .LE. 0) GO TO 570 VEL.POT 00366
CALL DCODER(1BOX,LBXCD, I,JCOLL, I,JCOLL, ,F., ICD) VEL.POT 00367
IF (ICD .EQ. 0) GO TO 570 VEL.POT 00368
C A CONTRIBUTING BOX HAS BEEN FOUND. GET THE AIC LOCATION VEL.POT 00369
IAIC = IAIC + MUA1 VEL.POT 00370
C GET LOCATION IN N ARRAYS FOR THE VALUES AT BOX (I,JCOLL) VEL.POT 00371
IDS = LOCSDW(I,JCOLL, IPNTDW, LPNTDW, 1, LPNTDW) VEL.POT 00372

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C
      COEF = (SPSIS2*W(KAIC) - SPSIS2*V(KAIC)) * SYMTY
      ENRULU(JUCENT) = ENRUS(IDS)*COEF + ENRULU(JUCENT)
      ENRLLL(JUCENT) = -ENRLS(IDS)*COEF + ENRLLL(JUCENT)
      DPHIL(JUCENT) = (ENRUS(IDS) - ENRLS(IDS)) * C(KAIC) * SYMTY +
      1
      DPHIL(JUCENT)

570 CONTINUE

C
      JCOLL = JCOLL + JINCR
580 CONTINUE
C      END OF LOOP FOR LEFT ROW CONTRIBUTIONS
C
585 CONTINUE
      I = I - 1
      IF (I .LT. IXBU) GO TO 600
      IAIC = INCAIC + IAIC
      INCAIC = INCAIC + 2
590 CONTINUE
C      END OF LOOP FORWARD ON ROWS, TO COMPUTE LEFT SURFACE OUT-OF-
C      PLANE EFFECTS, FROM 540
C
C      IF THIS IS AN ON-PLANFORM TAIL BOX, THERE ARE WING-TAIL
C      CONTRIBUTIONS
C
600 CONTINUE
      IF (IWING) GO TO 830
      ENRURW(JUCENT) = (0.,0.)
      ENRULW(JUCENT) = (0.,0.)
      IF (INCXEN .NE. 1) GO TO 860
C
C      COMPUTE THE RIGHT WING CONTRIBUTION TO THE TAIL BOX
      II = 1
      IF (PSIW .EQ. PSIT) II = 2
      CALL GETAIC(JUCENT,RWT,II,IR)
      IF (IR .NE. 0) GO TO 700
      NUBMIN = ABS(EL) + .5
      NUBMAX = IUCENT - 1
      I = IUCENT - NUBMIN
      YMUBAR = (JUCENT - .5) * COS(PSIDIF) + CAPL * SIN(PSIW)
      JBAR = YMUBAR
      IF (YMUBAR .GE. 0) JBAR = JBAR + 1
      IF (YBAR) 620,625,630
      620 JINCR = -1
      GO TO 635
625 IAIC = NUBMIN+2
      INCAIC = 2*NUBMIN + 1
      JINCR = 1
      GO TO 640
630 JINCR = 1
635 IAIC = NUBMIN+2 + NUBMIN
      INCAIC = 2*NUBMIN + 2
640 CONTINUE

C
C      LOOP FORWARD OVER THE RIGHT WING
      DO 690 NUBAR = NUBMIN,NUBMAX
      MUAIIC1 = MUAIIC(1,NUBAR+1)
      MUAIIC2 = MUAIIC(2,NUBAR+1)
      IF (MUAIIC2 .EQ. 0) GO TO 685

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VELPOT	00373
VELPOT	00374
VELPOT	00375
VELPOT	00376
VELPOT	00377
VELPOT	00378
VELPOT	00379
VELPOT	00380
VELPOT	00381
VELPOT	00382
VELPOT	00383
VELPOT	00384
VELPOT	00385
VELPOT	00386
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VELPOT	00426
VELPOT	00427
VELPOT	00428
VELPOT	00429

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IF (YBAR .GE. 0) GO TO 650
JCOLR = JBAR + NUBAR .. MUAI C2 + 1
GO TO 660
650 JCOLR = JBAR - NUBAR + MUAI C1 - 1
660 CONTINUE
C
C      LOOP ON A ROW OF WING BOXES, COMPUTING RIGHT HAND WING-TAIL
C      CONTRIBUTION
DO 680 MUAI = MUAI C1, MUAI C2
IF (JCOLR .LE. 0) GO TO 670
CALL DCODER(1BOXW,LBXCDW, I,JCOLR, I,JCOLR, .F., ICD)
IF (ICD .EQ. 0) GO TO 670
C      A CONTRIBUTING BOX HAS BEEN FOUND. GET THE AIC LOCATION
KAIC = IAIC + MUAI
C      GET THE NORMAL-WASH LOCATION
IDS = LOCSDW( I, JCOLR, IPNTDW, LPNTIW, 1, LPNTDW)
C      ADD THIS CONTRIBUTION TO N-HAT
IF (II .EQ. 2) GO TO 665
ENRURW(JUCENT) = (COS(PSIDIF)*W(KAIC) - SIN(PSIDIF)*Y(KAIC))
      * ENRUS(IDS) + ENRURW(JUCENT)
      GO TO 670
665 CONTINUE
ENRURW(JUCENT) = COS(PSIDIF)*W(KAIC) * ENRUS(IDS) +
      ENRURW(JUCENT)
670 CONTINUE
JCOLR = JCOLR + JINOR
680 CONTINUE
C      END OF LOOP FOR RIGHT WING ROW CONTRIBUTIONS
C
685 CONTINUE
I = I - 1
IF (I .LE. 0) GO TO 700
IAIC = IAIC + INCAIC
INCAIC = INCAIC + 2
690 CONTINUE
C      END OF LOOP FORWARD ON ROWS. TO COMPUTE RIGHT WING OUT-OF-
C      PLANE EFFECTS ON THE TAIL, FROM 640
C
C      DETERMINE WHETHER LEFT WING INFLUENCE IS TO BE COMPUTED
700 CONTINUE
IF (SYM .EQ. 0) GO TO 800
C      GET AIC ARRAYS W AND Y FOR LEFT WING INFLUENCE ON TAIL
II = 1
IF (-PSIW .EQ. PSIT) II = 2
CALL GETAIC(JUCENT, LWT, II, IR)
IF (IR .NE. 0) GO TO 800
NUBMIN = ABS(EL) + .5
NUBMAX = IUCENT - 1
I = IUCENT - NUBMIN
YMBAR = - COS(PSIW + PSIT)*(JUCENT-.5) + CARASIN(PSIW)
JBAR = YMBAR
IF (YMBAR .GE. 0) JBAR = JBAR + 1
IF (YBAR) 720,725,730
720 JINCR = 1
GO TO 730
725 IAIC = NUBMIN+2
INCAIC = 2*NUBMIN + 1

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VELPOT 00430
 VELPOT 00431
 VELPOT 00432
 VELPOT 00433
 VELPOT 00434
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 VELPOT 00474
 VELPOT 00475
 VELPOT 00476
 VELPOT 00477
 VELPOT 00478
 VELPOT 00479
 VELPOT 00480
 VELPOT 00481
 VELPOT 00482
 VELPOT 00483
 VELPOT 00484
 VELPOT 00485
 VELPOT 00486


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C      ED BY WAKE EFFECTS WHERE APPLICABLE.          VEL.POT 00544
C
C      - TAIL -
820 CONTINUE
C      IF (INCBOX .NE. 1) GO TO 860
C      GET DEFLECTION AND SLOPE OF UNSUBDIVIDED TAIL BOX CENTER
IDS = LOCSDW(IPCENT+IOVLPN, JUCENT, IPNTDW, LMODES, 1, LMODES)
DFL = DEFL(1, IDS)
SLP = DEFL(2, IDS)
C      COMPUTE TAIL NORMAL WASH VALUES
ENDIF = 2.0*( CMPLX(B1*SLP, XXVL*DFL) - ENRURW(JUCENT) )
1           - ENRULW(JUCENT) ) + ENRLLL(JUCENT) - ENRULU(JUCENT) VEL.POT 00545
ENSUM = - (ENRLLL(JUCENT) + ENRULU(JUCENT) ) VEL.POT 00546
C
C      IF (INCDCEN .NE. 1) GO TO 852
LOCNW = LOCSDW(IUCENT+IOVLPN, JUCENT, IPNTDW, LPNTDW, 1, LPNTDW)
ENRUS(LOCNW) = 0.5*(ENSUM + ENDIF) VEL.POT 00547
ENRLS(LOCNW) = 0.5*(ENSUM - ENDIF) VEL.POT 00548
GO TO 850 VEL.POT 00549
C
C      - WING --
830 CONTINUE
C      IF (INCBOX .NE. 1) GO TO 860
C      GET DEFLECTION AND SLOPE OF UNGSUBDIVIDED WING BOX CENTER
IDS = LOCSDW(IPCENT, JUCENT, IPNTDW, LPNTDW, 1, LPNTDW)
DFL = DEFL(1, IDS)
SLP = DEFL(2, IDS)
C      COMPUTE WING NORMAL WASH VALUES
ENSUM = -ENRULU(JUCENT) - ENRLLL(JUCENT)
ENDIF = ENSUM + (CMPLX(B1*SLP, XXVL*DFL) + ENRLLL(JUCENT)) * 2.0 VEL.POT 00550
C
C      IF (IPCENT .NE. IUCENT) GO TO 852
LOCNW = LOCSDW(IUCENT, JUCENT, IPNTDW, LPNTDW, 1, LPNTDW)
ENRUS(LOCNW) = (ENSUM + ENDIF) * 0.5 VEL.POT 00551
ENRLS(LOCNW) = (ENSUM - ENDIF) * 0.5 VEL.POT 00552
C
C      NORMAL-WASH IS AVAILABLE IF THE BOX IS ON-PLANFORM
850 CONTINUE
C      IF (.N. FROK) LOCNW = LOCSDW(IUCENT+IOVLPN, JUCENT, IPNTDW, LPNTDW, 1, LPNTDW)
1           1, LPNTDW) VEL.POT 00553
C      IF (SUBOFF) GO TO 855
C      GET THE SUBDIVIDED VALUE FOR THE NORMAL WASH TERMS
DELPHA = ENRUS(LOCNW) VEL.POT 00554
IF (DELPHA .EQ. XINIT) GO TO 830 VEL.POT 00555
DELPHB = ENRLS(LOCNW) VEL.POT 00556
GO TO 854 VEL.POT 00557
C
C      THE NEXT 2 STATEMENTS ARE ONLY HIT FOR A SUBDIVIDED PLANFORM
C      BOX WITH NO ASSOCIATED PLANFORM CONTROL POINT
832 CONTINUE
FULLBX(JUCENT) = .FALSE. VEL.POT 00558
DELPHA = (ENSUM + ENDIF) * 0.5 VEL.POT 00559
DELPHB = (ENSUM - ENDIF) * 0.5 VEL.POT 00560
854 CONTINUE
ENRUS = CMPLX( REAL(DELPHA)/XSUBDV, AIMAG(DELPHA)/XSUBDV +
1           XXVL*(IROW+1)CENT*REAL(DELPHA) ) VEL.POT 00561
ENRLS = CMPLX( REAL(DELPHB)/XSUBDV, AIMAG(DELPHB)/XSUBDV +
1           XXVL*(IROW+1)CENT*REAL(DELPHB) ) VEL.POT 00562
VEL.POT 00563
VEL.POT 00564
VEL.POT 00565
VEL.POT 00566
VEL.POT 00567
VEL.POT 00568
VEL.POT 00569
VEL.POT 00570
VEL.POT 00571
VEL.POT 00572
VEL.POT 00573
VEL.POT 00574
VEL.POT 00575
VEL.POT 00576
VEL.POT 00577
VEL.POT 00578
VEL.POT 00579
VEL.POT 00580
VEL.POT 00581
VEL.POT 00582
VEL.POT 00583
VEL.POT 00584
VEL.POT 00585
VEL.POT 00586
VEL.POT 00587
VEL.POT 00588
VEL.POT 00589
VEL.POT 00590
VEL.POT 00591
VEL.POT 00592
VEL.POT 00593
VEL.POT 00594
VEL.POT 00595
VEL.POT 00596
VEL.POT 00597
VEL.POT 00598
VEL.POT 00599
VEL.POT 00600

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IF (CBOX) GO TO 870
GO TO 1030
855 CONTINUE
  ENSRUS = ENSRUS (LOCNM)
  ENSRLS = ENSRLS (LOCNM)
  GO TO 870
860 CONTINUE
  FULLBX(JUCENT) = .F.

C
C   - - - - -
C
C      DETERMINE THE VELOCITY POTENTIAL CONTRIBUTIONS FROM
C      BOXES LYING A H E A D  OF THE CURRENT BOX
C
C      870 CONTINUE
C         IF (PSIS .NE. 0 .AND. DIHS) GO TO 880
C            THE SURFACE IS PLANAR.  GET FULL SURFACE CONTRIBUTIONS
C            DELPH = B(IROW,JCOL, PKERNL,SKERNL, IBOX,LBXCD, WING, .F. )
C            GO TO 890
C
C            DIHEDRAL ANGLE IS TO BE ACCOUNTED FOR.  GET THE PLANAR
C            (SUBDIVIDED) CONTRIBUTION OF THE RIGHT SURFACE
C      880 CONTINUE
C         DELPH = B(IROW,JCOL, PKERNL,SKERNL, IBOX,LBXCD, WING, .T. )
C         ADD THE SPATIAL LEFT SURFACE CONTRIBUTIONS
C         DELPH = DELPH + DPHL(JUCENT)
C      890 CONTINUE
C         IF (INCDBOX - 2) 910,1000,980
C
C   - - - - -
C
C      THE BOX IS ON-PLANFORM, CENTER.  COMPLETE THE CALCULATION OF
C      THE VELOCITY POTENTIAL
C
C      910 CONTINUE
C         IIS = LOCENW(JUCENT+10V), JUCENT, IPNTRM,LMODES, 1, LMODES)
C         IF (SUBOFF) GO TO 915
C         DELPHI(IIS) = (ENSRUS*ENSLRS) * SKERNL(1) + DELPH
C         GO TO 920
C      915 CONTINUE
C         DELPHI(IIS) = (ENSRUS-ENSLRS) / PKERNL(1) + DELPH
C      920 CONTINUE
C
C            COMPUTE ANY TRAILING EDGE VELOCITY POTENTIALS ASSOCIATED
C            WITH THIS UNSUBDIVIDED BOX
C
C            IS THIS A TRAILING EDGE BOX -
C            IF (.NOT. COPLAH) GO TO 930
C            IF (JEXLOC(JEXLOC) .LT. FLIRW) JEXLOC = JEXLOC + MYSBW
C            IF (JEXLOC(JEXLOC) .GT. FLIRW) JEXLOC = JEXLOC -, MYSBW
C      930 CONTINUE
C            JJ = JEXLOC - NSUBDV
C            TEXMIN = TEXLOC(JJ)
C            IF (NSUBDV .EQ. 1) GO TO 935
C            DO 932 J = 2,NSUBDV
C              JJ = JJ + 1
C              TEXMIN = AMINI(TEXMIN,TEXLOC(JJ))
C
C      VELPOT 00601
C      VELPOT 00602
C      VELPOT 00603
C      VELPOT 00604
C      VELPOT 00605
C      VELPOT 00606
C      VELPOT 00607
C      VELPOT 00608
C      VELPOT 00609
C      VELPOT 00610
C      VELPOT 00611
C      VELPOT 00612
C      VELPOT 00613
C      VELPOT 00614
C      VELPOT 00615
C      VELPOT 00616
C      VELPCF 00617
C      VELPOT 00618
C      VELPOT 00619
C      VELPOT 00620
C      VELPOT 00621
C      VELPOT 00622
C      VELPOT 00623
C      VELPOT 00624
C      VELPOT 00625
C      VELPOT 00626
C      VELPOT 00627
C      VELPOT 00628
C      VELPOT 00629
C      VELPOT 00630
C      VELPOT 00631
C      VELPOT 00632
C      VELPOT 00633
C      VELPOT 00634
C      VELPOT 00635
C      VELPOT 00636
C      VELPOT 00637
C      VELPOT 00638
C      VELPOT 00639
C      VELPOT 00640
C      VELPOT 00641
C      VELPOT 00642
C      VELPOT 00643
C      VELPOT 00644
C      VELPOT 00645
C      VELPOT 00646
C      VELPOT 00647
C      VELPOT 00648
C      VELPOT 00649
C      VELPOT 00650
C      VELPOT 00651
C      VELPOT 00652
C      VELPOT 00653
C      VELPOT 00654
C      VELPOT 00655
C      VELPOT 00656
C      VELPOT 00657

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932 CONTINUE VEL.POT 00658
933 IF ( TEXMIN .GE. FLIRCH + XSUBDV) GO TO 1030 VEL.POT 00659
C YES. GET THE BASIC VELOCITY POTENTIAL VEL.POT 00660
DELPHB = DELPHI(IDS) VEL.POT 00661
C TEST UNSUBDIVIDED BOX AHEAD OF CURRENT ONE - VEL.POT 00662
CALL DCODER(ibox,LBXCD, IR0+NSUBDV, JCOL, IR0+NSUBDV, JCOL,
1 .T., NCDA) VEL.POT 00663
IF (NCDA .NE. 1) GO TO 940 VEL.POT 00664
C IT IS ON PLANFORM. TEST FOR TIP CHORD - VEL.POT 00665
IF ( JEXLOC .EQ. JCOL .AND. JUENT .LT. HYBW) GO TO 950 VEL.POT 00666
IF ( JEXLOC .GT. JCOL .AND. JUENT .LT. HYBT) GO TO 950 VEL.POT 00667
C BOX IS ON THE TIP CHORD. CHECK FOR THIRD TIP BOX - VEL.POT 00668
IR = IR0 - 2*NSUBDV VEL.POT 00669
IF ( IR .LT. IXB) GO TO 940 VEL.POT 00670
CALL DCODER(ibox,LBXCD, IR, JCOL, IR, JCOL, .T., NCDC) VEL.POT 00671
IF (NCDC .EQ. 1) GO TO 950 VEL.POT 00672
C TRY MACH RAY EXTRAPOLATION. ARE THE 2 RAY BOXES ON-PLANFORM - VEL.POT 00673
940 CONTINUE VEL.POT 00674
VEL.POT 00675
JC = JCOL - NSUBDV VEL.POT 00676
CALL DCODER(ibox,LBXCD, IR0, JC, IR0, JC, .T., NCDD) VEL.POT 00677
IF (NCDD .NE. 1) GO TO 945 VEL.POT 00678
JC = JC - NSUBDV VEL.POT 00679
CALL DCODER(ibox,LBXCD, IR0+NSUBDV, JC, IR0+NSUBDV, JC, .T., NCDD) VEL.POT 00680
IF (NCDD .NE. 1) GO TO 945 VEL.POT 00681
C MACH RAY EXTRAPOLATION, FOLLOWED BY CHORDWISE LINEAR EXTRA- VEL.POT 00682
C POLATION VEL.POT 00683
IDPHM2 = LOCSDW(JCENT-1+ICMLP, JCENT-2, IPNTRM, LMODES, 1, 1 MODES) VEL.POT 00684
IDPHM1 = LOCSDW(JCENT+ICMLP, JCENT-1, IPNTRM, LMODES, 1, LMODES) VEL.POT 00685
DELPHA = DELPHI(IDPHM2) VEL.POT 00686
DELPHC = DELPHI(IDPHM1) VEL.POT 00687
DELPH = (2.0*DELPHC - DELPHA - DELPHB)/XSUBDV VEL.POT 00688
GO TO 955 VEL.POT 00689
VEL.POT 00690
C MACH RAY UNAVAILABLE. ARE THERE 2 BOXES ON THE CHORD - VEL.POT 00691
945 IF (NCDA .NE. 1) GO TO 3020 VEL.POT 00692
C VEL.POT 00693
C CHORDWISE LINEAR EXTRAPOLATION VEL.POT 00694
C VEL.POT 00695
950 CONTINUE VEL.POT 00696
IDPHM1 = LOCSDW(JCENT-1+ICMLP, JCENT, IPNTRM, LMODES, 1, LMODES) VEL.POT 00697
DELPH = (DELPHB - DELPHI(IDPHM1))/XSUBDV VEL.POT 00698
C VEL.POT 00699
C LOOP TO COMPUTE AND STORE TRAILING EDGE VELOCITY POTENTIALS VEL.POT 00700
955 CONTINUE VEL.POT 00701
VEL.POT 00702
JA = JEXLOC - NSUBD2 VEL.POT 00703
JB = JEXLOC + NSUBD2 VEL.POT 00704
DO 990 JJ = JA, JB VEL.POT 00705
XINOR = TEXLOC(JJ) - IR0 VEL.POT 00706
IF (XINOR .LT. -XSUBDV/2.0) GO TO 960 VEL.POT 00707
TVP(JJ) = DELPHB + XINOR*DELPH VEL.POT 00708
990 CONTINUE VEL.POT 00709
C ALL TRAILING EDGE VALUES HAVE BEEN COMPUTED FOR THIS VEL.POT 00710
C UNSUBDIVIDED BOX. VEL.POT 00711
GO TO 1030 VEL.POT 00712
C VEL.POT 00713
C VEL.POT 00714

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C      DETERMINE THE UNSUBDIVIDED NORMAL WASH VALUES
1040 IF (NCDCEN = 1) 1100,1050,1060
C      IF THE BOX IS COMPLETE, THE VALUE IS ALREADY STORED -
1050 IF (FULLBX(JCENT)) GO TO 1100
C      THE UNSUBDIVIDED VALUE EQUALS THE AVERAGE OF ALL ITS
C      SUBDIVIDED BOXES
1060 CONTINUE
    II = ICENT - NSUBD2
    JJJ = JCENT + NSUBD2
    ENSRUS = (0.,0.)
    ENSRLS = (0.,0.)
    DO 1080 I = 1,NSUBDV
        JJ = JCENT - NSUBD2
        CALL DCOPFR(IBOX,LBXCD, II,JJ, II, JJJ, .T., 1CODE )
        DO 1075 J = 1,NSUBDV
            IF (1CODE(J) .EQ. 0) GO TO 1070
            LOCNW = LOCSDW( II,JJ, IPNTST,IPNTIN,IPNTOT,IPNTLS)
            ENSRUS = ENSUBD(1,LOCNW) + ENSRUS
            ENSRLS = ENSUBD(2,LOCNW) + ENSRLS
1070 CONTINUE
    JJ = JJ + 1
1075 CONTINUE
    II = II + 1
1080 CONTINUE
    ENSRUS = ENSRUS/NSUBDV
    ENSRLS = ENSRLS/NSUBDV
C
C      RESTORE THE PARTIAL BOX FLAG FOR THE NEXT ROW
    FULLBX(JCENT) = .T.
C
C      STORE THE UNSUBDIVIDED NORMAL WASHES
1090 CONTINUE
    LOCNW = LOCSDW( JCENT+1,0, JCENT, 1PNTDW, LPNTDW,1,LPNTDW )
    IF (LOCNW .EQ. 0) GO TO 3040
    ENRUS(LOCNW) = ENSRUS
    IF (WING) GO TO 1095
    ENRLS(LOCNW) = ENSRLS
    GO TO 1100
1095 ENRLS(LOCNW) = ENSRLS
1100 CONTINUE
    JEXLOC = JEXLOC + 1
C
C
1200 CONTINUE
C      END OF LOOP ON (SUBDIVIDED) CHORDS, STARTING AT 355*
C
C
1300 CONTINUE
C      END OF LOOP ON (SUBDIVIDED) ROWS, STARTING AT 120
C
C
RETURN
C
C      -----
C
C      DIAGNOSTICS - ALL CALL FLUSH

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C		
3010 WRITE (NT6,9010)	VELPOT	00829
WRITE (NT6,9999) IR0W,JC0L, IUCENT, JUCENT	VELPOT	00830
GO TO 420	VELPOT	00831
3020 WRITE (NT6,9020)	VELPOT	00832
GO TO 3999	VELPOT	00833
3030 WRITE (NT6,9030)	VELPOT	00834
GO TO 3999	VELPOT	00835
3040 WRITE (NT6,9040)	VELPOT	00836
GO TO 3999	VELPOT	00837
C	VELPOT	00838
3999 WRITE (NT6,9999) IR0W,JC0L, IUCENT, JUCENT	VELPOT	00839
CALL FLUSH(1)	VELPOT	00840
C	VELPOT	00841
C	VELPOT	00842
9010 FORMAT(5SH0*** WARNING - NO PLANFORM CONTROL POINT FOUND FOR SUBDI	VELPOT	00843
1 52H DIVIDED BOX DURING VELOCITY POTENTIAL CALCULATIONS ***)	VELPOT	00844
9020 FORMAT(5SH0*** ERROR - THE TIP BOX PATTERN DOES NOT ALLOW TRAILING	VELPOT	00845
1 44H EDGE VELOCITY POTENTIALS TO BE COMPUTED ***)	VELPOT	00846
9030 FORMAT(5SH0*** ERROR - FAILURE IN STORING SUBDIVIDED NORMAL-WASHES	VELPOT	00847
1 4H ***)	VELPOT	00848
9040 FORMAT(5SH0*** ERROR - FAILURE IN STORING CONTROL POINT NORMAL-	VELPOT	00849
1 10H WASHES ***)	VELPOT	00850
9999 FORMAT(14X,16HSUBDIVIDED BOX (,I3,1H,I3,1H), CONTROL POINT (,	VELPOT	00851
1 I2,1H,I2,1H))	VELPOT	00852
C	VELPOT	00853
ENC	VELPOT	00854
	VELPOT	00855

COMPLEX FUNCTION B (IROW, JCOL, PKERNL, SKERNL, IBOX, LBXCD,
 1 WING, DH)
 C
 C COMPUTES B = SUM OVER NU(SUM OVER MU((DOWNWASH)*(KERNEL)))
 C (NU,MU .NE. 0)
 C B IS USED TO COMPUTE VELOCITY POTENTIALS OF ON-
 C PLANFORM BOXES, OR DOWNWASHES OF DIAPHRAGM BOXES
 C
 C PARAMETERS -
 C IROW = ROW LOCATION OF BOX FOR WHICH B IS TO BE
 C COMPUTED
 C JCOL = COLUMN LOCATION OF BOX
 C PKERNL = PRIMARY KERNEL ARRAY
 C SKERNL = SUBDIVIDED KERNEL ARRAY
 C IBOX = ARRAY OF BOX CODES
 C LBXCD = LENGTH OF BOX CODE ARRAY
 C WING = .T., WING. .F., TAIL
 C DH = .T., LEFT SIDE TO BE IGNORED (SURFACE HAS DIHEDR.
 C = .F., LEFT SIDE TO BE INCLUDED
 C
 C VALUES FROM COMMON -
 C NSUBDV = NUMBER OF SUBDIVISIONS
 C IXB = CENTER OF FIRST UNSUBDIVIDED BOX RELATIVE TO THE
 C SUBDIVIDED PATTERN
 C MXB = NUMBER OF UNSUBDIVIDED ROWS
 C MYBB = NUMBER OF UNSUBDIVIDED CHORDS, INCLUDING DIAPH.
 C MYBBS = NUMBER OF SUBDIVIDED CHORDS, INCLUDING DIAPH.
 C MXSKRN = SIZE OF SUBDIVIDED KERNEL
 C SYM = SYMMETRY INDICATOR
 C
 C LROT = NSUBDV + NSUBDV/2 + 1
 C ENSUBD = SUBDIVIDED NORMAL-WASHES
 C ENRUS, ENRLS = UNSUBDIVIDED NORMAL-WASHES
 C IPNTDW = POINTER ARRAY FOR UNSUBDIVIDED NORMAL WASHES
 C IPNTSD = POINTER ARRAY FOR SUBDIVIDED NORMAL WASHES (END-
 C AROUND
 C IPNTIN = NEXT AVAILABLE POINTER
 C IPNTOT = FIRST POINTER IN USE
 C IPNTLS = DIMENSION OF ARRAY IPNTSD
 C LIMWS = DIMENSION OF SUBDIVIDED NORMAL-WASH ARRAYS
 C
 C COMMON /GEOMTY/ COPLAN, NSUBDV, XSUBDV, NSUPDD, NSUBCN, NSURF,
 1 B1, B1BETA, B1S, B1BTAS, WLAX, WLAZ, PSTW,
 2 MXBW, MXBRW, MYBW, MYBRW, MXBSW, MYBSW, MYBBSW,
 3 IXBW, XCENTR
 C LOGICAL COPLAN
 C COMMON /GEOR2/ TLAX, YLAT, PS1T, IXBT, MYBT, MYBBT, MYST, MYBST,
 1 MYBSY, IXST, IXBT, CAPL
 C COMMON /MOCOM/ SYM, SYMT, MTYPW, MYPET
 C COMMON /SNWASH/ IPNTSD(2,50), ENSUBD(2,600), IPNTIN, IPNTOT, IPNTLS
 C IPNTSD(L'PNTSD), ENSUBD(2*L'SLW)
 C COMPLEX ENSUBD
 C COMMON /NWASHES/ IPNTDW(2,100), ENRUS(1275), ENRLS(1275), IONLAPN
 C COMPLEX ENRUS, ENRLS
 C COMMON /KERN/ ERR, MXSKRN, IPKERN, NPKRN, NSFATK, NRCHEA
 C DELPHI(L'MOCOM), TVP(L'TVP), TEXLOC(L'YY)
 C COMMON /DELTAP/ DELPHI(1080), TVP(250), TEXLOC(250), DELTAP 00003

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1           IPNTRH(2,100),NPNTRS, IOVLAP          DELTAP 00004
C           COMPLEX      DELPHI,      TVP          DELTAP 00005
C           DEFSL(2,LMODES)                      DELTAP 00006
C           DIMENSION DEFSL(2,1000)                DELTAP 00007
C           EQUIVALENCE (DELPHI(81), DEFSL)        DELTAP 00008
C           ARRAYS DELPHI AND DEFSL ARE          DELTAP 00009
C           EQUIVALENCED TO GIVE A 2 ROW UN-OVERLAPPED SECTION DELTAP 00010
C           COMMON /LROT / LROT                  B     00050
C
C           COMPLEX PKERNL(1),SKERNL(1)          B     00051
C           LOGICAL WING,DIM                  B     00052
C
C           DIMENSION IXCD(150)                 B     00053
C           LOGICAL LEFT, LSIDE                B     00054
C
C           IA = IRW
C           B = (0.,0.)                         B     00055
C
C           IF (WING) GO TO 20                  B     00056
C
C           IXB = IXBT
C           IXBS = IXBST
C           MXB = MXBT - (IXBT-IXBW)/NSUBDV
C           MYBBS = MYBBST
C           IXLPN = IOVLAPN
C           SYMTY = SYHT
C           GO TO 25
C
C 20 CONTINUE
C           IXB = IXBW
C           IXBS = 1
C           MXB = MXBBW
C           IF (COPLAN) MXB = MXBT
C           MYBBS = MYBBBW
C           IOVLPN = 0
C           SYMTY = SYM
C
C 25 CONTINUE
C           LSIDE = SYMTY .NE. 0 .AND .NOT. DIH
C
C           IS SUBDIVISION REQUESTED -
C           IF (NSUBDV .EQ. 1) GO TO 410
C
C           YES. DETERMINE THE NUMBER OF ROWS WHICH CAN BE HANDLED (MXB1)
C           NSRWNG = IPNTIN - IPNTOT - 1
C           IF (NSRWNG .LT. 0) NSRWG1 = NSRWG1 + IPNTLS
C           NSRWG1 = MIN(NSRWG1,MXSKRN-1)
C           IF (IA - IXBS .GT. NSRWG1) GO TO 120
C           ALL SUBDIVIDED. ALLOW TO JUMP BEYOND TO TRIGGER RETURN.
C           MXB1 = IA - IXBS > 1
C           GO TO 200
C           PARTIAL SUBDIVIDED
C
C 120 CONTINUE
C           I = IA - NSRWG1 - IXB
C           I = MOD(I,NSUBDV)
C           I = LROT - I
C           MXB1 = NSRWG1 - MOD(I,NSUBDV)
C
C

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C DETERMINE THE CONTRIBUTION TO B FROM A FORWARD CONE OF SUB- B 00100
 C DIVIDED BOXES B 00101
 C MU = ROW NUMBER OF CONTRIBUTING BOX RELATIVE TO RECEIVING B 00102
 C BOX. MU OF RECEIVING BOX = 0. B 00103
 C IA = ACTUAL ROW LOCATION OF CONTRIBUTING BOX, RELATIVE TO B 00104
 C SUBDIVIDED GRID. B 00105
 C B 00106
 200 CONTINUE B 00107
 IF (MXB1 .EQ. 0) GO TO 310 B 00108
 DO 300 MU = 1,MXB1 B 00109
 IA = IA - 1 B 00110
 C HAS THE FORWARD EDGE OF THE PATTERN BEEN REACHED - B 00111
 IF (IA .LT. IXBS) GO TO 600 B 00112
 C NO. GET BOX TYPE CODES FOR CURRENT ROW. B 00113
 LEFT = .T. B 00114
 IIA = MOD(IA-1,IPNTLS) + 1 B 00115
 IIAP1 = MOD(IA,IPNTLS) + 1 B 00116
 IPNTPI = IPNTSD(1,IIAP1) B 00117
 IF (IPNTPI .LE. 1) GO TO 208 B 00118
 NYB = IPNTPI - IPNTSD(1,IIA) + IPNTSD(2,IIA) - 1 B 00119
 GO TO 210 B 00120
 208 NYB = MYBBS B 00121
 210 CONTINUE B 00122
 CALL DCODER (IBOX,LBXCD, IA,1, IA,NYB, .T., IBXCD) B 00123
 IF (NYB . NE. MYBBS) GO TO 215 B 00124
 DO 212 I = 1,MYBBS B 00125
 IF (IBXCD(NYB) .NE. 0) GO TO 215 B 00126
 NYB = NYB - 1 B 00127
 212 CONTINUE B 00128
 215 CONTINUE B 00129
 C IBXCD = ROW OF BOX CODES B 00130
 C NYB = NUMBER FOUND B 00131
 C B 00132
 C GET LOCATION IN THE SUBDIVIDED DOWNWASH ARRAY FOR BOX(IA,JCOL) B 00133
 IDW = LOCSDW(IA,JCOL, IPNTSD,IPNTIN,IPNTOT,IPNTLS) B 00134
 N = (NU*(NU+1))/2 + 1 B 00135
 C KERNEL(NU,MU) = SKERNL((NU*(NU+1))/2 + ADS(MU) + 1), SO B 00136
 C N = SUBSCRIPT FOR KERNEL (STARTING WITH NU,0) B 00137
 C B 00138
 C CENTER BOX OF ROW IN CONE B 00139
 IF (JCOL .GT. NYB) GO TO 220 B 00140
 IF (IBXCD(JCOL) .EQ. 0) GO TO 220 B 00141
 B = B + SKERNL(N) * (ENSUBD(1,IDL) - ENUBD(2,IDL)) B 00142
 220 CONTINUE B 00143
 C B 00144
 C B 00145
 C GOING OUT FROM CENTER CHORD OF CONE IN BOTH DIRECTIONS B 00146
 C IDWR,IDL = POINTERS IN DOWNWASH ARRAY FOR RIGHT,LEFT SIDES B 00147
 C IBXR,IBXL = POINTERS IN BOX CODES ARRAY, AS ABOVE B 00148
 C L = LEFT SIDE POINTER INCREMENTER (CHANGES SIGN WHEN B 00149
 C THE PLANFORM CENTER-LINE IS ENCOUNTERED) B 00150
 C E = LEFT SIDE MULTIPLIER, USED TO DETERMINE SYM/ANTI B 00151
 C SYN. AFTER PLANFORM CENTER-LINE ENCOUNTERED. B 00152
 IDWR = IDW+1 B 00153
 IDWL = IDW-1 B 00154
 IBXR = JCOL+1 B 00155
 IBXL = JCOL-2 B 00156

N = N + 1	B	00157
L = -1	B	00158
E = 1.0	B	00159
C	B	00160
DO 280 MU = 1,NU	B	00161
C	B	00162
C RIGHT SIDE	B	00163
IF (IBXR .GT. NYB) GO TO 230	B	00164
IF (IBXCD(IBXR)) .EQ. 0) GO TO 230	B	00165
B = B + SKERNL(N) * (ENSUBD(1,ICWR) - ENSUBD(2,ICWR))	B	00166
230 CONTINUE	B	00167
C	B	00168
C LEFT SIDE	B	00169
C	B	00170
IF (.NOT. LEFT) GO TO 270	B	00171
C HAS PLATEFORM CENTER LINE BEEN ENCOUNTERED -	B	00172
IF (IBXL .GT. 0) GO TO 250	B	00173
C YES. SET PARAMETERS TO SWEEP BACK ACROSS RIGHT HALF AS A	B	00174
C SYMMETRIC/ANTISYMMETRIC IMAGE OF THE LEFT SIDE.	B	00175
LEFT = LSIDE	B	00176
IF (.NOT. LEFT) GO TO 270	B	00177
E = SYMTY	B	00178
L = 1	B	00179
IBXL = 1	B	00180
IDWL = IDWL + 1	B	00181
GO TO 260	B	00182
C IF THE CENTER LINE HAS PREVIOUSLY BEEN ENCOUNTERED, IBXL WILL	B	00183
C BE INCREASING. IF IBXL HAS EXCEEDED THE NUMBER OF BOXES ON	B	00184
C THIS ROW, THIS ROW IS COMPLETE, TRANSFER TO LOOP ON NU.	B	00185
260 IF (IBXL .GT. NYB) GO TO 270	B	00186
260 IF (IBXCD(IBXL)) .EQ. 0) GO TO 270	B	00187
B = B + SKERNL(N) * (ENSUBD(1,ICWL) - ENSUBD(2,ICWL)) * E	B	00188
270 CONTINUE	B	00189
C	B	00190
C SET COUNTERS FOR NEXT STEP OUTWARD	B	00191
IDWR = ICWR+1	B	00192
IDWL = ICWL+L	B	00193
IBXR = IBXR+1	B	00194
IBXL = IBXL+L	B	00195
N = N+1	B	00196
280 CONTINUE	B	00197
C END OF LOOP ON MU (SUBDIVIDED COLUMNS OUTBOARD)	B	00198
C	B	00199
300 CONTINUE	B	00200
C END OF LOOP ON NU (SUBDIVIDED ROWS FORWARD) FROM 200	B	00201
C	B	00202
C IS THERE AT LEAST ONE FULL UNSUBDIVIDED ROW LEFT AHEAD OF	B	00203
C CURRENT POSITION -	B	00204
310 CONTINUE	B	00205
IF (IA .LT. IAB) GO TO 600	B	00206
C	B	00207
C	B	00208
C UNSUBDIVIDED BOXES	B	00209
C	B	00210
C DETERMINE ROW AND COLUMN NUMBERS IN SUBDIVIDED ARRAYS CORRESPONDING TO UNSUBDIVIDED BOX CENTERS.	B	00211
C IA = ROW LOCATION OF CONTRIBUTING SUBDIVIDED BOX	B	00212
C	D	00213

N = N + 1	B	00271
C	B	00272
DO 480 MU = 1,NU	B	00273
C	B	00274
C RIGHT SIDE	B	00275
IF (IBXR .GT. NYB) GO TO 430	B	00276
IF (IBXCC(IBXR) .EQ. 0) GO TO 430	B	00277
B = B + PERNL(N) * (ENRUS(IDWR) - ENRLS(IDWR))	B	00278
430 CONTINUE	B	00279
C	B	00280
C LEFT SIDE	B	00281
C	B	00282
IF (.NOT. LEFT) GO TO 470	B	00283
C HAS PLANFORM CENTER LINE BEEN ENCOUNTERED -	B	00284
IF (IBXL .GT. 0) GO TO 450	B	00285
C YES. SET PARAMETERS TO SWEEP BACK ACROSS RIGHT SIDE	B	00286
LEFT = LSIDE	B	00287
IF (.NOT. LEFT) GO TO 470	B	00288
E = SYMTY	B	00289
L = 1	B	00290
IBXL = 1	B	00291
IDWL = IDWL + 1	B	00292
C TEST FOR ROW COMPLETE, AS IN SUBDIVIDED LOGIC	B	00293
450 IF (IBXL .GT. NYB) GO TO 470	B	00294
460 IF (IBXCD(IBXL) .EQ. 0) GO TO 470	B	00295
B = B + PERNL(N) * (ENRUS(IDWL) - ENRLS(IDWL)) * E	B	00296
470 CONTINUE	B	00297
C SET COUNTERS FOR NEXT STEP OUTWARD	B	00298
IDWR = IDWR+1	B	00299
IDWL = IDWL+L	B	00300
IBXR = IBXR+1	B	00301
IBXL = IBXL+1.	B	00302
N = N + 1	B	00303
480 CONTINUE	B	00304
C END OF LOOP ON MU (CHORDS OUTWARD)	B	00305
C	B	00306
500 CONTINUE	B	00307
C END OF LOOP ON NU (ROWS FORWARD) FROM 420	B	00308
C	B	00309
C	B	00310
600 RETURN	B	00311
C	B	00312
END	B	00313

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SUBROUTINE GETAIC(JUCENT, ITPE, ICODE, IR)          GETAIC 00002
C                                                 GETAIC 00003
C       GETS DESIRED AIC ARRAYS FROM DISK           GETAIC 00004
C                                                 GETAIC 00005
C       JUCENT = CHORD NUMBER                      GETAIC 00006
C       ITPE   = 1, WING/WING      3, RIGHT WING/TAIL    GETAIC 00007
C               2, TAIL/TAIL        4, LEFT WING/TAIL     GETAIC 00008
C       ICODE  = 0, C,V,W DESIRED                  GETAIC 00009
C               1, V,W DESIRED                   GETAIC 00010
C               2, W DESIRED                     GETAIC 00011
C       IR    = ERROR RETURN  0, SUCCESS            2, C,W NOT FOUND    GETAIC 00012
C               1, C NOT FOUND                 3, NOTHING FOUND    GETAIC 00013
C                                                 GETAIC 00014
C COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP, FILES 00002
1          IOLFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC FILES 00003
COMMON /TAPEIO/ NFS,NMS,LS,NMR,TD(20),NID,ITYPE,LRS,LWS,M,N, TAPEIO 00002
1          PARM(10),IRR TAPEIO 00003
DIMENSION IPARM(10) TAPEIO 00004
EQUIVALENCE (PARM,IPARM) TAPEIO 00005
COMMON /MUAICS/ YBAR,EL,MUAIC(2,50),NROWS,SURF, MUAICS 00002
1          YBARL,ELL,MUAICL(2,50),NROWSL,SURFL,PSIDIF MUAICS 00003
LOGICAL SURF,SURFL MUAICS 00004
COMMON /PAICS / NAK, NTK, NRWTK, NLWTK, PAIC(4,50) PAICS 00002
INTEGER PAIC PAICS 00003
DIMENSION NK(4) PAICS 00004
EQUIVALENCE (NAK,NK(1)) PAICS 00005
COMMON /AICS / XVL, C(1640),W(1640),V(1640) AICS 00002
COMPLEX          C, W, V AICS 00003
COMMON /ARRAYS/ KBXCDW,LBXCDW,LBXXC,KBXCDT,LBXCDT,KJALPH,LJALPH, ARRAYS 00002
1          KALPHA,KKERNEL,LKERNEL,KPNTRM,LPNTRM,KDEFSL,KELPHI, ARRAYS 00003
2          LMODES,KPNTSD,LPNNTSD,KSDW,LSDW,KPNTDW,LPNNTDW, ARRAYS 00004
3          KDW,LDW,KTVP,LTVP ARRAYS 00005
LOGICAL MXWRIT,RANDOU GETAIC 00022
DATA MXWRIT,RANDOU / .F.,.F. / GETAIC 00023
DATA IPAIC,IFAICL,IPNT, IFLAG,IFLAGL, ICCDEP / 640 / FTNDI 00061
C
IR = 0 GETAIC 00024
IF (IPNT .NE. 0) GO TO 160 GETAIC 00025
C
C       INITIAL CALL. SET UP FILES AND POINTERS GETAIC 00026
REWIND IAICSC GETAIC 00027
IPNT = 1 GETAIC 00028
C
EXPAND PAIC ARRAY GETAIC 00029
I = 4 GETAIC 00030
IF (NRWTK .EQ. 0) I = 3 GETAIC 00031
IF (NTK .EQ. 0) I = I - 1 GETAIC 00032
IF (NAK .EQ. 0) I = I - 1 GETAIC 00033
IF (NLWTK .EQ. 0) GO TO 120 GETAIC 00034
IF (I .EQ. 4) GO TO 140 GETAIC 00035
DO 110 J = 1,NLWTK GETAIC 00036
PAIC(4,J) = PAIC(I,J) GETAIC 00037
PAIC(I,J) = 0 GETAIC 00038
110 CONTINUE GETAIC 00039
I = I - 1 GETAIC 00040
120 CONTINUE GETAIC 00041
IF (NRWTK .EQ. 0) GO TO 130 GETAIC 00042
IF (I .EQ. 3) GO TO 140 GETAIC 00043
GETAIC 00044
GETAIC 00045

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DO 125 J = 1, NRWKR
PAIC(3,J) = PAIC(1,J)
PAIC(1,J) = 0
125 CONTINUE
I = I - 1
130 CONTINUE
IF (INTX .EQ. 0) GO TO 140
IF (I .EQ. 2) GO TO 140
DO 135 J = 1, NTK
PAIC(2,J) = PAIC(1,J)
PAIC(1,J) = 0
135 CONTINUE
140 CONTINUE
C      ZERO OUT THE AIC ARRAYS
DO 150 I = 1,LKERNL
C(I) = (0.,0.)
W(I) = (0.,0.)
V(I) = (0.,0.)
150 CONTINUE
C
C      GET THE AIC LOCATION
160 CONTINUE
IF (INC((ITPE) .LT. JUCENT)) GO TO 290
ILOC = PAIC(ITPE,JUCENT)
C      ARE THE DESIRED ARRAYS ALREADY IN CORE -
IF (ILOC .EQ. IPATC) GO TO 300
IPATC = ILOC
ILOC = (ILOC-1)*4 + 1
C
C      GET THE MUASIC ARRAY FROM THE NON-PLANAR AIC SCRATCH FILE
C
C      SPACE AND READ MUASIC
CALL RDINIT
IF ((ILOC - IPNT) 200,220,210
C      MUASIC ARE BEHIND CURRENT LOCATION
200 REWIND IAICSC
NNU = ILOC - 1
GO TO 220
C
C      REQUIRED MUASIC ARE AHEAD OF CURRENT POSITION
210 CONTINUE
NMS = ILOC - IPNT
C
C      READ MUASIC FROM IAICSC
220 CONTINUE
MXARRY = 9H    MUASIC
K = 2
CALL READMX( IAICSC, MXARRY, RANDOU, NF3, NMS, LS, NMR, K, MID, ID,
1           ITYPE, LRS, MUASIC, M,N, PARM, IRR)
IF (IRR .NE. 0) GO TO 3000
NROWS = N
IPNT = ILOC + 1
EL = PARM(5)
YBAR = PARM(4)
ICD = IPARM(6)
C
CALL RDINIT
GETAIC 00046
GETAIC 00047
GETAIC 00048
GETAIC 00049
GETAIC 00050
GETAIC 00051
GETAIC 00052
GETAIC 00053
GETAIC 00054
GETAIC 00055
GETAIC 00056
GETAIC 00057
GETAIC 00058
GETAIC 00059
GETAIC 00060
GETAIC 00061
GETAIC 00062
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GETAIC 00092
GETAIC 00093
GETAIC 00094
GETAIC 00095
GETAIC 00096
GETAIC 00097
GETAIC 00098
GETAIC 00099
GETAIC 00100
GETAIC 00101
GETAIC 00102

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C      IS THE C ARRAY DESIRED -          GETAIC 00103
C      IF (ICODE .NE. 0) GO TO 250      GETAIC 00104
C      YES. IS IT AVAILABLE -          GETAIC 00105
C      IF (ICD .EQ. 0) GO TO 240      GETAIC 00106
C      NO. SET THE ERROR FLAG AND CONTINUE
C      IR = 1                          GETAIC 00107
C      GO TO 250                      GETAIC 00108
C      READ THE C ARRAY FROM IAICSC    GETAIC 00109
240 CONTINUE                         GETAIC 00110
      MXARRY = 9HSpatial C            GETAIC 00111
      CALL READMX( IAICSC, MXWRIT,RANDOU, NFS,NMS, LS,NMR, K, NID, ID,
1           ITYPE, LRS, C, M,N, PARM, IRR)   GETAIC 00112
      IF (IRR .NE. 0) GO TO 3000        GETAIC 00113
      CALL RDINIT                      GETAIC 00114
      GO TO 260                      GETAIC 00115
250 CONTINUE                         GETAIC 00116
      NMS = 1                          GETAIC 00117
260 CONTINUE                         GETAIC 00118
      IPNT = IPNT + 1                GETAIC 00119
C      READ THE W ARRAY FROM IAICSC    GETAIC 00120
      MXARRY = 9HSpatial W            GETAIC 00121
      CALL READMX( IAICSC, MWWRIT,RANDOU, NFS,NMS, LS,NMR, K, NID, ID,
1           ITYPE, LRS, W, M,N, PARM, IRR)   GETAIC 00122
      IF (IRR .NE. 0) GO TO 3000        GETAIC 00123
      IPNT = IPNT + 1                GETAIC 00124
      CALL RDINIT                      GETAIC 00125
C      IS THE V ARRAY DESIRED -          GETAIC 00126
      IF (ICODE .EQ. 2) GO TO 300      GETAIC 00127
C      YES. IS IT AVAILABLE -          GETAIC 00128
      IF (ICD .NE. 2) GO TO 280      GETAIC 00129
C      NO. SET ERROR FLAG           GETAIC 00130
      IR = IR + 1                    GETAIC 00131
      GO TO 300                      GETAIC 00132
C      READ THE V ARRAY FROM IAICSC    GETAIC 00133
280 CONTINUE                         GETAIC 00134
      MXARRY = 9HSpatial V            GETAIC 00135
      CALL READMX( IAICSC, MVWRIT,RANDOU, NFS,NMS, LS,NMR, K, NID, ID,
1           ITYPE, LRS, V, M,N, PARM, IRR)   GETAIC 00136
      IF (IRR .NE. 0) GO TO 3000        GETAIC 00137
      IPNT = IPNT + 1                GETAIC 00138
      GO TO 300                      GETAIC 00139
C
C      NO AICS CAN BE FOUND OF THE TYPE DESIRED FOR THIS CHORD
290 IR = 3                          GETAIC 00140
C
300 CONTINUE                         GETAIC 00141
      RETURN                          GETAIC 00142
C
C      DIAGNOSTIC                     GETAIC 00143
C
3000 CONTINUE                         GETAIC 00144
      WRITE (WT6,9000) IAICSC,IRR    GETAIC 00145
      WRITE (WT6,9192) MXARRY, M,N    GETAIC 00146
      CALL FLUSH(1)
9000 FORMAT(49H*** ERROR - WHILE READING FROM SPATIAL AIC FILE ,A10,     GETAIC 00147
1           14H, ERROR CODE = 14, 4H *** )
9192 FORMAT(14X,A10,      20HARRAY, DIMENSIONED (14,2H,14,11H) WAS BEING    GETAIC 00148
                                         GETAIC 00149
                                         GETAIC 00150
                                         GETAIC 00151
                                         GETAIC 00152
                                         GETAIC 00153
                                         GETAIC 00154
                                         GETAIC 00155
                                         GETAIC 00156
                                         GETAIC 00157
                                         GETAIC 00158
                                         GETAIC 00159

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1 SH READ)
ENC

GETAIC 00160
GETAIC 00161

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SUBROUTINE STOSDW(IROW,JCOL, EN, IBOX,LBXCD, IXB,MXB$,$MYB, IRR) STOSDW 00002
C
C      STORES A COMPUTED DOWNWASH VALUE IN THE END-AROUND SUBDIVIDED
C      DOWNWASH ARRAY, AND UPDATES POINTERS WHEN NECESSARY STOSDW 00003
C
C      IROW    = BOX CHORDWISE LOCATION STOSDW 00004
C      JCOL    = BOX SPANWISE LOCATION STOSDW 00005
C      EN      = COMPLEX NORMAL-WASHES TO BY STORED STOSDW 00006
C      IBOX    = ARRAY OF BOX CODES STOSDW 00007
C      LBXCD   = LENGTH OF BOX CODE ARRAY STOSDW 00008
C      IXB     = FIRST SUBDIVIDED ROW OF THE PLANFORM STOSDW 00009
C      MXBS$   = MAXIMUM CHORD LENGTH OF SUBDIVIDED PATTERN STOSDW 00010
C      MYB     = MAXIMUM ROW LENGTH STOSDW 00011
C
C      RETURNS -
C          IRR    = ERROR RETURN, 0 = SUCCESSFUL STOSDW 00012
C          = 1, FUNCTION LOCSDW FOUND THE POINTER OUTSIDE STOSDW 00013
C              THE DEFINED SET OF DOWNWASHES STOSDW 00014
C          ENSUBD = SUBDIVIDED NORMAL-WASH ARRAY WITH ADDED VALUE STOSDW 00015
C
C      COMMON PARAMETERS USED STOSDW 00016
C          MXSKRN = MAXIMUM SIZE OF THE SUBDIVIDED KERNEL STOSDW 00017
C          LSDW   = DIMENSION OF SUBDIVIDED NORMAL WASH ARRAY STOSDW 00018
C          IPNTSD = POINTER ARRAY FOR SUBDIVIDED NORMAL WASH ARRAY STOSDW 00019
C          IPNTIN = NEXT AVAILABLE CELL IN IPNTSD STOSDW 00020
C          IPNTOT = FIRST CURRENTLY VALID CELL IN IPNTSD STOSDW 00021
C          IPNTLS = DIMENSION OF IPNTSD STOSDW 00022
C
C          COMMON / KERN / ERR, MXSKRN, IPKERN, NFKRN, NFATK, NRWEA KERN 00002
C          COMMON / ARRAYS / KBXCDW, LBXCDW, LBOXC, KBXC$T, LBXC$T, KJALPH, LJALPH, ARRAYS 00002
C          1          KALPHA, KKERNL, LKERNL, KPNTM, LPNTM, KDEFSL, KELPHI, ARRAYS 00003
C          2          LMODES, KPNTSD, LPNTSD, KSDW, LSDW, KPNTDW, LPNTDW, ARRAYS 00004
C          3          KDW, LDW, KTVP, LTVP ARRAYS 00005
C          COMMON / SNWASH / IPNTSD(2,50), ENSUBD(2,600), IPNTIN, IPNTOT, IPNTLS SNWASH 00002
C          IPNTSD(LPNTSD), ENSUBD(24LSDW) SNWASH 00003
C          COMPLEX ENSUBD SNWASH 00004
C          COMMON / CHECKPR / DPPCPR, GEOPR, MDCPR, AICLR, NSCPR, SMCPR, GAFPR CHECKPR 00002
C          LOGICAL DPPCPR, GEOPR, MDCPR, AICCR, NSCPR, SMCPR, GAFPR CHECKPR 00003
C          EQUIVALENCE (CHECKPR, NGCPR) STOSDW 00033
C          LOGICAL CHECKPR STOSDW 00034
C          DIMENSION TITL(3) STOSDW 00035
C
C          COMPLEX EN(2) STOSDW 00036
C
C          IRR = 0 STOSDW 00037
C          IS THIS THE INITIAL CALL - STOSDW 00038
C          IF (IROW .EQ. IXB .AND. JCOL .EQ. 1) GO TO 700 STOSDW 00039
C              NO. IS A NEW ROW BEING CONSIDERED - STOSDW 00040
C              STOSDW 00041
C          IF (IROW .GT. IROW) GO TO 200 STOSDW 00042
C              NO. GET THE LOCATION FOR THE VALUE IN THE SUBDIVIDED DOWN-
C              WASH ARRAY STOSDW 00043
C              IJ = LOCSDW(IROW, JCOL, IPNTSD, IPNTIN, IPNTOT, IPNTLS) STOSDW 00044
C              IF (IJ) 900,900,550 STOSDW 00045
C
C              MUST UPDATE POINTERS AND ADD A ROW TO THE SUBDIVIDED BOX ARRAY STOSDW 00046
C              STOSDW 00047
C              STOSDW 00048
C              STOSDW 00049
C
C              200 CONTINUE STOSDW 00050
C              IROWW = IROW STOSDW 00051
C

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C      SET THE NEXT VALUES OF THE POINTER ARRAY (1 ROW)          ST05DW 00053
C      INCREMENT IPNTIN, ALLOWING FOR END-AROUND INCREMENTAL    ST05DW 00054
C      INM1 = IPNTIN                                              ST05DW 00055
C      I1 = IPNTSD(1,IPNTIN)                                         ST05DW 00056
C      CALL PTRNTR (IROW, 1,MYB, .T.,.T., IBOX,LBXCD, IPNTLS,
C           I1, IPNTIN,IPNTSD)                                         ST05DW 00057
C
C      LOOP INCREMENTING IPNTOT, IF OVER-LAP OCCURS.          ST05DW 00058
220 CONTINUE
C      IF (IPNTIN-IPNTOT) 230,225,235                                ST05DW 00059
225 IPNTOT = MOD(IPNTOT,IPNTLS) +1                               ST05DW 00060
C      GO TO 220
230 ISROWS = IPNTIN-IPNTOT+IPNTLS                                ST05DW 00061
C      GO TO 240
235 ISROWS = IPNTIN-IPNTOT                                         ST05DW 00062
240 CONTINUE
C      IF (ISROWS .GT. MXSKRN) IPNTOT = MOD(IPNTOT+ISROWS-MXSKRN-1,
C           1 IPNTLS) + 1                                         ST05DW 00063
C
C      KEEP SUBDIVIDED DOWNWASHES END-AROUND.                  ST05DW 00064
C      HAS THE ARRAY LIMIT BEEN EXCEEDED -
C      ICOLD = IPNTSD(1,INM1)                                         ST05DW 00065
C      IF (IPNTSD(1,IPNTIN) .LE. LSTM + 1) GO TO 405             ST05DW 00066
C      LIMIT EXCEEDED BY CURRENT ROW. PLACE AT BEGINNING OF THE ARRAY ST05DW 00067
C      IF (.NOT. CHECKPR) GO TO 400
TITL(1) = 10HEN SUBDIVI                                         ST05DW 00068
TITL(2) = 10HDFD, UPPER                                         ST05DW 00069
TITL(3) = 10H, PARTIAL                                         ST05DW 00070
C      IF (IPNTIN .LT. IPNTOT) GO TO 395
CALL PRINTR(TITL,0,ENSUBD,2,IPNTOT,IPNTIN-1,MYB,IPNTSD)        ST05DW 00071
C      GO TO 400
395 CALL PRINTR(TITL,0,ENSUBD,2,IPNTOT,IPNTLS-1, MYB,IPNTSD)
CALL PRINTR(TITL,0,ENSUBD,2, 1,  IPNTIN-1,   MYB,IPNTSD)
400 CONTINUE
C      IPNTSD(1,IPNTIN) = IPNTSD(1,IPNTIN) - ICOLD + 1
C      IPNTSD(1,INM1) = 1
C
C      405 CONTINUE
C      IPM = IPNTSD(1,IPM1)
C      IPT = IPNTSD(1,IPNTIN)
C      IPO = IPNTSD(1,IPNTOT)
C
C      IF (ICOLD .GT. IPO) GO TO 430
C
C      ARRAY WAS ALREADY END-AROUND PRIOR TO LATEST ADDITION
C      IF (ICOLD .EQ. IPM) GO TO 420
C      ADDED ROW WENT END-AROUND AS WELL
410 IPNTOT = MOD(IPNTOT,IPNTLS) + 1
C      IF (IPNTSD(1,IPNTOT) .NE. 1 ) GO TO 410
C      IPNTOT = MOD(IPNTOT,IPNTLS) + 1
C      IPO = IPNTSD(1,IPNTOT)
C      GO TO 440
C
C      ARRAY WAS SEQUENTIAL. CHECK WHETHER IT HAS GONE END-AROUND
430 CONTINUE
C      IF (ICOLD .EQ. IPM) GO TO 500
C      IT HAS GONE END-AROUND

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C
440 CONTINUE
C      HAS THE ARRAY BEEN OVER-WRITTEN IN GOING END-AROUND -
IF (IPO .GT. IPI) GO TO 500
C      YES. MOVE IPNTOT UNTIL CLEAR.
IPNTOT = MOD(IPNTOT,IPNTLS) + 1
IPO = IPNTSD(1,IPNTOT)
IF (IPO .NE. 1) GO TO 440
C
C      ALL POINTERS HAVE BEEN RESET. GET LOCATION
500 CONTINUE
IJ = LOCSDW(IROW,ICOL,IPNTSD,IPNTIN,IPNTOT,IPNTLS)
IF (IJ .EQ. 0) GO TO 900
C
C      STORE THE DOWNASH VALUE
550 CONTINUE
ENSUBD(1,IJ) = EN(1)
ENSUBD(2,IJ) = EN(2)
600 RETURN
C
C
C      INITIAL CALL
700 CONTINUE
IPNTOT = MOD(IXB-1,IPNTLS) + 1
IPNTIN = IPNTOT
MX = MIN(MXB-IXB+1, MXSKRN, IPNTLS-1)
C      SET UP POINTER ARRAY FOR FIRST PASS
CALL PTRNTR(IXB, MX, MYB, .T., .T., IBOX, LBXCD, IPNTLS,
1, IPNTIN, IPNTSD)
720 CONTINUE
IF (IPNTSD(1,IPNTIN) .LE. LSDW ) GO TO 730
IPNTIN = IPNTIN - 1
MX = MX - 1
GO TO 720
730 CONTINUE
IROW = MX + IXB - 1
C
ENSUBD(1,1) = EN(1)
ENSUBD(2,1) = EN(2)
GO TO 800
*
C
C      ERROR
900 IRR = 1
GO TO 600
C
END

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STOSDW 00110
STOSDW 00111
STOSDW 00112
STOSDW 00113
STOSDW 00114
STOSDW 00115
STOSDW 00116
STOSDW 00117
STOSDW 00118
STOSDW 00119
STOSDW 00120
STOSDW 00121
STOSDW 00122
STOSDW 00123
STOSDW 00124
STOSDW 00125
STOSDW 00126
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STOSDW 00151
STOSDW 00152
STOSDW 00153
STOSDW 00154
STOSDW 00155

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SUBROUTINE PRINTR( TITL,IMODE,ARRAY,K,IXB,MXB,MYB,IPNTRM)          PRNTNW 00002
C
C      TITL  - TITLE TO PRINT FOR THE ARRAY                         PRNTNW 00003
C      IMODE - MODE SHAPE NUMBER                                     PRNTNW 00004
C      ARRAY - ARRAY TO BE PRINTED                                    PRNTNW 00005
C
C      DIMENSION ARRAY(K,1), TITL(3)                                 PRNTNW 00006
C      COMPLEX ARRAY                                                 PRNTNW 00007
C      DIMENSION IPNTRM(2,50)                                         PRNTNW 00008
C      COMMON /CTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT, CTRL 00002
C      I                      DEFAULT                                CTRL 00003
C      LOGICAL    PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT                 CTRL 00004
C      COMMON /PROBLM/ XMACH,NMODOES,NTSLOP,NKVALS,SMOOTH,NDEG,CRCFIT, PROBLM 00002
C      I                      EXAIC,SUBDV,PLYWOOD                  PROBLM 00003
C      LOGICAL    SMOOTH,CRCFIT,EXAIC,SUBDV,PLYWOOD                PROBLM 00004
C      COMMON /FILES / NTS,NT6,INTAPE,INFSR,NPLAIC,NSPAIC,NOUTP, FILES 00002
C      I                      IOUFSR,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC FILES 00003
C      COMMON /XVAL / IKVAL,XXVAL(20), XKS(20)                      KVAL 00002
C      DIMENSION PC(2)                                              PRNTNW 00016
C      DIMENSION S(50),D(50)                                         PRNTNW 00020
C      EQUIVALENCE (S(1),BUFF(1)),(D(1),BUFF(1251))                 PRNTNW 00021
C      REAL K1                                                       PRNTNW 00022
C      INTEGER PAGE                                                 PRNTNW 00023
C      COMMON /RWCUFF/ BFCODE,IBFCNT,  BUFF(3280)                   RWBUFF 00002
C      DATA  PC  / 10HPAGE CONTI,4HNUED /
C      DATA  BLANK / 1H /
C      DATA  XINIT / -1.0 /
C      K1 = XXVAL(IKVAL)
C      IF(XKS(IKVAL).NE.XINIT) K1 = XKS(IKVAL)

C
C      PAGE = 0                                                       PRNTNW 00026
C      N =1                                                       PRNTNW 00027
C      M =4                                                       PRNTNW 00028
C      IF(M.GT.MYB) M = MYB                                       PRNTNW 00029
100   LINE = 100
200   DO 1400 I = IXB,MXB
      DO 300 J=N,M
         S(J) = 0.0
         D(J) = 0.0
300   CONTINUE
      IF(LINE.LE.50) GO TO 900
      PAGE = PAGE + 1
      LINE = 4
      WRITE (NT6,9001) TITLE,TITL, XMACH, K1, IMODE
C
      IF(PAGE.EQ.1) GO TO 700
      WRITE (NT6,9005) PC
      GO TO 800
700   WRITE(NT6,9005)
800   CONTINUE
      WRITE(NT6,6006) (BLANK,J,J=N,M)
      WRITE(NT6,6007) (BLANK, J=N,M)
900   CONTINUE
      JS = IPNTRM(2,1)
      IF (JS .LE. 0) GO TO 1400
      IX = IPNTRM(1,1)
      PRNTNW 00042
      PRNTNW 00043
      PRNTNW 00044
      PRNTNW 00045
      PRNTNW 00046
      PRNTNW 00047
      PRNTNW 00048
      PRNTNW 00049
      PRNTNW 00050
      PRNTNW 00051
      PRNTNW 00052
      PRNTNW 00053

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JE = IPNTRM(1,I+1) - IDX + JS -1          PRNTNW 00054
IF(JE.EQ.0) GO TO 1400                   PRNTNW 00055
DO 1000 J=JS,JE                          PRNTNW 00056
S(J) = REAL(ARRAY(1,IDX) )                PRNTNW 00057
D(J) = AIMAG(ARRAY(1,IDX) )              PRNTNW 00058
IDX = IDX +1                           PRNTNW 00059
1000 CONTINUE                            PRNTNW 00060
DO 1200 J=N,M                           PRNTNW 00061
IF(S(J)) 1300,1100,1300                 PRNTNW 00062
1100 CONTINUE                            PRNTNW 00063
IF(D(J)) 1300,1200,1300                 PRNTNW 00064
1200 CONTINUE                            PRNTNW 00065
GO TO 1400                               PRNTNW 00066
1300 WRITE (NT6,9013) I,(S(J),D(J),J=N,M)  PRNTNW 00067
LINE = LINE + 1                         PRNTNW 00068
1400 CONTINUE                            PRNTNW 00069
N = N+4                                 PRNTNW 00070
N = N+4                                 PRNTNW 00071
IF(N.GT.MYB) GO TO 1500                 PRNTNW 00072
IF(M.GT.MYB) M=MYB                      PRNTNW 00073
IF(LINE.GT.45) GO TO 100                PRNTNW 00074
WRITE(N16,6006) (BLANK,J,J=N,M)        PRNTNW 00075
WRITE(N16,6007) (BLANK, J=N,M)          PRNTNW 00076
LINE = LINE+3                           PRNTNW 00077
GO TO 200                                PRNTNW 00078
1500 CONTINUE                            PRNTNW 00079
RETURN                                  PRNTNW 00080
9001 FORMAT(1H1,20X,8A10/ 50X,3A10/ 46X, 'H( HACH F5.3,5X,10HRED. FREQ. PRNTNW 00081
   1 *-* ,FB.5, * )* /52X,*MCDE  SHAPE* ,I3) PRNTNW 00082
9005 FORMAT(44X,42(IH-),20X,A10,A4)      PRNTNW 00083
9006 FORMAT(4HROW, A1,14X,SHCHORD,I3, 3(A1,22X,SHCHRD,I3) ) PRNTNW 00084
9007 FORMAT(3X, 4(A1,9X,4HREAL,8X,9HIMAGINARY) ) PRNTNW 00085
9013 FORMAT(14.8E16.8)                   PRNTNW 00086
END                                     PRNTNW 00087

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SUBROUTINE DCODE(IBOX,LBOX, IA, JA, IL,JL, SUBD, ICODE)      DCODNW 00002
DIMENSION IBOX(LBOX,1), ICODE(1)                            DCODNW 00003
C                                                               DCODNW 00004
C       IBOX - ARRAY OF BOX CODES IN PACKED WORD FORMAT      DCODNW 00005
C       LBOX - ROW DIMENSION OF BOX CODES ARRAY             DCODNW 00006
C       IA - I-TH INDEX OF FIRST CODE TO RETRIEVE          DCODNW 00007
C       JA - J-TH INDEX OF FIRST CODE TO RETRIEVE          DCODNW 00008
C       IL - LAST BOX CODE ON THE JA-TH CHORD TO RETRIEVE   DCODNW 00009
C       JL - LAST BOX ON THE IA-TH ROW TO RETRIEVE          DCODNW 00010
C       SUBD - T., SUBDIVIDED BOX CODES DESIRED, .F. UNSUBDIVIDED. DCODNW 00011
C       ICODE - ARRAY INTO WHICH BOX CODE WILL BE STORED.    DCODNW 00012
C                                                               DCODNW 00013
C                                                               DCODNW 00014
C       COMMENT ON USAGE                                     DCODNW 00015
C       BOX CODES CAN BE RETRIEVED FOR ONE BOX, A ROW OR PART OF DCODNW 00016
C       A ROW, OR A COLUMN OR PART OF A COLUMN. A ROW AND COLUMN CAN DCODNW 00017
C       NOT BE RETRIEVED AT THE SAME TIME. IF ONLY 1 BOX IS DESIRED DCODNW 00018
C       SET IL = IA AND JL = JA. IF BOTH IL .NE. IA AND JL .NE. DCODNW 00019
C       JA, ONE ROW WILL BE RETURNED, IL BEING IGNORED.        DCODNW 00020
C                                                               GEOMTY 00002
C       COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF. 1   GEOMTY 00003
1           B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,                GEOMTY 00004
2           MXBW,MXBBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW,            GEOMTY 00005
3           IXBW,XCENTR                                         GEOMTY 00006
LOGICAL COPLAN
LOGICAL SUBD
INTEGER SHIFT
DATA NBWRD /20/
MASK = 7
IB = 1
IF (SUBD) GO TO 50
I = NSUBDV * (IA-1) + IXBW
J = NSUBDV * (JA-1) + NSUBCN
ISKIP = NSUBDV
IEND = NSUBDV * (IL-1) + IXBW
JEND = NSUBDV * (JL-1) + NSUBCN
GO TO 60
50 CONTINUE
I = IA
J = JA
ISKIP = 1
IEND = IL
JEND = JL
60 CONTINUE
IF (JL .EQ. JA) GO TO 1100
C
C       PROGRAM WILL RETRIEVE NI BOXES FROM ROW I
100 CONTINUE
DO 1000 JJ = J,JEND,ISKIP
JB = (JJ-1)/NBWRD + 1
IJWORD = IBOX(I,JB)
JR = (NBWRD - MOD(JJ,NBWRD)) * 3
IF (JB.EQ.60) JB = 0
C       JB = NUMBER OF BITS TO SHIFT LEFT.
IJMASK= BSHIFT(MASK,JB)
IJCODE = IJWORD.AND.IJMASK
NJB = -JB
ICODE(IB) = SHIFT(IJCODE,NJB)      DCODNW 00042
DCODNW 00043
DCODNW 00044
DCODNW 00045
DCODNW 00046
DCODNW 00047
DCODNW 00048
DCODNW 00049
DCODNW 00050
DCODNW 00051
DCODNW 00052
DCODNW 00053
DCODNW 00054

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IB = IB + 1	DCODNW 00055
1000 CONTINUE	DCODNW 00056
GO TO 3000	DCODNW 00057
C	DCODNW 00058
C PROGRAM WILL RETRIEVE NJ BOXES FROM CHORD J	DCODNW 00059
1100 CONTINUE	DCODNW 00060
JSB = (J-1)/NBWRD + 1	DCODNW 00061
JB = (NBWRD - MOD(J,NBWRD)) * 3	DCODNW 00062
IF(JB.EQ.60) JB = 0	DCODNW 00063
IJMASK = SHIFT(MASK,JB)	DCODNW 00064
NJB = -JB	DCODNW 00065
D~ 2000 II = I,IEND,ISKIP	DCODNW 00066
IJWORD = IBOX(II,JSB)	DCODNW 00067
IJCODE = IJWORD.AND.IJMASK	DCODNW 00068
ICODE(IB) = SHIFT(IJCODE,NJB)	DCODNW 00069
IB = IB + 1	DCODNW 00070
2000 CONTINUE	DCODNW 00071
C	DCODNW 00072
3000 CONTINUE	DCODNW 00073
RETURN	DCODNW 00074
END	DCODNW 00075

```

SUBROUTINE POINTR( IX,MX, MYB,           SUBD, DIAPH, IBOX,LBXCD,
1                  MXIR,  IPNTIN,IPNTR )
C
C      GENERATES A POINTER ARRAY WHICH SERVES AS AN INDEX TO A
C      CONDENSED ARRAY OF BOX VALUES (MOTES, DOWNWASHES, ETC.)
C
C      IX      = CENTER OF FIRST BOX TO USE
C      MX      = NUMBER OF ROWS TO PROCESS
C      MYB     = MAXIMUM ROW LENGTH
C      SUBD   = .T., SUBDIVIDED BOXES TO BE USED
C              = .F., UNSUBDIVIDED BOXES TO BE USED
C      DIAPH   = .T., DIAPHRAGMS TO BE INCLUDED
C              = .F., ONLY PLANFORM BOXES
C      IBOX    = ARRAY OF BOX CODES
C      LBXCD  = ROW DIMENSION OF BOX CODES ARRAY
C      MXIR   = SIZE OF IPNTR ARRAY
C      IPNTIN = VALUE TO BE USED FOR FIRST POINTER (NORMALLY 1)
C      IN/OUT -
C          IPNTR = LOCATION OF NEXT ABAILABLE LOCATION IN IPNTR ARRAY
C                  (OVERLAP OF TAIL IS ACCOUNTED FOR HERE.)
C      OUT PUTS -
C          IPNTR = POINTER ARRAY - OUTPUT FROM THE SUBROUTINE
C          IPNTR(1,I) = LOCATION OF THE FIRST VALUE FOR ROW I
C          IPNTR(2,I) = CHORD LOCATION (SUBSCRIPT J) OF THAT VALUE
C
C      COMMON VALUES USED -
C
C      DIMENSION IBOX(LBXCD,1),  IPNTR(2,1), ICODE(150)
C      LOGICAL  DIAPH,SUBD,WING
C
C      ITOTBX = IPNTIN
C      IL = IX + MX - 1
C      DO 100 IROW = IX,IL
C      IPBSUM=0
C      IUSUM=0
C      IPBX=1
C      IBXSUM = 0
C      CALL DC005R(IBOX,LBXCD, IROW,1, IROW,MYB, SUBD, ICODE)
C
C      FIND LAST BOX ON ROW
C      M = MYB
C      IF (DIAPH) GO TO 20
C      ONLY PLANFORM BOXES DESIRED
C      DO 15 JCOL = 1,MYB
C      IF (ICODE(M) .EQ. 1) GO TO 30
C      M = M - 1
C 15 CONTINUE
C      GO TO 92
C      PLANFORM AND DIAPHRAGM DESIRED
C 20 CONTINUE
C      DO 2 JCOL = 1,MYB
C      IF (ICODE(M) .NE. 0) GO TO 30
C      M = M - 1
C 25 CONTINUE
C      GO TO 92
C
C      POINTR 00002
C      POINTR 00003
C      POINTR 00004
C      POINTR 00005
C      POINTR 00006
C      POINTR 00007
C      POINTR 00008
C      POINTR 00009
C      POINTR 00010
C      POINTR 00011
C      POINTR 00012
C      POINTR 00013
C      POINTR 00014
C      POINTR 00015
C      POINTR 00016
C      POINTR 00017
C      POINTR 00018
C      POINTR 00019
C      POINTR 00020
C      POINTR 00021
C      POINTR 00022
C      POINTR 00023
C      POINTR 00024
C      POINTR 00025
C      POINTR 00026
C      POINTR 00027
C      POINTR 00028
C      POINTR 00029
C      POINTR 00030
C      POINTR 00031
C      POINTR 00032
C      POINTR 00033
C      POINTR 00034
C      POINTR 00035
C      POINTR 00036
C      POINTR 00037
C      POINTR 00038
C      POINTR 00039
C      POINTR 00040
C      POINTR 00041
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C      POINTR 00043
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C      POINTR 00048
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C      POINTR 00050
C      POINTR 00051
C      POINTR 00052
C      POINTR 00053
C      POINTR 00054
C      POINTR 00055
C      POINTR 00056
C      POINTR 00057
C      POINTR 00058

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C      LOOP ON CHORDS IN THE ROW
30 CONTINUE
DO 90 JCOL = 1,M
IF (ICODE(JCOL) - 1) 35,40,50
C
C      ICODE = 0
C
35 IF (IDISUM.NE.0) GO TO 40
IPBX = IPBX + 1
GO TO 90
C
C      ICODE = 1
40 IPBSUM = IDISUM + 1
C
C      ICODE = 2 OR 3
50 CONTINUE
IDISUM = IDISUM + 1
90 CONTINUE
C
92 CONTINUE
IPNTR(1,IPNTIN) = ITOTBX
IPNTR(2,IPNTIN) = IPBX
IPNTIN = MOD(IPNTIN,MXIR) + 1
IF (DIAPH) GO TO 95
ITOTBX = ITOTBX + IPBSUM
GO TO 100
95 ITOTBX = ITOTBX + IDISUM
100 CONTINUE
IPNTR(1,IPN,IN) = ITOTBX
IPNTR(2,IPNTIN) = 0
RETURN
END

```

POINTR	00059
POINTR	00060
POINTR	00061
POINTR	00062
POINTR	00063
POINTR	00064
POINTR	00065
POINTR	00066
POINTR	00067
POINTR	00068
POINTR	00069
POINTR	00070
POINTR	00071
POINTR	00072
POINTR	00073
POINTR	00074
POINTR	00075
POINTR	00076
POINTR	00077
POINTR	00078
POINTR	00079
POINTR	00080
POINTR	00081
POINTR	00082
POINTR	00083
POINTR	00084
POINTR	00085
POINTR	00086
POINTR	00087
POINTR	00088
POINTR	00089
POINTR	00090

```

FUNCTION LOCSDW(IROW,JCOL,IPNTSD,IPNTIN,IPNTOT,IPNTLS) LOCSDW 00002
C
C      RETURNS THE LOCATION OF THE WORD IN THE END-AROUND SUBDIVIDED LOCSDW 00003
C      DOWNWASH ARRAY CORRESPONDING TO BOX(IROW,JCOL) OF THE SUB- LOCSDW 00004
C      DIVIDED BOX ARRAY LOCSDW 00005
C
C          IROW = BOX CHORDWISE LOCATION LOCSDW 00006
C          JCOL = BOX SPANWISE LOCATION LOCSDW 00007
C          IPNTSD = ARRAY OF POINTERS LOCSDW 00008
C          IPNTIN = NEXT AVAILABLE (UNUSED) CELL IN IPNTSD (END- LOCSDW 00009
C          AROUND)
C          IPNTOT = FIRST CURRENTLY AVAILABLE CELL IN IPNTSD LOCSDW 00010
C          IPNTLS = LAST CELL OF IPNTSD (LENGTH OF ARRAY) LOCSDW 00011
C
C      RETURN -
C          LOCSDW = LOCATION OF DESIRED DOWNWASH, IF SUCCESSFUL LOCSDW 00012
C          = 0, IF LOCPNT LIES OUTSIDE THE DEFINED AREA. LOCSDW 00013
C
C      DIMENSION IPNTSD(2,IPNTLS) LOCSDW 00014
C
C      LOCPNT = MOD(IROW+1,IPNTLS) + 1 LOCSDW 00015
C          LOCPNT = LOCATION OF CELL IN IPNTSD WHICH WAS OR IS TO BE LOCSDW 00016
C          USED LOCSDW 00017
C          IF(IPNTIN - IPNTOT) 100, 300, 200 LOCSDW 00018
C          END AROUND HAS OCCURRED LOCSDW 00019
100 IF (LOCPNT - IPNTIN) 400, 300, 150 LOCSDW 00020
C          NOT IN UPPER PART. IS LOCPNT WITHIN BOTTOM PART - LOCSDW 00021
150 IF (LOCPNT - IPNTOT) 300, 400, 400 LOCSDW 00022
C
C          NO END AROUND, NORMAL SEQUENCE LOCSDW 00023
200 IF (LOCPNT - IPNTIN) 250, 300, 300 LOCSDW 00024
C          LESS THAN UPPER LIMIT. IS LOCPNT .GE. LOWER LIMIT - LOCSDW 00025
250 IF (LOCPNT .GE. IPNTOT) GO TO 400 LOCSDW 00026
C
C          ERROR OR INITIAL CONDITION ENCOUNTERED (SHOULD NEVER OCCUR) LOCSDW 00027
300 LOCSDW = 0 LOCSDW 00028
GO TO 500 LOCSDW 00029
C
C          SUCCESSFUL, BOX HAS BEEN DEFINED LOCSDW 00030
400 IFB = IPNTSD(2,LOCPNT) LOCSDW 00031
IF(JCOL.LT.IFB) GO TO 300 LOCSDW 00032
LOCSDW = IPNTSD(1,LOCPNT) + JCOL-IFB LOCSDW 00033
C
500 CONTINUE LOCSDW 00034
RETURN LOCSDW 00035
END LOCSDW 00036

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SUBROUTINE SMPLW( IBOX,LBXCD, JCHRD, JT, IFRST,ILAST)          SMPLW 00002
C
C      COMPUTES DOWNWASH, SIDEWASH AND VELOCITY POTENTIAL FOR A   SMPLW 00003
C      SAMPLE CHORD LOCATED IN THE WING FLOW FIELD                 SMPLW 00004
C
C      IBOX    = ARRAY OF BOX CODES FOR THE WING                      SMPLW 00005
C      LBXCD  = LENGTH OF BOX CODE ARRAY                         SMPLW 00006
C      JCHRD  = SAMPLE-WASH CHORD NUMBER                      SMPLW 00007
C      JT     = J-LOCATION OF THE CHORD                      SMPLW 00008
C      IFRST  = NUMBER OF FIRST SAMPLE BOX                   SMPLW 00009
C      ILAST  = NUMBER OF LAST SAMPLE BOX                  SMPLW 00010
C
C      DIMENSION IBOX(LBXCD,1)                                     SMPLW 00011
COMMON /CONTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT, CONTRL 00002
1           DEFAULT                                              CONTRL 00003
LOGICAL PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT                     CONTRL 00004
COMMON /PROBLW/ XMAHC,NMOCES,NTSLOF,NKVALS,SMOOTH,NDEG,CDFIT, PROBLW 00002
1           EXAIC,SUBDV,PLYWOOD                                PROBLW 00003
LOGICAL SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD                     PROBLW 00004
COMMON /SNASH/ IPNTSD(2,50), ENSUD(2,600), IPNTIN,IPNTOT,IPNTLS SNASH 00002
C           IPNTSD(IPNTSD), ENSUD(2*LSDW)                      SNASH 00003
COMPLEX ENSUD
COMMON /MUAICS/ YBAR,EL,MUAIC(2,50),NROWS,SURF,               MUAICS 00002
1           YBARL,ELL,MUAICL(2,50),NROWSL,SURFL,PSIDIF        MUAICS 00003
LOGICAL SURF,SURFL
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,    GEOMTY 00002
1           B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,                GEOMTY 00003
2           MXBW,MXBW,MYBW,MYBBW,MXBSW,MYZSW,MYBBSW,            GEOMTY 00004
3           IXBW,XCENTR                                         GEOMTY 00005
LOGICAL COPLAN
COMMON /FILES/ NT5,NT6,INTAPE,INFSP,NFLAIC,NSFAIC,NJTP, FILES 00002
1           ICFLSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC        FILES 00003
COMMON /IOCONT/ OPLAIC,OSPAIC,WGEOM,WGNAF,WTSI,WTSI,PRBOX, IOCONT 00002
1           PRPAIC,PRSAIC,PRMDS,PRCOEF,PRDW,PRSW,PRVP,        IOCONT 00003
2           PRBL,PRDCP,PRGNAF,PRGNAC,PRSL,PRLW,PRNW,PRCM      BCSFRB 00001
EQUIVALENCE (PRW,PRDW)
LOGICAL OPLAIC,OSPAIC,WGEOM,WGNAF,WTSI,WTSI,PRBOX,PRPAIC, IOCONT 00005
1           PRSAIC,PRMDS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAF, IOCONT 00006
2           PRDCP,PRGNAC,PRW,PRLW,PRNW,PRCM                  BCSFRB 00002
COMMON /MOCOM/ SYM,SYMT,MYPW,MYPET                           MOCOM 00002
COMMON /NWASHES/ IPNTLW(2,100),ENRS(1275), ENRLS(1275),IOMLAPN NWASHES 00002
COMPLEX ENRS,ENRLS
COMMON /AICS / XKVL, C(1640),W(1640),V(1640)                AICS 00002
COMPLEX C, W, V
COMPLEX DW(52),SW(50),LW(50),PHI(50)                         SMPLW 00025
EQUIVALENCE (SW,IPNTSD),(LW,ENSUD),(DW,ENSUD(1,26))
EQUIVALENCE (PHI,ENSUD(1,52))
CC .EX WSUM,VSUM,PHISUM,EN
IN,EVER RWT,LWT
DATA RWT,LWT / 3.4 /
C
C      SET CONSTANTS
CPSI = COS(PSIW)
SPSI = SIN(PSIW)
BINV = 1./B1
C
C      COMPUTE THE RIGHT WING CONTRIBUTION TO THE SAMPLE CHORD

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C          GET THE NECESSARY AIC ARRAYS           SMPLW 00038
C          CALL GETAIC(JOHRD,RWT,0,IR)           SMPLW 00039
IF (IR .NE. 0) GO TO 800                      SMPLW 00040
YMBAR = ( JT-.5 + EL*SPSI ) / CPSI           SMPLW 00041
JBAR = YMBAR                                     SMPLW 00042
IF (YMBAR .GE. 0) JBAR = JBAR + 1             SMPLW 00043
NUBMIN = ABS(EL) + .5                         SMPLW 00044
IBX = 1                                         SMPLW 00045
SMPLW 00046
C          LOOP ON BOXES ALONG THE SAMPLE CHORD   SMPLW 00047
DO 200 IBXX = 1FRST,ILAST                   SMPLW 00048
NUBMAX = IPXX - 1                           SMPLW 00049
I = IBXX - NUBMIN                         SMPLW 00050
SMPLW 00051
C          ZERO OUT THE SUMMATION VARIABLES      SMPLW 00052
WSUM = (0.,0.)                                SMPLW 00053
VSUM = (0.,0.)                                SMPLW 00054
RHISUM = (0.,0.)                               SMPLW 00055
SMPLW 00056
SMPLW 00057
C          IF (YBAR) 120,125,130                 SMPLW 00058
120 JINCR = -1                                SMPLW 00059
GO TO 135                                     SMPLW 00060
125 IAIC = NUBMIN+2                          SMPLW 00061
INCAIC = 2*NUBMIN + 1                         SMPLW 00062
JINCR = 1                                     SMPLW 00063
GO TO 140                                     SMPLW 00064
130 JINCR = 1                                SMPLW 00065
135 IAIC = NUBMIN+2 + NUBMIN                SMPLW 00066
INCAIC = 2*NUBMIN + 2                         SMPLW 00067
140 CONTINUE                                  SMPLW 00068
SMPLW 00069
C          LOOP FORWARD OVER THE RIGHT WING       SMPLW 00070
DO 190 NUBAR = NUBMIN,NUBMAX                 SMPLW 00071
MUAIC1 = MUAIC(1,NUBAR+1)                     SMPLW 00072
MUAIC2 = MUAIC(2,NUBAR+1)                     SMPLW 00073
IF (MUAIC2 .EQ. 0) GO TO 185                SMPLW 00074
IF (YBAR .GE. 0) GO TO 150                  SMPLW 00075
JCCLR = JBAR + NUBAR - MUAIC2 + 1           SMPLW 00076
GO TO 160                                     SMPLW 00077
150 JCCLR = JBAR - NUBAR + MUAIC1 - 1        SMPLW 00078
160 CONTINUE                                  SMPLW 00079
SMPLW 00080
C          LOOP ON A ROW OF WING BOXES, COMPUTING   SMPLW 00081
RIGHT WING CONTRIBUTION                      SMPLW 00082
DO 180 MUA1 = MUAIC1,MUAIC2                 SMPLW 00083
IF (JCCLR .LE. 0) GO TO 170                 SMPLW 00084
CALL DCODE(IBOX,LBXCD,I,JCCLR,I,JCCLR,.F.,ICD)
IF (ICD .EQ. 0) GO TO 170                 SMPLW 00085
C          A CONTRIBUTING BOX HAS BEEN FOUND. GET   SMPLW 00086
THE AIC LOCATION                            SMPLW 00087
C          GET THE NORMAL-WASH LOCATION          SMPLW 00088
IDB = LOCSDW(I,JCCLR,IPNTDW,LPNTDW,1,LPNTDW)
C          ADD THIS CONTRIBUTION TO THE SUMS       SMPLW 00089
IF (EL .LT. 0) GO TO 163                  SMPLW 00090
EN = ENRUS(IDS)                           SMPLW 00091
GO TO 165                                     SMPLW 00092
165 EN = ENRLS(IDS)                        SMPLW 00093
SMPLW 00094

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165 CONTINUE
    WSUM = WSUM + W(KAIC) * EN
    VSUM = VSUM + V(KAIC) * EN
    PHISUM = PHISUM + C(KAIC) * EN
170 CONTINUE
    JCCLR = JCCLR + JINCR
180 CONTINUE
C           END OF LOOP FOR RIGHT WING ROW CONTRIBUTIONS
C
185 CONTINUE
    I = I - 1
    IF (I .LE. 0) GO TO 195
    IAIC = IAIC + INCAIC
    INCAIC = INCAIC + 2
190 CONTINUE
C           END OF LOOP FORWARD ON RIGHT WING ROWS, FROM 140*
C
195 CONTINUE
    DW(IBX) = BINV * (CPSI*WSUM + SPSI*VSUM)
    SW(IBX) = BINV * (CPSI*VSUM - SPSI*WSUM)
    PHI(IBX) = PHISUM
    IBX = IBX + 1
200 CONTINUE
C           END OF LOOP ON RECEIVING BOXES, FOR RIGHT WING CONTRIBUTIONS
C
NBXS = IBX - 1
C           IS LEFT WING CONTRIBUTION NEEDED -
IF (SYM .EQ. 0) GO TO 310
C           YES. GET THE AIC ARRAYS FOR LEFT WING CONTRIBUTIONS.
CALL GETAIC(JCHRD, LWT, 0, IR)
IF (IR .NE. 0) GO TO 800
NUBMIN = ABS(EL) + .5
IBX = 1
YMBAR = (-JT+.5 + EL*SPSI) / CPSI
JBAR = YMBAR
IF (YMBAR .GE. 0) JBAR = JBAR + 1
C           LOOP ON BOXES ALONG THE SAMPLE CHORD
DO 300 IBXX = IFRST,ILAST
    NUBMAX = IBXX - 1
    I = IBXX - NUBMIN
    C           ZERO OUT THE SUMMATION VARIABLES
    WSUM = (0.,0.)
    VSUM = (0.,0.)
    PHISUM = (0.,0.)
C
    IF (YBAR) 220,225,230
220 JINCR = 1
    GO TO 235
225 IAIC = NUBMIN*2
    INCAIC = 2*NUBMIN + 1
    JINCR = -1
    GO TO 240
230 JINCR = -1
235 IAIC = NUBMIN*2 + NUBMIN
    INCAIC = 2*NUBMIN + 2
240 CONTINUE
C

```

SMPLW 00095
 SMPLW 00096
 SMPLW 00097
 SMPLW 00098
 SMPLW 00099
 SMPLW 00100
 SMPLW 00101
 SMPLW 00102
 SMPLW 00103
 SMPLW 00104
 SMPLW 00105
 SMPLW 00106
 SMPLW 00107
 SMPLW 00108
 SMPLW 00109
 SMPLW 00110
 SMPLW 00111
 SMPLW 00112
 SMPLW 00113
 SMPLW 00114
 SMPLW 00115
 SMPLW 00116
 SMPLW 00117
 SMPLW 00118
 SMPLW 00119
 SMPLW 00120
 SMPLW 00121
 SMPLW 00122
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 SMPLW 00144
 SMPLW 00145
 SMPLW 00146
 SMPLW 00147
 SMPLW 00148
 SMPLW 00149
 SMPLW 00150
 SMPLW 00151

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C      LOOP FORWARD OVER THE LEFT WING                      SMPLW 00152
DO 290 NUBAR = NUBMIN,NUBMAX                           SMPLW 00153
MUAIC1 = MUAIC(1,NUBAR+1)                             SMPLW 00154
MUAIC2 = MUAIC(2,NUBAR+1)                             SMPLW 00155
IF (MUAIC2 .LE. 0) GO TO 285                          SMPLW 00156
IF (YBAR .GE. 0) GO TO 250                           SMPLW 00157
JCOLL = JBAR -NUBAR +MUAIC1 -1                         SMPLW 00158
GO TO 260                                              SMPLW 00159
250 JCOLL = JBAR +NUBAR -MUAIC1 +1                     SMPLW 00160
260 CONTINUE                                           SMPLW 00161
C
C      LOOP ON A ROW OF WING BOXES, COMPUTING LEFT WING CONTRIBUTIONS SMPLW 00162
DO 280 MUAI = MUAIC1,MUAIC2                           SMPLW 00163
IF (JCOLL .LE. 0) GO TO 270                           SMPLW 00164
CALL DCODER(IBOX,LBXCD, I,JCOLL, I,JCOLL, .F., ICD)  SMPLW 00165
IF (ICD .EQ. 0) GO TO 270                           SMPLW 00166
C      A CONTRIBUTING BOX HAS BEEN FOUND. GET THE AIC LOCATION  SMPLW 00167
KAIC = IAIC + MUAI                                    SMPLW 00168
C      GET THE NORMAL-WASH LOCATION                   SMPLW 00169
IDS = LCCSDW( I,JCOLL, IPNTDW, LPNTDW, 1, LPNTDW)   SMPLW 00170
C      ADD THIS CONTRIBUTION TO THE SUMS              SMPLW 00171
IF (EL .LT. 0) GO TO 263                           SMPLW 00172
EN = ENRUS(IDS)                                     SMPLW 00173
GO TO 265                                              SMPLW 00174
263 EN = ENRLS(IDS)                                 SMPLW 00175
265 CONTINUE                                           SMPLW 00176
WSUM = WSUM + W(KAIC) * EN                           SMPLW 00177
VSUM = VSUM + V(KAIC) * EN                           SMPLW 00178
PHISUM = PHISUM + C(KAIC) * EN                        SMPLW 00179
270 CONTINUE                                           SMPLW 00180
JCOLL = JCOLL + JINC
280 CONTINUE                                           SMPLW 00181
C      END OF LOOP FOR LEFT WING ROW CONTRIBUTIONS  SMPLW 00182
C
285 CONTINUE                                           SMPLW 00183
I = I - 1                                             SMPLW 00184
IF (I .LE. 0) GO TO 295                           SMPLW 00185
IAIC = IAIC + INCAIC                            SMPLW 00186
INCAIC = INCAIC + 2                                SMPLW 00187
290 CONTINUE                                           SMPLW 00188
C      END OF LOOP FORWARD ON LEFT WING ROWS, FROM 240  SMPLW 00189
C
C
295 CONTINUE                                           SMPLW 00190
DW(IBX) = DW(IBX) + BINV*(CPSI*WSUM - SPSI*VSUM) * SYM  SMPLW 00191
SW(IBX) = SW(IBX) + BINV*(CPSI*VSUM + SPSI*WSUM) * SYM  SMPLW 00192
PHI(IBX) = PHI(IBX) + PHISUM*SYM                  SMPLW 00193
IBX = IBX + 1                                       SMPLW 00194
300 CONTINUE                                           SMPLW 00195
C      END OF LOOP ON RECEIVING BOXES, FOR LEFT WING CONTRIBUTIONS SMPLW 00196
C
C      DETERMINE WHAT TO PRINT                         SMPLW 00197
310 CONTINUE                                           SMPLW 00198
WRITE (NT6,601) TITLE, XMACH, XKVL, JT, IFRST,ILAST  SMPLW 00199
IF (.NOT. PRCW) GO TO 330                           SMPLW 00200
WRITE (NT6,6010)
WRITE (NT6,6011, (DW(I),I = 1,NDXS))                SMPLW 00201
C

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330 CONTINUE          SMPLW 00209
    IF (.NOT. PRSM) GO TO 340
    WRITE (NTG,6011)          SMPLW 00210
    WRITE (NTG,6013) (SW(I),I = 1,NBXS)          SMPLW 00211
340 CONTINUE          SMPLW 00212
    IF (.NOT. PRLM) GO TO 400          SMPLW 00213
    IF (NBXS .LT. 2) GO TO 400          SMPLW 00214
    ISUB = 0          SMPLW 00215
    PHI (ISUB) = 2.*PHI (1) - PHI (2)          SMPLW 00216
    PHI (NBXS+1) = 2.*PHI (NBXS) - PHI (NBXS-1)          SMPLW 00217
    FACTOR = B1BETA/2.0 * BINV**2          SMPLW 00218
    DO 350 I = 1,NBXS          SMPLW 00219
    LW(I) = FACTOR*(PHI (I+1) - PHI (I-1))          SMPLW 00220
350 CONTINUE          SMPLW 00221
    WRITE (NTG,6012)          SMPLW 00222
    WRITE (NTG,6013) (LW(I),I=1,NBXS)          SMPLW 00223
C          SMPLW 00224
400 RETURN          SMPLW 00225
C          SMPLW 00226
800 WRITE (NTG,8000) IR          SMPLW 00227
    GO TO 400          SMPLW 00228
C          SMPLW 00229
C          SMPLW 00230
6001 FORMAT(1H1,2DX,8A10/ 51X,18HFLOWFIELD SAMPLING /40X,7H( MACH ,          SMPLW 00231
    1      F3.3,5X,12HRED. FREQ. =,FB.5, 2H )/ 41X,16HSAMPLED AT CHORD          SMPLW 00232
    2      I3, 8H, BOX I2, 8H TO BOX I2 / 1H0,2X, 4(10X,4HREAL,8X,          SMPLW 00233
    3      9HIMAGINARY) )          SMPLW 00234
6010 FORMAT(1SH0 UF-WASHES - )          SMPLW 00235
6011 FORMAT(1SH0SIDE-WASHES - )          SMPLW 00236
6012 FORMAT(22H0LONGITUDINAL-WASHES - )          SMPLW 00237
6013 FORMAT(4X,8E16.8)          SMPLW 00238
C          SMPLW 00239
6000 FORMAT(54H0*** WARNING - PROBLEMS ENCOUNTERED WHILE GETTING AICS          SMPLW 00240
    1      39H FOR FLOW-FIELD SAMPLING. ERROR CODE = ,15, 4H *** )          SMPLW 00241
    END          SMPLW 00242

```

PROGRAM SMTH

C THIS PROGRAM WILL FIT A LEAST SQUARES POLYNOMIAL SURFACE
C THROUGH THE VELOCITY POTENTIALS FOR A PLANFORM. A NEW SET OF
C VELOCITY POTENTIALS WILL BE CALCULATED FROM THE POLYNOMIALS
C AND WRITTEN ON THE IVPSC FILE.

COMMON /ARRAYS/ KBXCW,LBXCDW,LBOXC,KBXCDT,LBXCDT,KJALPH,LJALPH,
1 KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFSL,KELPHI,
2 LMODES,KPNTSD,LPNTSD,KSDW,LSDW,KPNTDW,LPNTDW,
3 KDW,LDW,KTVP,LTVP

COMMON /FILES/ NTS,NTG,IHTAPE,INFSP,NPLAIC,NSPAIC,NOUTP,
1 IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC

COMMON /IOCONT/ OPLAIC,OSPAIC,WGEOM,WGNAF,WTSI,WTBL,PRBOX,
1 PRPAIC,PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,
2 PRBL,PRDCP,PRGNAC,PRGNAC,PRSL,PRLW,PRNW,PRCM

EQUIVALENCE (PRUW,PRDW)

LOGICAL OPLAIC,OSPAIC,WGEOM,WGNAF,WTSI,WTBL,PRBOX,PRPAIC,
1 PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAC,
2 PRDCP,PRGNAC,PRUW,PRLW,PRNW,PRCM

COMMON /PROBLM/ XMACH,NMODES,NTSOP,NKVALS,SMOOTH,NDEG,CDFIT,
1 EXAIC,SUBDV,PLYWOOD

LOGICAL SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD

COMMON /KVAL/ IKVAL,XKVAL(20),XKS(20)

COMMON /GEMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,
1 B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,
2 MXCW,MXBW,MYBW,MYBBW,MXBW,MYBSW,MYBBSW,
3 IXBW,XCENTR

LOGICAL COPLAN

COMMON /GEOM2/ TLAX,TLAZ,FSIT,MXB1,MYBT,MYBBT,MXBST,MYBBS1,
1 MYBBST,IXBT,IXBST,CAPL

COMMON /TAPEIO/ NFS,NMS,LS,NMR,IC(20),NIC,ITYPE,LRS,LWS,M,N,
1 PARM(10),IRR

DIMENSION IPARM(10)

EQUIVALENCE (PARM,IPARM)

COMMON /CHECKPR/ DPPCP,R,GEOPR,MODCP,AICCP,R,NSCP,R,SMCP,R,GACP,R

LOGICAL DPPCP,R,GEOPR,MODCP,AICCP,R,NSCP,R,SMCP,R,GACP,R

EQUIVALENCE (CHECKPR,SMCP,R)

LOGICAL CHECKPR

C DELPHI(NBOXES), TVP(NCOLS1 + NCOLS2 * NSUBDV)
COMPLEX DELPHI(1000), TVP(250), AVPS(1250)
C X(NO. DELPHI + NO. TVP), Y(SAME)
COMPLEX SDELPHI
DIMENSION X(1250), Y(1250)
C A(NO. COEFF.)
COMPLEX A(66)

COMMON /INDEX/ IS(100),NO(100),JS(100),JO(100)
C XP(NO. COEFF. + 1), YP(SAME)
DIMENSION XP(1), YP(1)
C FEXLOC((MYBW+MYBT)*NSUBDV), TEXLOC(SAME)
DIMENSION FEXLOC(250), TEXLOC(250)
C IPNTRM(2,NCOLS*NSUBDV)
DIMENSION IPNTRM(2,150)
DIMENSION TITL(3)

C

FTNDX1 00066
SMOOTH 00004
SMOOTH 00005
SMOOTH 00006
SMOOTH 00007
SMOOTH 00008
SMOOTH 00009
ARRAYS 00002
ARRAYS 00003
ARRAYS 00004
ARRAYS 00005
FILES 00002
FILES 00003
IOCONT 00002
IOCONT 00003
BCSFRB 00001
IOCONT 00005
IOCONT 00006
IOCONT 00007
BCSFRB 00002
PROBLM 00002
PROBLM 00003
PROBLM 00004
KVAL 00002
GEMTY 00002
GEMTY 00003
GEMTY 00004
GEMTY 00005
GEMTY 00006
GEOM2 00002
GEOM2 00003
TAPEIO 00002
TAPEIO 00003
TAPEIO 00004
TAPEIO 00005
CHECKPR 00002
CHECKPR 00003
SMOOTH 00020
SMOOTH 00021
SMOOTH 00022
SMOOTH 00023
SMOOTH 00024
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SMOOTH 00037
SMOOTH 00038
SMOOTH 00039

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REAL K1
COMPLEX VP
LOGICAL MXREAD,MXWRIT,RANDIN,RANDOU
K1 = XKVAL(1KVAL)
MXREAD = .FALSE.
RANDIN = .FALSE.
MXWRIT = .FALSE.
RANDOU = .FALSE.
MXB = MXBW
IF(COPLAN) MXB = MXBT
C
C
C      PUT NAME OF SCRATCH FILE FOR SMOOTHED VALUES INTO PLACE
C
NEVPSC = 1AICSC
REWIND NEVPSC
C
C      GET THE PLANFORM POINTERS FROM THE MODESC FILE
C
REWIND MODESC
CALL RDINIT
ITYPE = SHMIXED
MXARRY = SHIPNTRM
CALL READMX(MODESC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,II,ITYPE,
1           LRS,IPNTRM,2,NPNTRS,PARM,IRR)
IOLAP = IPARM(3)
IF(IRR.NE.0) GO TO 6020
C
CALL RDINIT
ITYPE = SHMIXED
NFS = 1
MXARRY = SHIS PT.
CALL READMX(MODESC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,II,ITYPE,
1           LRS,IS,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6020
C
C      READ THE FEXLOC AND TEXLOC ARRAYS FROM THE GEOMETRY SCRATCH
C      FILE. THESE ARE NEEDED TO INTERPOLATE VELOCITY POTENTIALS AT
C      BOX EDGES.
C
REWIND 1GEOSC
CALL RDINIT
NMS = 2
IF(NSURF.EQ.1.OR.COPLAN) NMS=1
ITYPE = SHMIXED
MXARRY = SHFEXLOC
CALL READMX(1GEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,II,ITYPE,
1           LRS,FEXLOC,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6019
C
CALL RDINIT
ITYPE = SHMIXED
MXARRY = SHTEXLOC
CALL READMX(1GEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,II,ITYPE,
1           LRS,TEXLOC,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6010

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SMOOTH 00040
 SMOOTH 00041
 SMOOTH 00042
 SMOOTH 00043
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 SMOOTH 00096

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C
C      REORDER THE FEXLOC AND TEXLOC ARRAYS SO THAT THERE ARE      SMOOTH 00097
C      VALUES FOR UNSUBDIVIDED CHORDS ONLY.                      SMOOTH 00098
C
C      IF(NSUBDV.EQ.1) GO TO 120                                SMOOTH 00099
XSLIDE = NSUBDV - IXBW                                         SMOOTH 00100
JCOL = NSUBCN                                                 SMOOTH 00101
NCOLS = MXBW + MYBT                                           SMOOTH 00102
DO 110 I=1,NCOLS                                         SMOOTH 00103
TEXLOC(I) = (TEXLOC(JCOL) + XSLIDE)/NSUBDV                  SMOOTH 00104
FEXLOC(I) = (FEXLOC(JCOL) + XSLIDE)/NSUBDV                  SMOOTH 00105
JCOL = JCOL + NSUBDV                                         SMOOTH 00106
110 CONTINUE                                              SMOOTH 00107
120 CONTINUE                                              SMOOTH 00108
C
C
C      LOOP ON NUMBER OF MODES (ALSO NO. OF V.P.)                SMOOTH 00109
REWIND IVPSC                                               SMOOTH 00110
DO 2000 NM=1,NMODES                                         SMOOTH 00111
C
C      READ DELPHI ARRAY FROM IVPSC. THE TVP ARRAY MUST BE SKIPPED SMOOTH 00112
C      IF NM IS NOT 1                                         SMOOTH 00113
C
CALL RDINIT                                              SMOOTH 00114
C
C
ITYPE = SHMIXED                                         SMOOTH 00115
CALL READMX(IPWSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE, SMOOTH 00116
1          LRS,DELPHI,M,N,PARM,IRR)                         SMOOTH 00117
IF(IRR.NE.0) GO TO 6740                                    SMOOTH 00118
C
CALL RDINIT                                              SMOOTH 00119
ITYPE = SHMIXED                                         SMOOTH 00120
CALL READMX(IPWSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE, SMOOTH 00121
1          LRS,TVP,M,N,PARM,IRR)                         SMOOTH 00122
IF(IRR.NE.0) GO TO 6040                                    SMOOTH 00123
C
C      LOOP ON NSURF TO FIT EACH PLANFORM INDEPENDENTLY.        SMOOTH 00124
C
DO 1000 NS=1,NSURF                                         SMOOTH 00125
C
C      MOVE DELPHI FOR PLANFORM NS TO AVF ARRAY DELETING ZERO    SMOOTH 00126
C      VALUES AND OBTAINING THE (X,Y) COORDINATES IN (I,J) INDICES SMOOTH 00127
C
IF(NS.EQ.2) GO TO 100                                     SMOOTH 00128
C
FIRST PLANFORM                                              SMOOTH 00129
IBEG = ?                                                    SMOOTH 00130
ILIM = MXBW                                                 SMOOTH 00131
IC = 0                                                       SMOOTH 00132
NCH = 0                                                       SMOOTH 00133
GO TO 200
C
SECOND PLANFORM                                             SMOOTH 00134
100 CONTINUE                                              SMOOTH 00135
IFBT = (IXBT-IXBW/NSUBDV + 1                               SMOOTH 00136
IBEG = IFBT                                                 SMOOTH 00137
C

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ILIM = MYBT          SMOOTH 00154
IC = 0               SMOOTH 00155
NCH = MYBW           SMOOTH 00156
IUP = MYBW           SMOOTH 00157
IF(COPLAN) IUP = IFBT - 1   SMOOTH 00158
DO 125 I=1,IUP      SMOOTH 00159
IC = IC + JOC(I)    SMOOTH 00160
125 CONTINUE        SMOOTH 00151
200 CONTINUE        SMOOTH 00162
ICS = IC            SMOOTH 00163
INO = 0              SMOOTH 00164
DO 400 IX=IBEG,ILIM  SMOOTH 00165
I = IX              SMOOTH 00166
IF(NG.EQ.2) I = IX + IOVLAP  SMOOTH 00167
JST = JS(I)          SMOOTH 00168
JEND = JS(I) + JOC(I)- 1   SMOOTH 00169
DO 400 J=JST,JEND    SMOOTH 00170
IC = IC + 1          SMOOTH 00171
IB = IS(J+NCH)       SMOOTH 00172
IT = IB + KOC(J+NCH) -1  SMOOTH 00173
IF(IX.LT.IB) GO TO 400  SMOOTH 00174
IF(IX.GT.IT) GO TO 400  SMOOTH 00175
INO = INO + 1        SMOOTH 00176
AVPS(INO) = DELPHI(IC)  SMOOTH 00177
X(INO) = I            SMOOTH 00178
Y(INO) = J            SMOOTH 00179
400 CONTINUE        SMOOTH 00180
C
C      ADD THE LEADING EDGE VELOCITY POTENTIAL TO THE AVP ARRAY
C      VEL. POT. = 0. UNLESS IT IS FOR SECOND PLANEFORM IN COPLANAR
C      ANALYSIS
C
JLAST = MYBW          SMOOTH 00181
IF(NS.EQ.2) JLAST = MYBT  SMOOTH 00182
DO 600 J=1,JLAST      SMOOTH 00183
IB = IS(J+NCH)         SMOOTH 00184
INO= INO +1            SMOOTH 00185
X(INO) = FEXLOC(J)     SMOOTH 00186
Y(INO) = J              SMOOTH 00187
AVPS(INO) = (0.,0.)    SMOOTH 00188
IF(NR.EQ.1) GO TO 600  SMOOTH 00189
IF(.NOT.COPLAN) GO TO 600  SMOOTH 00190
XDKVL = (FEXLOC(J+MYBW)-TEXLOC(J)) * K1
JT = J
IF(NSUBDV.NE.1) JT = NSUBDV * (J-1) + NSUBCN
AVPS(INO) = TVP(JT) * CMPLX(COS(XDKVL),-SIN(XDKVL))
600 CONTINUE        SMOOTH 00191
C
C      CALL LEAST SQUARES SURFACE FITTING ROUTINE
C
IDIM = 2              SMOOTH 00192
CN = 1.0               SMOOTH 00193
IDEGL = NDEG            SMOOTH 00194
IF(NDEG.NE.0) GO TO 675  SMOOTH 00195
DO 650 I=1,10          SMOOTH 00196
IDEGL = 10 -I+1         SMOOTH 00197
IM = IDEGL + 1          SMOOTH 00198

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XME = XM/2.0                      SMOOTH 00211
NC = XM + XME + XME + 1.0E-04    SMOOTH 00212
NC = (3*NC) / 2                   SMOOTH 00213
IF(NC.LE.INO) GO TO 675           SMOOTH 00214
650 CONTINUE
675 CONTINUE
CALL FITTER(IDEF,IND,X,Y,AVPS,A,CN,IDIM)
C
C      EVALUATE THE POLYNOMIAL EQUATION FOR DELPHI
C
MDEG = IDEG + 1                  SMOOTH 00215
IC = ICS                         SMOOTH 00216
DO 900 IX=IBEG,ILIM              SMOOTH 00217
I = IX                           SMOOTH 00218
IF(NS.EQ.2) I = IX + IOVLAP     SMOOTH 00219
XP(I) = 1.                         SMOOTH 00220
DO 700 IP=2,MDEG                SMOOTH 00221
700 XP(IP) = EXP(IP-1) * FLOAT(I)
JI = JS(I)                        SMOOTH 00222
JT = JOC(I) + JI - 1             SMOOTH 00223
DO 900 J=JI,JT                   SMOOTH 00224
IC = IC + 1                       SMOOTH 00225
IB = IS(J+NCH)                   SMOOTH 00226
IT = IB + NOC(J+NCH) - 1         SMOOTH 00227
IF(IX.LT.IB) GO TO 900           SMOOTH 00228
IF(IX.GT.IT) GO TO 900           SMOOTH 00229
YP(1) = 1.0                        SMOOTH 00230
DO 800 JP=2,MDEG                SMOOTH 00231
800 YP(JP) = YP(JP-1) * FLOAT(J)
VP = A(I)                         SMOOTH 00232
IA = 1                            SMOOTH 00233
DO 850 L2=2,MDEG                SMOOTH 00234
DO 850 L3=1,L2                   SMOOTH 00235
L4 = L2 - L3 + 1                 SMOOTH 00236
IA = IA + 1                       SMOOTH 00237
VP = VP + XP(L4)*YP(L3)*A(IA)   SMOOTH 00238
850 CONTINUE
DELPHI(IC) = VP                  SMOOTH 00239
900 CONTINUE
C
C      CALCULATE THE TRAILING EDGE VELOCITY POTENTIALS (TVF ARRAY)
C
IF(NS.EQ.2) GO TO 910
NTST = 1
NTVPS = MYBSW
JJ = 0
GO TO 920
910 CONTINUE
NTST = NTVPS + 1
NTVPS = MYBSW + MYBST
920 CONTINUE
DO 930 J=NTST,NTVPS
930 TVP(J) = (0.,0.)
C
NBEG = NTST + NSUBD2
NEND = NTVPS - NSUBD2
JC = 0

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DO 960 J=NSEG,NEND,NSUBDV
C      J = LOCATION IN THE TVP ARRAY (SUBDIVIDED VALUES)
C      JJ = LOCATION IN ARRAYS TEXLOC, NOC, IS, ETC.
C      JC = UNSUBDIVIDED CHORD NUMBER
C      IROW = ROW NUMBER OF LAST BOX ON THE CHORD
C      I      = ROW LOCATION OF IROW IN ARRAY IPNTRM
      JJ = JJ + 1
      JC = JC + 1
      I = TEXLOC(JJ)
      IROW = I
      XINCR = TEXLOC(JJ) - I
      IF (.NOT. COPLAN .AND. J .GT. MYBW) I = I + IOWAP
      INDB = IPNTRM(1,I) + JC - IPNTRM(2,I)

C      TEST FOR 3 BOXES ON CHORD JJ
      IF(NOC(JJ).LT.3) GO TO 940

C      2 BOXES AND NO MACH RAY AVAILABLE, OR
C      3 BOXES OR MORE. DO LINEAR EXTRAPOLATION.
935 CONTINUE
      INDE = IPNTRM(1,I-1) + JC - IPNTRM(2,I-1)
      SDELPH = DELPHI(INDB) - DELPHI(INDE)
      GO TO 950

C      TEST FOR MACH RAY EXTRAPOLATION.
940 CONTINUE
      IB = IS(JJ-1)
      IX = IB + NOC(JJ-1) + 1
      IF (IROW .LT. IB .OR. IROW .GT. IX) GO TO 945
      IB = IS(JJ-2)
      IX = IB + NOC(JJ-2) + 1
      IMI = IROW-1
      IF (IMI .GE. IB .AND. IMI .LE. IX) GO TO 948

C      MACH RAY CANNOT BE USED. TEST FOR 2 BOXES ON CHORD JJ
945 CONTINUE
      IF (NOC(JJ) .LT. 2) GO TO 7010
      GO TO 935

C      MACH RAY CAN BE USED
948 CONTINUE
      INDA = IPNTRM(1,I-1) + JC - IPNTRM(2,I-1) - 2
      INDC = IPNTRM(1,I) + JC - IPNTRM(2,I) - 1
      SDELPH = 2.0 * DELPHI(INDC) - DELPHI(INDA) - DELPHI(INDB)
950 CONTINUE
      JT = JJ
      IF(NSUBDV.NE.1) JT = NSUBCN + NSUBDV * (JJ-1)
      TVP(JT) = DELPHI(INDB) + XINCR + SDELPH
960 CONTINUE

C      5000 CONTINUE

C      WRITE THE DELPHI AND TVP ARRAY ON THE NIVPSC FILE
      CALL RDINIT
      ITYPE = SHMIXED
      N = IPNTRM(1,NPNTR) - 1
      MXARRY = SHDELPHI

```

SMOOTH 00268
BCSSMA 00001
BCSSMA 00002
BCSSMA 00003
BCSSMA 00004
BCSSMA 00005
SMOOTH 00269
SMOOTH 00270
SMOOTH 00271
BCSSMA 00005
SMOOTH 00273
SMOOTH 00274
BCSSMA 00007
SMOOTH 00275
SMOOTH 00276
SMOOTH 00277
SMOOTH 00278
BCSSMA 00008
SMOOTH 00279
BCSSMA 00009
SMOOTH 00280
SMOOTH 00281
SMOOTH 00282
SMOOTH 00283
SMOOTH 00284
SMOOTH 00285
SMOOTH 00286
SMOOTH 00287
BCSSMA 00010
SMOOTH 00289
SMOOTH 00290
BCSSMA 00011
BCSSMA 00012
BCSSMA 00013
BCSSMA 00014
BCSSMA 00015
BCSSMA 00016
BCSSMA 00017
SMOOTH 00293
SMOOTH 00294
BCSSMA 00018
SMOOTH 00295
SMOOTH 00296
SMOOTH 00297
SMOOTH 00298
SMOOTH 00299
SMOOTH 00300
SMOOTH 00301
SMOOTH 00302
SMOOTH 00303
SMOOTH 00304
SMOOTH 00305
SMOOTH 00306
SMOOTH 00307
SMOOTH 00308
SMOOTH 00309
SMOOTH 00310

```

CALL WRTEMX(NIVPSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1           DELPHI,I TYPE,2,N,PARM,IRR)                      SMOOTH 00311
IF(IRR.NE.0) GO TO 6040                                     SMOOTH 00312
C
N = NTVPS
MXARRY = GHTVP
CALL WRTEMX(NIVPSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,ID,
1           TVP, I TYPE,2,N,PARM,IRR)                      SMOOTH 00313
IF(IRR.NE.0) GO TO 6040                                     SMOOTH 00314
C
C
C      PRINT THE SMOOTHED VEL. POT. ARRAY
IF(.NOT.PRVP) GO TO 1500                                     SMOOTH 00315
TITL(1) = 8K   WING
TITL(2) = 10HVELOCITY P
TITL(3) = 10HOTENTIALS
IF(COPLAN) TITL(1) = 10HMIN/TAIL
CALL PRINTR(TITL, NM, DELPHI, 1, 1, MXB, MYBW, IPNTRM)
IF(NSURF.EQ.1.OR.COPLAN) GO TO 1500
TITL(1) = 8H   TAIL
CALL PRINTR(TITL, NM, DELPHI, 1, IFBT, MXST, MYBT,
1           IPNTRM(1,IOMLAP+1) )
1500 CONTINUE
IF (.NOT. CHECKPR) GO TO 2000
WRITE (NT6,9400) (TVP(I),I=1,NTVPS)
9400 FORMAT(U/ (IX,8E16.8) )
C
2000 CONTINUE
C
END FILE NIVPSC
REWIND NIVPSC
C
CHANGE FILE NAMES
C
IAICSC = IVPSC
IVPSC = NIVPSC
C
RETURN
6010 CONTINUE
WRITE (NT6,9010) 1GEOSC,IRR
WRITE (NT6,9011) MXARRY
GO TO 6100
6020 CONTINUE
WRITE (NT6,9010) MODESC,IRR
WRITE (NT6,9011) MXARRY
GO TO 6100
6040 CONTINUE
WRITE (NT6,9010) IVPSC,IRR
WRITE (NT6,9041) NM
6100 CONTINUE
WRITE (NT6,9101) ID(1),ID(2)
WRITE (NT6,9102) PARM,IPARM
WRITE (NT6,9103) NFS,NMS
WRITE (NT6,9104) I TYPE,M,N
WRITE (NT6,9900)
GO TO 8000
C
7010 CONTINUE

```

```

      WRITE (NT6,9020)                                     SMOOTH 00368
      WRITE (NT6,9021) IROW,JC                           BCSSMA 00019
C
      8000 CONTINUE                                         SMOOTH 00370
      CALL FLUSH(1)                                       SMOOTH 00371
C
      9020 FORMAT(7HG*** ERROR - NO TIP TRAILING EDGE VELOCITY POTENTIAL CAN SMOOTH 00374
      1 BE COMPUTED. *** )                               SMOOTH 00375
      9021 FORMAT(5X,13HCOORDINATES (I2,IH,I2,IH) )       SMOOTH 00376
      9010 FORMAT(5SH*** ERROR - WHILE READING THE GEOMETRY SCRATCH FILE A10 SMOOTH 00377
      1, 1SH, ERROR CODE = I4,4H *** )                  SMOOTH 00378
      9011 FORMAT(5X,32HAN ATTEMPT WAS MADE TO READ THE A6, 8H MATRIX.//) SMOOTH 00379
      9041 FORMAT(5X,54HAN ATTEMPT WAS MADE TO READ THE VEL. POT. ARRAY NUMBER SMOOTH 00380
      1R I3,1H. )                                       SMOOTH 00381
      9050 FORMAT(5H0*** ERROR - WHILE WRITING ON THE VEL. POT. SCRATCH FILE SMOOTH 00382
      1          A10, 1SH, ERROR CODE = I4,4H *** )        SMOOTH 00383
      9051 FORMAT(5X,36HATTEMPTING TO WRITE VEL. POT. NUMBER I3 ) SMOOTH 00384
      9101 FORMAT(5X,*PARAMETERS *,10E11.3, / 10X,*(INTEGER)*, I7, 9I11 ) SMOOTH 00385
      9102 FORMAT(5X,*FILE SPACING = *,I3,* MATRIX SPACING = *,I3 ) SMOOTH 00386
      9103 FORMAT(5X,*MATRIX TYPE =*,A10,* , DIMENSIONED (*I4,2H X,I4,1H) ) SMOOTH 00387
      9900 FORMAT(40      ERROR OCCURRED IN SMOOTHING SECTION. *) FTNX1 00068
      END                                                 SMOOTH 00390

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SUBROUTINE PRINTR( TITL,IMODE,ARRAY,K,IXB,MXB,MYB,IPNTRM)          PRNTVP 00002
C
C      TITL - TITLE TO PRINT FOR THE ARRAY                         PRNTVP 00003
C      IMODE - MODE SHAPE NUMBER                                PRNTVP 00004
C      ARRAY - ARRAY TO BE PRINTED                               PRNTVP 00005
C
C      DIMENSION ARRAY(K,1), TITL(3)                            PRNTVP 00006
C      COMPLEX ARRAY                                         PRNTVP 00007
C      DIMENSION IPNTRM(2,100)                                 PRNTVP 00008
C      COMMON /CTRL/  PREVEX,XMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT,   CTRL  00002
C      I                   DEFAULT                                CTRL  00003
C      LOGICAL      PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT           CTRL  00004
C      COMMON /PROBLM/ XMACH,NMODOES,NTSLOP,NKVALS,SMOOTH,NDEG,CRCFIT,   PROBLM 00002
C      I                   EXAIC,SUBDV,PLYWOOD                  PROBLM 00003
C      LOGICAL      SMOOTH,CRCFIT,EXAIC,SUBDV,PLYWOOD           PROBLM 00004
C      COMMON /FILES/  NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP,   FILES  00002
C      I                   ICUSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC    FILES  00003
C      COMMON /KVAL / IKVAL,XXVAL(20), XKS(20)                 KVAL   00002
C      DIMENSION PC(2)                                         PRNTVP 00016
C      DIMENSION S(50),D(50)                                    PRNTVP 00020
C      EQUIVALENCE (S(1),BUFF(1)),(D''),BUFF(1251))            PRNTVP 00021
C      REAL K1                                              PRNTVP 00022
C      INTEGER PAGE                                         PRNTVP 00023
C      COMMON /RWBUFF/ BFCCODE,IBFCNT, BUFF(3280)              RWBUFF 00002
C      DATA PC / 10HPAGE CONTI,4HNUED /
C      DATA BLANK / 1H /
C      DATA XINIT / -1.0 /
C      K1 = XXVAL(IKVAL)                                     PRNTVP 00024
C      IF (XKS(IKVAL) .NE. XINIT) K1 = XKS(IKVAL)             PRNTVP 00025
C
C      PAGE = 0                                              PRNTVP 00026
C      N =1                                                 PRNTVP 00027
C      M =4                                                 PRNTVP 00028
C      IF(M.GT.MYB) M = MYB                                PRNTVP 00029
C
100 LINE = 100                                         PRNTVP 00030
200 DO 1400 I = IXB,MXB
DO 300 J=N,M
S(J) = 0.0
D(J) = 0.0
300 CONTINUE
IF(LINE.LE.50) GO TO 900
PAGE = PAGE + 1
LINE = 4
WRITE (NT6,9001) TITLE,TITL, XMACH, K1, IMODE
C
IF(PAGE.EQ.1) GO TO 700
WRITE (NT6,9005) PC
GO TO 800
700 WRITE(NT6,9005)
800 CONTINUE
WRTE(NT6,6006) (BLANK,J,J=N,M)
WRTE(NT6,6007) (BLANK, J=N,M)
900 CONTINUE
JS = IPNTRM(2,I)
IDX = IPNTRM(1,I)
JE = IPNTRM(1,I+1) - IDX + JS -1

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IF(JE.EQ.0) GO TO 1400          PRNTVP 00054
DO 1000 J=JS,JE                 PRNTVP 00055
S(J) = REAL(ARRAY(1,IDX) )      PRNTVP 00056
D(J) = AIMAG(ARRAY(1,IDX) )    PRNTVP 00057
IDX = IDX +1                   PRNTVP 00058
1000 CONTINUE                   PRNTVP 00059
DO 1200 J=N,M                   PRNTVP 00060
IF(S(J)) 1300,1100,1300        PRNTVP 00061
1100 CONTINUE                   PRNTVP 00062
IF(D(J)) 1300,1200,1300        PRNTVP 00063
1200 CONTINUE                   PRNTVP 00064
GO TO 1400                     PRNTVP 00065
1300 WRITE (NT6,9013) I,(S(J),D(J),J=N,M)   PRNTVP 00066
LINE = LINE + 1                 PRNTVP 00067
1400 CONTINUE                   PRNTVP 00068
M = M+4                         PRNTVP 00069
N = N+4                         PRNTVP 00070
IF(N.GT.NYB) GO TO 1500         PRNTVP 00071
IF(M.GT.NYB) M=NYB              PRNTVP 00072
IF(LINE.GT.45) GO TO 100        PRNTVP 00073
WRITE(NT6,6006) (BLANK,J,J=N,M)  PRNTVP 00074
WRITE(NT6,6007) (BLANK, J=N,M)   PRNTVP 00075
LINE = LINE+3                   PRNTVP 00076
GO TO 200                       PRNTVP 00077
1500 CONTINUE                   PRNTVP 00078
RETURN                         PRNTVP 00079
9001 FORMAT(1H1,20X,8A10/ 46X,*SMOOTHED *,3A10/ 46X,7H( MACH F5.3,5X,
     1 12HRED. FREQ. =,FB.5, * )* / 52X,*MODE SHAPE*, 13)  PRNTVP 00080
9005 FORMAT(44X,42(1H-),20X,A10,A4)  PRNTVP 00081
9006 FORMAT(4HORCH, A1,14X,SHCHRD,I3, 3(A1,22X,SHCHRD,I3) )  PRNTVP 00082
9007 FORMAT(3X, 4(A1,9X,4HREAL,8X,9HIMAGINARY) )  PRNTVP 00083
9013 FORMAT(I4,8E16.8)           PRNTVP 00084
END                            PRNTVP 00085
                                PRNTVP 00086

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SUBROUTINE FITTER(M,N,X,Y,Z,C,CN,CDIM) FITTER 00002
DIMENSION X(100), Y(100), Z(100), C(CDIM,66) FITTER 00003
DIMENSION AI(68),A(66,68),XP(11),YP(11) FITTER 00004
DIMENSION VS(10) FITTER 00005
LOGICAL COMPLX FITTER 00006
C FITTER 00007
C M - DEGREE OF POLYNOMIAL EQUATION FITTER 00008
C N - NUMBER OF DATA POINTS TO FIT CURVE THROUGH FITTER 00009
C X - X COORDINATE OF DATA POINT FITTER 00010
C Y - Y COORDINATE OF DATA POINT FITTER 00011
C Z - Z COORDINATE OF DATA POINT FITTER 00012
C C - OUTPUT COEFFICIENT ARRAY FITTER 00013
C CN - SCALE FACTOR FITTER 00014
C CN - SCALE FACTOR FITTER 00015
C CDIM - INDICATOR OF REAL OR COMPLEX FUNCTION FITTER 00016
C = 1, FUNCTION IS REAL FITTER 00017
C = 2, FUNCTION IS COMPLEX FITTER 00018
C IF COMPLEX SET DIMENSIONS OF FUNCTION AND COEFFICIENTS FITTER 00019
C TO (CDIM * --- )
C FITTER 00020
C DETERMINE NUMBER OF COEFFICIENTS FITTER 00021
C FITTER 00022
C FITTER 00023
C EPS = 1.0E-04 FITTER 00024
C COMPLX = .FALSE.
C IF(CDIM.EQ.2) COMPLX = .TRUE.
C FITTER 00025
C SCALE DATA TO REDUCE MAGNITUDE OF MATRIX TERMS.
C SHOULD AVOID BOMB OUTS DUE TO OVERFLOW CONDITIONS.
C IF(CN.EQ.0) CN=1.0 FITTER 00026
C IF(CN.EQ.1.0) GO TO 15 FITTER 00027
DO 5 I=1,N FITTER 00028
X(I) = X(I)/CN FITTER 00029
Y(I) = Y(I)/CN FITTER 00030
5 CONTINUE FITTER 00031
15 CONTINUE FITTER 00032
XM = M + 1 FITTER 00033
XMR= XM/2.
NC = XM*XMR + XMR + EPS FITTER 00034
IF(NC.LE.N) GO TO 25 FITTER 00035
M = M-1 FITTER 00036
GO TO 15 FITTER 00037
25 CONTINUE FITTER 00038
C NC = NC FITTER 00039
C DETERMINE THE MAXIMUM DEGREE THAT CAN BE COMPUTE 'N
C EACH DIRECTION AND SET UP ORDER OF SOLUTION.
C FITTER 00040
C NCV = 1 FITTER 00041
MDX = M FITTER 00042
VS(1) = X(1) FITTER 00043
DO 60 I=1,M FITTER 00044
DO 70 J=1,NCV FITTER 00045
IF(X(I).EQ.VS(J)) GO TO 35 FITTER 00046
90 CONTINUE FITTER 00047
NCV = NCV + 1 FITTER 00048
VS(NCV) = X(I) FITTER 00049
FITTER 00050
FITTER 00051
FITTER 00052
FITTER 00053
FITTER 00054
FITTER 00055
FITTER 00056
FITTER 00057
FITTER 00058

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      IF(NDV-1.EQ.M) GO TO 65          FITTER 00059
55 CONTINUE                         FITTER 00060
60 CONTINUE                         FITTER 00061
      MDX = NDV -1                     FITTER 00062
63 CONTINUE                         FITTER 00063
C                                     FITTER 00064
      NDY = 1                          FITTER 00065
      NDY = M                          FITTER 00066
      VS(1) = Y(1)                     FITTER 00067
      DO 80 I=1,N                      FITTER 00068
      DO 70 J=1,NDV                   FITTER 00069
      IF(Y(I).EQ.VS(J)) GO TO 75       FITTER 00070
70 CONTINUE                         FITTER 00071
      NDV = NDV + 1                   FITTER 00072
      VS(NDV) = Y(I)
      IF(NDV-1.EQ.M) GO TO 85         FITTER 00073
75 CONTINUE                         FITTER 00074
80 CONTINUE                         FITTER 00075
      MDY = NDV - 1                   FITTER 00076
85 CONTINUE                         FITTER 00077
C                                     FITTER 00078
      ITOT = NC +1                   FITTER 00079
      ITOT1 = ITOT                    FITTER 00080
      IF(COMPLX) ITOT = ITOT + 1      FITTER 00081
C                                     FITTER 00082
C                                     ZERO OUT THE A ARRAY
C                                     FITTER 00083
      DO 95 I=1,NC                  FITTER 00084
      ' ,I) = 0.0                     FITTER 00085
      IF(.NOT.COMPLX) GO TO 90        FITTER 00086
      C(2,I) = 0.0                     FITTER 00087
90 CONTINUE                         FITTER 00088
      DO 95 J=1,ITOT                 FITTER 00089
      95 A(I,J) = 0.0                 FITTER 00090
C                                     FITTER 00091
C                                     DETERMINE DEVIATION EQUATION AND SQUARE THE EQUATION
C                                     FITTER 00092
      AI(1) =1.0                      FITTER 00093
      XP(1) =1.0                      FITTER 00094
      YP(1) =1.0                      FITTER 00095
      MM = M + 1                      FITTER 00096
      DO 200 K=1,N                    FITTER 00097
      DO 10 L=2,MM                    FITTER 00098
      X (L) = XP(L-1)*X(K)           FITTER 00099
      YP(L) = YP(L-1)*Y(K)
10 CONTINUE                         FITTER 00100
C                                     FITTER 00101
      I = 1                           FITTER 00102
      DO 40 L=2,MM                    FITTER 00103
      DO 20 LL=1,L                    FITTER 00104
      IL= L - LL +1                  FITTER 00105
      IF(LL-1.GT.NDY) GO TO 30       FITTER 00106
      IF(IL-1.GT.MDX) GO TO 20       FITTER 00107
      I = I + 1                      FITTER 00108
      AI(I) = XP(IL)*YP(LL)
20 CONTINUE                         FITTER 00109
30 CONTINUE                         FITTER 00110

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40 CONTINUE          FITTER 00116
    AI(I+1) = Z(1,K)          FITTER 00117
    IF(COMPLX) AI(I+2) = Z(2,K)          FITTER 00118
    IF(K.GT.1) GO TO 45          FITTER 00119
    NC = I          FITTER 00120
    ITOT = NC + 1          FITTER 00121
    ITOT1 = ITOT          FITTER 00122
    IF(COMPLX) ITOT = ITOT + 1          FITTER 00123
45 CONTINUE          FITTER 00124
C
    DO 1100 I=1,NC          FITTER 00125
    DO 1100 J=I,ITOT          FITTER 00126
    ASAV = AI(I)*AI(J)          FITTER 00127
    A(I,J)=A(I,J)+ASAV          FITTER 00128
1100 CONTINUE          FITTER 00129
200 CONTINUE          FITTER 00130
C
C      SQUARE ROOT METHOD          FITTER 00131
C      INTERMEDIATE MATRIX          FITTER 00132
    DO 1200 I=1,NC          FITTER 00133
    IM1 = I-1          FITTER 00134
    TMP=0.0          FITTER 00135
    IF(I.EQ.1) GO TO 1150          FITTER 00136
    DO 1120 L=1,IM1          FITTER 00137
1120 TMP= TMP+ A(L,I)*A(L,I)          FITTER 00138
1150 CONTINUE          FITTER 00139
    T = A(I,I) - TMP          FITTER 00140
    IF(T.GT.EPS) GO TO 4          FITTER 00141
    A(I,I) = 0.0          FITTER 00142
    GO TO 1200          FITTER 00143
4 CONTINUE          FITTER 00144
    A(I,I) = SQRT(T)          FITTER 00145
    IF(A(I,I).GT.EPS) GO TO 1155          FITTER 00146
    A(I,ITOT) = 0.0          FITTER 00147
    GO TO 1200          FITTER 00148
1155 CONTINUE          FITTER 00149
C
    JS = I+1          FITTER 00150
    DO 1180 J = JS,ITOT          FITTER 00151
    TMP= 0.0          FITTER 00152
    IF(I.EQ.1) GO TO 1175          FITTER 00153
    DO 1160 L=1,IM1          FITTER 00154
1160 TMP = TMP + A(L,I)*A(L,J)          FITTER 00155
1175 A(I,J) = (A(I,J)-TMP)/A(I,I)          FITTER 00156
1180 CONTINUE          FITTER 00157
1200 CONTINUE          FITTER 00158
C
C      BACK SUBSTITUTE FOR COEFFICIENTS          FITTER 00159
    DO 1400 K=1,NC          FITTER 00160
    I = NC - K + 1          FITTER 00161
    IP1=I+1          FITTER 00162
    TMP1 = 0.0          FITTER 00163
    TMP2 = 0.0          FITTER 00164
    IF(A(I,I).GT.EPS) GO TO 1325          FITTER 00165
    C(I,I) = 0.0          FITTER 00166
    IF(COMPLX) C(2,I) = 0.0          FITTER 00167

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GO TO 1400
1325 CONTINUE
IF(I.EQ.NC) GO TO 1375
DO 1350 L=IP3,NC
TMP1 = TMP1 + A(I,L)*C(1,L)
IF(.NOT.COMPLX) GO TO 1350
TMP2 = TMP2 + A(I,L)*C(2,L)
1350 CONTINUE
1375 CONTINUE
C(1,I) =(A(I,ITOT)-TMP1)/A(I,I)
IF(.NOT.COMPLX) GO TO 1400
C(2,I) =(A(I,ITOT) -TMP2)/A(I,I)
1400 CONTINUE
C
C
C      REORDER THE COEFFICIENTS IN CORRECT POWERS
C      OF X AND Y.
C
IF(NAC.EQ.NC) GO TO 1475
C
IZ = 1
I = 1
DO 1440 L=2,MM
DO 1420 LL=1,L
IL = L -LL +1
I = I +1
IF(LL-1.LE.MDY.AND.IL-1.LE.MDX) GO TO 1410
X(I) = 0.0
Y(I) = 0.0
GO TO 1420
1410 CONTINUE
IZ = IZ + 1
X(I) = C(1,IZ)
IF(COMPLX) Y(I) = C(2,IZ)
1420 CONTINUE
1440 CONTINUE
C
DO 1450 I=2,NAC
C(1,I) = X(I)
IF(COMPLX) C(2,I) = Y(I)
1450 CONTINUE
1475 CONTINUE
C
C      ELIMINATE THE SCALE FACTOR FROM THE COEFFICIENTS.
C
IF(CN.EQ.1.0) GO TO 1700
I=1
CP= 1.0/CN
DO 1600 L1=2,MM
DO 1500 L2=1,L1
I = I+1
C(1,I) = C(1,I)*CP
C(2,I) = C(2,I)*CP
1500 CONTINUE
CP= CP/CN
1600 CONTINUE
1700 CONTINUE

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FITTER 00173
FITTER 00174
FITTER 00175
FITTER 00176
FITTER 00177
FITTER 00178
FITTER 00179
FITTER 00180
FITTER 00181
FITTER 00182
FITTER 00183
FITTER 00184
FITTER 00185
FITTER 00186
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FITTER 00210
FITTER 00211
FITTER 00212
FITTER 00213
FITTER 00214
FITTER 00215
FITTER 00216
FITTER 00217
FITTER 00218
FITTER 00219
FITTER 00220
FITTER 00221
FITTER 00222
FITTER 00223
FITTER 00224
FITTER 00225
FITTER 00226
FITTER 00227
FITTER 00228
FITTER 00229

C
C THE C ARRAY NOW CONTAINS THE COEFFICIENTS.
C
RETURN
END

FITTER 00230
FITTER 00231
FITTER 00232
FITTER 00233
FITTER 00234

OVERLAY(4,FBBOX,1,7)
PROGRAM CHORDF

C
C THIS PROGRAM WILL FIT A CURVE THROUGH THE VELOCITY POTENTIALS
C ALONG EACH CHORD INDEPENDENTLY. THESE CURVES WILL BE USED TO
C EVALUATE A SMOOTHED VELOCITY POTENTIAL.
C
COMMON /ARRAYS/ KBXCDW,LBXCDW,LBXXC,KBXCDT,LBXCDT,KJALPH,LJALPH,
1 KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFLSL,KELPHI,
2 LMODES,KPNTSD,LPNTRM,KSCW,LSCW,KPNTCW,LPNTRM,
3 KDW,LDW,KTVP,LTVP
COMMON /FILES/ NT5,NT6,INTAPE,INFSP,NPLAIC,NPLAIC,NOUTP,
1 IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC
COMMON /ICONT/ OPLAIC,OSFAIC,WTGEOM,WTGNAF,WTSI,WTBL,PRBOX,
1 PRPAIC,PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,
2 PRBL,PRDCP,PRGNAF,PRGNAC,PRSL,PRLW,PRNW,PRCM
EQUIVALENCE (PRUW,PRDW)
LOGICAL OPLAIC,OSFAIC,WTGEOM,WTGNAF,WTSI,WTBL,PRBOX,PRPAIC,
1 PRSAIC,PRMODS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAF,
2 PRDCP,PRGNAC,PRUW,PRLW,PRNW,PRCM
COMMON /PROBLM/ XMAXH,NMODES,NTSLCF,NKVALS,SMOOTH,NDEG,CDFIT,
1 EXAIC,SUBDV,PLYWOOD
LOGICAL SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD
COMMON /KVAL/ IKVAL,XKVAL(20),XKS(20)
COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,
1 B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSIW,
2 MXBW,MXBW,MYBW,MYBBW,MXBSW,MYBSW,MYBBSW,
3 IXBW,XCENTR
LOGICAL COPLAN
COMMON /GEOR2/ TLAX,TLAZ,PSIT,MXB1,MYBT,MYBBT,MXBST,MYBST,
1 MYBBST,IXBT,IXBT,CAPL
COMMON /TAPEIO/ NFS,NMS,LS,NMR,ID(20),NIC,IYPE,LRS,LWS,M,N,
1 PARM(10),IRR
DIMENSION IPARM(10)
EQUIVALENCE (PARM,IPARM)
COMMON /CHECKPR/ DPPCP,R,GEOPR,MDCPR,AICCP,NSCP,R,SMCP,R,GFCP,R
LOGICAL DPPCP,R,GEOPR,MDCPR,AICCP,NSCP,R,SMCP,R,GFCP,R
EQUIVALENCE (CHECKPR,SMCP,R)
LOGICAL CHECKPR
C
C DELPHI(NBOLES), TVP(NCOLS1 + NCOLS2 * NSUBDV)
COMPLEX DELPHI(1000), TVP(250), AVPS(52)
C X(NO. DELPHI + NO. TVP), Y(SAME)
DIMENSION X(52), Y(52)
C A(NO. COEFF.)
COMPLEX A(21)
C
COMMON /INDEX/ IS(100),NOC(100),JS(100),JOC(100)
C PEXLOC((MYBW+MYBT)*NSUBDV), TEXLOC(SAME)
DIMENSION PEXLOC(250), TEXLOC(250)
C IPNTRM(2,NROWS+NSUBDV)
DIMENSION IPNTRM(2,150)
DIMENSION TITL(3)
C
REAL K1
COMPLEX VP, SDELPH, VC, AVPA, AVPB
LOGICAL MXREAD,MXRWIT,RANDIN,RANDOU

CHORDF 00002
CHORDF 00003
CHORDF 00004
CHORDF 00005
CHORDF 00006
CHORDF 00007
CHORDF 00008
ARRAYS 00002
ARRAYS 00003
ARRAYS 00004
ARRAYS 00005
FILES 00002
FILES 00003
ICONT 00002
ICONT 00003
BCSFRB 00001
ICONT 00005
ICONT 00006
ICONT 00007
BCSFRB 00002
PROBLM 00002
PROBLM 00003
PROBLM 00004
KVAL 00002
GEOMTY 00002
GEOMTY 00003
GEOMTY 00004
GEOMTY 00005
GEOMTY 00006
GEOR2 00002
GEOR2 00003
TAPEIO 00002
TAPEIO 00003
TAPEIO 00004
TAPEIO 00005
CHECKPR 00002
CHECKPR 00003
CHORDF 00019
FTNXL 00073
CHORDF 00020
CHORDF 00021
CHORDF 00022
CHORDF 00023
CHORDF 00024
CHORDF 00025
CHORDF 00026
CHORDF 00027
CHORDF 00028
CHORDF 00029
CHORDF 00030
CHORDF 00031
CHORDF 00032
CHORDF 00033
CHORDF 00034
CHORDF 00035
CHORDF 00036
CHORDF 00037

```

K1 = XKVAL(IKVAL)
MXREAD = .FALSE.
RANDIN = .FALSE.
NWRIT = .FALSE.
RANDOU = .FALSE.
MWB = MWBW
IF(COPLAN) MWB = MXBT
C
C
C      PUT NAME OF SCRATCH FILE FOR SMOOTHED VALUES INTO PLACE
C
NIVPSC = IAICSC
REWIND NIVPSC
C
C      GET THE PLANFORM POINTERS FROM THE MODESC FILE
C
REWIND MODESC
CALL RDINIT
ITYPE = SHMIXED
MXARRY = SHIPNTRM
CALL READMX(MODESC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,1D,ITYPE,
1           LRS,IPNTRM,2,NPNTRS,PARM,IRR)
IOLAP = IPARM(3)
IF(IRR.NE.0) GO TO 6020
C
CALL RDINIT
ITYPE = SHMIXED
NFS = 1
MXARRY = GHIS PT.
CALL READMX(MODESC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,1D,ITYPE,
1           LRS,IS,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6020
C
C
C      READ THE FEXLOC AND TEXLOC ARRAYS FROM THE GEOMETRY SCRATCH
C      FILE. THESE ARE NEEDED TO INTERPOLATE VELOCITY POTENTIALS AT
C      BOX EDGES.
C
REWIND IGEOSC
CALL RDINIT
NMS = 2
IF(NSURF.EQ.1.OR.COPLAN) NMS=1
ITYPE = SHMIXED
MXARRY = SHFEXLOC
CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,1D,ITYPE,
1           LRS,FEXLOC,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6010
C
CALL RDINIT
ITYPE = SHMIXED
MXARRY = SHTEXLOC
CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,1D,ITYPE,
1           LRS,TEXLOC,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6010
C
C      REORDER THE FEXLOC AND TEXLOC ARRAYS SO THAT THERE ARE
C      VALUES FOR UNSUBDIVIDED CHORDS ONLY.

```

CHORDF 00038
 CHORDF 00039
 CHORDF 00040
 CHORDF 00041
 CHORDF 00042
 CHORDF 00043
 CHORDF 00044
 CHORDF 00045
 CHORDF 00046
 CHORDF 00047
 CHORDF 00048
 CHORDF 00049
 CHORDF 00050
 CHORDF 00051
 CHORDF 00052
 CHORDF 00053
 CHORDF 00054
 CHORDF 00055
 CHORDF 00056
 CHORDF 00057
 CHORDF 00058
 CHORDF 00059
 CHORDF 00060
 CHORDF 00061
 CHORDF 00062
 CHORDF 00063
 CHORDF 00064
 CHORDF 00065
 CHORDF 00066
 CHORDF 00067
 CHORDF 00068
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 CHORDF 00073
 CHORDF 00074
 CHORDF 00075
 CHORDF 00076
 CHORDF 00077
 CHORDF 00078
 CHORDF 00079
 CHORDF 00080
 CHORDF 00081
 CHORDF 00082
 CHORDF 00083
 CHORDF 00084
 CHORDF 00085
 CHORDF 00086
 CHORDF 00087
 CHORDF 00088
 CHORDF 00089
 CHORDF 00090
 CHORDF 00091
 CHORDF 00092
 CHORDF 00093
 CHORDF 00094

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C
IF(NSUBDV.EQ.1) GO TO 120
XSLIDE = NSUBDV - IXBW
JCOL = NSUBCN
NCLS = MYBW + MYBT
DO 110 I=1,NCLS
TEXLOC(I) = (TEXLOC(JCOL) + XSLIDE)/NSUBDV
FEXLOC(I) = (FEXLOC(JCOL) + XSLIDE)/NSUBDV
JCOL = JCOL + NSUBDV
110 CONTINUE
120 CONTINUE
C
IFBT = (IXBT-IXBW)/NSUBDV + 1
C
C      LOOP ON NUMBER OF MODES (ALSO NO. OF V.P.)
REWIND IVPSC
DO 2000 NM=1,MMODES
C
C      READ DELPHI ARRAY FROM IVPSC. THE TVP ARRAY MUST BE SKIPPED
C      IF NM IS NOT 1
C
CALL RDINIT
ITYPE = SHMIXED
NMS = 1
IF(NM.EQ.1) NMS = 0
CALL READMX(IPSPC,MREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,IC,ITYPE,
1           LRS,DELPHI,M,N,FARM,IRR)
IF(IRR.NE.0) GO TO 6040
C
C      LOOP ON NUMBER OF CHORDS
C
NCHRDS = MYBW
IF(NSURF.EQ.2) NCHRDS = MYBW + MYBT
DO 1000 J=1,NCHRDS
NC = 1
IF(J.GT.MYBW) NC = MYBW + 1
IST = IS(J)
NK = IST + NOC(J) - 1
JSUM = 0
ITROW = IST
IF(.NOT.COPLAN.AND.J.GT.MYBW) ITROW = IST + IOVLAP
DO 100 I=1,ITROW
100 JSUM = JSUM + JOC(I)
JSUM = JSUM - JOC(ITROW) + 1
IND = 0
DO 200 I=IST,NK
IX = I
IND = IND + 1
IF(.NOT.COPLAN.AND.J.GT.MYBW) IX = I + IOVLAP
ISUB = JSUM + J - JS(IX) - NC + 1
X(IND) = I
AVPS(IND) = DELPHI(ISUB)
JSUM = JSUM + JOC(IX)
200 CONTINUE
C
C      FIND THE DERIVATIVE OF DELPHI, AND SMOOTH THESE
C

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CHORDF 00095
 CHORDF 00096
 CHORDF 00097
 CHORDF 00098
 CHORDF 00099
 CHORDF 00100
 CHORDF 00101
 CHORDF 00102
 CHORDF 00103
 CHORDF 00104
 CHORDF 00105
 CHORDF 00106
 CHORDF 00107
 CHORDF 00108
 CHORDF 00109
 CHORDF 00110
 CHORDF 00111
 CHORDF 00112
 CHORDF 00113
 CHORDF 00114
 CHORDF 00115
 CHORDF 00116
 CHORDF 00117
 CHORDF 00118
 CHORDF 00119
 CHORDF 00120
 CHORDF 00121
 CHORDF 00122
 CHORDF 00123
 CHORDF 00124
 CHORDF 00125
 CHORDF 00126
 CHORDF 00127
 CHORDF 00128
 CHORDF 00129
 CHORDF 00130
 CHORDF 00131
 CHORDF 00132
 CHORDF 00133
 CHORDF 00134
 CHORDF 00135
 CHORDF 00136
 CHORDF 00137
 CHORDF 00138
 CHORDF 00139
 CHORDF 00140
 CHORDF 00141
 CHORDF 00142
 CHORDF 00143
 CHORDF 00144
 CHORDF 00145
 CHORDF 00146
 CHORDF 00147
 CHORDF 00148
 CHORDF 00149
 CHORDF 00150
 CHORDF 00151

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VC = AVPS(1)          CHORDF 00152
INDM1 = INC -1        CHORDF 00153
AVPA = 0.5 * (AVPS(1) + AVPS(2))  CHORDF 00154
AVPS(1) = (AVPS(2) - AVPS(1))/(X(2) - X(1))  CHORDF 00155
DO 350 I=2,INDM1      CHORDF 00156
AVPB = 0.5 * (AVPS(I) + AVPS(I-1))  CHORDF 00157
AVPS(I) = AVPB - AVPA  CHORDF 00158
AVPA = AVPB  CHORDF 00159
350 CONTINUE          CHORDF 00160
AVPS(IND) = (AVPS(IND) - AVPA)/0.5  CHORDF 00161
IND = IND + 1         CHORDF 00162
INDM1 = INDM1 + 1     CHORDF 00163
AVPS(IND) = AVPS(INDM1)  CHORDF 00164
X(IND) = X(INDM1) + 0.5  CHORDF 00165
C          CHORDF 00166
XINC = X(1)          CHORDF 00167
DO 375 I=1,IND       CHORDF 00168
X(I) = X(I) - XINC  CHORDF 00169
375 CONTINUE          CHORDF 00170
IDEG = NDEG          CHORDF 00171
C          CHORDF 00172
C          CALL FITTING ROUTINE LEAST SQUARES ERROR CURVE.  CHORDF 00173
C          CALL CURVE(IDEG,IND,X,AVPS,A)  CHORDF 00174
C          EVALUATE THE CURVE FOR SMOOTH DELPHI VALUES  CHORDF 00175
C          MDEG = IDEG + 1  CHORDF 00176
JSUM = 0              CHORDF 00177
DO 400 I=1,ITROW      CHORDF 00178
400 JSUM = JSUM + JCC(I)  CHORDF 00179
C          JSUM = JSUM - JCC(ITROW) + 1  CHORDF 00180
DO 500 I = IST,NK    CHORDF 00181
IX = I                CHORDF 00182
IF(.NOT.COPLAN.AND.J.GT.MYBW) IX = I + IOVLAP  CHORDF 00183
ISUB = JSUM + J - JS(IX) - NC + 1  CHORDF 00184
VP = VC                CHORDF 00185
XV = FLOAT(I) - XINC  CHORDF 00186
XP = 1.0                CHORDF 00187
DO 450 L = 1,MDEG    CHORDF 00188
XP = XP * XV          CHORDF 00189
XD = L                CHORDF 00190
XPI = XP/XD           CHORDF 00191
VP = VP + A(L) * XPI  CHORDF 00192
450 CONTINUE          CHORDF 00193
DELPHI(ISUB) = VP     CHORDF 00194
JSUM = JSUM + JCC(IX)  CHORDF 00195
500 CONTINUE          CHORDF 00196
C          CALCULATE THE TRAILING EDGE VELOCITY POTENTIALS (TVP ARRAY)  CHORDF 00197
C          L = J          CHORDF 00198
IF(NSUBDV.NE.1) L = NSUBDV * (J-1) + NSUBCN  CHORDF 00199
TVP(L) = (0.,0.)      CHORDF 00200
JJ = J                CHORDF 00201
JC = J                CHORDF 00202
C          L = J          CHORDF 00203
IF(NSUBDV.NE.1) L = NSUBDV * (J-1) + NSUBCN  CHORDF 00204
TVP(L) = (0.,0.)      CHORDF 00205
JJ = J                CHORDF 00206
JC = J                CHORDF 00207
C          L = J          CHORDF 00208

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IF(J.GT.MYBW) JC = J - MYBW          CHORDF 00209
I = TEXLOC(JJ)                      CHORDF 00210
XINCR = TEXLOC(JJ) - I              CHORDF 00211
IROW = I                            BCSCFA 00001
IF(.NOT.COPLAN.AND.J.GT.MYBW) I = I + IOVLAP
INDB = IPNTRM(1,I) + JC - IPNTRM(2,I)          CHORDF 00213
                                                CHORDF 00214
C                                              CHORDF 00215
C      TEST FOR 3 BOXES ON CHORD JJ
IF(NOC(JJ).LT.3) GO TO 940          CHORDF 00216
C
C      2 BOXES AND NO MACH RAY AVAILABLE, OR
C      3 BOXES OR MORE. DO LINEAR EXTRAPOLATION.
930 CONTINUE
INDE = IPNTRM(1,I-1) + JC - IPNTRM(2,I-1)          BCSCFA 00003
SDELPH = DELPHI(INDB) - DELPHI(INDE)          CHORDF 00219
GO TO 950                                      CHORDF 00220
C
C      TEST FOR MACH RAY EXTRAPOLATION.
940 CONTINUE
IB = IS(JJ-1)                      CHORDF 00221
IX = IB + NOC(JJ-1) + 1            CHORDF 00222
IF (IROW.LT. IB .OR. IROW.GT. IX) GO TO 945
IB = IS(JJ-2)                      CHORDF 00223
IX = IB + NOC(JJ-2) + 1            CHORDF 00224
IMI = IROW-1                      BCSCFA 00004
IF (IMI.GE. IB .AND. IMI.LE. IX) GO TO 948
C
C      MACH RAY CANNOT BE USED. TEST FOR 2 BOXES ON CHORD JJ
945 CONTINUE
IF (NOC(JJ).LT. 2) GO TO 7010
GO TO 950
C
C      MACH RAY CAN BE USED
948 CONTINUE
INDA = IPNTRM(1,I-1) + JC - IPNTRM(2,I-1) - 2          BCSCFA 00009
INDC = IPNTRM(1,I) + JC - IPNTRM(2,I) - 1          BCSCFA 00010
SDELPH = 2.0*DELPHI(INDC) - DELPHI(INDA) - DELPHI(INDB)          BCSCFA 00011
950 CONTINUE
JT = JJ
IF(NSUBDV.NE.1) JT = NSUBDV * (JJ-1) + NSUBCN
TVP(JT) = DELPHI(INDB) + XINCR*SDELPH
950 CONTINUE
C
1000 CONTINUE
C
C      WRITE THE DELPHI AND TVP ARRAY ON THE NIVPSC FILE
CALL RDINIT
ITYPE = SHMIXED
N = IPNTRM(1,NPNTRS) - 1
MXARRY = GHDELPHI
CALL WRTENX(NIVPSC,MXWRIT,RANXCU,NFS,NHS,LS,NMR,LWS,2,TD,
1               DELPHI,ITYPE,2,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6040
C
MTVPS = NSUBDV + NCARDS
+ HTVPS
MXARRY = GHHTVP

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CALL WRTEMX(NIVPSC,MXWRIT,RANDOU,NFS,NMS,LS,NMR,LWS,2,IC,
1           TVP,    ITYPE,2,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6040
C
C       PRINT THE SMOOTHED VEL. FCT. ARRAY
C
IF(.NOT.PRVP) GO TO 1500
TITL(1) = 8H   WING
TITL(2) = 10HVELOCITY P
TITL(3) = 10HOTENTIALS
IF(COPLAN) TITL(1) = 10H WING/TAIL
CALL PRINTR(TITL,NM,DELPHI,1,1,MXB,MYBW,IPNTRM)
IF(NSURF.EQ.1.OR.COPLAN) GO TO 1500
TITL(1) = 8H   TAIL
CALL PRINTR(TITL,NM,DELPHI,1,IFBT,MXB,T,IPNTRM(1,10MLAP+1))
1500 CONTINUE
IF(.NOT.CHECKPR) GO TO 2000
WRITE(NT6,9400) (TVP(I),I=1,NTVPS)
9400 FORMAT(//(1X,8E16.8))
C
2000 CONTINUE
C
END FILE NIVPSC
REWIND NIVPSC
C       CHANGE FILE NAMES
C
IAICSC = IVPSC
IVPSC = NIVPSC
C
RETURN
8010 CONTINUE
  WRITE (NT6,9010) IGEOSC,IRR
  WRITE (NT6,9011) MXARRY
  GO TO 6100
8020 CONTINUE
  WRITE (NT6,9010) MODESC,IRR
  WRITE (NT6,9011) MXARRY
  GO TO 6100
8040 CONTINUE
  WRITE (NT6,9010) IVPSC,IRR
  WRITE (NT6,9041) NM
6100 CONTINUE
  WRITE (NT6,9101) ID(1),ID(2)
  WRITE (NT6,9102) FARM,IPARM
  WRITE (NT6,9103) NFS,NMS
  WRITE (NT6,9104) ITYPE,M,N
  WRITE (NT6,9900)
  GO TO 8000
C
7010 CONTINUE
  WRITE (NT6,8020)
  WRITE (NT6,8021) IRW,JC
C
8000 CONTINUE
  CALL FLUSH(1)
C
8020 FORMAT(7SHN*** ERROR - NO TIP TRAILING EDGE VELOCITY POTENTIAL CAN CHORDF 00313

```

1 BE COMPUTED. ***)	CHORDF 00314
8021 FORMAT(5X,13HCOORDINATES (I2,1H,I2,1H))	CHORDF 00315
9010 FORMAT(53H*** ERROR - WHILE READING THE GEOMETRY SCRATCH FILE A1G 1, 1SH, ERROR CODE = I4,4H ***)	CHORDF 00316
9011 FORMAT(5X,32HAN ATTEMPT WAS MADE TO READ THE A6, 8H MATRIX.//)	CHORDF 00318
9041 FORMAT(5X,54HAN ATTEMPT WAS MADE TO READ THE VEL. POT. ARRAY NUMBE 1R I3,1H.)	CHORDF 00319
9050 FORMAT(56H*** ERROR - WHILE WRITING ON THE VEL. POT. SCRATCH FILE 1 A10, 1SH, ERROR CODE = I4,4H ***)	CHORDF 00321
9051 FORMAT(5X,36HAFTER ATTEMPTING TO WRITE VEL. POT. NUMBER I3)	CHORDF 00322
9101 FORMAT(5X,*MATRIX ID = *, A10, I10)	CHORDF 00323
9102 FORMAT(5X,*PARAMETERS *,10E11.3, / 10X,*(INTEGER)*, I7, 9I11)	CHORDF 00324
9103 FORMAT(5X,*FILE SPACING = *,I3,* MATRIX SPACING = *,I3)	CHORDF 00325
9104 FORMAT(5X,*MATRIX TYPE =*,A10,*, DIMENSIONED (*I4,2H X,I4,1H))	CHORDF 00326
9900 FORMAT(*0 ERROR OCCURRED IN CHORD-FIT SMOOTHING SECTION. *) END	FTNXL 00074
	CHORDF 00329

```

SUBROUTINE PRINTR(TITL,IMODE,ARRAY,K,IXB,MXB,MYB,IPNTRM)          PRINTR 00002
C
C      TITL - TITLE TO PRINT FOR THE ARRAY                         PRINTR 00003
C      IMODE - MODE SHAPE NUMBER                                PRINTR 00004
C      ARRAY - ARRAY TO BE PRINTED                               PRINTR 00005
C
C      DIMENSION ARRAY(K,1), TITL(3)                             PRINTR 00006
C      COMPLEX ARRAY                                         PRINTR 00007
C      COMMON /FILES / NTS,NTS,INTAPE,INFSF,NPLAIC,NSFAIC,NOUTP,   FILES 00002
C                           IOUFSF,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC   FILES 00003
C      COMMON /CTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT, CTRL 00002
C                           DEFAULT                                 CTRL 00003
C      LOGICAL      PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT           CTRL 00004
C      COMMON /PROBLM/ XNACH,NMODOES,NTSLOP,NKVALS,SMOOTH,NDEG,CDFIT, PROBLM 00002
C                           EXAIC,SUBDV,PLYWOOD                   PROBLM 00003
C      LOGICAL      SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD            PROBLM 00004
C      COMMON /KVAL / IKVAL,XXVAL(20), XXS(20)                  KVAL  00002
C      DIMENSION IPNTRM(2,50)                                     PRINTR 00015
C      DIMENSION PC(2)                                         PRINTR 00016
C      DIMENSION S(50),D(50)                                     PRINTR 00020
C      EQUIVALENCE (S(1),BUFF(1)),(D(1),BUFF(1251))             PRINTR 00021
C      REAL K1                                              PRINTR 00022
C      INTEGER PAGE                                         PRINTR 00023
C      COMMON /RWBUFF/ BFCCDE,IBFCNT,  BUFF(3280)                RWBUFF 00002
C      DATA  PC    / 10H PAGE CONTI,4HNUED /
C      DATA  BLANK / 1H /
C      DATA  XINIT / -1.0 /
C      K1 = XXVAL(IKVAL)
C      IF(XXS(IKVAL).NE.XINIT) K1 = XXS(IKVAL)
C      PAGE = 0
C      N=1
C      M=4
C      IF(M.GT.MYB) M = MYB
100 LINE = 100
200 CONTINUE
      DO 1400 I=IXB,MXB
      DO 300 J=N,M
      S(J) = 0.0
      D(J) = 0.0
300 CONTINUE
      IF(LINE.LE.50) GO TO 900
      PAGE = PAGE + 1
      LINE = 4
      WRITE (NT6,9001) TITLE,TITL,XNACH,K1,IMODE
C
      IF(PAGE.EQ.1) GO TO 700
      WRITE (NT6,9005) PC
      GO TO 800
700 WRITE(NT6,9005)
800 CONTINUE
      WRITE(NT6,6006) (BLANK,J,J=N,M)
      WRITE(NT6,6007) (BLANK, J=N,M)
900 CONTINUE
      JS = IPNTRM(2,I)
      IF(JS .LE. 0) GO TO 1400
      IDX = IPNTRM(1,I)
      JE = IPNTRM(1,(+1)) - IDX + JS -1

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IF(JE.EQ.0) GO TO 1400          PRINTR 00054
DO 1000 J=JS,JE                 PRINTR 00055
S(J) = REAL(ARRAY(1,IDX))       PRINTR 00056
D(J) = AIMAG(ARRAY(1,IDX))      PRINTR 00057
IDX = IDX +1                   PRINTR 00058
1000 CONTINUE                    PRINTR 00059
DO 1200 J=N,M                   PRINTR 00060
IF(S(J)) 1300,1100,1300         PRINTR 00061
1100 CONTINUE                    PRINTR 00062
IF(D(J)) 1300,1200,1300         PRINTR 00063
1200 CONTINUE                    PRINTR 00064
GO TO 1400                      PRINTR 00065
1300 WRITE (NT6,9013) I,(S(J),D(J),J=N,M)   PRINTR 00066
LINE = LINE + 1                  PRINTR 00067
1400 CONTINUE                    PRINTR 00068
M = M+4                          PRINTR 00069
N = N+4                          PRINTR 00070
IF(N.GT.MYB) GO TO 1500         PRINTR 00071
IF(M.GT.MYB) M = MYB            PRINTR 00072
IF(LINE.GT.45) GO TO 100          PRINTR 00073
WRITE(NT6,8006) (BLANK,J,J=N,M)  PRINTR 00074
WRITE(NT6,8007) (BLANK, J=N,M)   PRINTR 00075
LINE = LINE+3                   PRINTR 00076
GO TO 200                        PRINTR 00077
1500 CONTINUE                    PRINTR 00078
RETURN                          PRINTR 00079
9001 FORMAT(1H1,20X,8A10/ 46X,*SMOOTHED *,3A10/ 46X,7H( MACH F5.3,5X,
1 12HRED. FREQ. =,F8.5, * )* / 52X,*MODE SHAPE*, I3)  PRINTR 00080
9005 FORMAT(44X,42(1H-),20X,A10,A4)        PRINTR 00081
8006 FORMAT(4HOROW, A1,14X,SHCHORD,I3, 3(A1,22X,SHCHORD,I3) )  PRINTR 00082
8007 FORMAT(3X, 4(A1,8X,4HREAL,8X,9HIMAGINARY) )  PRINTR 00083
9013 FORMAT(I4,8E16.8)           PRINTR 00084
END                            PRINTR 00085
                                         PRINTR 00086

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SUBROUTINE CURVE(M,N,X,Z,C)
DIMENSION X(50), Z(2,50), C(2,21)
DIMENSION A1(23), A(21,23), XP(6)

C
C      M - DEGREE OF POLYNOMIAL EQUATION
C      N - NUMBER OF DATA POINTS TO FIT CURVE THROUGH
C      X - X COORDINATE OF DATA POINT
C      Z - Z COORDINATE
C      C - OUTPUT COEFFICIENT ARRAY

C
C      EPS = 1.0E-04
IF(N.LT.M+1) M = N-1
NC = M + 1

C      ZERO OUT THE ARRAYS NEEDED
C
ITOT = NC + 2
DO 100 I=1,NC
C(I,1) = 0.0
C(2,I) = 0.0
100 CONTINUE
DO 100 J=1,ITOT
A(I,J) = 0.0
100 CONTINUE

C      DETERMINE DEVIATION EQUATION AND SQUARE THE EQUATION
C
AI(1) = 1.00
DO 400 K=1,N
AI(1) = 1.0
DO 200 L=2,NC
AI(L) = AI(L-1) * X(K)
200 CONTINUE
AI(NC+1) = Z(1,K)
AI(NC+2) = Z(2,K)

C
DO 300 I=1,NC
DO 300 J=1,ITOT
ASAV = AI(I) * AI(J)
A(I,J) = A(I,J) + ASAV
300 CONTINUE
400 CONTINUE

C      SQUARE ROOT METHOD INTERMEDIATE MATRIX
C
DO 1200 I=1,NC
IM1 = I-1
TMP = 0.0
IF(I.EQ.1) GO TO 600
DO 500 L=1,IM1
500 TMP = TMP + A(L,I) ** 2
500 CONTINUE
T= A(I,I) - TMP
IF(T.GT.EPS) GO TO 700
A(I,I) = 0.0
GO TO 1200
700 CONTINUE

```

CURVE	00002
CURVE	00003
CURVE	00004
CURVE	00005
CURVE	00006
CURVE	00007
CURVE	00008
CURVE	00009
CURVE	00010
CURVE	00011
CURVE	00012
CURVE	00013
CURVE	00014
CURVE	00015
CURVE	00016
CURVE	00017
CURVE	00018
CURVE	00019
CURVE	00020
CURVE	00021
CURVE	00022
CURVE	00023
CURVE	00024
CURVE	00025
CURVE	00026
CURVE	00027
CURVE	00028
CURVE	00029
CURVE	00030
CURVE	00031
CURVE	00032
CURVE	00033
CURVE	00034
CURVE	00035
CURVE	00036
CURVE	00037
CURVE	00038
CURVE	00039
CURVE	00040
CURVE	00041
CURVE	00042
CURVE	00043
CURVE	00044
CURVE	00045
CURVE	00046
CURVE	00047
CURVE	00048
CURVE	00049
CURVE	00050
CURVE	00051
CURVE	00052
CURVE	00053
CURVE	00054
CURVE	00055
CURVE	00056
CURVE	00057
CURVE	00058

```

A(I,I) = SQRT(T)
IF(A(I,I).GT.EPS) GO TO 800
A(I,ITOT) = 0.0
GO TO 1200
800 CONTINUE
C
JS = I+1
DO 1100 J=JS,ITOT
TMP = 0.0
IF(I.EQ.1) GO TO 1000
DO 900 L=1,IM1
900 TMP = TMP + A(L,I)*A(L,J)
1000 A(I,J) = (A(I,J)-TMP)/A(I,I)
1100 CONTINUE
1200 CONTINUE
C
C      BACK SUBSTITUTE FOR COEFFICIENTS
C
DO 1600 K=1,NC
I = NC - K + 1
IP1 = I + 1
TMP1 = 0.0
TMP2 = 0.0
IF(A(I,I).GT.EPS) GO TO 1300
C(1,I) = 0.0
C(2,I) = 0.0
GO TO 1600
1300 CONTINUE
IF(I.EQ.NC) GO TO 1500
DO 1400 L=IP1,NC
TMP1 = TMP1 + A(I,L) * C(1,L)
TMP2 = TMP2 + A(I,L) * C(2,L)
1400 CONTINUE
1500 CONTINUE
C(1,I) = (A(I,NC+1)-TMP1)/A(I,I)
C(2,I) = (A(I,ITOT)-TMP2)/A(I,I)
1600 CONTINUE
RETURN
END

```

CURVE	00059
CURVE	00060
CURVE	00061
CURVE	00062
CURVE	00063
CURVE	00064
CURVE	00065
CURVE	00066
CURVE	00067
CURVE	00068
CURVE	00069
CURVE	00070
CURVE	00071
CURVE	00072
CURVE	00073
CURVE	00074
CURVE	00075
CURVE	00076
CURVE	00077
CURVE	00078
CURVE	00079
CURVE	00080
CURVE	00081
CURVE	00082
CURVE	00083
CURVE	00084
CURVE	00085
CURVE	00086
CURVE	00087
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CURVE	00089
CURVE	00090
CURVE	00091
CURVE	00092
CURVE	00093
CURVE	00094
CURVE	00095
CURVE	00096
CURVE	00097

```

OVERLAY (AFMBOX,1,10)                                FORCES 00002
PROGRAM FORCES                                     FORCES 00003
FORCES 00004
FORCES 00005
FORCES 00006
FORCES 00007
FORCES 00008
FORCES 00009
FORCES 00010
FORCES 00011
FORCES 00012
FORCES 00013
FORCES 00014
FORCES 00015
FORCES 00016
FORCES 00017
FORCES 00018
FORCES 00019
FORCES 00020
FORCES 00021
FORCES 00022
FORCES 00023
FORCES 00024
FORCES 00025
FORCES 00026
FORCES 00027
FORCES 00028
FORCES 00029
FORCES 00030
FORCES 00031
FORCES 00032
FORCES 00033
FORCES 00034
FORCES 00035
FORCES 00036
FORCES 00037
FORCES 00038

C      THIS PROGRAM CALCULATES BOX LIFTS, SECTION LIFTS, TOTAL LIFT,  

C      AND GENERALIZED AIR FORCES. THE PROGRAM MUST READ INFORMATION  

C      FROM THE GEOMETRY SCRATCH FILE AND THE MODE SCRATCH FILE.  

C

COMMON /ARRAYS/ KBXCDW,LBXCDW,LBCXC,KBXCDT,LBXCDT,KJALPH,LJALPH,  

1      KALPHA,KKERNL,LKERNL,KPNTRM,LPNTRM,KDEFSL,KELPHI,  

2      LMODES,KPNTSD,LPNTSO,KSCW,LSCW,KPNTDW,LPNTDW,  

3      KDW,LDW,KTVP,LTVP  

COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSFAIC,NOUTP,  

1      IOUFSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC  

EQUIVALENCE (IWTFSC,ITSLS)  

COMMON /IOCONT/ OPLAIC,OSFAIC,WTCOM,WTCNAF,WTSI,WTLB,PRBOX,  

1      PRPAIC,PRSAIC,PRMCDS,PRCOEF,PRDW,PRSW,PRVP,  

2      PRBL,PRDCP,PRGNAF,PRGNAC,PRSL,PRLW,PRNW,PRCM  

EQUIVALENCE (PRUW,PRDW)  

LOGICAL OPLAIC,OSFAIC,WTCOM,WTCNAF,WTSI,WTLB,PRBOX,PRPAIC,  

1      PRSAIC,PRMCDS,PRCOEF,PRDW,PRSW,PRVP,PRBL,PRSL,PRGNAF,  

2      PRDCP,PRGNAC,PRUW,PRLW,PRNW,PRCM  

COMMON /KERN / ERR,MXSKRN,IPKERN,NPLKRN,NSPATK,NRONEA  

COMMON /KVAL / IKVAL,XKVAL(20),XKS(20)  

COMMON /PROBLW/ XHACH,NMODOES,NTSLOC,NKVALS,SMOOTH,NDEG,CDFIT,  

1      EXAIC,SUBDV,PLYWOOD  

LOGICAL SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD  

COMMON / MODES/ SYM,SYMT,MTYPEW,MTYPEF  

COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,  

1      B1,B1BETA,B1S,B1BTAS,TLAX,TLAZ,PSIW,  

2      MXBW,MXBHW,MYBW,MYEBW,MXESW,MYBSW,MYBBSW,  

3      IXBW,XCENTR  

LOGICAL COPLAN  

COMMON /GEOM2/ TLAX,TLAZ,PSIT,MXB1,MYBT,MYBBT,MXBST,MYBST,  

1      MYBBST,IXBT,IXBST,CAFL  

COMMON /TAPEIO/ NFS,NMS,LS,NMR,ID(20),NID,ITYPE,LRS,LWS,N,N,  

1      PARM(10),IRR  

DIMENSION IPARM(10)  

EQUIVALENCE (PARM,IPARM)  

COMMON /CHECKPR/ DPPCPR,GEOPCR,MODCPR,AICCPR,NMSCPR,SMCPR,GAFCP  

LOGICAL DPPCPR,GEOPCR,MODCPR,AICCPR,NMSCPR,SMCPR,GAFCP  

EQUIVALENCE (CHECKPR,GAFCP)  

LOGICAL CHECKPR  

DIMENSION RWBF(1250)  

EQUIVALENCE (RWBF,BUFF)  

COMPLEX RWPF  

COMMON /LOCL/ XXVL  

COMMON     BLANK(1)  

DIMENSION TITL(2)

C      FELOC((MYBW+MYBT)*NSUBDV),TEXLOC(SAME)
DIMENSION FELOC(250),TEXLOC(250)
C      IPNTRM(2,NROWS)
DIMENSION IPNTRM(2,100)
C      IBOXW(NROWS,150/20),IBOXT(90,150/20)
DIMENSION IBOXW(150,8),IBOXT(90,8)
C      IBXCDF(NCOLS),IBXCD(NCOLS),IBXCDA(NCOLS)
DIMENSION IBXCDF(150),IBXCD(150),IBXCDA(150)

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C      DEFSL(2,NBOXES),  DEFLTE(NCOLS)          FORCES 00039
C      DIMENSION DEFSL(2,1000),  DEFLTE(50)        FORCES 00040
C      ALPHA(NCOLS#2#NSURF),  IJALPH(SAME)       FORCES 00041
C      TSLFN(NBOXES)                            FORCES 00042
C      DIMENSION TSLFN(1000)                      FORCES 00043
C      DIMENSION ALPHA(200),  IJALPH(200)         FORCES 00044
C      DELPHI(NBOXES), TVP(NCOLS1+NCOLS2+NSUBDV) FORCES 00045
C      COMPLEX DELPHI(1000), TVP(250)            FORCES 00046
C      COMPLEX BXLIFT(NBOXES), SLIFT(NCOLS#NMODES), GENAF(NMODOES#NMODE) FORCES 00047
C      COMPLEX BXLIFT(1000), SLIFT(100), TLIFT, GENAF(400), AFROW(NMODOES) FORCES 00048
C      AFROW(NMODOES)                           FORCES 00049
1      AFROW(20), BL2, TLIFT1,TLIFT2           FORCES 00050
C      DELCP(NBOXES)                           FORCES 00051
C      COMPLEX DELCP(1000)                      FORCES 00052
C      GRAFC(NMODOES#NMODOES), GFFAFC(SAME)     FORCES 00053
C      DIMENSION GFAFC(400), GFFAFC(400)         FORCES 00054
C      DIMENSION AFC(2)                         FORCES 00055
C      EQUIVALENCE (AFCSTR,AFC)                 FORCES 00056
C      COMPLEX SECWOM(100), GAF                  FORCES 00057
C      VPTE(NCOLS)                            FORCES 00058
C      COMPLEX VPTE(50),VPLE, TEMP1, TEMP2, TEMP3, BL FORCES 00059
LOGICAL MXREAD,RANDIN,MXRIT,RANDOU           FORCES 00060
LOGICAL BLNEED                                FORCES 00061
COMPLEX XINDEF                                FORCES 00062
DIMENSION XINDEF(2)                          FTNXI 00079
EQUIVALENCE (XINDEF,XINDEF)                   FTNXI 00080
COMMON /RWBUFF/ BFCODE,IBFCNT,BUFF(3280)      RWBUFF 00082
MXREAD = .FALSE.                             FORCES 00084
RANDIN = .FALSE.                            FORCES 00085
MXRIT = .FALSE.                            FORCES 00086
RANDOU = .FALSE.                            FORCES 00087
C      FORCES 00088
C      FORCES 00089
C      XKVL = XKVAL(IKVAL)                     FORCES 00090
BLNEED = WTBL .OR. PRBL .OR. PRSL .OR. PRDCP .OR. PRCM BCSFRB 00011
TWCET = (2.0*B1BETA)/B1                      FORCES 00072
TMBB = TWCET/B1                               FORCES 00073
C      CONSTANTS FOR AGARD GENERALIZED AERODYNAMIC COEFFICIENTS, BCSFRA 00001
C      BASED ON WING SEMI-SPAN                  BCSFRA 00002
8      = MYBW * B1BETA                        BCSFRA 00003
S3      = S4*S3                                FORCES 00076
S4      = S4*S3                                FORCES 00077
BS3BET = -B1BETA/S3                          FORCES 00078
BKSB4BT = 0.0                                  BCSFRA 00004
IF (XKVL .EQ. 0.) GO TO 5                    BCSFRA 00005
BKSB4BT = -1. +(B1*B1BETA)/(XKVL*S4)        FORCES 00079
5 CONTINUE                                     BCSFRA 00006
NVP8 = NMODES                                FORCES 00082
MAX = NVP8 * NMODES                          FORCES 00083
C      FORCES 00084
C      FORCES 00085
C      FORCES 00086
C      FORCES 00087
REWIND MODESC                                FORCES 00088
NSPCE = 0                                     FORCES 00098

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C          READ THE POINTERS FROM THE MODESC FILE.          FORCES 00089
C          CALL RDINIT                                     FORCES 00090
C          ITYPE = SHMIXED                                FORCES 00091
C          MXARRY = GHIPNTRM                             FORCES 00092
C          CALL READMX(MODESC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,IPNTRM,M,N,FARM,IRR)                   FORCES 00093
C          ICVLAP = IPARM(3)                               FORCES 00094
C          NPNTRS = N                                     FORCES 00095
C          MXB = NPNTRS - 1                               FORCES 00096
C          MYB = MAXD(MYBW,MYBT)                          FORCES 00097
C          NBOXES = MYB * MXB                            FORCES 00098
C          IF(IRR.NE.0) GO TO 6020                         FORCES 00099
C
C          REWIND IGEOSC                                 FORCES 00100
C
C          READ BOX CODES INTO STORAGE FROM GEOMETRY SCRATCH FILE   FORCES 00101
C
C          CALL RDINIT                                     FORCES 00102
C          ITYPE = GHMIXED                                FORCES 00103
C          MXARRY = GHIBOXW                             FORCES 00104
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,150,NID,ID,ITYPE,
1           LRS,IBOXW,M,N,FARM,IRR)                    FORCES 00105
C          IF(IRR.NE.0) GO TO 6010                         FORCES 00106
C
C          NPLS = 1                                       FORCES 00107
C          IF(NSURF.EQ.1.OR.COPLAN) GO TO 10              FORCES 00108
C          NPLS = 2                                       FORCES 00109
C
C          CALL RDINIT                                     FORCES 00110
C          ITYPE = GHMIXED                                FORCES 00111
C          MXARRY = GHIBOXW                             FORCES 00112
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,150,NID,ID,ITYPE,
1           LRS,IBOXW,M,N,FARM,IRR)                    FORCES 00113
C          IF(IRR.NE.0) GO TO 6010                         FORCES 00114
C
C          NPLS = 1                                       FORCES 00115
C          IF(NSURF.EQ.1.OR.COPLAN) GO TO 10              FORCES 00116
C          NPLS = 2                                       FORCES 00117
C
C          CALL RDINIT                                     FORCES 00118
C          ITYPE = GHMIXED                                FORCES 00119
C          MXARRY = GHIBOXT                             FORCES 00120
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,90,NID,ID,ITYPE,
1           LRS,IBOXT,M,N,FARM,IRR)                    FORCES 00121
C          IF(IRR.NE.0) GO TO 6010                         FORCES 00122
C
C          10 CONTINUE                                     FORCES 00123
C
C          READ THE TEXLOC AND FEXLOC ARRAYS FROM THE GEOMETRY SCRATCH   FORCES 00124
C          FILE. THESE ARE NEEDED TO INTERPOLATE VELOCITY POTENTIALS AT   FORCES 00125
C          BOX EDGES.                                         FORCES 00126
C
C          CALL RDINIT                                     FORCES 00127
C          ITYPE = SHMIXED                                FORCES 00128
C          MXARRY = GHFEXLOC                            FORCES 00129
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
1           LRS,FEXLOC,M,N,PARM,IRR)                   FORCES 00130
C          IF(IRR.NE.0) GO TO 6010                         FORCES 00131
C
C          CALL RDINIT                                     FORCES 00132
C          ITYPE = SHMIXED                                FORCES 00133
C          MXARRY = GHTEXLOC                            FORCES 00134
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
1           LRS,TEXLOC,M,N,FARM,IRR)                   FORCES 00135
C          IF(IRR.NE.0) GO TO 6010                         FORCES 00136
C
C          CALL RDINIT                                     FORCES 00137
C          ITYPE = SHMIXED                                FORCES 00138
C          MXARRY = GHTEXLOC                            FORCES 00139
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
1           LRS,TEXLOC,M,N,FARM,IRR)                   FORCES 00140
C          IF(IRR.NE.0) GO TO 6010                         FORCES 00141
C
C          CALL RDINIT                                     FORCES 00142
C          ITYPE = SHMIXED                                FORCES 00143
C          MXARRY = GHTEXLOC                            FORCES 00144
C          CALL READMX(IGEOSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
1           LRS,TEXLOC,M,N,FARM,IRR)                   FORCES 00145
C          IF(IRR.NE.0) GO TO 6010                         FORCES 00146

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```

C
C      READ AREAS AND POINTERS FOR AREAS FROM GEOMETRY SCRATCH FILE.
C      THESE ARE USED IN CALCULATION OF BOX LIFTS AND GEN. FORCES.      FORCES 00146
C
C      CALL RDINIT
C      ITYPE = SHMIXED
C      MARRAY = GH1PHAS
C      CALL READMX(I_GEOSC,MREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
C      1           LRS,ALPHA,M,N,PARM,IRR)      FORCES 00147
C      IF(IRR.NE.0) GO TO 6010      FORCES 00148
C
C      CALL RDINIT
C      ITYPE = SHMIXED
C      MARRAY = GH1JALPH
C      CALL READMX(I_GEOSC,MREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
C      1           LRS,I_JALPH,M,N,PARM,IRR)      FORCES 00149
C      NALPH = N      FORCES 00150
C      NALPHW = I_PARM(3)
C      NALPHH = N - NALPHW      FORCES 00151
C      IF(IRR.NE.0) GO TO 6010      FORCES 00152
C
C      REORDER THE FEXLOC AND TEXLOC ARRAYS SO THAT THERE ARE      FORCES 00153
C      VALUES FOR UNSUBDIVIDED CHORDS ONLY.      FORCES 00154
C
C      IF(NSUBDV.EQ.1) GO TO 120      FORCES 00155
C      XSLIDE = NSUBDV-IXBW      FORCES 00156
C      JCCL = NSUBCN      FORCES 00157
C      NCOLS = MYBW + MYBT      FORCES 00158
C      DO 110 I=1,NCOLS      FORCES 00159
C      TEXLOC(I) = (TEXLOC(JCCL)+XSLIDE)/NSUBDV      FORCES 00160
C      FEXLOC(I) = (FEXLOC(JCCL)+XSLIDE)/NSUBDV      FORCES 00161
C      JCCL = JCCL + NSUBDV      FORCES 00162
C
C      110 CONTINUE      FORCES 00163
C      120 CONTINUE      FORCES 00164
C
C      LOOP ON THICKNESS SLOPE FUNCTIONS (IF NONE WERE REQUESTED,
C      ONE DUMMY SET OF ONES WILL HAVE BEEN GENERATED.)      FORCES 00165
C
C      REWIND ITSLSLC      FORCES 00166
C      DO 750 ITSLOP=1,NTSLOP      FORCES 00167
C
C      READ THICKNESS SLOPE FUNCTIONS      FORCES 00168
C
C      CALL RDINIT
C      ITYPE = 4HREAL
C      MARRAY = GHTSLFN
C      CALL READMX(ITSLSLC,MREAD,RANDIN,NFS,NMS,LS,NMR,1,NID,ID,ITYPE,
C      1           LRS,TSLFN,M,N,PARM,IRR)      FORCES 00169
C      IF(IRR.NE.0) GO TO 6040      FORCES 00170
C
C      ZERO OUT THE AIR FORCES ARRAY      FORCES 00171
C
C      DO 150 J=1,MAX      FORCES 00172
C      150 GENAF(J) = (0.,0.)      FORCES 00173
C
C      LOOP ON NUMBER OF MODE SHAPES      FORCES 00174
C      DO 650 NM=1,NMOCES      FORCES 00175

```

```

C
C      GET MODE SHAPE NM FROM MODESC SCRATCH FILE
CALL RDINIT
IF(NM.EQ.1) NMS = NMSPCE
ITYPE = 4HREAL
MXARRY = 8HDEFSL
CALL READMX(MODESC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,DEFSL,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6020

C
C
C      LOOP ON VELOCITY POTENTIALS
REWIND IVPSC

C      DO 600 JVP=1,NTVPS

C      READ ONE SET OF VELOCITY POTENTIALS
CALL RDINIT
ITYPE = 4HREAL
MXARRY = 8HDELPHI
CALL READMX(IPVSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,DELPHI,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6030

C      CALL RDINIT
ITYPE = 4HREAL
MXARRY = 3HTVP
CALL READMX(IPVSC,MXREAD,RANDIN,NFS,NMS,LS,NMR,2,NID,ID,ITYPE,
1           LRS,TVP,M,N,PARM,IRR)
IF(IRR.NE.0) GO TO 6030

C      CONDENSE THE TRAILING EDGE VELOCITY POTENTIAL ARRAY TO
C      UNSUBDIVIDED BOXES.
C      ALSO ZERO OUT THE SECTIONAL GENERALIZED FORCES.

C      NTVPS = MYBW + MYBT
DO 210 I=1,NTVPS
SECMOM(I) = (0.,0.)
210 CONTINUE

C      IF(NSUBDV.EQ.1) GO TO 220
JCOL = NSUBCN
DO 215 I=1,NTVPS
TVP(I) = TVP(JCOL)
JCOL = JCOL + NSUBDV
215 CONTINUE
220 CONTINUE

C      ZERO OUT THE BOX LIFT ARRAY

C      IF(NM.NE.1) GO TO 240
IF (.NOT. BLNED) GO TO 240
NEX = IPNTRM(1,NPNTRS) - 1
DO 230 I=1,NEX
DELCP(I) = (0.,0.)

```

```

FORCES 00203
FORCES 00204
FORCES 00205
FORCES 00206
FORCES 00207
FORCES 00208
FORCES 00209
FORCES 00210
FORCES 00211
FORCES 00212
FORCES 00213
FORCES 00214
FORCES 00215
FORCES 00216
FORCES 00217
FORCES 00218
FORCES 00219
FORCES 00220
FORCES 00221
FORCES 00222
FORCES 00223
FORCES 00224
FORCES 00225
FORCES 00226
FORCES 00227
FORCES 00228
FORCES 00229
FORCES 00230
FORCES 00231
FORCES 00232
FORCES 00233
FORCES 00234
FORCES 00235
FORCES 00236
FORCES 00237
BCSFRB 00012
BCSFRB 00013
BCSFRB 00014
BCSFRB 00015
BCSFRB 00016
BCSFRB 00017
FORCES 00238
FORCES 00239
FORCES 00240
FORCES 00242
FORCES 00243
FORCES 00244
FORCES 00245
FORCES 00246
FORCES 00247
FORCES 00248
FORCES 00249
FORCES 00250
FORCES 00251
FORCES 00252
FORCES 00253
FORCES 00254

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```

230 BXLIFT(I) = (0.,0.)
240 CONTINUE
C
C
C*****+
C      THE FOLLOWING BLOCK OF CODE COMPUTES
C      BXLIFT - BOX LIFTS
C      SLIFT - SECTION (CHORD) LIFTS
C      TLIFT - TOTAL LIFT
C      GENAF - GENERALIZED AIRFORCES
C
C*****+
C
C      AFROW(JVP) = (0.,0.)
C          AFROW = CURRENT ROW OF GENERALIZED AIRFORCES
C
C      SET UP INITIAL CONDITIONS FOR DOUBLE LOOP OVER THE ENTIRE
C      BOX PATTERN
C          IBXCDF = FORWARD
C          IBXCD = CENTER ROWS OF BOX CODES, EXPANDED
C          IBXCDA = AFT /
C          VPLE = VELOCITY POTENTIAL AT BOX LEADING EDGE
C          VPTE = ARRAY OF BOX TRAILING EDGE VELOCITY POTENTIALS
C
C
DO 565 NP=1,NPLS
IF (NP.EQ.2) GO TO 245
ISROWA = 1
NBXA = IPNTRM(1,2)
CALL DCODER(IBOXW,150,ISROWA,1,ISROWA,NBX,.F.,IBXCDA)
NBX = NBXA
GO TO 250
245 CONTINUE
ISROWA = (IXBT-IXBW)/NSUBDV + 1
IXBUT = ISROWA
IDEX = ISROWA + IOVLAP
NBXA = IPNTRM(1,IDEV+1) - IPNTRM(1,IDEV)
ISUBT = 2-IXBST
CALL DCODER(IBOKT(ISUBT,1),LBXCDT,ISROWA,1,ISROWA,NBX,.F.,IBXCDA)
NBX = NBXA
250 CONTINUE
DO 270 JCOL = 1,NBXA
IBXCD(JCOL) = IBXCDA(JCOL)
IF (IBXCDA(JCOL) .EQ. 1) GO TO 260
VPTE(JCOL) = XINDEF
DEFLTE(JCOL) = XINDEF
GO TO 270
260 CONTINUE
VPTE(JCOL) = (0.,0.)
IDC = JCOL
DEFLTE(JCOL) = DEFSL(1, IDC) + DEFSL(2, IDC)*81*(FEXLOC(JCOL)-1.0)
270 CONTINUE
C
C
C      LOOP ON ROWS OF THE BOX PATTERN
IF(NP.EQ.2) GO TO 275
IRS = 1

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NGBT = NXBT
IF(COPLAN) NGBT = NXBT
NROWS = NXBT
GO TO 280
275 CONTINUE
IRS = (IXBT-IXBW)/NSUBDV + 1 + IOVLAP
NGBT = NXBT + IOVLAP
280 CONTINUE
DO 560 IROW=IRS,NGBT
C
C      TRANSFER BOX CODES TO CORRECT ARRAYS FOR NEW ROW
NBXF = NBX
NBX = NBXA
IF (NBXF .EQ. 0) GO TO 315
DO 310 JCCL = 1,NBXF
310 IBXCDF(JCCL) = IBXCD(JCCL)
315 CONTINUE
IF (NBX .EQ. 0) GO TO 325
DO 320 JCCL = 1,NBX
320 IBXCD(JCCL) = IBXDA(JCCL)
325 CONTINUE
IF (IROW .EQ. NGBT) GO TO 335
ISROWA = ISROWA + 1
IF(NP.EQ.2) GO TO 330
NBXA = IPNTRM(1,ISROWA+1)-IPNTRM(1,ISROWA)+IPNTRM(2,ISROWA)-1
CALL DCODER(IBOXW,150,ISROWA,1,ISROWA,NBXA,.F.,IBXDA)
GO TO 340
330 CONTINUE
IDEX = ISROWA + IOVLAP
NSYA = IPNTRM(1,IDEINDEX) - IPNTRM(1,IDEINDEX) + IPNTRM(2,IDEINDEX) - 1
ISUBT = 2-IXBST
CALL DCODER(IBOXT(ISUBT,1),LBXCOT,ISROWA,1,ISROWA,NBXA,.F.,IBXDA)
GO TO 340
335 CONTINUE
NBXA = 0
340 CONTINUE
C
ITROW = IROW
IF(NP.EQ.2) ITROW = IROW - IOVLAP
C
C      LOOP ON CHORDS OF THE BOX PATTERN
IF (NBX .EQ. 0) GO TO 560
DO 550 JCCL = 1,NBX
C
IF (IBXCD(JCCL) .NE. 1) GO TO 550
IDC = LOCSDW(ITROW,JCCL, IPNTRM,LPNTRM,1,LPNTRM)
IF (IDC .EQ. 0) GO TO 970
C
C      GET THE SUBSCRIPT TO USE IN THE EDGE ARRAYS, JJ
JJ = JCCL
IF (NP .EQ. 2) GO TO 350
IF (.NOT. COPLAN) GO TO 355
IF (JJ .GT. NYBT) GO TO 355
IF (YEXLOC(JJ) .GE. FLOAT(ITROW)) GO TO 355
350 JJ = JJ + NYBW
355 CONTINUE
C

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FORCES 00312
FORCES 00313
FORCES 00314
FORCES 00315
FORCES 00316
FORCES 00317
FORCES 00318
FORCES 00319
FORCES 00320
FORCES 00321
FORCES 00322
FORCES 00323
FORCES 00324
FORCES 00325
FORCES 00326
FORCES 00327
FORCES 00328
FORCES 00329
FORCES 00330
FORCES 00331
FORCES 00332
FORCES 00332?
FORCES 00334
FORCES 00335
FORCES 00336
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FORCES 00338
FORCES 00339
FORCES 00340
FORCES 00341
FORCES 00342
FORCES 00343
FORCES 00344
FORCES 00345
FORCES 00346
FORCES 00347
FORCES 00348
FORCES 00349
FORCES 00350
FORCES 00351
FORCES 00352?
FORCES 00353
FORCES 00354
FORCES 00355
FORCES 00356
FORCES 00357
FORCES 00358
FORCES 00359
FORCES 00360
FORCES 00361
FORCES 00362
FORCES 00363
FORCES 00364
FORCES 00365
FORCES 00366
FORCES 00367
FORCES 00368

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C      DETERMINE BOX LEADING EDGE VALUES          FORCES 00369
C      IF (FEXLOC(JJ) .GT. FLOAT(ITROW+1) ) GO TO 410   FORCES 00370
C      BOX LEADING EDGE IS INTERNAL TO THE PLANFORM   FORCES 00371
C      VPLE = VPTE(JCOL)   FORCES 00372
C      DEFLE = DEFLTE(JCOL)   FORCES 00373
C      GO TO 440   FORCES 00374
C      BOX IS ON PLANFORM LEADING EDGE. IS IT INFLUENCED BY THE WAKE-   FORCES 00375
C      410 CONTINUE   FORCES 00376
C      IF ( (JJ .EQ. JCOL .OR. .NOT. COPLAN) GO TO 420   FORCES 00377
C      LEADING EDGE OF SECOND PLANFORM. VELOCITY POTENTIAL   FORCES 00378
C      COMPUTED FRO: WAKE EQUATION.   FORCES 00379
C      XDKVL = (FEXLOC(JJ) - TEXLOC(JCOL)) * XKVL   FORCES 00380
C      VPLE = TVP(JCOL) * CMPLX(COS(XDKVL),-SIN(XDKVL))   FORCES 00381
C      GO TO 425   FORCES 00382
C      LEADING EDGE OF SECOND PLANFORM OF SPATIAL ANALYSIS   FORCES 00383
C      OR LEADING EDGE OF FIRST PLANFORM (WING)   FORCES 00384
C      420 CONTINUE   FORCES 00385
C      VPTE = (0.,0.)   FORCES 00386
C      425 CONTINUE   FORCES 00387
C      TEST FOR SINGLE BOX   FORCES 00388
C      IF (TEXLOC(JJ) .LT. FLOAT(ITROW+1) ) GO TO 430   FORCES 00389
C      BOX IS A SIMPLE LEADING EDGE BOX   FORCES 00390
C      IDA = LOCSDW(ITROW+1,JCOL, IPNTRM,LPNTRM,1,LPNTRM)   FORCES 00391
C      IF ((IDA .EQ. 0) GO TO 970   FORCES 00392
C      SLOPE = B1*DEFSL(2, IDC)   FORCES 00393
C      X0IF = FLOAT(ITROW) - FEXLOC(JJ)   FORCES 00394
C      DEFLE = DEFSL(1, IDC) - SLOPE * X0IF   FORCES 00395
C      GO TO 450   FORCES 00396
C      SINGLE BOX. GET LEADING AND TRAILING VALUES   FORCES 00397
C      430 CONTINUE   FORCES 00398
C      SLOPE = B1* DEFSL(2, IDC)   FORCES 00399
C      DEFLE = DEFSL(1, IDC) - SLOPE * (FLOAT(ITROW) - FEXLOC(JJ))   FORCES 00400
C      VPTE(JCOL) = TVP(JJ)   FORCES 00401
C      DEFLTE(JCOL) = DEFSL(1, IDC) + SLOPE * (TEXLOC(JJ) - FLOAT(ITROW))   FORCES 00402
C      GO TO 500   FORCES 00403
C      DETERMINE BOX TRAILING EDGE VALUES   FORCES 00404
C      440 CONTINUE   FORCES 00405
C      IF (TEXLOC(JJ) .LT. FLOAT(ITROW+1) ) GO TO 460   FORCES 00406
C      BOX TRAILING EDGE IS INTERNAL TO THE PLANFORM   FORCES 00407
C      IDA = LOCSDW(ITROW+1,JCOL, IPNTRM,LPNTRM, 1, LPNTRM)   FORCES 00408
C      450 CONTINUE   FORCES 00409
C      VPTE(JCOL) = .5 * (DELPHI(IDC) + DELPHI(IDA))   FORCES 00410
C      DEFLTE(JCOL) = 0.5*(DEFSL(1, IDC) + DEFSL(1, IDA))   FORCES 00411
C      GO TO 500   FORCES 00412
C      BOX IS ON SURFACE TRAILING EDGE   FORCES 00413
C      460 CONTINUE   FORCES 00414
C      VPTE(JCOL) = TVP(JJ)   FORCES 00415
C      SLOPE = B1*DEFSL(2, IDC)   FORCES 00416
C      DEFLTE(JCOL) = DEFSL(1, IDC) + SLOPE * (TEXLOC(JJ) - FLOAT(ITROW))   FORCES 00417
C      GO TO 500   FORCES 00418
C      BOX LEADING AND TRAILING EDGE VALUES ARE COMPUTED. GET   FORCES 00419
C                                         FORCES 00420
C                                         FORCES 00421
C                                         FORCES 00422
C                                         FORCES 00423
C                                         FORCES 00424
C                                         FORCES 00425

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C      ALPHA, THE AREA MULTIPLIER
500 CONTINUE
IF(NP.EQ.2) GO TO 503
NAS = 1
NAL = NALPHW
GO TO 506

C
503 CONTINUE
NAS = NALPHW + 1
NAL = NALPH
506 CONTINUE

C
ALPH = 1.0
JCOMP = JCCL+512
JCOMP1 = JCOMP+512
DO 510 I=NAS,NAL
IF (IJALPH(I) .LT. JCOMP) GO TO 510
IF (IJALPH(I) .GT. JCOMP1 ) GO TO 520
IF (IJALPH(I) .NE. JCOMP+ITROW) GO TO 510
ALPH = ALPHA(I)
GO TO 520
510 CONTINUE
520 CONTINUE
IF(PLYWOOD) ALPH = 1.0

C
C      COMPUTE TEMP1 = K1*ALPHA*(I)*(DELTA PHI)
TEMP1 = CMPLX(-AIMAG(DELPHI(IDC)), REAL(DELPHI(IDC)) )
TEMP1 = TEMP1 * (XXVL*ALPH)

C
IF(NM.NE.1) GO TO 530
C      ARE BOX LIFTS DESIRED -
IF (.NOT. BLNED) GO TO 530
BXLIFT(IDC) = (TEMP1 + ALPH*(VPTC(JCCL)-VPLC))*TWC BET* TSLFN(IDC)
DELCP(IDC) = BXLIFT(IDC)/(ALPH*B1)
530 CONTINUE

C
TEMP2 = DEFLLC(IDC)*VPTC(JCCL) - DEFLLC* VPLC
TEMP3 = (B1*ALPH*DEFSL(2, IDC)) * DE_9YI(IDC)
GAF = (TEMP1*DEFSL(1, IDC) + TEMP2 - TEMP3) * TSLFN(IDC)
AFROW(JVP) = AFROW(JVP) + GAF

C
SECMON(JJ) = SECMON(JJ) + GAF

C
530 CONTINUE
C      END OF LOOP ON CHORDS OF THE BOX PATTERN, FROM 340*
560 CONTINUE
C      END OF LOOP ON ROWS OF THE BOX PATTERN, FROM 270*
C
565 CONTINUE
C      END OF LOOP ON NUMBER OF PLATFORMS

C
SET UP TO WRITE RESULTS ON TAPE
CALL RDINIT
ITYPE = 7HCOMPLEX
PARM(1) = XXVL
PARM(2) = B1

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FORCES 00426
FORCES 00427
FORCES 00428
FORCES 00429
FORCES 00430
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FORCES 00456
FORCES 00457
FORCES 00458
FORCES 00459
FORCES 00460
FORCES 00461
FORCES 00462
FORCES 00463
BCSFRB 00018
BCSFRB 00019
BCSFRB 00020
BCSFRB 00021
BCSFRB 00022
FORCES 00466
FORCES 00467
FORCES 00468
FORCES 00469
FORCES 00470
FORCES 00471
FORCES 00472
FORCES 00473
FORCES 00474
FORCES 00475
FORCES 00476
FORCES 00477
FORCES 00478
FORCES 00479

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      PARM(3) = XMACH          FORCES 00480
C
C      IF THIS IS THE FIRST WEIGHTING FUNCTION, IT MAY BE NECESSARY  FORCES 00481
C      TO WRITE AND/OR PRINT BOX LIFTS, ETC  FORCES 00482
C      IF (.NOT. BLNEED) GO TO 600  FORCES 00483
C      IF(INM,NE,1) GO TO 600  FORCES 00484
C
C      IF (.NOT. PRBL) GO TO 570  FORCES 00485
C      PRINT BOX LIFTS  FORCES 00486
C      TITL(1) = 0H   WING  FORCES 00487
C      TITL(2) = 10HBOX LIFTS  FORCES 00488
C      TITL(3) = 2H  FORCES 00489
C      IF(COPLAN) TITL(1) = 10HWING/TAIL  FORCES 00490
C      CALL PRNTBL(TITL,JVP,BXLIFT,1,NROWS,MYBW,IPNTRM)  FORCES 00491
C      IF(NSURF.EQ.1 .OR. COPLAN) GO TO 570  FORCES 00492
C      TITL(1) = 0H   TAIL  FORCES 00493
C      CALL PRNTBL(TITL,JVP,BXLIFT,IXBUT,MXBUT,MYBT,IPNTRM(1,ICVLAP+1))  FORCES 00494
C      570 CONTINUE  FORCES 00495
C
C      IF (.NOT. PRDCP) GO TO 572  FORCES 00496
C
C      PRINT PRESSURE DIFFERENTIAL  FORCES 00497
C      TITL(1) = 0H   WING  FORCES 00498
C      TITL(2) = 10HPRESS. DI  FORCES 00499
C      TITL(3) = 10HDIFFERENCE  FORCES 00500
C      IF(COPLAN) TITL(1) = 10HWING/TAIL  FORCES 00501
C      CALL PRNTBL(TITL,JVP,DELCP,1,NROWS, MYBW,IPNTRM)  FORCES 00502
C      IF (NSURF.EQ.1 .OR. COPLAN) GO TO 572  FORCES 00503
C      TITL(1) = 0H   TAIL  FORCES 00504
C      CALL PRNTBL(TITL,JVP,DELCP,IXBUT,MXBUT,MYBT,IPNTRM(1,ICVLAP+1))  FORCES 00505
C      572 CONTINUE  FORCES 00506
C
C      IF(.NOT. (WTBL.OR.PRSL)) GO TO 600  FORCES 00507
C
C-----+
C      EXPAND BOX LIFTS FOR WRITING ON TAPE. WRTETP FORMAT ONLY  FORCES 00508
C      IF (MBRIT) WRITE (NT6,9999)  FORCES 00509
C      INITIALIZE COUNTERS FOR PASSING OVER ARRAY BACKWARDS  FORCES 00510
C      IJKL = CURRENT LOCATION IN INPUT (COMPRESSED) ARRAY  FORCES 00511
C      IJ = CURRENT LOCATION IN OUTPUT (EXPANDED) ARRAY  FORCES 00512
C      IJFST = FIRST LOCATION FOR CURRENT ROW IN INPUT ARRAY  FORCES 00513
C      IJPRV = FIRST LOCATION FOR PREVIOUS ROW IN INPUT ARRAY  FORCES 00514
C
C      IRW = MXB  FORCES 00515
C      IJFST = IPNTRM(1,MXB)  FORCES 00516
C      LOCFST = IPNTRM(2,MXB)  FORCES 00517
C      IJPRV = IPNTRM(1,MXB+1)  FORCES 00518
C      IJKL = IJPRV  FORCES 00519
C      IJ = MXBOXES - MYB + IJPRV-IJFST + LOCFST  FORCES 00520
C      DO 573 I = 1, MXBOXES  FORCES 00521
C      RNBF(I) = (0.,0.)  FORCES 00522
C
C      573 CONTINUE  FORCES 00523
C
C      LOOP BACK HERE ON ROWS, AND ON CHORDS WITHIN A ROW  FORCES 00524
C
C      500 CONTINUE  FORCES 00525
C      IJ = IJ - 1  FORCES 00526
C      IJKL = IJKL - 1  FORCES 00527
C
C      IJ = IJ - 1  FORCES 00528
C      IJKL = IJKL - 1  FORCES 00529
C
C      IJ = IJ - 1  FORCES 00530
C      IJKL = IJKL - 1  FORCES 00531
C
C      IJ = IJ - 1  FORCES 00532
C      IJKL = IJKL - 1  FORCES 00533
C
C      IJ = IJ - 1  FORCES 00534
C      IJKL = IJKL - 1  FORCES 00535
C
C      IJ = IJ - 1  FORCES 00536

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BL = BXLIFT(IJKL)	FORCES 00537
BXLIFT(IJKL) = (0.,0.)	FORCES 00538
RWBF(IJ) = BL	FORCES 00539
IF (IJKL .GT. IJFST) GO TO 583	FORCES 00540
C END OF LOOP ON CHORDS WITHIN ONE ROW. STEP TO NEXT ROW	FORCES 00541
IJPRV = IJFST	FORCES 00542
LOCPV = LOCFST	FORCES 00543
C LOOP BACK HERE ON EMPTY ROWS (COPLANAR CASE)	FORCES 00544
582 CONTINUE	FORCES 00545
IROW = IRW - 1	FORCES 00546
IJ = IJ - MYB	FORCES 00547
C DETERMINE WHETHER DONE -	FORCES 00548
IF (IROW .EQ. 0) GO TO 584	FORCES 00549
C IS THE ROW EMPTY -	FORCES 00550
IF (IPNTRM(1,IROW) .EQ. IJPRV) GO TO 582	FORCES 00551
IJFST = IPNTRM(1,IROW)	FORCES 00552
LOCFST = IPNTRM(2,IROW)	FORCES 00553
IJ = IJ - LOCPV + IJPRV-IJFST + LOCFST	FORCES 00554
GO TO 580	FORCES 00555
C END OF LOOP ON ROWS.	FORCES 00556

C 584 CONTINUE	FORCES 00557
C ARE THE SECTION LIFTS TO BE PRINTED OR WRITTEN -	FORCES 00558
IF (.NOT. PRSL) GO TO 595	FORCES 00559
C COMPUTE SECTION LIFTS	FORCES 00560
TLIFT = (0.,0.)	FORCES 00561
TLIFT1 = (0.,0.)	FORCES 00562
TLIFT2 = (0.,0.)	FORCES 00563
DO 590 JCCL = 1,MYB	FORCES 00564
BL = (0.,0.)	FORCES 00565
BL2 = (0.,0.)	FORCES 00566
IROW = 0	FORCES 00567
DO 587 IJ = JCCL,NBOXES,MYB	FORCES 00568
IROW = IRW + 1	FORCES 00569
IF (IROW.GT.TEXLOC(JCCL)) GO TO 586	FORCES 00570
BL = BL + RWBF(IJ)	FORCES 00571
GO TO 587	FORCES 00572
586 BL2 = BL2 + RWBF(IJ)	FORCES 00573
587 CONTINUE	FORCES 00574
TLIFT1 = TLIFT1 + BL	FORCES 00575
TLIFT2 = TLIFT2 + BL2	FORCES 00576
SLIFT(JCCL+MYBW) = BL2	FORCES 00577
SLIFT(JCCL) = BL	FORCES 00578
590 CONTINUE	FORCES 00579
C	FORCES 00580
IF (.NOT. PRSL) GO TO 595	FORCES 00581
C PRINT SECTION LIFTS AND TOTAL LIFT	FORCES 00582
CALL PRNTSL(JVP,SLIFT,TLIFT1,TLIFT2,MYBW,MYBT)	FORCES 00583
C	FORCES 00584
595 CONTINUE	FORCES 00585
IF (.NOT. PRCM) GO TO 599	FORCES 00586
DO 596 I=1,NTVPS	FORCES 00587
SECHCM(I) = SECHCM(I) * TWOBET	BCSFRB 00023
596 CONTINUE	BCSFRB 00024
NCH = -JVP	BCSFRB 00025
CALL PRNTSL(NCH,SECHCM,TLIFT1,TLIFT2,MYBW,MYBT)	BCSFRB 00026
	BCSFRB 00027
	BCSFRB 00028

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C
      599 CONTINUE
      IF (.NOT. WTEL) GO TO 600
C           WRITE BOX LIFTS ONTO THE BINARY OUTPUT FILE
      ITYPE = 7HCOMPLEX
      M = MXB
      K = - M
      N = MYB
      MXARRY = 7HBXLJFTS
      ID(2) = 1000000+IKVAL + JVP
      CALL WRTEMX( NOUTP, MXWRIT,RANDOU, NFS, NMS,LS, NMR,LWS, K, ID,
      1           RWBF,ITYPE, M, N, PARM, IRR )
      IF (IRR .NE. 0) GO TO 922
C
      600 CONTINUE
C           END OF LOOP ON VELOCITY POTENTIALS
C
C           STORE THE ROW OF GENERALIZED AIRFORCES INTO THE FULL MATRIX
      IJ = NM
      DO 620 JVP = 1,NVPS
      GENAF(IJ) = AFROW(JVP)*TWOBT
      GPFAC(IJ) = BS3BT * REAL(GENAF(IJ))
      GPFAC(IJ) = 0.
      IF(XKL.NE.0.) GPPAFC(IJ) = BKS4BT * AIMAG(GENAF(IJ))
      IJ = IJ + NMODES
      620 CONTINUE
      630 CONTINUE
C
C           IF(.NOT.WTGNMF) GO TO 670
      K = NMODES
      M = NMODES
      N = NMODES
      ID(2) = IKVAL
      CALL WRTEMX( NOUTP, MXWRIT,RANDOU, NFS, NMS,LS, NMR,LWS, K, ID,
      1           GENAF, ITYPE, M,N, PARM, IRR )
      IF (IRR .NE. 0) GO TO 922
C
C           ARE THE FORCES TO BE PRINTED -
      670 CONTINUE
      IF (.NOT. PRGNMF) GO TO 700
C
      CALL PRNTAF(GENAF,PRGNAC,GPFAC,GPPAFC)
      700 CONTINUE
C
      REWIND MODESC
      NNSPCE = 1
      750 CONTINUE
C
      RETURN
C
C           DIAGNOSTICS - ALL CALL FLUSH
C
C           READING FROM SCRATCH FILE
      8010 CONTINUE
      WRITE (NTE,9100) IGEOSC
      GO TO 950

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BCSFNB	00029
BC3FNB	00030
FORCES	00588
FORCES	00589
FORCES	00590
FORCES	00591
FORCES	00592
FORCES	00593
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FORCES	00595
FORCES	00596
FORCES	00597
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FORCES	00599
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FORCES	00639
FORCES	00640
FORCES	00641
FORCES	00642

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6020 CONTINUE
    WRITE (NT6,9120) MODESC
    GO TO 550
8030 CONTINUE
    WRITE (NT6,9180) IVPSC
    GO TO 950
C
6040 CONTINUE
    WRITE (NT6,9140) ITSLSC
    GO TO 950
C
C           WRITING ON THE OUTPUT TAPE
922 CONTINUE
    WRITE (NT6,9220) NOUTP
    GO TO 952
928 CONTINUE
    WRITE (NT6,9280) NOUTP
    GO TO 952
C           INCORRECT DIMENSIONS READ
930 CONTINUE
    I = 1
    GO TO 932
931 I = 2
932 WRITE (NT5,930G) I
    IF(MXREAD) GO TO 950
    GO TO 962
C           ERROR DETECTED READING A MATRIX
950 CONTINUE
    WRITE (NT6,9500) IRR
    IF(MXREAD) GO TO 950
    GO TO 962
C           ERROR DETECTED WRITING A MATRIX
952 CONTINUE
    WRITE (NT6,9520) IRR
    IF(MXWRIT) GO TO 950
    GO TO 962
C           MATRIX DESCRIPTION
960 CONTINUE
    WRITE (NT6,9600) (ID(I),I=1,10),(ID(I),I=1,10)
    WRITE (NT6,9622) FARM,FARM
    WRITE (NT6,9614) NMR,NMR,LRS,LWS
    GO TO 964
962 WRITE (NT6,9620) ID(1),ID(2)
    WRITE (NT6,9622) FARM,FARM
    WRITE (NT6,9624) NFS,NMS
964 WRITE (NT6,9640) ITYPE,M,N
    WRITE (NT6,9650) MXARRY
    GO TO 990
970 CONTINUE
    WRITE (NT6,9700) IROW, JCOL
    GO TO 990
C
990 CC. INUE
    WRITE (NT6,9900)
C
    CALL FLUSH(1)
C
FORCES 00643
FORCES 00644
FORCES 00645
FORCES 00646
FORCES 00647
FORCES 00648
FORCES 00649
FORCES 00650
FORCES 00651
FORCES 00652
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FORCES 00699

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C	DIAGNOSTIC FORMATS	
9100	FORMAT(47H0*** ERROR WHILE READING GEOMETRY SCRATCH FILE ,A10,	FORCES 00700
1	4H ***)	FORCES 00701
9120	FORMAT(44H0*** ERROR WHILE READING MODES SCRATCH FILE ,A10,	FORCES 00702
1	4H ***)	FORCES 00703
9140	FORMAT(54H0*** ERROR WHILE READING THICKNESS SLOPE SCRATCH FILE	FORCES 00704
1	A10, 4H ***)	FORCES 00705
9180	FORMAT(51H0*** ERROR WHILE READING VELOCITY POTENTIAL SCRATCH	FORCES 00706
1	SH FILE ,A10,4H ***)	FORCES 00707
	DATA XINOFF /2* 6000000000200377777B /	FTNXI 00082
9220	FORMAT(49H0*** ERROR WHILE WRITING BOX LIFTS ON OUTPUT TAPE,I2,	FORCES 00711
1	4H ***)	FORCES 00712
9280	FORMAT(56H0*** ERROR WHILE WRITING GENERALIZED AIRFORCES ON OUTPUT	FORCES 00713
1	SH TAPE,I2,4H ***)	FORCES 00714
9300	FORMAT(1H0, 48H*** MATRIX READ ERROR. THE M DIMENSION SHOULD	FORCES 00715
1	4H BE ,I2, 4H ***)	FORCES 00716
9500	FORMAT(16H0 *** ERROR CODE ,15, 28H WHILE READING THE FOLLOWING	FORCES 00717
1	11H MATRIX ***)	FORCES 00718
9520	FORMAT(16H0 *** ERROR CODE ,15, 28H WHILE WRITING THE FOLLOWING	FORCES 00719
1	11H MATRIX ***)	FORCES 00720
9600	FORMAT(5X, #MATRIX ID = *, 10A10 / (20X,10A10))	FTNXI 00083
9614	FORMAT(5X,22HMATRIX INDEX (NAME) = ,15,2H (A10,1H) /	FORCES 00722
1	5X,33HLEVEL NUMBER READ (OR WRITTEN) = 02,3H, (,02,1H))	FTNXI 00084
9620	FORMAT(5X, #MATRIX ID = *, A10, I10)	FORCES 00724
9622	FORMAT(5X,11HPARAMETERS, 10E11.3 /10X, 9H(INTEGER), I7,9I11)	FORCES 00725
9624	FORMAT(5X,19HFILE SPACING = ,I3, 19H, MATRIX SPACING = ,I3)	FORCES 00726
9640	FORMAT(5X, #MATRIX TYPE - *,;10, *, DIMENSIONED (*I4,* X*,I4,*))	FORCES 00727
9650	FORMAT(5X, #ARRAY - *, A10)	FORCES 00728
9700	FORMAT(37H0*** POINTER ARRAY EXCEEDED FOR BOX (,I4,1H,I4,SH) ***)	FORCES 00729
9800	FORMAT(48H0*** ERROR OCCURRED DURING GENERALIZED AIRFORCES	FORCES 00730
1	17H CALCULATIONS ***)	FORCES 00731
9899	FORMAT(54H0*** W A R N I N G - BOX LIFTS CANNOT BE WRITTEN IN	FORCES 00732
1	16HSNARK FORMAT ***)	FORCES 00733
C		FORCES 00734
	END	FORCES 00735

```

SUBROUTINE DCODER(IBOX,LBOX, IA,JA, IL,JL, SUBD, ICODE)      DCODAF 00002
DIMENSION IBOX(LBOX,1), ICODE(1)                                DCODAF 00003
DCODAF 00004
DCODAF 00005
DCODAF 00006
DCODAF 00007
DCODAF 00008
DCODAF 00009
DCODAF 00010
DCODAF 00011
DCODAF 00012
DCODAF 00013
DCODAF 00014
DCODAF 00015
DCODAF 00016
DCODAF 00017
DCODAF 00018
DCODAF 00019
DCODAF 00020
GEOMTY 00002
GEOMTY 00003
GEOMTY 00004
GEOMTY 00005
GEOMTY 00006
DCODAF 00022
DCODAF 00023
DCODAF 00024
DCODAF 00025
DCODAF 00026
DCODAF 00027
DCODAF 00028
DCODAF 00029
DCODAF 00030
DCODAF 00031
DCODAF 00032
DCODAF 00033
DCODAF 00034
DCODAF 00035
DCODAF 00036
DCODAF 00037
DCODAF 00038
DCODAF 00039
DCODAF 00040
DCODAF 00041
DCODAF 00042
DCODAF 00043
DCODAF 00044
DCODAF 00045
DCODAF 00046
DCODAF 00047
DCODAF 00048
DCODAF 00049
DCODAF 00050
DCODAF 00051
DCODAF 00052
DCODAF 00053
DCODAF 00054

C     IBOX - ARRAY OF BOX CODES IN PACKED WORD FORMAT
C     LBOX - ROW DIMENSION OF BOX CODES ARRAY
C     IA   - I-TH INDEX OF FIRST CODE TO RETRIEVE
C     JA   - J-TH INDEX OF FIRST CODE TO RETRIEVE
C     IL   - LAST BOX CODE ON THE JA-TH CHORD TO RETRIEVE
C     JL   - LAST BOX ON THE IA-TH ROW TO RETRIEVE
C     SUBD - .T., SUBDIVIDED BOX CODES DESIRED, .F. UNSUBDIVIDED.
C     ICODE - ARRAY INTO WHICH BOX CODE WILL BE STORED.

C     COMMENT ON USAGE
C     BOX CODES CAN BE RETRIEVED FOR ONE BOX, A ROW OR PART OF
C     A ROW, OR A COLUMN OR PART OF A COLUMN. A ROW AND COLUMN CAN
C     NOT BE RETRIEVED AT THE SAME TIME. IF ONLY 1 BOX IS DESIRED
C     SET IL = IA AND JL = JA. IF BOTH IL .NE. IA AND JL .NE.
C     JA, ONE ROW WILL BE RETURNED, IL BEING IGNORED.

C     COMMON /GEOMTY/ COPLAN,NSUBDV,XSUBDV,NSUBD2,NSUBCN,NSURF,
1           B1,B1BETA,B1S,B1BTAS,WLAX,WLAZ,PSTW,
2           MXBW,MXBBW,MYBW,MYBBW,MXBW,MYSBW,MYBSBW,
3           IXBW,XCENTR

LOGICAL COPLAN
LOGICAL SUBD
INTEGER SHIFT
DATA NBWRD /20/
MASK = 7
IB = 1
IF (SUBD) GO TO 50
I = NSUBDV * (IA-1) + IXBW
J = NSUBDV * (JA-1) + NSUBCN
ISKIP = NSUBDV
IEND = NSUBDV * (IL-1) + IXBW
JEND = NSUBDV * (JL-1) + NSUBCN
GO TO 60
50 CONTINUE
I = IA
J = JA
ISKIP = 1
IEND = IL
JEND = JL
60 CONTINUE
IF (JL .EQ. JA) GO TO 1100

C     PROGRAM WILL RETRIEVE NI BOXES FROM ROW I
100 CONTINUE
DO 1000 JJ = J,JEND,ISKIP
JSB =(JJ-1)/NBWRD + 1
IJWORD = IBOX(I,JSB)
JB = (NBWRD - MOD(JJ,NBWRD)) * 3
IF (JB.EQ.60) JB = 0
C     JB = NUMBER OF BITS TO SHIFT LEFT.
IJMASK= SHIFT(MASK,JB)
IJCODE = IJWORD.AND.IJMASK
NJB = -JB
ICODE(IB) = SHIFT(IJCODE,NJB)


```

```

IB = IB + 1
1000 CONTINUE
GO TO 3000
C
C      PROGRAM WILL RETRIEVE NU BOXES FROM CHORD J
1100 CONTINUE
JSB = (J-1)/NBWRD + 1
JB = (NBWRD - MOD(J,NBWRD)) * 3
IF(JB.EQ.60) JB = 0
IJMASK = SHIFT(MASK,JB)
NUB = -JB
DO 2000 II = I,IEND,ISKIP
IJWORD = IBOX(II,JSB)
IJCODE = IJWORD.AND.IJMASK
ICODE(IB) = SHIFT(IJCODE,NUB)
IB = IB + 1
2000 CONTINUE
C
3000 CONTINUE
RETURN
END

```

```

DCODAF 00055
DCODAF 00056
DCODAF 00057
DCODAF 00058
DCODAF 00059
DCODAF 00060
DCODAF 00061
DCODAF 00062
DCODAF 00063
DCODAF 00064
DCODAF 00065
DCODAF 00066
DCODAF 00067
DCODAF 00068
DCODAF 00069
DCODAF 00070
DCODAF 00071
DCODAF 00072
DCODAF 00073
DCODAF 00074
DCODAF 00075

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SUBROUTINE PRNTBL(TITL,IMODE,ARRAY,IXB,MXB,MYB,IPNTRM)          PRNTBL 00002
C
C   PRINTS BOX LIFTS.  USES /RWBUFF/ FOR INTERMEDIATE SCRATCH      PRNTBL 00003
C   IMODE - MODE SHAPE NUMBER                                     PRNTBL 00004
C   ARRAY - ARRAY TO BE PRINTED                                  PRNTBL 00005
C   IPNTRM - POINTER ARRAY FOR ROWS IN -ARRAY-                  PRNTBL 00006
C   NPNTRS - NUMBER OF POINTERS                                PRNTBL 00007
C
C   COMPLEX ARRAY(1)
DIMENSION TITL(3)                                              PRNTBL 00008
DIMENSION IPNTRM(2,50)                                         PRNTBL 00009
C
COMMON /CONTRL/ PREVEX,OMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT, CONTRL 00002
1           DEFAULT                                              CONTRL 00003
LOGICAL     PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT                 CONTRL 00004
COMMON /PROBLW/ XMACH,NMODOES,NTSLOP,NKVALS,SMOOTH,NDEC,CDFIT, PROBLW 00002
1           EXAIC,SUBDV,PLYWOOD                               PROBLW 00003
LOGICAL     SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD                 PROBLW 00004
COMMON /KVAL / IKVAL,XKVAL(20), XKS(20)                      KVAL  00002
COMMON /FILES / NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP, FILES 00002
1           IOUFSP,MODESC,IVPSC,IGEOSC,IWFSC,IAICSC            FILES 00003
C
C   INTEGER PAGE
DIMENSION S(1),D(1)                                              PRNTBL 00019
EQUIVALENCE (S,BUFF),(D,BUFF(1251))                           PRNTBL 00020
COMPLEX TLIFT
COMPLEX TLIFT1,TLIFT2
DIMENSION PC(2), IPNT(2)                                         PRNTBL 00021
EQUIVALENCE (IPNT, TLIFT)                                       PRNTBL 00022
COMMON /RWBUFF/ BFCCDE,IBFCNT, BUFF(3280)                      RWBUFF 00002
DATA PC / 10H PAGE CONTI,4HNUED /
DATA BLANK /1H /
DATA XINIT / -1.0 /
DATA LINEMX /50 /
C
C
XXVL = XKVAL(IKVAL)
IF(XKS(IKVAL).NE.XINIT) XXVL = XKS(IKVAL)                   PRNTBL 00033
PAGE = 0
N = 1
M = 4
IF(M.GT.MYB) M = MYB                                         PRNTBL 00034
C
100 LINE = 100
200 DO 1400 I=IXB,MXB
    DO 300 J=N,M
        S(J) = 0.0
        D(J) = 0.0
300 CONTINUE
    IF(LINE.LE.50) GO TO 900
    PAGE = PAGE + 1
    LINE = 4
    WRITE (NT6,9001) TITLE,TITL,XMACH,XXVL,IMODE
    IF(PAGE.EQ.1) GO TO 700
    WRITE (NT6,9005) PC
    GO TO 800
700 WRITE (NT6,9005)

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800 CONTINUE          PRNTBL 00054
    WRITE(NT6,6006) (BLANK,J,J=N,M)
    WRITE(NT6,6007) (BLANK, J=N,M)

C
900 CONTINUE          PRNTBL 00055
    JS = IPNTRM(2,I)
    IF(JS.LE.0) GO TO 1400
    IDX = IPNTRM(1,I)
    JE = IPNTRM(1,I+1) - IDX + JS-?
    IF(JE.EQ.0) GO TO 1400
    DO 1000 J=JS,JE
    S(J) = REAL(ARRAY(IDX))
    D(J) = AIMAG(ARRAY(IDX))
    IDX = IDX + 1
1000 CONTINUE          PRNTBL 00056
    DO 1200 J =N,M
    IF(S(J)) 1300,1100,1300
1100 CONTINUE          PRNTBL 00057
    IF(D(J)) 1300,1200,1300
1200 CONTINUE          PRNTBL 00058
    GO TO 1400
1300 WRITE(NT6,9013) I,(S(J),D(J),J=N,M)
    LINE = LINE + 1
1400 CONTINUE          PRNTBL 00059
C
    M = M+4            PRNTBL 00060
    N = N+4            PRNTBL 00061
    IF(N.GT.MYB) GO TO 1500
    IF(M.GT.MYB) M = MYB
    IF(LINE.GT.45) GO TO 100
    WRITE(NT6,6006) (BLANK,J,J=N,M)
    WRITE(NT6,6007) (BLANK, J=N,M)
    LINE = LINE+3
    GO TO 200
1500 CONTINUE          PRNTBL 00062
    RETURN              PRNTBL 00063
9001 FORMAT(1H1,20X,8A10,/50X,3A10,/ 46X,7H( MACH F5.3,5X,15HRED.FREQ.
    1  *  ,F8.5, * )* /52X,44HDE SHAPE*, I3)      PRNTBL 00064
9005 FORMAT(44X,42(IH-),20X,A10,A4)             PRNTBL 00065
9006 FORMAT(4HROW,A1,14X,SHCHORD,I3,3(A1,22X,SHCHORD,I3))  PRNTBL 00066
9007 FORMAT(3X, 4(A1,9X,4HREAL,8X,9HIMAGINARY) )  PRNTBL 00067
9013 FORMAT(14,8E16.8)                           PRNTBL 00068
    END                PRNTBL 00069
                                PRNTBL 00070
                                PRNTBL 00071
                                PRNTBL 00072
                                PRNTBL 00073
                                PRNTBL 00074
                                PRNTBL 00075
                                PRNTBL 00076
                                PRNTBL 00077
                                PRNTBL 00078
                                PRNTBL 00079
                                PRNTBL 00080
                                PRNTBL 00081
                                PRNTBL 00082
                                PRNTBL 00083
                                PRNTBL 00084
                                PRNTBL 00085
                                PRNTBL 00086
                                PRNTBL 00087
                                PRNTBL 00088
                                PRNTBL 00089
                                PRNTBL 00090
                                PRNTBL 00091
                                PRNTBL 00092
                                PRNTBL 00093
                                PRNTBL 00094
                                PRNTBL 00095
                                PRNTBL 00096

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SUBROUTINE PRNTSL(IMODE,SLIFT,TLIFT1,TLIFT2,MYBW,MYBT)          PRNTBL 00097
C
C      PRINTS THE SECTION LIFTS AND TOTAL LIFTS                  PRNTBL 00098
C
C      IMODE - MODE SHAPE NUMBER                                PRNTBL 00099
C      SLIFT - SECTION LIFT ARRAY                               PRNTBL 00100
C      TLIFT1 - WING TOTAL LIFT                                PRNTBL 00101
C      TLIFT2 - TAIL TOTAL LIFT                               PRNTBL 00102
C
C      IF IMODE IS NEGATIVE THE PROGRAM WILL OUTPUT SECTION MOMENTS PRNTBL 00103
C
C
C      COMMON /PROBLM/ XMACH,MODES,NTSLOP,NKVALS,SMOOTH,NDEG,CRDFIT,    PROBLM 00002
1           EXAIC,SUBDV,PLYWOOD                                PROBLM 00003
LOGICAL      SMOOTH,CRDFIT,EXAIC,SUBDV,PLYWOOD                PROBLM 00004
COMMON /CTRL/  PREVEX,XMACH, TITLE(8), PRVGEOM,PRVMODE,DIHW,DIHT,   CTRL  00002
1           DEFAULT                                         CTRL  00003
LOGICAL      PRVGEOM,PRVMODE,DIHW,DIHT,DEFAULT               CTRL  00004
COMMON /KVAL /  IKVAL,XXVAL(20), XKS(20)                      KVAL   00002
COMMON /FILES /  NT5,NT6,INTAPE,INFSP,NPLAIC,NSPAIC,NOUTP,       FILES 00002
1           IOUTSP,MODESC,IVPSC,IGEOSC,IWTFSC,IAICSC            FILES 00003
C
COMPLEX SLIFT()
COMPLEX TLIFT1,TLIFT2,TLIFT                                PRNTBL 00109
LOGICAL PRCM
DATA BLANK/1H/
DATA XINIT / -1.0 /
C
IF (IMODE.LT.0) GO TO 100
PRCM = .FALSE.
GO TO 200
100 CONTINUE
PRCM = .TRUE.
IMODE = -IMODE
200 CONTINUE
C
XXVL = XXVAL(IKVAL)
IF (XKS(IKVAL).NE.XINIT) XXVL = XKS(IKVAL)
IF (PRCM) GO TO 300
WRITE (NT5,6010) XMACH,XXVL,IMODE
GO TO 400
300 CONTINUE
WRITE (NT5,6010) XMACH,XXVL,IMODE
400 CONTINUE
WRITE (NT5,6008)
WRITE (NT5,6005)
WRITE (NT5,6020)
WRITE (NT5,6007) BLANK,BLANK,BLANK,BLANK
DO 600 I=1,MYBW,4
II = I + 3
IF (II .GT. MYBW) II = MYBW
WRITE (NT5,6030) I,(SLIFT(IJ),IJ=I,II)
600 CONTINUE
IF (PRCM) GO TO 650
WRITE (NT5,6023) TLIFT1
IF (MYBT.EQ.0) GO TO 800
C

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        WRITE (NT6,6009)
        GO TO 675
650 CONTINUE
        IF (MYBT.EQ.0) GO TO 800
        WRITE (NT6,6009)
675 CONTINUE
        WRITE (NT6,6020)
        WRITE (NT6,6021) BLANK,BLANK,BLANK,BLANK
        DO 700 I=1,MYBT,4
        II = I + 3
        IF (II .GT. MYBT) II = MYBT
        I2 = I + MYBW
        I12 = II + MYBW
        WRITE (NT6,6030) I,(SLIFT(IJ),IJ=I2,I12)
700 CONTINUE
C
        IF (PRCM) GO TO 800
        WRITE (NT6,6024) TLIFT2
        TLIFT = TLIFT1 + TLIFT2
        WRITE (NT6,6025) TLIFT
800 CONTINUE
        RETURN
C
C
6005 FORMAT(44X,32(1H-),30X,A10,A4)
6007 FORMAT(3X, 4(A1,9X,4HREAL,8X,9HIMAGINARY) )
6008 FORMAT(58X,4#WING# )
6009 FORMAT(1HD,// 53X,4SECTION LIFTS#/ 58X,*TAIL#/ 44X,32(1H-)/ )
6010 FORMAT(1HD,53X, 14SECTION LIFTS /44X,* ( MACH *,F3.3,5X,
     1 #RED. FREQ. == F8.5,* )* / 52X,* MODE  SHAPE*,13)
6020 FORMAT(5H0CHORD)
6023 FORMAT(1HD,44X,* TOTAL LIFT - WING */ 1HD 40X,2E16.8 )
6024 FORMAT(1HD,44X,* TOTAL LIFT - TAIL */ 1HD 40X,2E16.8 )
6025 FORMAT(1HD,53X,* TOTAL LIFT */ 1HD,40X,2E16.8 )
6030 FORMAT(14,8E16.8)
9009 FORMAT(1HD// 45X,*SECTIONAL MOMENT COEFFICIENTS*/58X,*TAIL*/#
     1 44X, 32(1H-)/ )
9010 FORMAT(1HD///,45X,*SECTIONAL MOMENT COEFFICIENTS*/44X,* (MACH *,
     1  F3.3,5X,*RED. FREQ. == F8.5,* )* /52X,* MODE  SHAPE*,13)
      END

```

PRNTBL	00130
BCSFRB	00050
BCSFRB	00051
BCSFRB	00052
BCSFRB	00053
BCSFRB	00054
PRNTBL	00131
PRNTBL	00132
PRNTBL	00133
PRNTBL	00134
PRNTBL	00135
PRNTBL	00136
PRNTBL	00137
PRNTBL	00138
PRNTBL	00139
PRNTBL	00140
BCSFRB	00055
PRNTBL	00141
PRNTBL	00142
PRNTBL	00143
BCSFRB	00056
PRNTBL	00144
PRNTBL	00145
PRNTBL	00146
PRNTBL	00147
PRNTBL	00148
PRNTBL	00149
PRNTBL	00150
PRNTBL	00151
PRNTBL	00152
PRNTBL	00153
PRNTBL	00154
PRNTBL	00155
PRNTBL	00156
PRNTBL	00157
BCSFRB	00057
BCSFRB	00058
BCSFRB	00059
BCSFRB	00060
PRNTBL	00158

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SUBROUTINE PRNTAF(ARRAY,PRGNAC,GPAFC,GPPAFC)          PRNTAF 00002
C      PRINTS COMPLEX GENERALIZED AIRFORCES, FROM COMPACT FORTRAN    PRNTAF 00003
C      STORAGE                                                       PRNTAF 00004
C
C      ARRAY - ARRAY OF GENERALIZED AIR FORCES                  PRNTAF 00005
C      PRGNAC - LOGICAL FLAG FOR PRINT OPTION                   PRNTAF 00006
C      GPAFC - AGARD GENERALIZED AERODYNAMIC COEFFICIENT     PRNTAF 00007
C      GPPAFC - AGARD GENERALIZED AERODYNAMIC COEFFICIENT     PRNTAF 00008
C
C      COMPLEX ARRAY(1)
DIMENSION GPAFC(1),GPPAFC(1)                         PRNTAF 00009
LOGICAL PRGNAC                                         PRNTAF 00010
C
C      DIMENSION PC(2)
COMMON /PRCBLM/ XMACH,NMODES,NTSLOP,NKVALS,SMOOTH,NDEG,CDFIT,   PRNTAF 00011
1           EXAIC,SUBDV,PLYWOOD                           PROBLM 00002
LOGICAL SMOOTH,CDFIT,EXAIC,SUBDV,PLYWOOD             PROBLM 00003
COMMON /KVAL / IKVAL,XXVAL(20),XKS(20)                KVAL   00004
COMMON /FILES / NTS,NT6,INTA'E,INFSP,INPLAIC,NSFAIC,NOUTF,   FILES  00002
1           IOUFSP,MODESC,?VRSC,IGEOSC,IWTFSC,IAICSC        FILES  00003
INTEGER PAGE
DATA PC / 10HPAGE CONTI,4HNUED /
DATA BLANK / 1H /
DATA XINIT / -1.0 /
XXVL = XXVAL(IKVAL)
IF(XKS(IKVAL).NE.XINIT) XXVL = XKS(IKVAL)
LINEX = 50
PAGE = 0
J1 = 1
J2 = 4
IF (NMODES .LT. J2) J2 = NMODES
IJ1 = 1
IJ2 = (J2-1)*NMODES + 1
C
100 LINE = LINEX + 10
110 DO 200 I = 1,NMODES
  IF (LINE .LE. LINEX) GO TO 170
  PAGE = PAGE + 1
  LINE = 8
  WRITE (NT6,6001) XMACH,XXVL
  IF (PAGE .EQ. 1) GO TO 150
  WRITE (NT6,6005) PC
  GO TO 180
150 WRITE (NT6,6005)
160 CONTINUE
C
  WRITE (NT6,6006) (BLANK,J, J=J1,J2)
  WRITE (NT6,6007) (BLANK, J=J1,J2)
C
170 CONTINUE
  WRITE (NT6,6010) I, (ARRAY(IJ),IJ= IJ1,IJ2,NMODES)
  LINE = LINE + 1
  IJ1 = IJ1 + 1
  IJ2 = IJ2 + 1
200 CONTINUE
C
  J1 = J1 + 4

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J2 = J2 + 4          PRNTAF 00056
IF (J1 .GT. NMODES) GO TO 300  PRNTAF 00057
IF (J2 .GT. NMODES) J2 = NMODES  PRNTAF 00058
IJ1 = IJ1 + 3*NMODES  PRNTAF 00059
IJ2 = IJ1 + (J2-J1)*NMODES  PRNTAF 00060
IF (LINE .GT. LINEMX -6) GO TO 100  PRNTAF 00061
WRITE (NT6,6006) (BLANK,J, J = J1,J2)  PRNTAF 00062
LINE = LINE+3  PRNTAF 00063
GO TO 110  PRNTAF 00064
C  PRNTAF 00065
300 CONTINUE  PRNTAF 00066
C  PRNTAF 00067
C      PRINT THE GENERALIZED AERODYNAMIC COEFFICIENTS  PRNTAF 00068
C      IF DESIRED.  PRNTAF 00069
C  PRNTAF 00070
IF(.NOT.PRNAC) GO TO 1400  PRNTAF 00071
PAGE = 0  PRNTAF 00072
DO 1300 IPR = 1,2  PRNTAF 00073
J1 = 1  PRNTAF 00074
J2 = 8  PRNTAF 00075
IF (NMODES.LT.J2) J2 = NMODES  PRNTAF 00076
IJ1 = 1  PRNTAF 00077
IJ2 = (J2-1)*NMODES + 1  PRNTAF 00078
C  PRNTAF 00079
1100 LINE = LINEMX + 10  PRNTAF 00080
1110 DO 1200 I=1,NMODES  PRNTAF 00081
    IF(LINE.LE.LINEMX) GO TO 1170  PRNTAF 00082
    PAGE = PAGE + 1  PRNTAF 00083
    LINE = 8  PRNTAF 00084
    WRITE (NT6,7001) XMACH,XVRL  PRNTAF 00085
    IF(IPR.EQ.2) GO TO 1140  PRNTAF 00086
    IF(PAGE.EQ.1) GO TO 1130  PRNTAF 00087
    WRITE (NT6,7005) PC  PRNTAF 00088
    GO TO 1180  PRNTAF 00089
1130 CONTINUE  PRNTAF 00090
    WRITE(NT6,7005)  PRNTAF 00091
    GO TO 1180  PRNTAF 00092
1140 CONTINUE  PRNTAF 00093
    IF (PAGE.EQ.1) GO TO 1150  PRNTAF 00094
    WRITE (NT6,7015) PC  PRNTAF 00095
    GO TO 1180  PRNTAF 00096
1150 WRITE(NT6,7015)  PRNTAF 00097
1160 CONTINUE  PRNTAF 00098
C  PRNTAF 00099
    WRITE (NT6,7006)  PRNTAF 00100
    WRITE (NT6,7007) (J,J=J1,J2)  PRNTAF 00101
C  PRNTAF 00102
1170 CONTINUE  PRNTAF 00103
    IF(IPR.EQ.2) GO TO 1180  PRNTAF 00104
    WRITE(NT6,6010) I,(GPAFC(IJ),IJ=IJ1,IJ2,NMODES)  PRNTAF 00105
    GO TO 1180  PRNTAF 00106
1180 CONTINUE  PRNTAF 00107
    WRITE(NT6,6010) I,(GPPAFC(IJ),IJ=IJ1,IJ2,NMODES)  PRNTAF 00108
1190 CONTINUE  PRNTAF 00109
    LINE = LINE + 1  PRNTAF 00110
    IJ1 = IJ1 + 1  PRNTAF 00111
    IJ2 = IJ2 + 1  PRNTAF 00112

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1200 CONTINUE          PRNTAF 00113
C                      PRNTAF 00114
J1 = J1 + 8           PRNTAF 00115
J2 = J2 + 8           PRNTAF 00116
IF(J1.GT.NMNODES) GO TO 1300  PRNTAF 00117
IF(J2.GT.NMNODES) J2 = NMNODES  PRNTAF 00118
IJ1 = IJ1 + 3*NODES  PRNTAF 00119
IJ2 = IJ1 + (J2-J1)* NMNODES  PRNTAF 00120
IF (LINE.GT.LINEMX-6) GO TO 1100  PRNTAF 00121
WRITE (NT6,7006)          PRNTAF 00122
LINE = LINE + 3          PRNTAF 00123
GO TO 1110             PRNTAF 00124
C
1300 CONTINUE          PRNTAF 00125
1400 CONTINUE          PRNTAF 00126
      RETURN            PRNTAF 00127
C                      PRNTAF 00128
8001 FORMAT(1H1,51X,18HGENERALIZED FORCES /44X,6H(MACH ,F5.3,5X,*RED. F PRNTAF 00130
     1REQ. =* F8.5,1H) )  PRNTAF 00131
8005 FORMAT(35X,50(1H-),20X,A10,A4)  PRNTAF 00132
8006 FORMAT(5H0 WT. / 6H FUNCT, 4(A2,23HVELOCITY POTENTIAL MODE,I3,3X)) PRNTAF 00133
8007 FORMAT(2X,4(A10,4HREAL,8X, 9HIMAGINARY) )  PRNTAF 00134
8010 FORMAT( I4, 8E16.8 )  PRNTAF 00135
7001 FORMAT(1H1,44X,*GENERALIZED AERODYNAMIC COEFFICIENTS*, / PRNTAF 00136
     1    44X,6H(MACH , F5.3,5X,12HRED.FREQ. = F8.5,1H) )  PRNTAF 00137
7006 FORMAT(5H0 WT. / 6H FUNCT,27X,*VELOCITY POTENTIAL MODES*,/ 1 PRNTAF 00138
7007 FORMAT(I13,7I16)  PRNTAF 00139
7005 FORMAT(50X,*REAL PART*,/35X,50(1H-),20X,A10,A4)  PRNTAF 00140
7015 FORMAT(50X,*IMAGINARY PART*,/35X,50(1H-),20X,A10,A4)  PRNTAF 00141
      END               PRNTAF 00142

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FUNCTION LOCSDW(IROW,JCOL,IPNTSD,IPNTIN,IPNTOT,IPNTLS)      LOCSDW 00002
C                                                               LOCSDW 00003
C       RETURNS THE LOCATION OF THE WORD IN THE END-AROUND SUBDIVIDED LOCSDW 00004
C       DOWNWASH ARRAY CORRESPONDING TO BOX(IROW,JCOL) OF THE SUB- LOCSDW 00005
C       DIVIDED BOX ARRAY LOCSDW 00006
C                                                               LOCSDW 00007
C               IROW = BOX CHORDWISE LOCATION LOCSDW 00008
C               JCOL = BOX SPANWISE LOCATION LOCSDW 00009
C               IPNTSD = ARRAY OF POINTERS LOCSDW 00010
C               IPNTIN = NEXT AVAILABLE (UNUSED) CELL IN IPNTSD (END- LOCSDW 00011
C                   AROUND) LOCSDW 00012
C               IPNTOT = FIRST CURRENTLY AVAILABLE CELL IN IPNTSD LOCSDW 00013
C               IPNTLS = LAST CELL OF IPNTSD (LENGTH OF ARRAY) LOCSDW 00014
C               RETURN - LOCSDW 00015
C               LOCSDW = LOCATION OF DESIRED DOWNWASH, IF SUCCESSFUL LOCSDW 00016
C                   = 0, IF LOPNT LIES OUTSIDE THE DEFINED AREA. LOCSDW 00017
C                                                               LOCSDW 00018
C               DIMENSION IPNTSD(2,IPNTLS) LOCSDW 00019
C                                                               LOCSDW 00020
C               LOPNT = MOD(IROW+1,IPNTLS) + 1 LOCSDW 00021
C               LOPNT = LOCATION OF CELL IN IPNTSD WHICH WAS OR IS TO BE LOCSDW 00022
C                   USED LOCSDW 00023
C               IF(IPNTIN - IPNTOT) 100, 300, 200 LOCSDW 00024
C                   END AROUND HAS OCCURRED LOCSDW 00025
C               100 IF (LOPNT - IPNTIN) 400, 300, 150 LOCSDW 00026
C                   NOT IN UPPER PART. IS LOPNT WITHIN BOTTOM PART - LOCSDW 00027
C               150 IF (LOPNT - IPNTOT) 300, 400, 400 LOCSDW 00028
C                                                               LOCSDW 00029
C               NO END AROUND, NORMAL SEQUENCE LOCSDW 00030
C               200 IF (LOPNT - IPNTIN) 250, 300, 300 LOCSDW 00031
C                   LESS THAN UPPER LIMIT. IS LOPNT .GE. LOWER LIMIT - LOCSDW 00032
C               250 IF (LOPNT .GE. IPNTOT) GO TO 400 LOCSDW 00033
C                                                               LOCSDW 00034
C               ERROR OR INITIAL CONDITION ENCOUNTERED (SHOULD NEVER OCCUR)
C               350 LOCSDW = 0 LOCSDW 00035
C                   GO TO 500 LOCSDW 00036
C                                                               LOCSDW 00037
C               SUCCESSFUL, BOX HAS BEEN DEFINED LOCSDW 00038
C               400 IF(JFB = IPNTSD(2,LOPNT))
C                   IF(JCOL.LT.JFB) GO TO 300 LOCSDW 00039
C                   LOCSDW = IPNTSD(1,LOPNT) + JCOL-JFB LOCSDW 00040
C
C               500 CONTINUE LOCSDW 00041
C               RETURN LOCSDW 00042
C               END LOCSDW 00043
C                                                               LOCSDW 00044
C                                                               LOCSDW 00045
C                                                               LOCSDW 00046

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13. ABSTRACT <p>The Mach box technique has been extended to include wing and tail with dihedral angles and vertical separation. A digital computer program, written in FORTRAN, is presented. The program provides for up to nine sweep angles of the leading and trailing edges of each surface. First order piston theory thickness correction is available as an option and two refinement procedures are provided, subdivision with averaging and velocity potential smoothing. For a maximum of twenty oscillatory mode shapes the program calculates normal washes, velocity potentials, lifts, pressures and generalized forces matrices. If only one surface is being analyzed, sampling of wake up-wash, side-wash and longitudinal wash is available.</p> <p>The methods described in this report are intended to be used by airplane designers to calculate with improved accuracy, the unsteady aerodynamic loads that act on a lifting surface being propelled at supersonic speeds. The new feature of these calculations is that the aerodynamic interference between the wing and tail has been taken into account. These calculations are an essential ingredient of flutter analyses and will improve the confidence level of such calculations in preventing wing-tail flutter. The general requirement for such calculations are contained in Military Specification MIL-A-8870A (USAF).</p>		

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